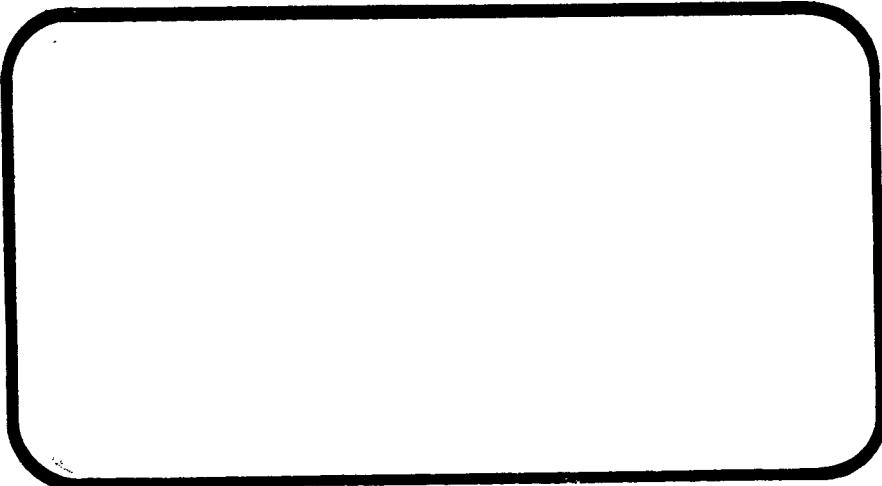


NATIONAL AERONAUTICS AND SPACE ADMINISTRATION



(NASA-CR-134072) MISALIGNMENT STUDIES ON  
SPICE SHUTTLE INTEGRATED VEHICLE (1A31PC)  
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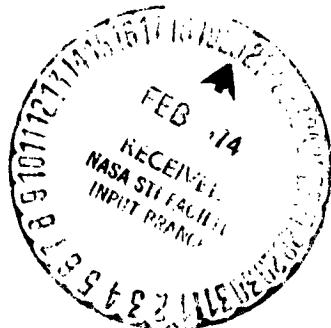
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SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT

JOHNSON SPACE CENTER

HOUSTON, TEXAS

DATA MANAGEMENT SERVICES

SPACE DIVISION



CHRYSLER  
CORPORATION

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MISALIGNMENT STUDIES ON  
SPACE SHUTTLE INTEGRATED VEHICLE  
(IA3IFC)

By

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Prepared under NASA Contract Number NAS9-13247

by

Data Management Services  
Chrysler Corporation Space Division  
New Orleans, La. 70189

for

Engineering Analysis Division

Johnson Space Center  
National Aeronautics and Space Administration  
Houston, Texas

WIND TUNNEL TEST SPECIFICS:

Test Number: MSFC TWT 573  
NASA Series No.: IA31FC  
Occupancy Hours: 36 Hours  
Test Date: July 10, 1973

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Chrysler Corporation Space Division assumes no responsibility for the data presented herein other than its display characteristics.

MISALIGNMENT STUDIES ON THE SPACE SHUTTLE  
INTEGRATED VEHICLE

By  
P. Ramsey\*, T. Mc Means\*\*, and T. Davis\*\*

ABSTRACT

Recent test series at Marshall Space Flight Center's 14" x 14" Transonic Wind Tunnel have raised questions concerning the effects of test model element misalignments on test results. TWT-573 was designed to study these misalignment effects in detail by purposely misaligning model elements a known amount. Misalignments in TWT-573 were achieved by use of special offset spacers and mounting hardware to change yaw, pitch and roll on various elements of a 0.004-Scale PRR Baseline Space Shuttle Configuration model. By comparing the misalignment runs to the nominal configuration with no misalignment, relative effects of the misalignment could be seen. This data was obtained over a Mach range of .9 to 1.46 using a angle of attack sweep of -10° to +10° in 2° increments. The Test Program consisted of 40 runs, which required approximately 4 days tunnel occupancy to complete.

\* NASA/MSFC

\*\* Northrop Services, Inc.

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**SCHEDULE OF PLOTTED COEFFICIENTS:**

- 3
- A) CN, CIM, CA, CAF, CABE, CABO, CABS Vs. ALPHA  
 B; CN, CIM, CA, CAF, CABE, CABO, CABS, CY, CYN, CBL Vs. ALPHA

NOMENCLATURE  
General

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
$c$		speed of sound; m/sec, ft/sec
$C_p$	CP	pressure coefficient; $(p_1 - p_\infty)/q$
$M$	MACH	Mach number; $V/a$
$p$		pressure; N/m <sup>2</sup> , psf
$q$	$Q(NSM)$ $Q(PSF)$	dynamic pressure; $1/2\rho V^2$ , N/m <sup>2</sup> , psf
RN/L	RN/L	unit Reynolds number; per m, per ft
$V$		velocity; m/sec, ft/sec
$\alpha$	ALPHA	angle of attack, degrees
$\beta$	BETA	angle of sideslip, degrees
$\psi$	PSI	angle of yaw, degrees
$\phi$	PHI	angle of roll, degrees
$\rho$		mass density; kg/m <sup>3</sup> , slugs/ft <sup>3</sup>
<u>Reference &amp; C.G. Definitions</u>		
$A_b$		base area; m <sup>2</sup> , ft <sup>2</sup>
$b$	BREF	wing span or reference span; m, ft
c.g.		center of gravity
$\frac{l}{c}_{REF}$	LREF	reference length or wing mean aerodynamic chord; m, ft
$s$	SREF	wing area or reference area; m <sup>2</sup> , ft <sup>2</sup>
	MRP	moment reference point
	XMRP	moment reference point on X axis
	YMRP	moment reference point on Y axis
	ZMRP	moment reference point on Z axis
<u>SUBSCRIPTS</u>		
$b$		base
$l$		local
$s$		static conditions
$t$		total conditions
$\infty$		free stream

NOMENCLATURE (Continued)

Body-Axis System

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
$C_N$	CN	normal-force coefficient; $\frac{\text{normal force}}{qS}$
$C_A$	CA	axial-force coefficient; $\frac{\text{axial force}}{qS}$
$C_Y$	CY	side-force coefficient; $\frac{\text{side force}}{qS}$
$C_{A_b}$	CAB	base-force coefficient; $\frac{\text{base force}}{qS}$ $-A_b(p_b - p_\infty)/qS$
$C_{A_f}$	CAF	forebody axial force coefficient, $C_A - C_{A_b}$
$C_m$	CLM	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS I_{REF}}$
$C_n$	CYN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qSb}$
$C_l$	CLL	rolling-moment coefficient; $\frac{\text{rolling moment}}{qSb}$

Stability-Axis System

$C_L$	CL	lift coefficient; $\frac{\text{lift}}{qS}$
$C_D$	CD	drag coefficient; $\frac{\text{drag}}{qS}$
$C_{D_b}$	CDB	base-drag coefficient; $\frac{\text{base drag}}{qS}$
$C_{D_f}$	CDF	forebody drag coefficient; $C_D - C_{D_b}$
$C_Y$	CY	side-force coefficient; $\frac{\text{side force}}{qS}$
$C_m$	CLM	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS I_{REF}}$
$C_n$	CLN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qSb}$
$C_l$	CSL	rolling-moment coefficient; $\frac{\text{rolling moment}}{qSb}$
L/D	L/D	lift-to-drag ratio; $C_L/C_D$
L/D <sub>f</sub>	L/DF	lift to forebody drag ratio; $C_L/C_{D_f}$

## ADDITIONS TO NOMENCLATURE

$\beta$	BETA	angle of sideslip of reference body; degrees
$\psi$	PSI	angle of yaw of reference body; degrees
$\psi_{SB}$	SRBYAW	angle of yaw of SRB relative to reference body; degrees
$\psi_o$	ORBYAW	angle of yaw of orbiter relative to reference body; degrees
$\phi$	PHI	angle of roll of reference body; degrees
$\phi_o$	ORBROL	angle of roll of orbiter relative to reference body; degrees
$\gamma$		angle of pitch of reference body; degrees
$\gamma_{SB}$	SRBPIT	angle of pitch of SRB relative to reference body; degrees
$i_0$	ORBINC	incidence angle between orbiter and external tank centerlines; degrees
$x$		horizontal separation distance between orbiter and tank aft attach points; inches
$z$	DELTAZ	vertical separation distance between orbiter and tank aft attach points; inches
$G$		denotes the application of grit to model components
$\delta_r$	RUDDER	rudder deflection
$\delta_e$	ELEVON	elevon deflection
$C_{Nbo}$	CNBO	normal force coefficient component of orbiter base drag
$C_{Abo}$	CABO	orbiter base axial force coefficient
$C_{Abe}$	CABE	external tank base axial force coefficient
$C_{Abs}$	CABS	solid rocket booster base axial force coefficient

## TEST FACILITY DESCRIPTION

The Marshall Space Flight Center 14" x 14" Trisonic Wind Tunnel is an intermittent blowdown tunnel which operates by high pressure air flowing from storage to either vacuum or atmospheric conditions. A Mach number range from .2 to 5.85 is covered by utilizing two interchangeable test sections. The transonic section permits testing at Mach 0.20 through 2.50, and the supersonic section permits testing at Mach 2.74 through 5.85. Mach numbers between .2 and .9 are obtained by using a controllable diffuser. The range from .95 to 1.3 is achieved through the use of plenum suction and perforated walls. Mach numbers of 1.44, 1.93 and 2.50 are produced by interchangeable sets of fixed contour nozzle blocks. Above Mach 2.50 a set of fixed contour nozzle blocks are tilted and translated automatically to produce any desired Mach number in .25 increments.

Air is supplied to a 6000 cubic foot storage tank at approximately -40°F dew point and 500 psi. The compressor is a three-stage reciprocating unit driven by a 1500 hp motor.

The tunnel flow is established and controlled with a servo actuated gate valve. The controlled air flows through the valve diffuser into the stilling chamber and heat exchanger where the air temperature can be controlled from ambient to approximately 180°F. The air then passes through the test section which contains the nozzle blocks and test region.

Downstream of the test section is a hydraulically controlled pitch sector that provides a total angle of attack range of 20° ( $\pm 10^\circ$ ). Sting offsets are available for obtaining various maximum angles of attack up to 90°.

## MODEL DESCRIPTION AND TEST HARDWARE

The model used in TWT 573 was a 0.004 scale model of the PRR Baseline Launch Configuration Space Shuttle. Except for the orbiter wings, the model was constructed from stainless steel with the wings made from aluminum. All model elements were built according to MCR 0074 Baseline Drawings by Rockwell, dated 15 January 1973.

For arrangement of model elements in test configuration see Figure 2. Dimensions of each model element are presented in the Component Description Sheets and on the figure.

Since a means for misalignment was needed, offset spacers were used. By attaching the Orbiter and SRB's directly to the External tank via these spacers and normal mounting hardware, a means for misalignments was readily available. Pitch, yaw and roll changes were then made by removing one set of spacers and mounting hardware and installing a different set.

As mentioned above, all model elements were mounted on the external tank element. This arrangement was used to obtain total vehicle data needed to check effects of misalignments. This data was gathered by a single six-component strain-gage balance (MSFC 232) mounted in the external tank using Balance to Adapter 80M32593. Supporting the model in the Test Section was sting 98 and straight extension S-3.

All model control surfaces were in the streamline position for this test series.

The total vehicle model had an approximate cross-section area of 4.0 square inches. This gave a tunnel blockage of 2.0% at  $\alpha = 0^\circ$ . With each

increase in  $\alpha$  the blockage percentage also increased.

Some test runs were conducted using grit. See Figure 8 for sketch of grit application.

## CONFIGURATIONS INVESTIGATED

Using the offset spacers various misalignments were tested to determine effect on total vehicle aerodynamic data. The various configurations investigated were:

1. Basic configuration using no misalignment

O<sub>3</sub> PRR Baseline orbiter less abort solid rocket motors

T<sub>9</sub> 324 inch Diameter External Tank with ogive nose

S<sub>3</sub> 142 inch Diameter Solid Rocket motor with 18° nose cone

2. Orbiter yawed 1° relative to external tank centerline.

O<sub>3</sub>(ψ<sub>0</sub> = +1°) Orbiter at +1° yaw

T<sub>9</sub> See baseline configuration

S<sub>3</sub> See baseline configuration

3. Orbiter rolled +1° relative to external tank centerline.

O<sub>3</sub>(φ<sub>0</sub> = +1°) Orbiter at +1° of roll

T<sub>9</sub> See baseline configuration

S<sub>3</sub> See baseline configuration

4. SRB pitched +1° relative to external tank centerline.

O<sub>3</sub> See baseline configuration

T<sub>9</sub> See baseline configuration

S<sub>3</sub>(γ<sub>SB</sub> = +1°) SRB pitched +1°

5. SRB pitch -1° relative to external tank centerline.

O<sub>3</sub> See baseline configuration

T<sub>9</sub> See baseline configuration

S<sub>3</sub>(γ<sub>SB</sub> = -1°) SRB pitched -1°

6. SRB yawed +1° relative to external tank centerline.

$O_3$  See baseline configuration

$T_9$  See baseline configuration

$S_3(\psi_{SB} = +1^\circ)$  SRB yaw +1°

7. SRB yawed -1° relative to external tank centerline.

$O_3$  See baseline configuration

$T_9$  See baseline configuration

$S_3(\psi_{SB} = -1^\circ)$  SRB yaw -1°

8. Grit study of baseline configuration

$O_3 G$

$T_9 G$  same as baseline  
with Grit added.

$S_3 G$

9. Data repeatability study using Grit configuration model.

$O_3 G$

$T_9 G$  same as baseline  
with Grit added.

$S_3 G$

The  $O_3$  Orbiter model consists of several parts and may be represented as  $B_{10}C_5D_7F_4M_3W_8E_{18}V_5R_5$ . These parts are defined as follows:

$B_{10}$  Double delta wing fuselage with 57.0 in. radius nose

$C_5$  Canopy

$D_7$  Manipulator housing for lightweight orbiter

$F_4$  Body flap  
 $M_3$  OMS pods for lightweight orbiter  
 $W_{87}$  Double delta wing for lightweight orbiter  
 $E_{18}$  Elevon  
 $V_5$  Vertical tail for lightweight orbiter  
 $R_5$  Rudder

The ET(T9) and SRB(S3) were not broken into sub assemblies for this test.

## DATA REDUCTION

The body axis system was used in TWT 573, with all coefficients being presented in non-dimensional form. See the axis system used in Figure 1.

A list of reference dimensions and areas are:

	<u>Symbol</u>	<u>Full Scale</u>	<u>0.004-Scale</u>
Reference area	$S_{ref}$	2690 ft <sup>2</sup>	6.198 in. <sup>2</sup>
Reference length	$l_{ref}$	1328 in.	5.313 in.
Reference span	$b_{ref}$	1328 in.	5.313 in.
Orbiter base area	$A_{b_0}$	427.8 ft <sup>2</sup>	0.9857 in. <sup>2</sup>
External Tank base area	$A_{b_E}$	572.55 ft <sup>2</sup>	1.319 in. <sup>2</sup>
SRB base area (1 SRB)	$A_{b_S}$	365.87 ft <sup>2</sup>	0.843 in. <sup>2</sup>
BMC location (from base of ET)			3.200 in.
MRP location (forward of the BMC)			1.564 in.

The Moment Reference Point (MRP) was located on the external tank centerline at the orbiter nose as shown in Figure 2. The Balance Moment Center (BMC) is also on the external tank centerline, 3.200 inches from the base of the tank. Therefore, the MRP is 1.564 inches forward of the BMC.

In addition to the standard force and moment data coefficients another set of coefficients were required. These were obtained by subtracting out the forces and moments due to base pressures from the standard data set.

The equations used in the calculations are:

$$CAF = CA - CAB_O - (2) CAB_S - CAB_E$$

$$CAB_O = CPB_O \frac{A_{bO}}{S_{ref}} \quad (\text{orbiter base axial force coefficient})$$

$$CAB_S = CPB_S \frac{A_{bS}}{S_{ref}} \quad (\text{SRB base axial force coefficient})$$

$$CAB_E = CPB_E \frac{A_{bE}}{S_{ref}} \quad (\text{external tank base axial force coefficient})$$

Where:

$$CPB_O = \frac{P_{bO} - P_\infty}{q}$$

$$CPB_S = \frac{P_{bS} - P_\infty}{q}$$

$$CPB_E = \frac{P_{bE} - P_\infty}{q}$$

All data presented was corrected for sting deflections and weight  
tares.

Table 1  
MODEL COMPONENT DIMENSIONAL DATA

MODEL COMPONENT: Body B10

GENERAL DESCRIPTION: Double Delta Wing Fuselage Per Lines VL70-000093.

with 57.0 in Radius Nose.

2A Configuration Lt. Wt. Orbiter

Scale Model = .004

DRAWING NUMBER: V172-000061  
VL70-000093

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length, in.	<u>1328.3</u>	<u>5.313</u>
Max. Width $X_0$ 560 to $X_0$ 1307	<u>216.0</u>	<u>0.864</u>
Max. Depth	<u>239.0</u>	<u>0.956</u>
Fineness Ratio	<u>5.495</u>	<u>5.495</u>
Area, $FT^2$		
Max. Cross-Sectional	<u>319.556</u>	<u>0.005</u>
Planform		
Wetted		
Base		

Table 1. (Cont.)

MODEL COMPONENT: Canopy - C5

GENERAL DESCRIPTION: 2A Configuration Per NR Lines VL70-000092.

Scale Model = .004

DRAWING NUMBER: VL70-000092

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Sta. Fwd Bulkhead	<u>391.00</u>	<u>1.564</u>
Sta. T.E.	<u>560.0</u>	<u>2.240</u>
Canopy Intersects Body ML	<u>391.00</u>	<u>1.564</u>
Fineness Ratio	—	—
Area	—	—
Max. Cross-Sectional	—	—
Planform	—	—
Wetted	—	—
Base	—	—

Table 1. (Cont.)

C  
MODEL COMPONENT Manipulator Housing D-7GENERAL DESCRIPTION: 2A Configuration, Light WT. Orbiter Per LinesVL70-000093Scale Model = .004DRAWING NUMBER:VL70-000093DIMENSIONS:FULL-SCALEMODEL SCALE

Length , in.

881.003.524

Max. Width, in.

51.000.204

Max. Depth, in.

20.000.080

Fineness Ratio

Area

Max. Cross-Sectional

Planform

Wetted

Base

$\epsilon$  Fuselage, BP = 0.0  
 WP = 500.0 INFS  
 $x_o$  426.0 to 1307.0

Table 1. (Cont.)

MODEL COMPONENT: F4 Body Flap

GENERAL DESCRIPTION: 2A Configuration Per NR Lines VL70-000094 "A"

Scale Model = .004

DRAWING NUMBER: VL70-000094A

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length, in.	<u>84.70</u>	<u>0.3388</u>
Max. Width , in.	<u>265.00</u>	<u>1.060</u>
Max. Depth	_____	_____
Fineness Ratio	_____	_____
Area	_____	_____
Max. Cross-Sectional	_____	_____
Planform, ft <sup>2</sup>	<u>142.63715</u>	<u>0.002282</u>
Wetted	_____	_____
Base	_____	_____

Table 1. (Cont.)

MODEL COMPONENT: OMS PODS-M3GENERAL DESCRIPTION: 2A Light WT Configuration; per MC120074,  
Per NR Lines VL70-000094.Scale Model = .004DRAWING NUMBER: VL70-000094

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length, in.	<u>346.0</u>	<u>1.440</u>
Max. Width, in.	<u>108.0</u>	<u>0.432</u>
Max. Depth, in.	<u>72.8</u>	<u>0.291</u>
Fineness Ratio	—	—
Area	—	—
Max. Cross-Sectional	—	—
Planform	—	—
Wetted	—	—
Base	—	—

E of OMS POD

$$WP = 463.9 \text{ inches FS}; WP 400.0 + 63.9 = 463.90 \text{ INFS}$$

$$1.600 + .2556 = 1.8556 \text{ INMS}$$

$$BP = 80.0 \text{ in. FS}; 0.320 \text{ INMS}$$

$$\text{From Fuselage Station } 1214.0 \text{ to } 1560 \text{ INFS} = 346.0 \text{ INFS}$$

$$4.856 \text{ to } 6.240 = 1.384 \text{ INMS}$$

Table 1. (Cont.)

MODEL COMPONENT: Wing W-87 New Light Weight

GENERAL DESCRIPTION: Orbiter Configuration per lines VL70-000093

Scale Model = .004

DRAWING NUMBER: VL70-000093

DIMENSIONS:FULL-SCALEMODEL SCALETOTAL DATAArea, FT<sup>2</sup> (W.R.P.)

Planform

2689.38

0.043

Wetted

--

--

## Span (equivalent), FT

77.12

0.308

## Aspect Ratio

2.214

2.214

## Rate of Taper

1.176

1.176

## Taper Ratio

0.209

0.209

## Dihedral Angle, degrees @ 75.33% element line

3.860

3.860

Incidence Angle, degrees @ 425 $\frac{1}{4}$  tol. 0.00

3.000

3.000

## Aerodynamic Twist, degrees

--

--

## Toe-In Angle

--

--

## Cant Angle

--

--

## Sweep Back Angles, degrees

Leading Edge

44.873

44.873

Trailing Edge

-10.242

-10.242

0.25 Element Line

35.050

35.050

## Chords:

Root (Wing Sta. 0.0)

690.19

2.761

Tip, (equivalent)

144.30

0.577

MAC

476.76

1.907

Fus. Sta. of .25 MAC

1136.12

4.544

W.P. of .25 MAC

289.44

1.158

B.L. of .25 MAC

181.03

0.724

## Airfoil Section

Root

--

--

Tip

--

--

EXPOSED DATAArea, FT<sup>2</sup>

1746.87

6.987

## Span, (equivalent), FT

59.16

0.237

## Aspect Ratio

2.004

2.004

## Taper Ratio

0.256

0.256

## Chords

Root

562.77

2.251

Tip

144.30

0.577

MAC

394.81

1.579

Fus. Sta. of .25 MAC

1185.17

4.741

W.P. of .25 MAC

291.56

1.156

B.L. of .25 MAC

250.54

1.002

## LEADING EDGE CUFF (data for (1) side)

Plan form area, FT<sup>2</sup> (BP 108.0)

120.333

0.0019

## L.E. Intersect Fus ML @ STA

560.0

2.240

## L.E. Intersects Wing @ STA

1035.0

4.140

Table 1. (Cont.)

MODEL COMPONENT: Elevon E-18

GENERAL DESCRIPTION: 2A Configuration Per W-87, NR Lines VL70-000093

Data for (1) or (2) Sides

Model Scale = .004

DRAWING NUMBER:VL70-000093DIMENSIONS:

	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area, FT <sup>2</sup>	<u>205.517</u>	<u>0.0033</u>
Span (equivalent), in.	<u>353.34</u>	<u>1.413</u>
Inb'd equivalent chord	<u>114.78</u>	<u>0.459</u>
Outb'd equivalent chord	<u>55.00</u>	<u>0.220</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>.208</u>	<u>.208</u>
At Outb'd equiv. chord	<u>.400</u>	<u>.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>0.00</u>	<u>0.00</u>
Tailing Edge	<u>-10.02</u>	<u>-10.02</u>
Hingeline	<u>0.00</u>	<u>0.00</u>
Area Moment (Normal to hinge line), FT <sup>3</sup> Product of area moment	<u>1548.07</u>	<u>0.00010</u>

Table 1. (Cont.)

MODEL COMPONENT: Vertical Tail V5 (Light Wt. Orbiter Config)GENERAL DESCRIPTION: Center Line Vertical Tail on the Double Delta Configuration with Double Wedge Airfoil and Rounded Leading Edge, Tot.Data Includes Void Area Listed Below      Scale Model = .004DRAWING NUMBER:VL70-000095DIMENSIONS:FULL-SCALEMODEL SCALETOTAL DATA

Area, FT <sup>2</sup>	386.05	0.006
* Void (included above), FT <sup>2</sup>	13.17	0.0002
Blanketed included above, FT <sup>2</sup>	12.67	0.0002
Span (equivalent), FT	24.37	0.097
Aspect Ratio	1.590	1.590
Rate of Taper	0.507	0.507
Taper Ratio	0.426	0.426
Diehedral Angle, degrees	--	--
Incidence Angle, degrees	--	--
Aerodynamic Twist, degrees	--	--
Toe-In Angle	0.0	0.0
Cant Angle	0.0	0.0
Sweep Back Angles, degrees	45.000	45.000
Leading Edge	26.249	26.249
Trailing Edge	41.130	41.130
0.25 Element Line		
Chords:		
Root (Wing Sta. 0.0)	257.99	1.032
Tip, (equivalent)	109.78	0.439
MAC	193.84	0.775
Fus. Sta. of .25 MAC	1473.64	5.895
W.P. of .25 MAC	647.31	2.589
B.L. of .25 MAC	0.0	0.0
Airfoil Section		
Root		
Tip		

EXPOSED DATA

Area		
Span, (equivalent)		
Aspect Ratio		
Taper Ratio		
Chords		
Root		
Tip		
MAC		
Fus. Sta. of .25 MAC		
W.P. of .25 MAC		
B.L. of .25 MAC		

\*Void area noted is the area located at lower aft portion of tail surface.

Table 1. (Cont.)

MODEL COMPONENT: Rudder R5GENERAL DESCRIPTION: 2A Configuration Per NR Lines VL70-000095.Scale Model - .004DRAWING NUMBER: VL70-000095

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area , FT <sup>2</sup>	<u>98.67</u>	<u>0.0116</u>
Span (equivalent), in.	<u>201.0</u>	<u>0.804</u>
Inb'd equivalent chord	<u>91.585</u>	<u>0.366</u>
Outb'd equivalent chord	<u>50.833</u>	<u>0.203</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
At Outb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>34.83314</u>	<u>34.83314</u>
Tailing Edge	<u>26.24915</u>	<u>26.24915</u>
Hingeline	<u>34.83314</u>	<u>34.83314</u>
Area Moment (Normal to hinge line), FT <sup>3</sup>	<u>526.125</u>	<u>0.00003</u>
Product of area and mean chord		

TABLE 1. (Continued)

MODEL COMPONENT: BODY - External Tank T<sub>9</sub>GENERAL DESCRIPTION: 2A Configuration Per NR Lines VL72-000061BVL70-000018                   ;Body of RevolutionScale = .004DRAWING NUMBER: VL78-000018

<u>DIMENSIONS:</u>	<u>THEORETICAL</u>	<u>ACTUAL MEASURED</u>	
	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>	<u>MODEL SCALE</u>
Length	<u>1989.0</u>	<u>7.956</u>	
Max. Width	<u>324.0</u>	<u>1.296</u>	
Max. Depth			
Fineness Ratio	<u>6.13889</u>	<u>6.13889</u>	
Area			
Max. Cross-Sectional	<u>572.555</u>	<u>0.00916</u>	
Planform			
Wetted			
Base	<u>572.555</u>	<u>0.00916</u>	

## REF.

FS (Orbiter) 0.00 = Tank Station 751.0 INFS

WP (ET) = 400 - 344.413 = 55.587 INFS

BP (Orbiter) 0.00 = 0.00 ET

Table 1. (Cont.)

MODEL COMPONENT: S3 Booster Solid Rocket MotorGENERAL DESCRIPTION: 2A Configuration Per NR Lines VL77-000012 and  
VL72-000061BBody of Revolution, Data for (1) or (2) SidesScale Model = .004DRAWING NUMBER: VL77-000012Data for (1) or (2) Sides  
DIMENSIONS:THEORETICALACTUAL MEASURED

	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>	<u>MODEL SCALE</u>
Length	<u>1758.00</u>	<u>7.032</u>	
Max. Width	<u>142.00</u>	<u>0.568</u>	
Max. Depth	<u>259.00</u>	<u>1.036</u>	
Fineness Ratio	<u>6.788</u>	<u>6.788</u>	
Area			
Max. Cross-Sectional	<u>109.978</u>	<u>0.00176</u>	
Planform			
Wetted			
Base	<u>365.870</u>	<u>0.00585</u>	

## REF

FS (Orbiter) 0.00 = 751.0 in. ET = 202.0 BSRM

WP (BSRM) = 400 - 344.413 = 55.587 INFS

BP (Orbiter) 0.00 = 243.0 BSRM

Table 2.

**TEST CONDITIONS**

BALANCE UTILIZED: MSFC 232

**CAPACITY:**

**ACCURACY:**

**COEFFICIENT  
TOLERANCE:** At  $q = 10 \text{ lbs/in.}^2$

NF	300	1bs
SF	143	1bs
AF	50	1bs
PM	400	in.-1bs
YM	192	in.-1bs
RM	100	in.-1bs

+1.50 lbs  
+0.72 lbs.  
+0.25 lbs.  
+2.00 in.-lbs  
+0.96 in.-lbs  
+0.50 in.-lbs

+0.024  
+0.012  
+0.004  
+0.006  
+0.003  
+0.002

**COMMENTS:**

## TEST: MSFC TWT - 573

TABLE 3.  
DATA SET/RUN NUMBER COLLATION SUMMARY  
PARAMETER/VALUES

DATA SET IDENTIFIER	CONFIGURATION	SCHED.	TEST RUN NUMBERS						NO. OF ALTERNATE INDEPENDENT VARIABLE	
			$\alpha$	$\beta$	$C_0$	$C_1$	$C_2$	$C_3$		
R900000	O <sub>3</sub> T <sub>9</sub> S <sub>3</sub>	A	0	-5	-14	49	0	0	5	.9
R90100	O <sub>3</sub> T <sub>9</sub> S <sub>3</sub>									1
R90200	O <sub>3</sub> T <sub>9</sub> S <sub>3</sub> ( $\phi_0 = +1^\circ$ )									4
R90201	O <sub>3</sub> T <sub>9</sub> S <sub>3</sub> ( $\psi_0 = +1^\circ$ )									6
R90300	O <sub>3</sub> T <sub>9</sub> S <sub>3</sub> ( $\psi_{SB} = +1^\circ$ )									18
R90302	O <sub>3</sub> T <sub>9</sub> S <sub>3</sub> ( $\psi_{SB} = -1^\circ$ )									20
R90400	O <sub>3</sub> T <sub>9</sub> S <sub>3</sub> ( $\phi_{SB} = +1^\circ$ )									23
R90402	O <sub>3</sub> T <sub>9</sub> S <sub>3</sub> ( $\phi_{SB} = -1^\circ$ )									24
R90500	O <sub>3</sub> T <sub>9</sub> S <sub>3</sub>									26
										27
										28
										37
										39
										40
										32
										31
										41
										33
										34
										35
										36
										37
										43
										49
										55
										61
										67
										75 76
										10
										NDV
										COVAR(1)
										COVAR(2)
										SCHEDULES

$\alpha$  OR  $\beta$  COEFFICIENTS  
 $\Delta x = A = -10^6$  TO  $10^6$   $\Delta x = -1$  COEFFICIENTS

- Notes:**
1. Positive directions of force coefficients, moment coefficients, and angles are indicated by arrow
  2. For clarity, origins of wind and stability axes have been displaced from the center of gravity

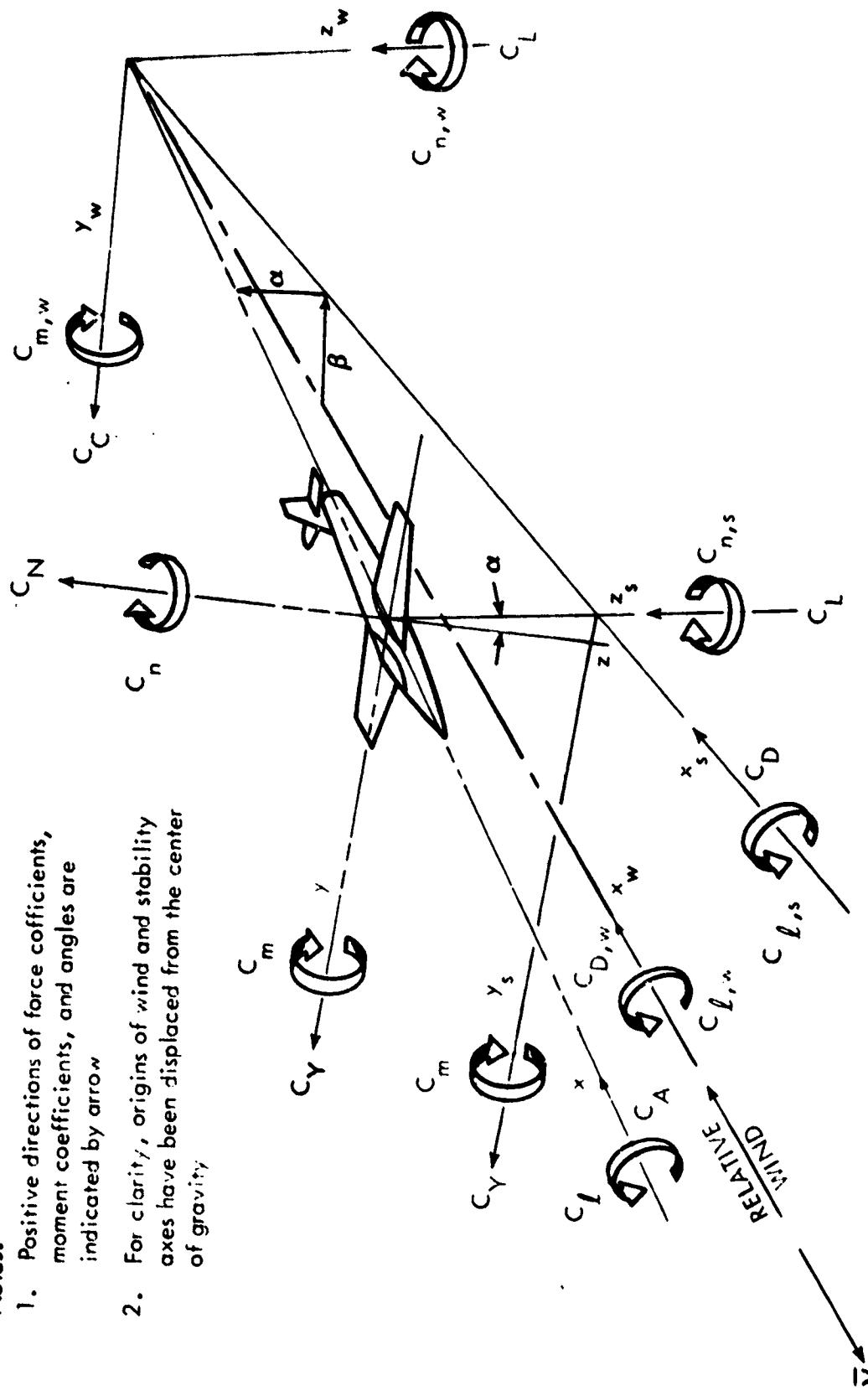


Figure 1. Axis systems.

ALL DIMENSIONS ARE INCHES MODEL SCALE

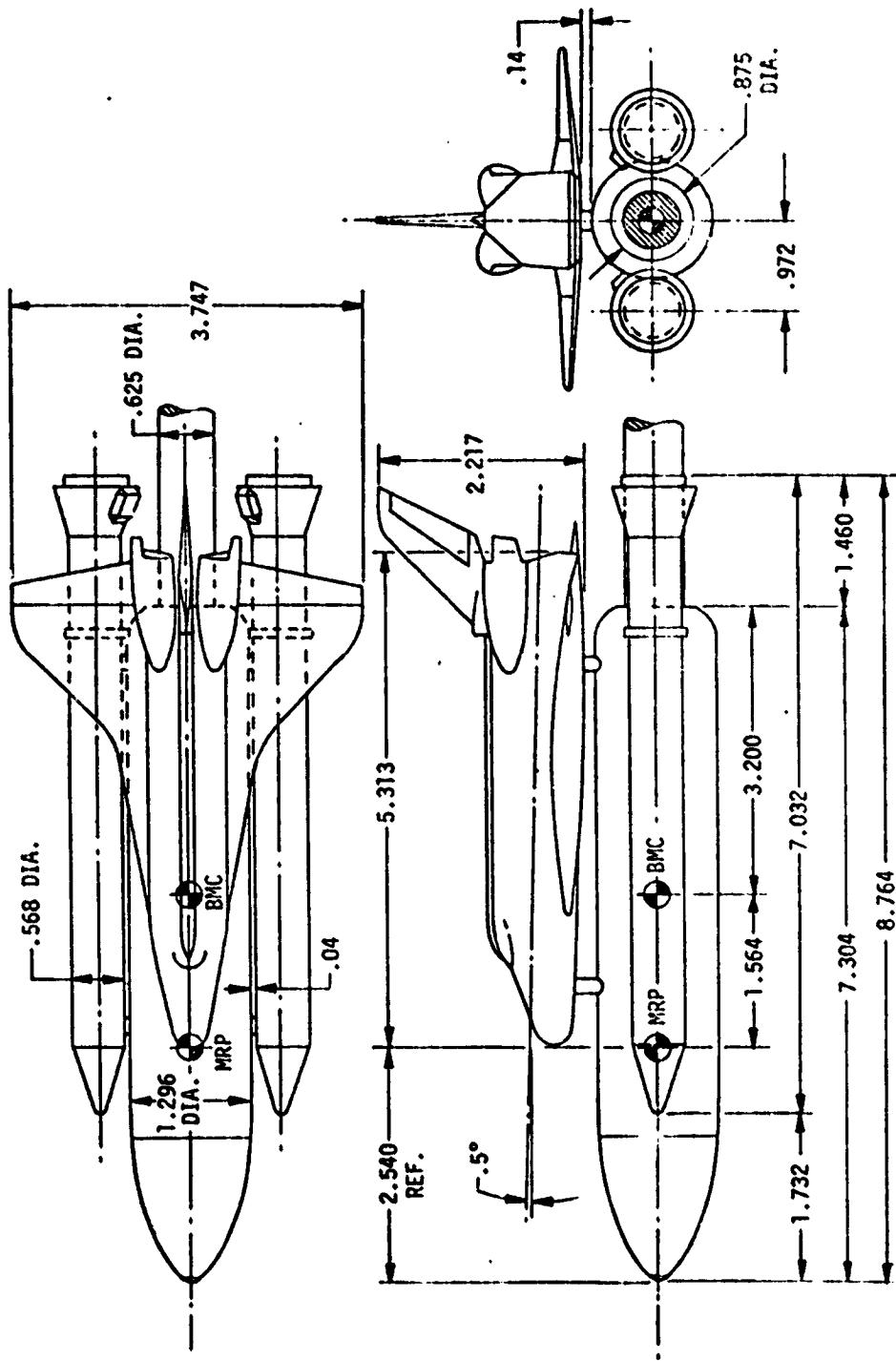


Figure 2. General Arrangement of the Model Configuration 02 TPS1

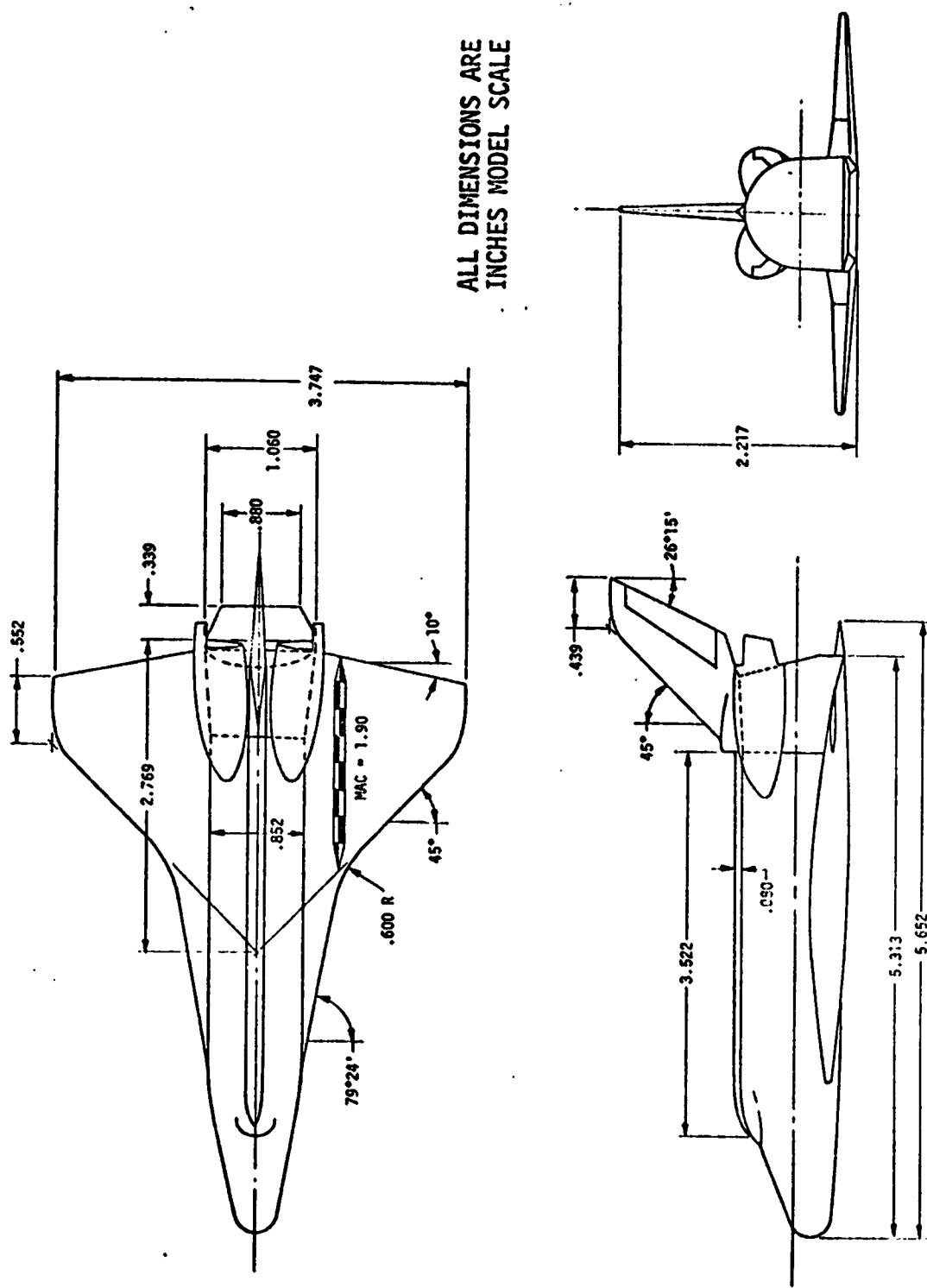


Figure 3. General Arrangement of PRR Orbiter Model 0<sub>3</sub>

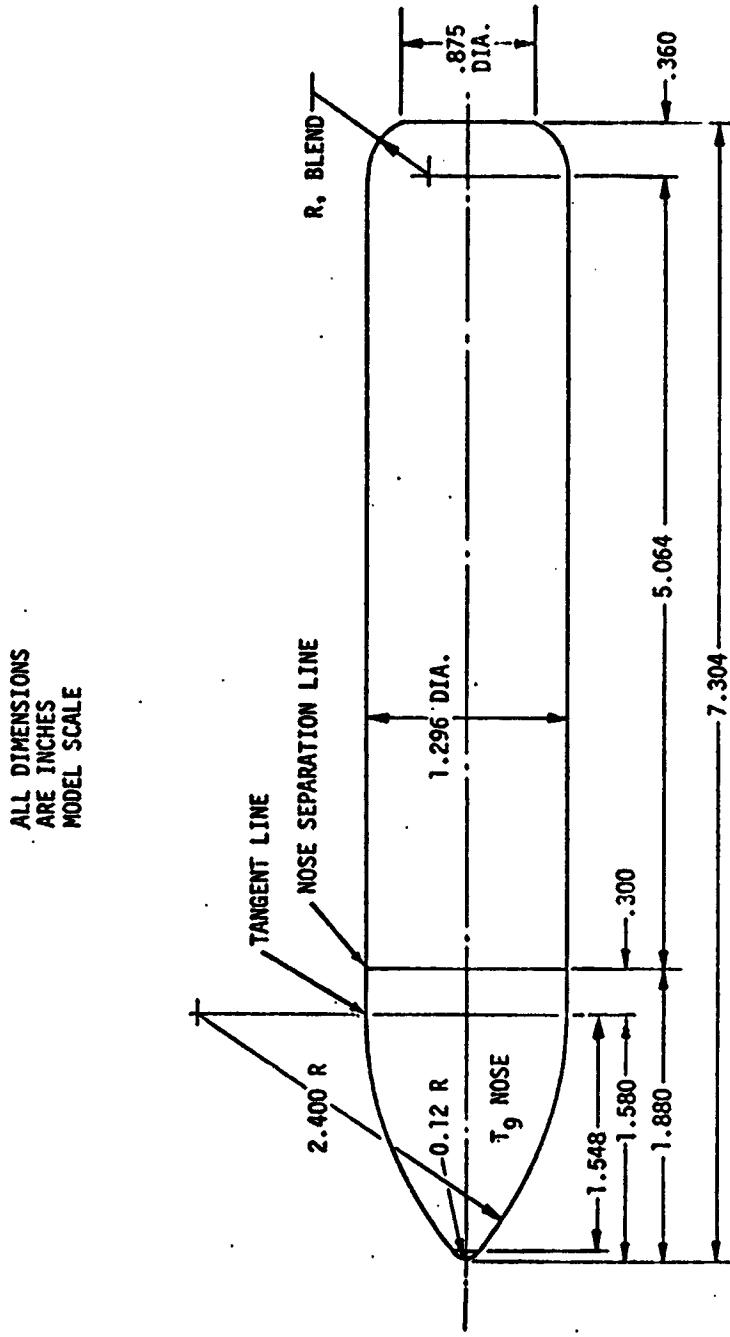


Figure 4. General Arrangement of the External Tank T<sub>8</sub> and T<sub>9</sub>

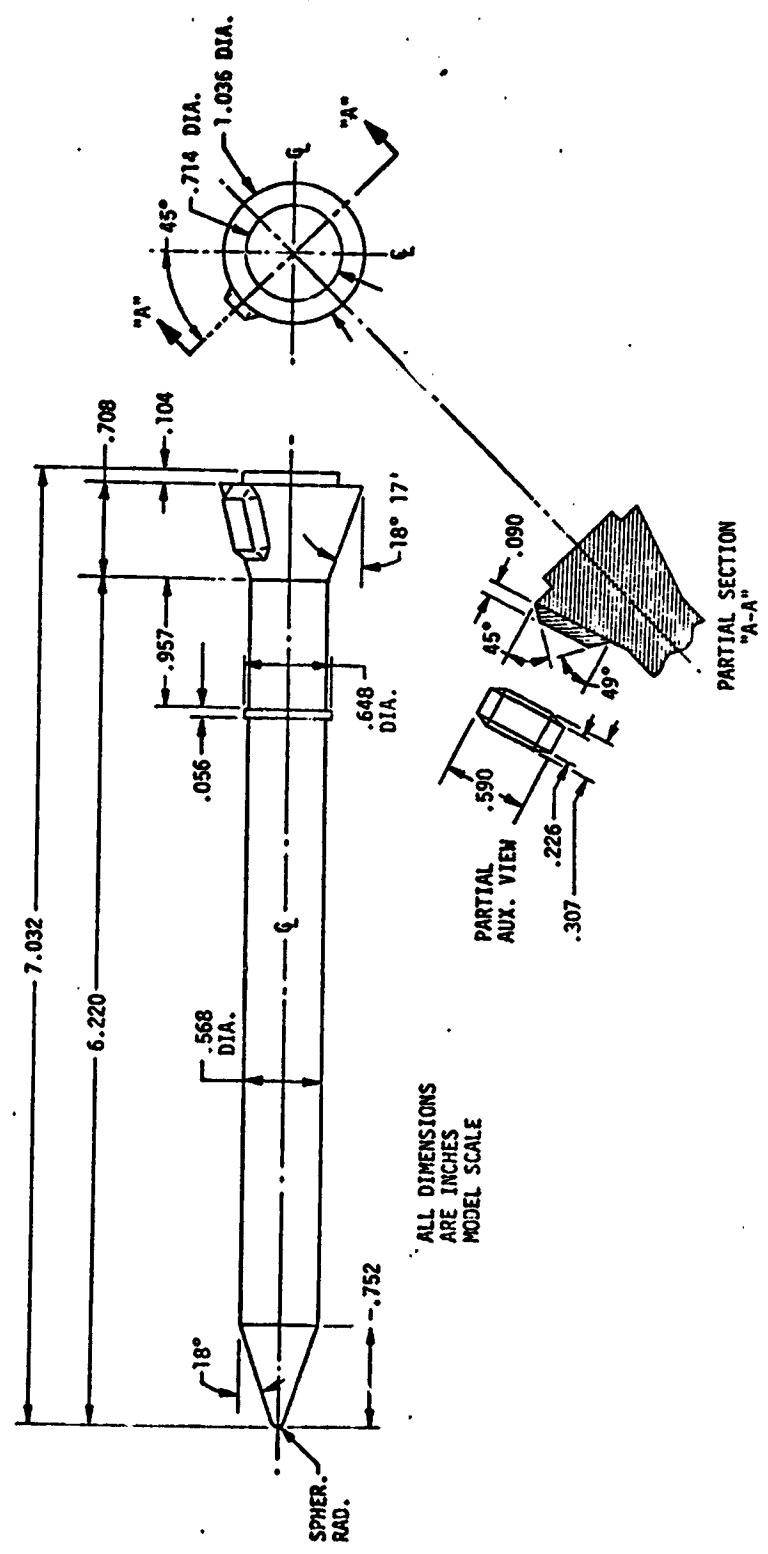
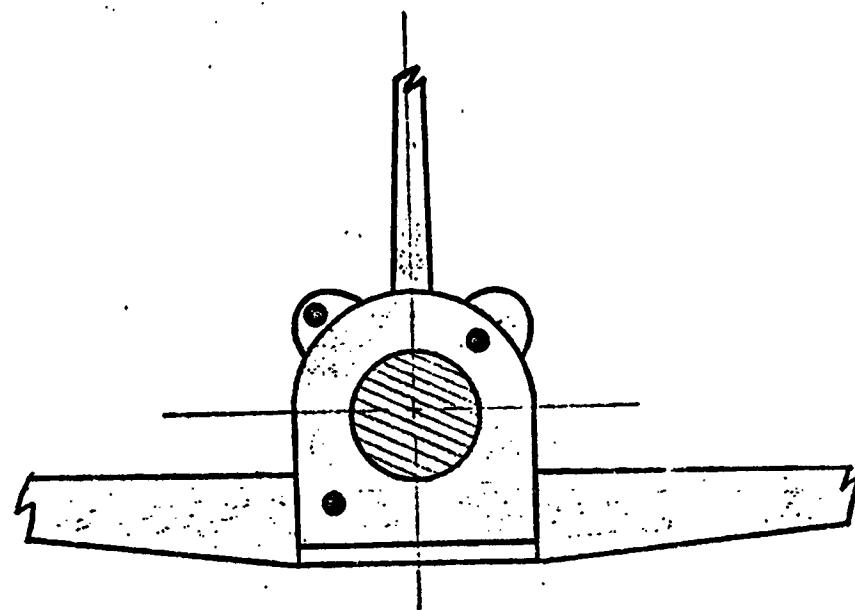


Figure 5. General Arrangement of the SRB S<sub>3</sub>



●— BASE PRESSURE MEASUREMENTS

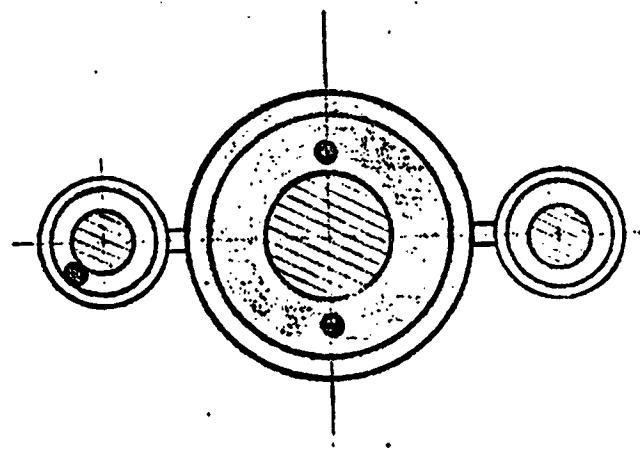


Figure 6. Base Pressure Probe Locations

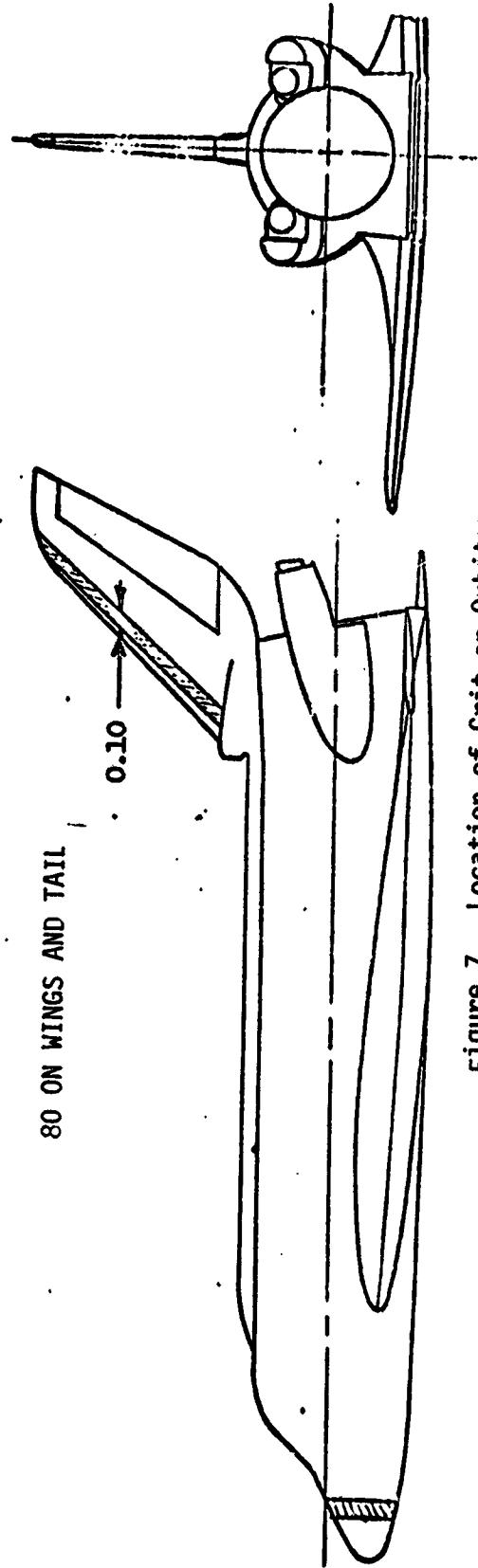
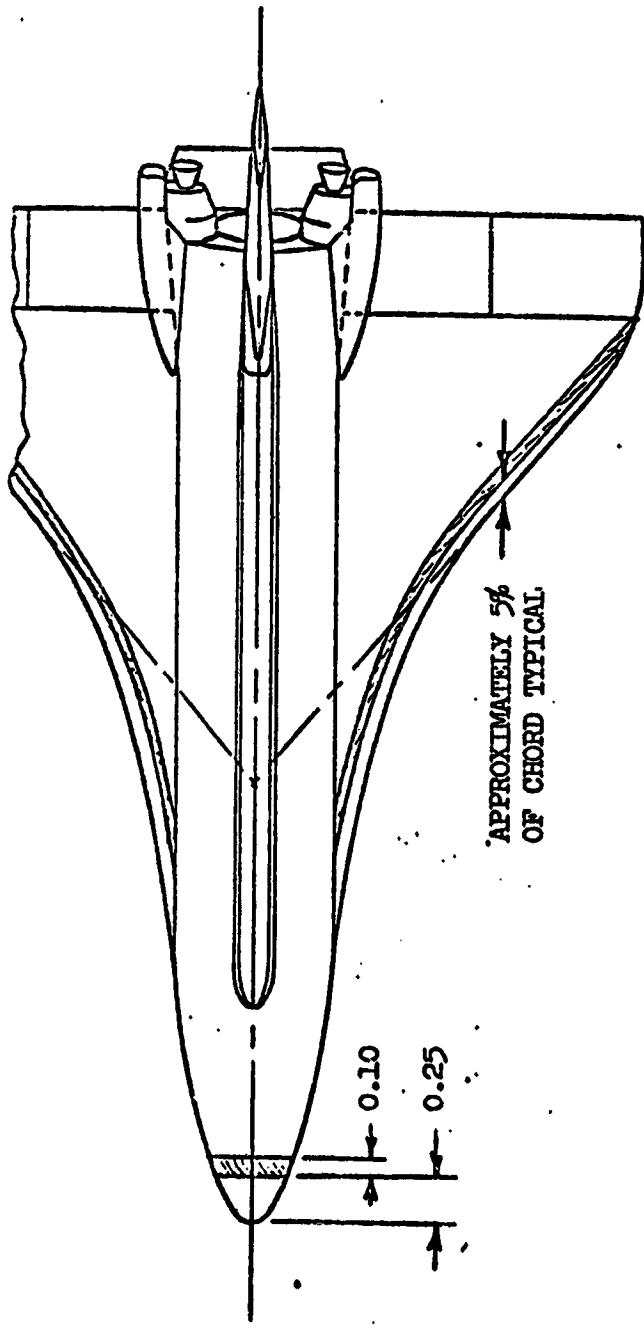


Figure 7. Location of Grit on Orbiter

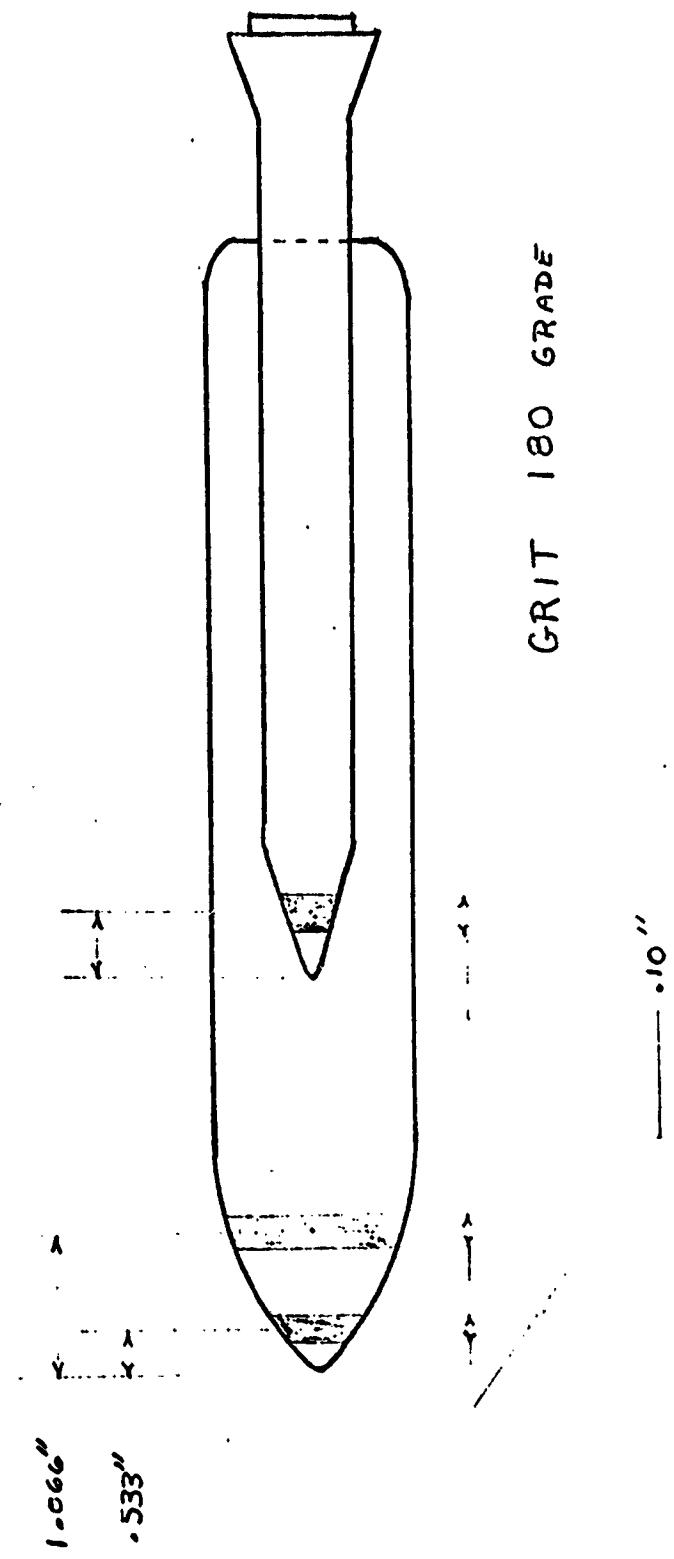


Figure 8. Location of Grit on ET and SRB

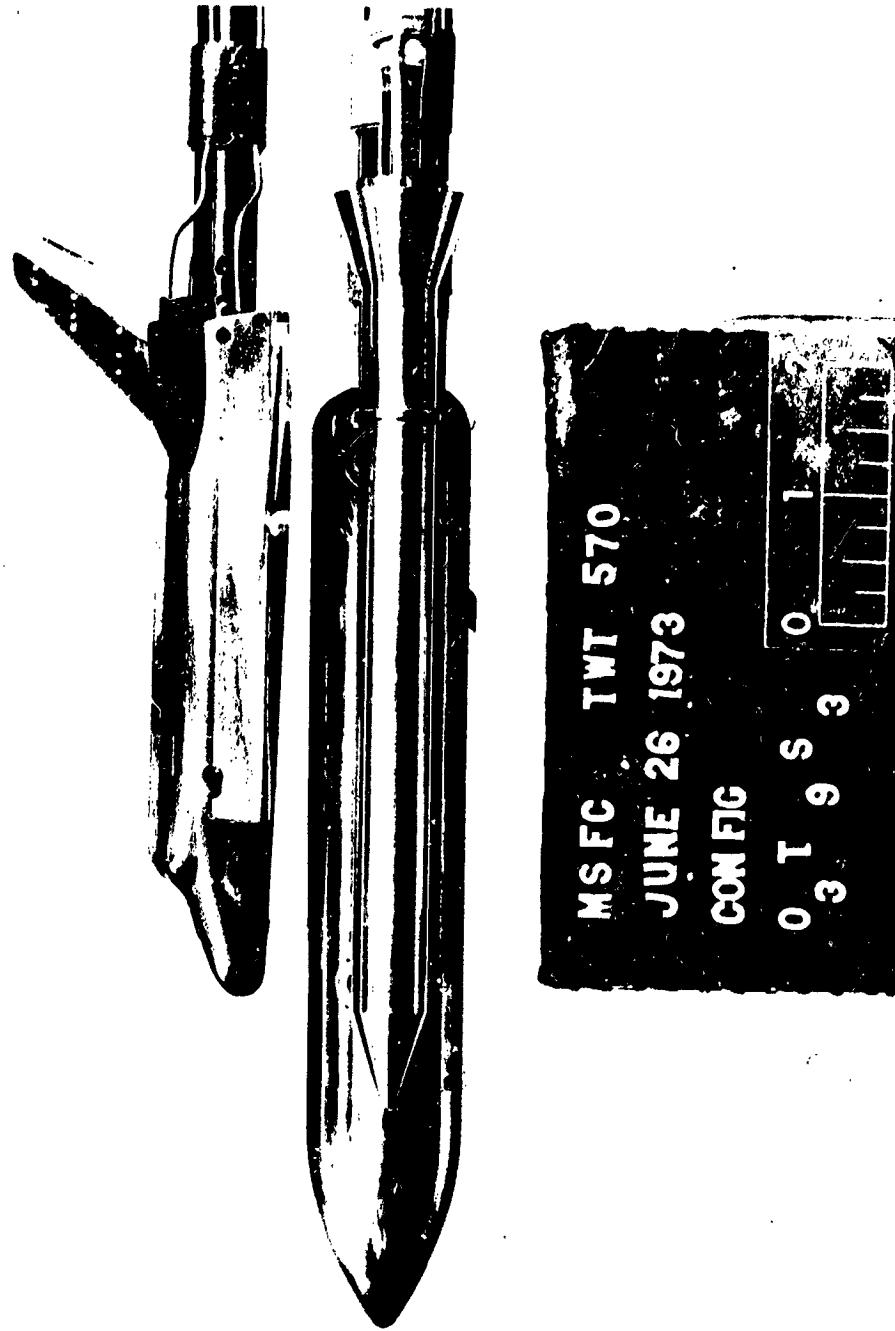


Figure 9. - Photograph of Tunnel Installation of Basic Configuration.

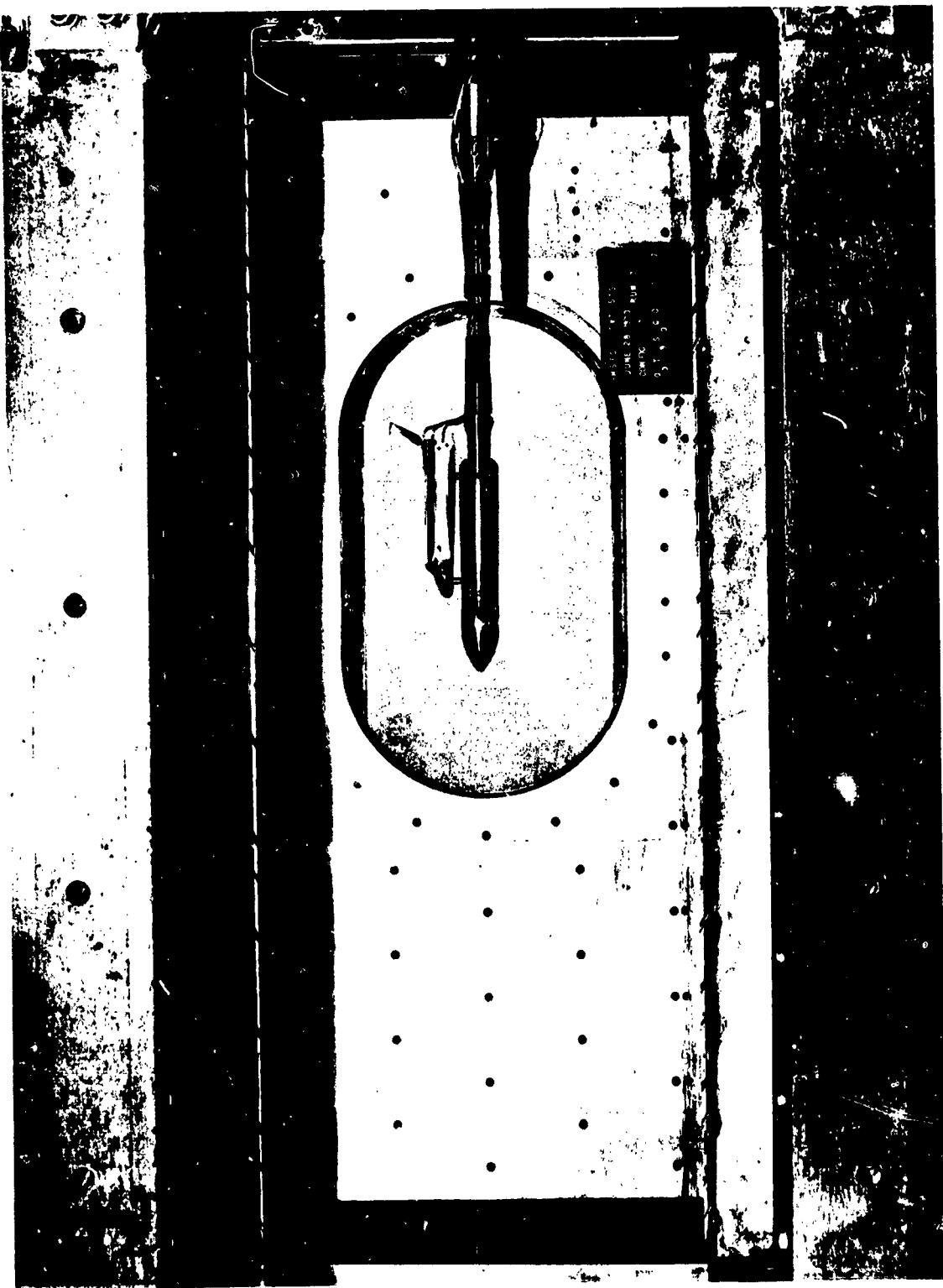


Figure 10. - Photograph of Tunnel Installation of Basic Configuration  
With Attach Hardware.

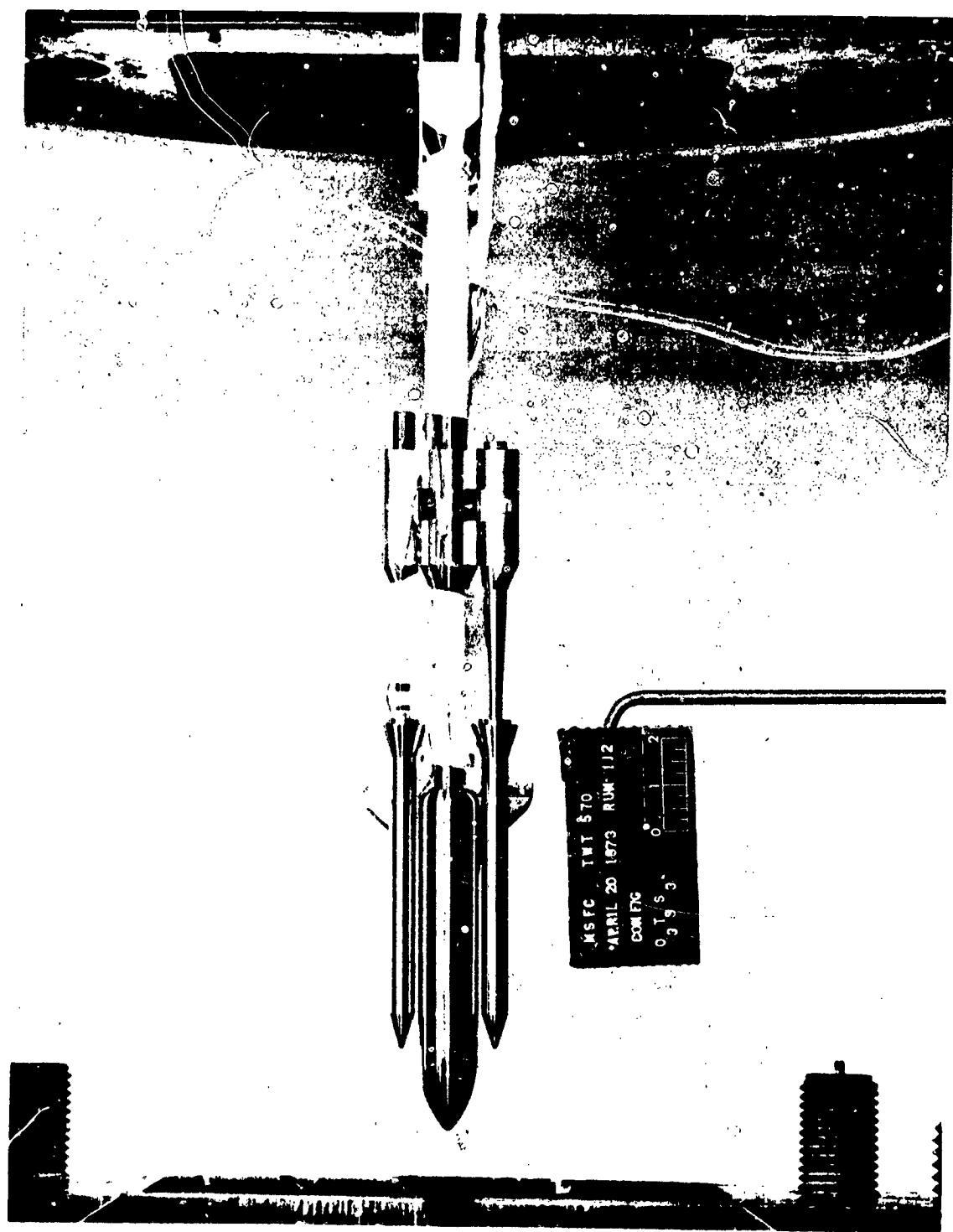


Figure 11. - Photograph of Tunnel Installation of Basic Configuration ( $\phi = 90^\circ$ )

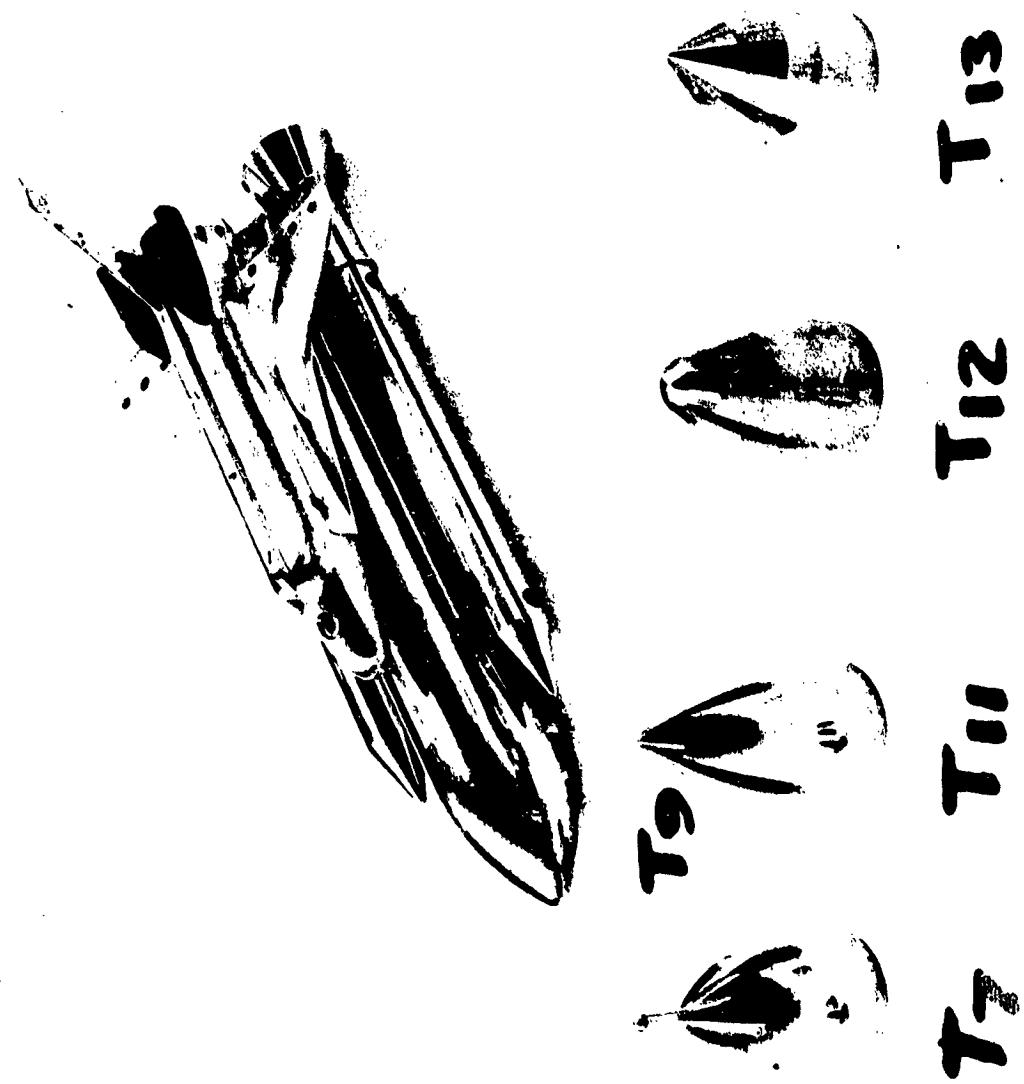
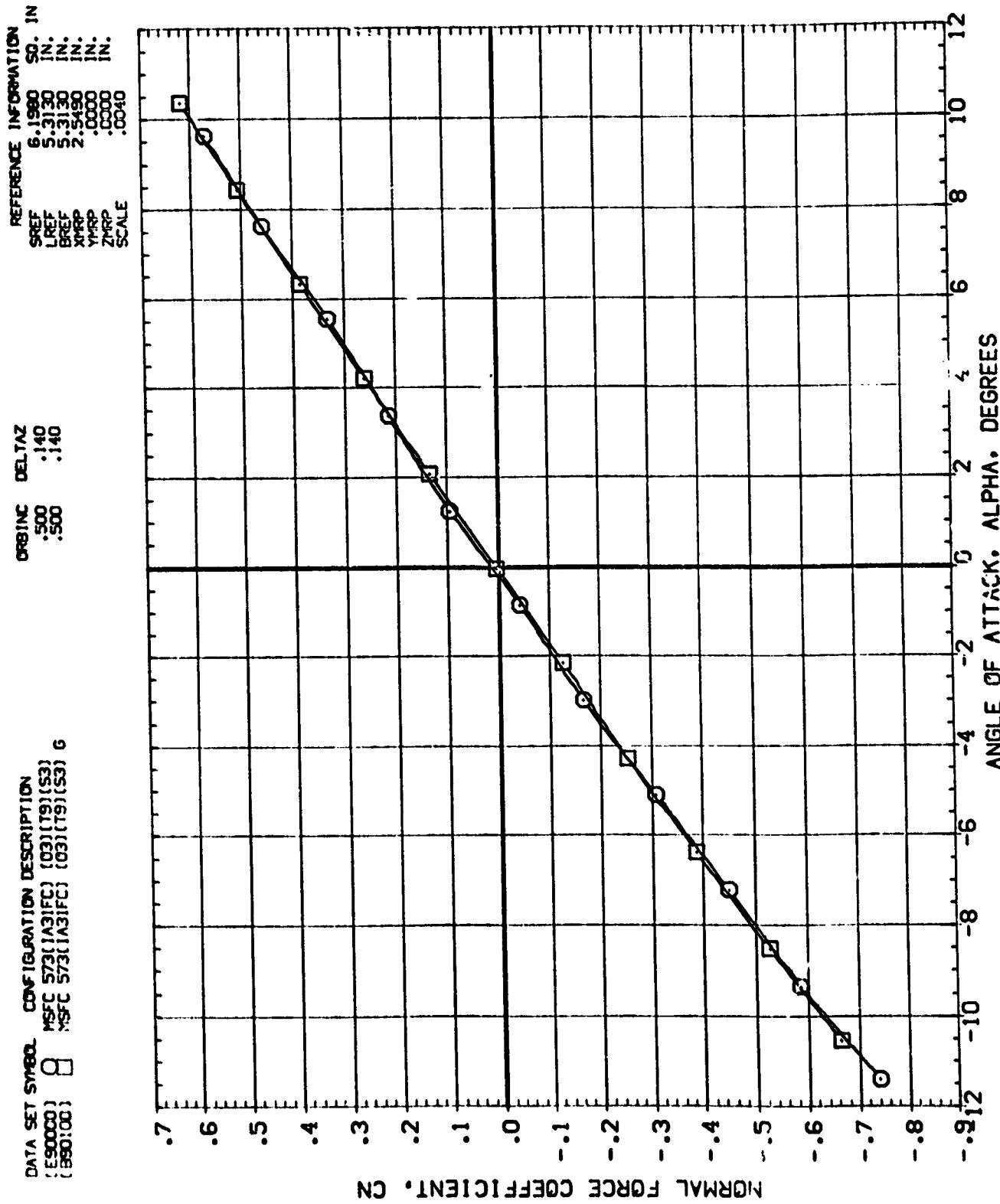


Figure 12. - Photograph of Basic Configuration With Different E.I. Nose Shapes.

## DATA FIGURES

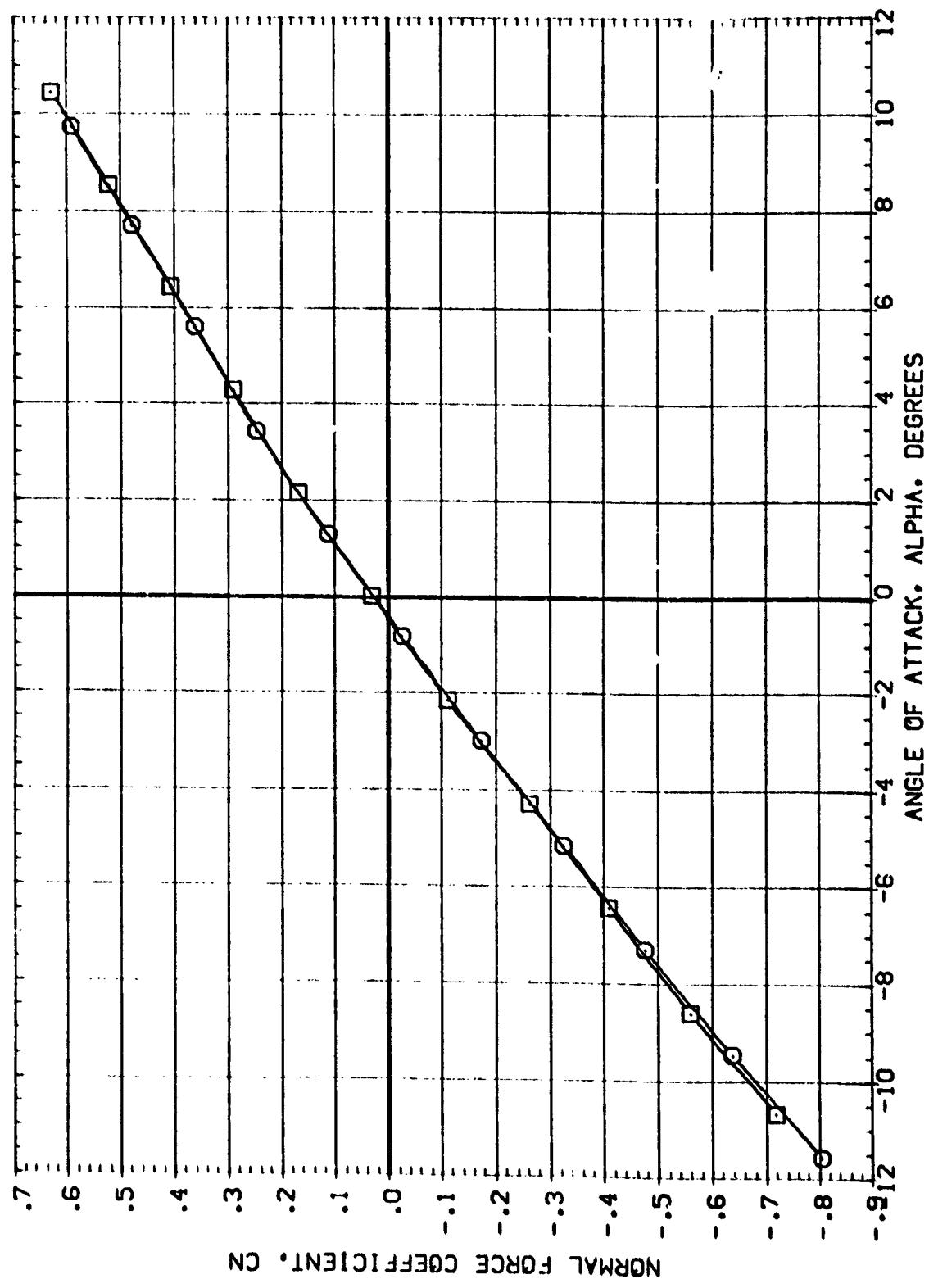
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(AJMACH = .90



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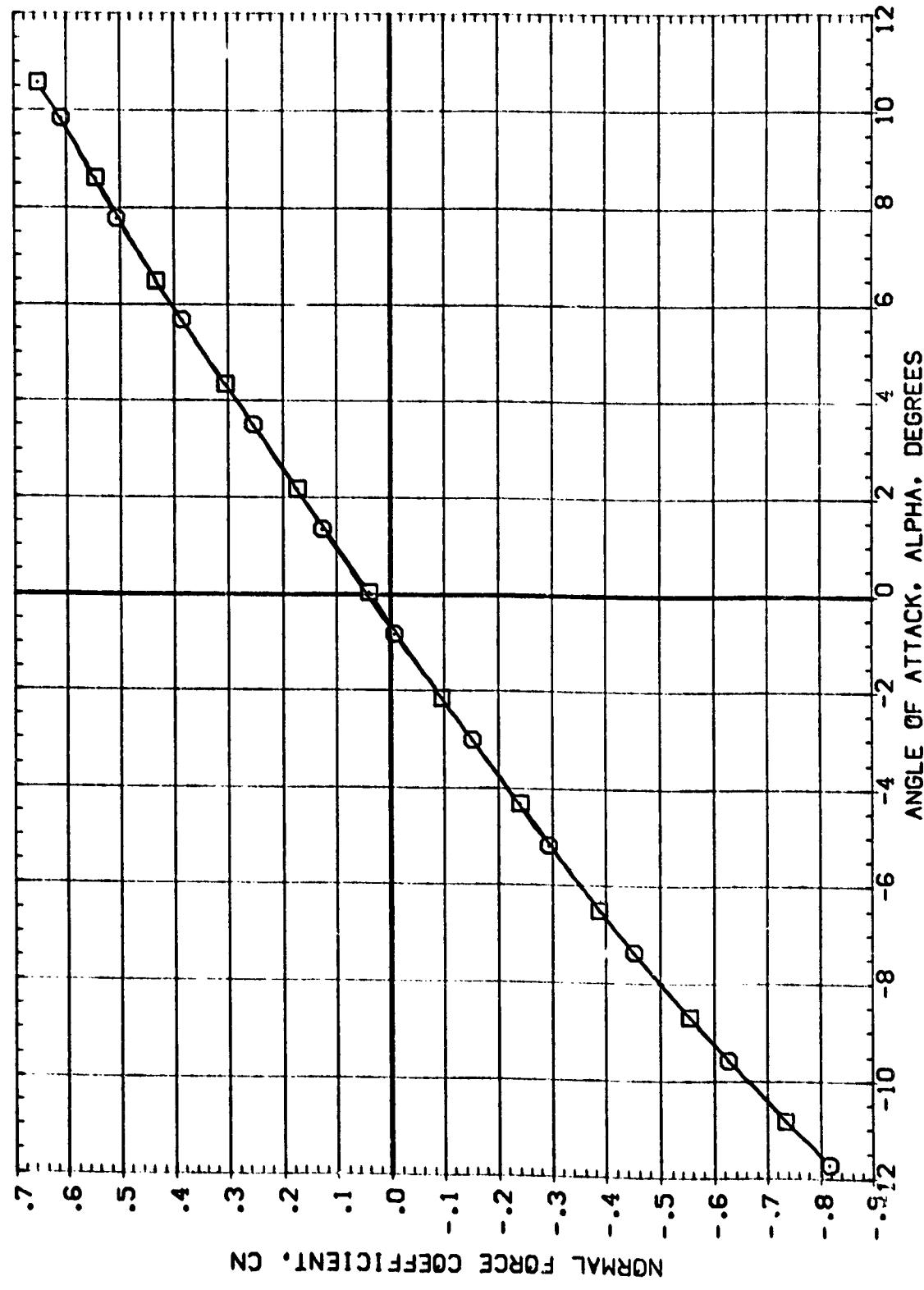
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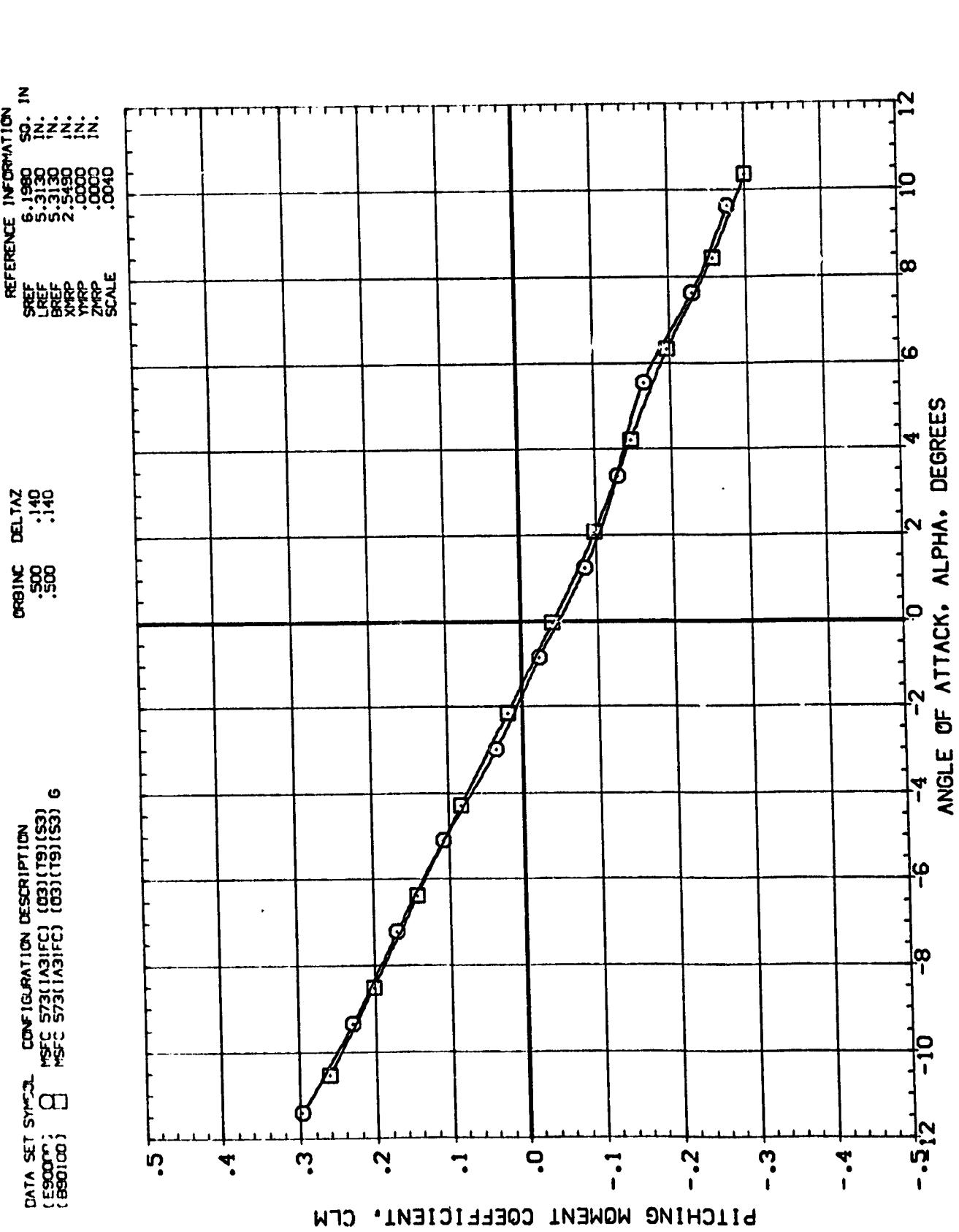
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 (890120)

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 BREF 5.3130 IN  
 XMRP 2.5120 IN  
 YMRP .0000 IN  
 ZMRP .0000 IN  
 SCALE .0000



EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS  
 $(C)MACH = 1.25$



EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS  
 $(\Delta MACH = .90)$

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DELTAZ  
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 .500 .140

REFERENCE INFORMATION  
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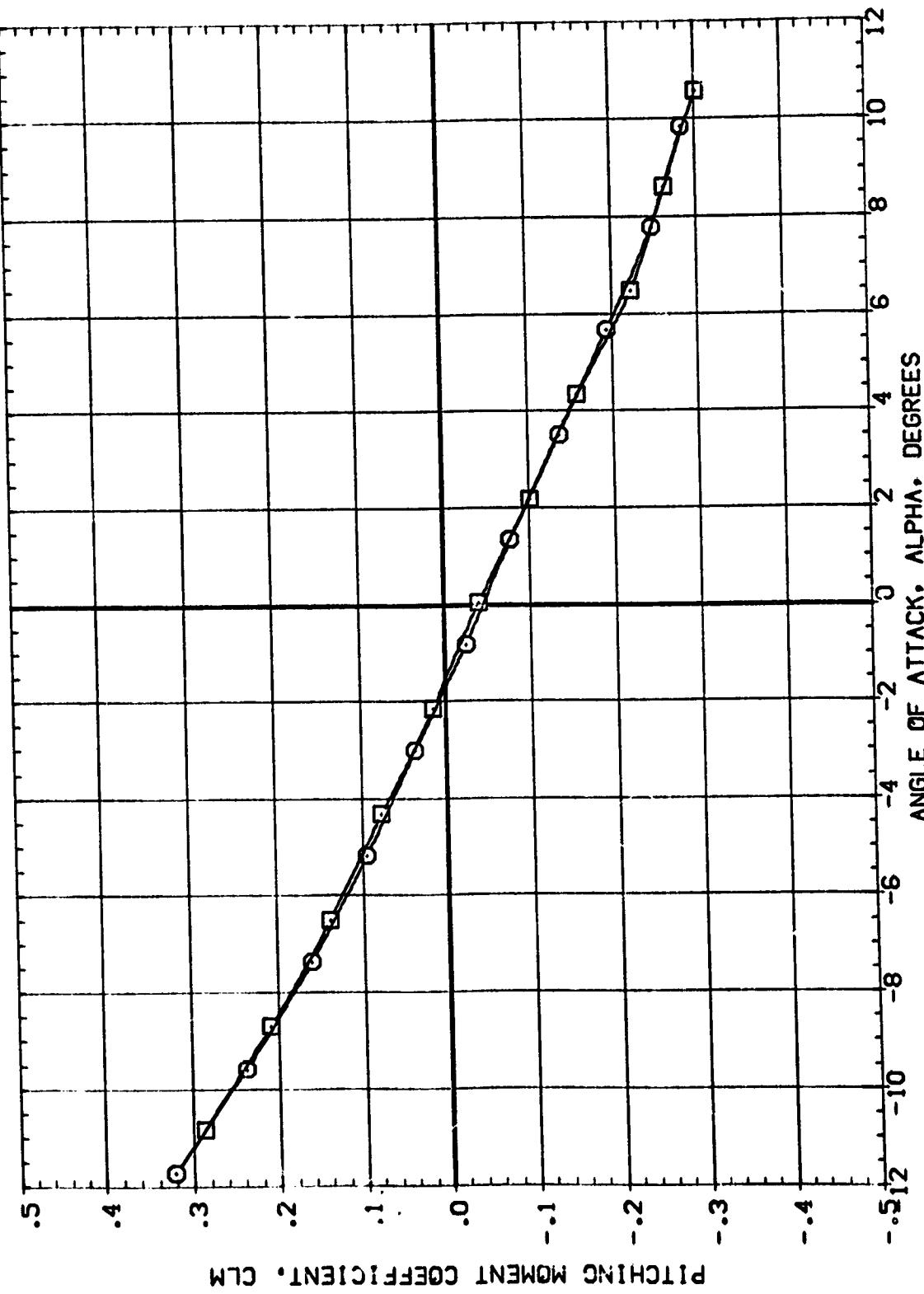
PITCHING MOMENT COEFFICIENT, CLM

ANGLE OF ATTACK, ALPHA, DEGREES  
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EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS  
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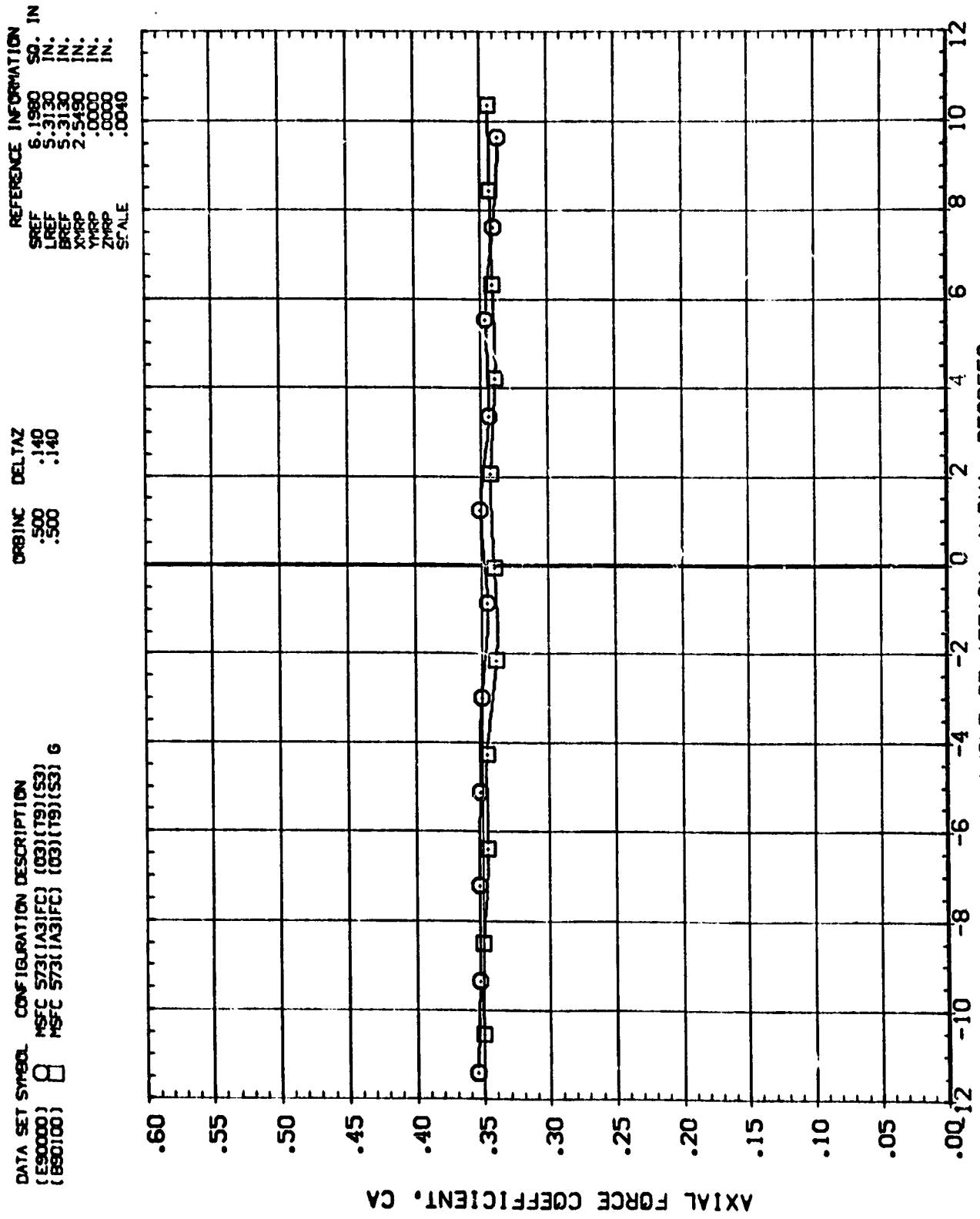
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 SCALE



### EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS

(C)<sub>MACH</sub> = 1.25

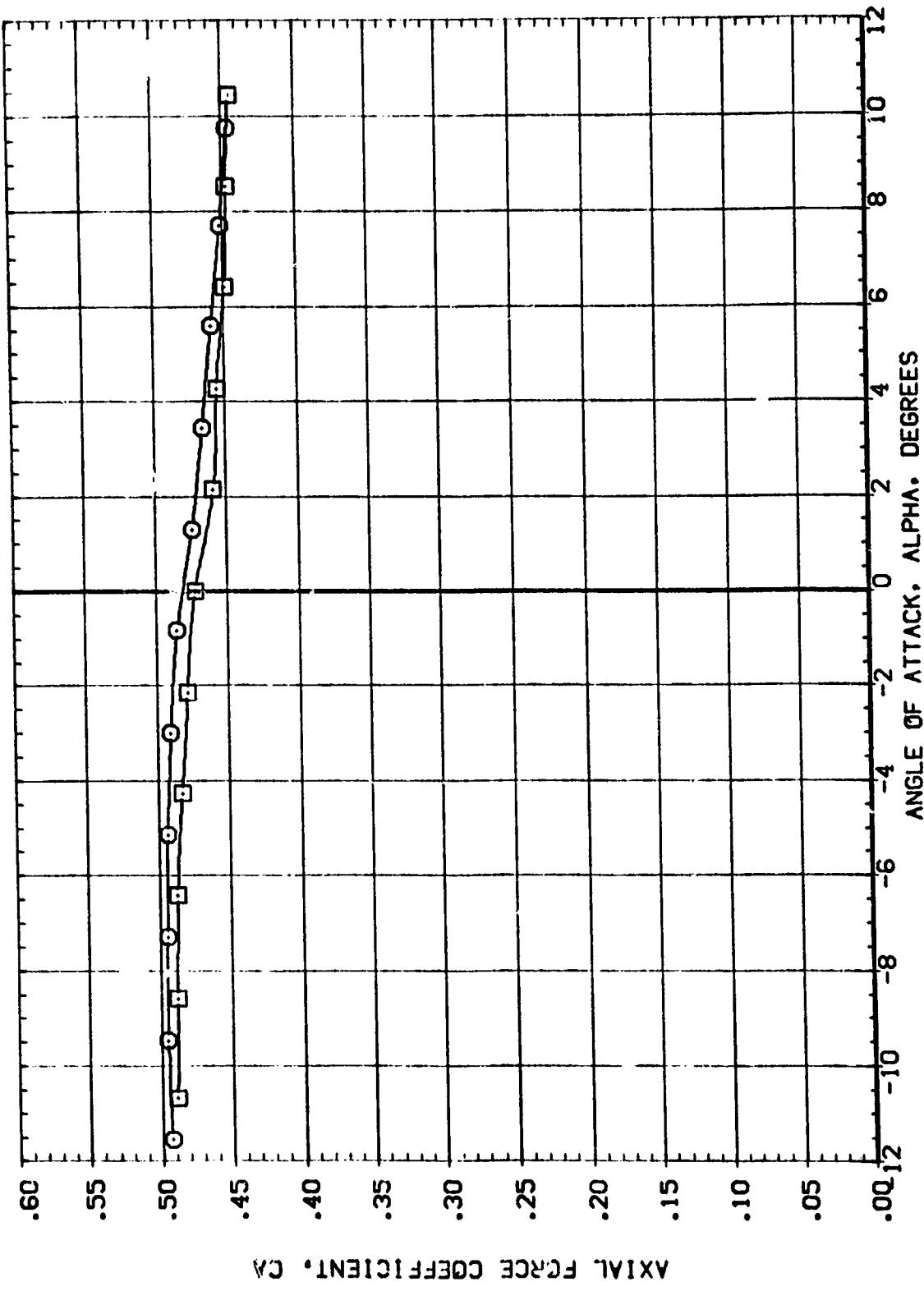
EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS  
 $(\text{MACH} = .90)$



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REFERENCE INFORMATION  
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YRP .0000 IN.  
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SCALE .0040

ORB INC DELTAZ  
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.500 .140

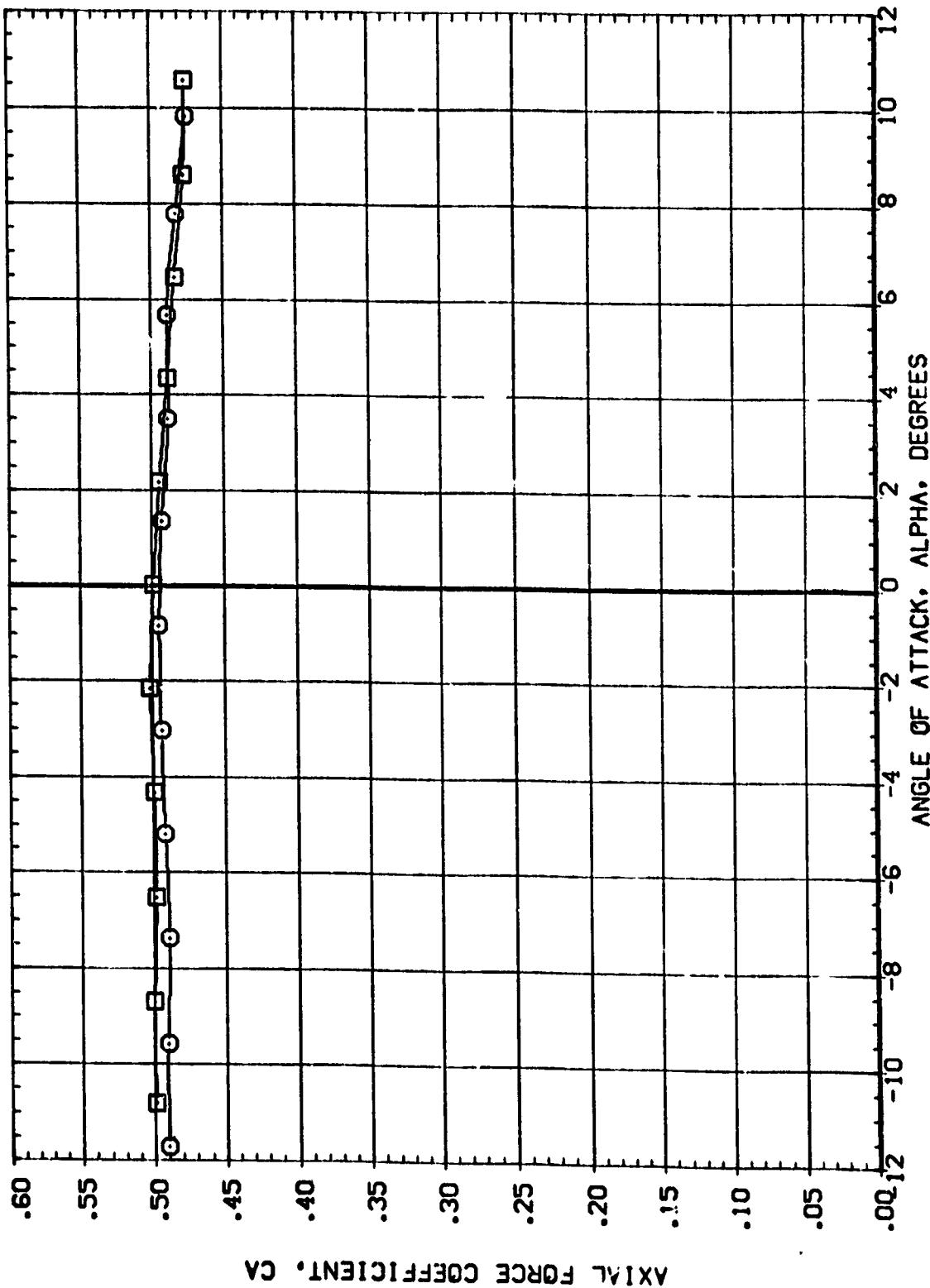


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[ES0100] NSFC 573(1A1FC) (G3)(T9)(S3)

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SCALE .0040



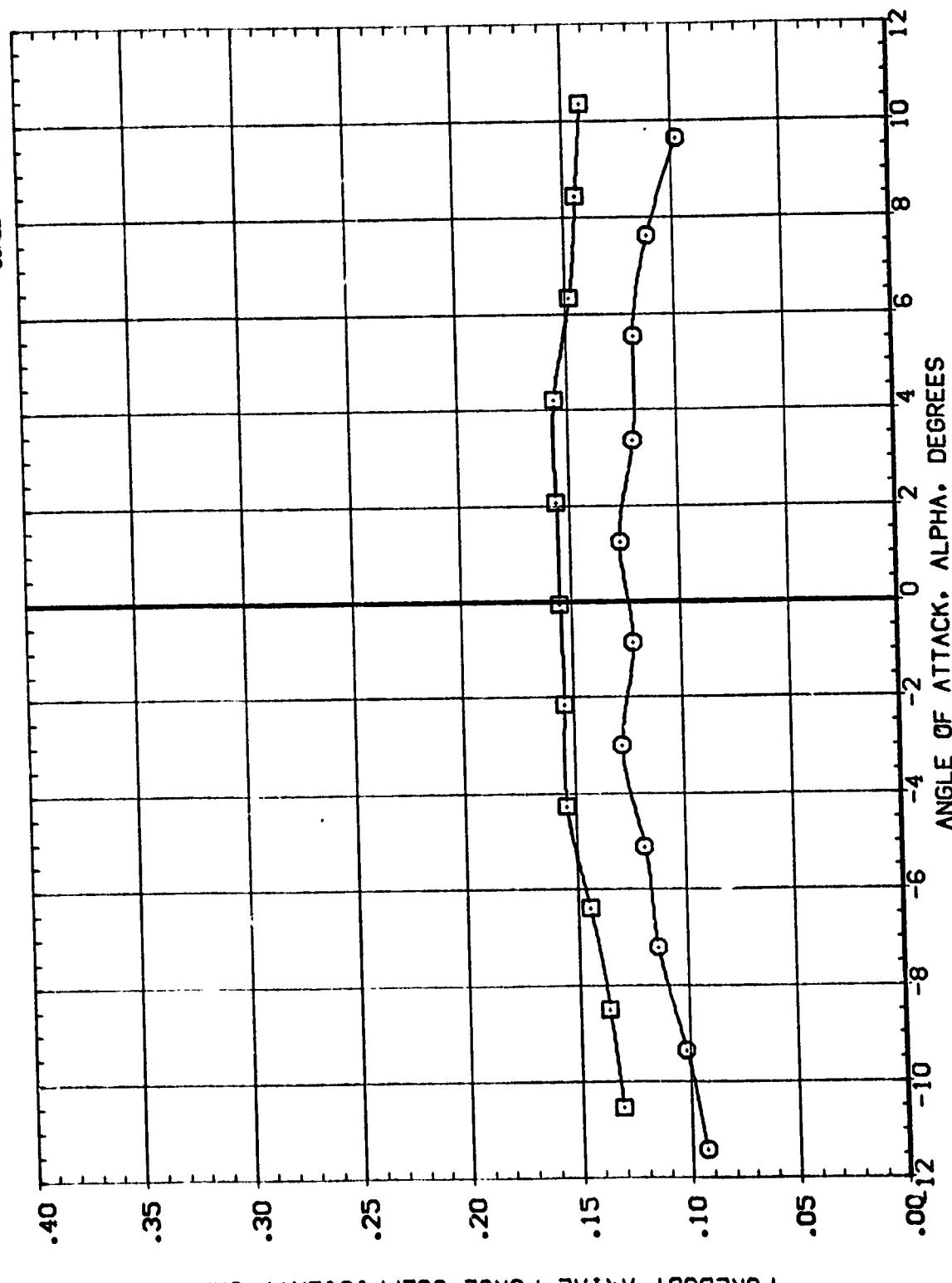
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REFERENCE INFORMATION  
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XREF 2.5490 IN.  
YREF .0000 IN.  
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SCALE

ORBIT INC DELTA Z  
.500 .140  
.500 .140



FORCEBODY AXIAL FORCE COEFFICIENT, CAF

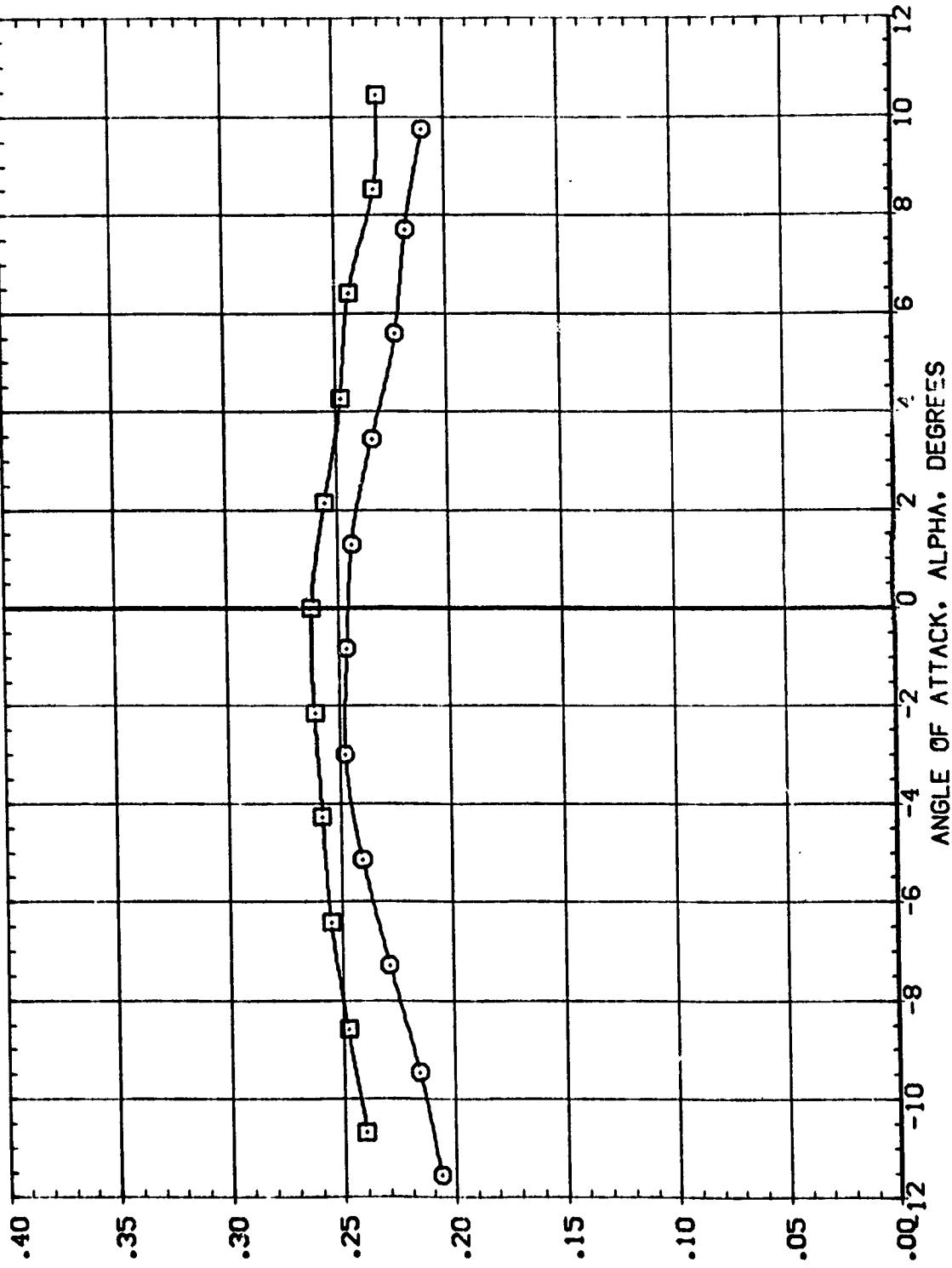
EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS

(MACH = .90

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DRINC .500 .140  
.500 .140



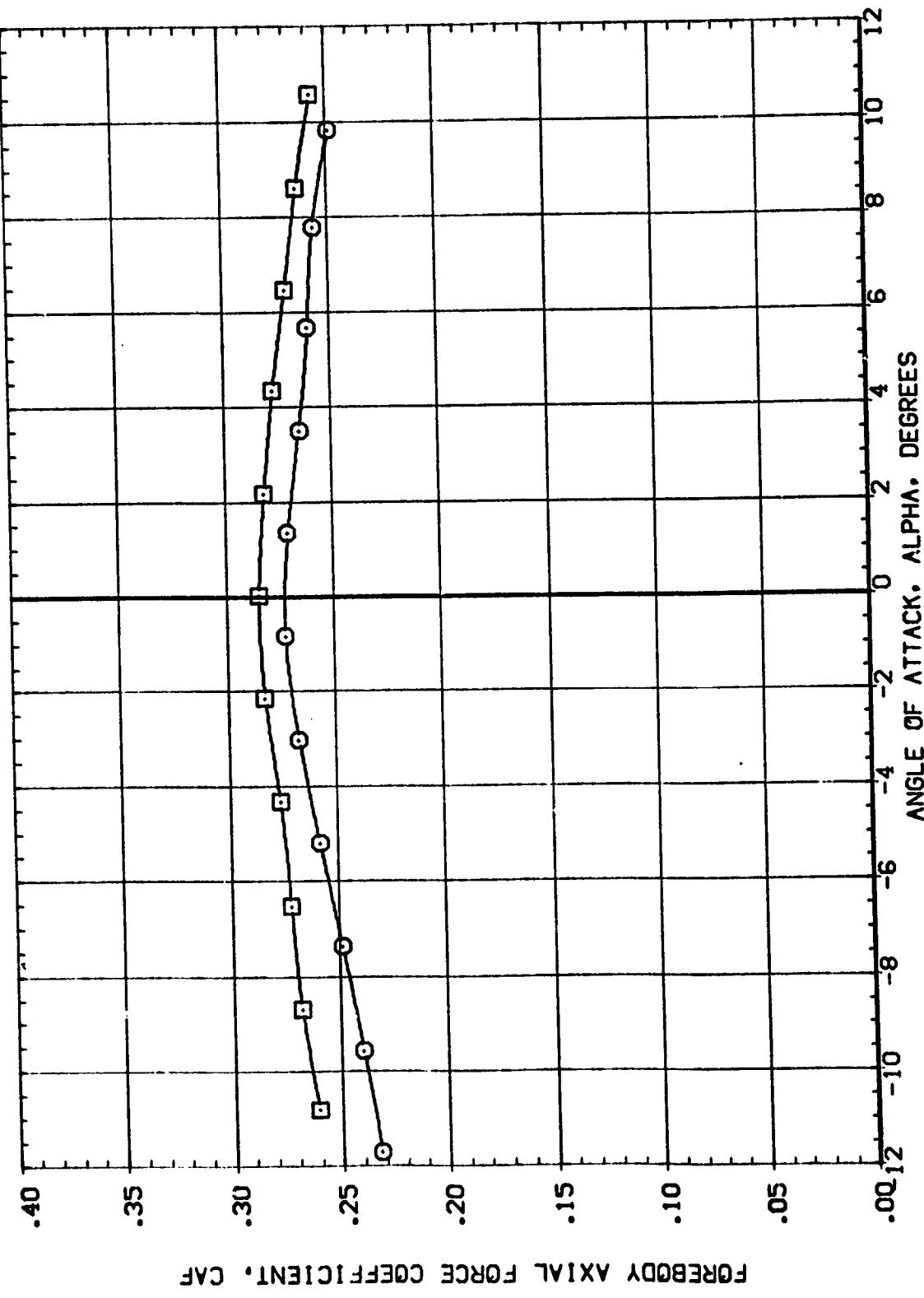
FOREBODY AXIAL FORCE COEFFICIENT, CAF

EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

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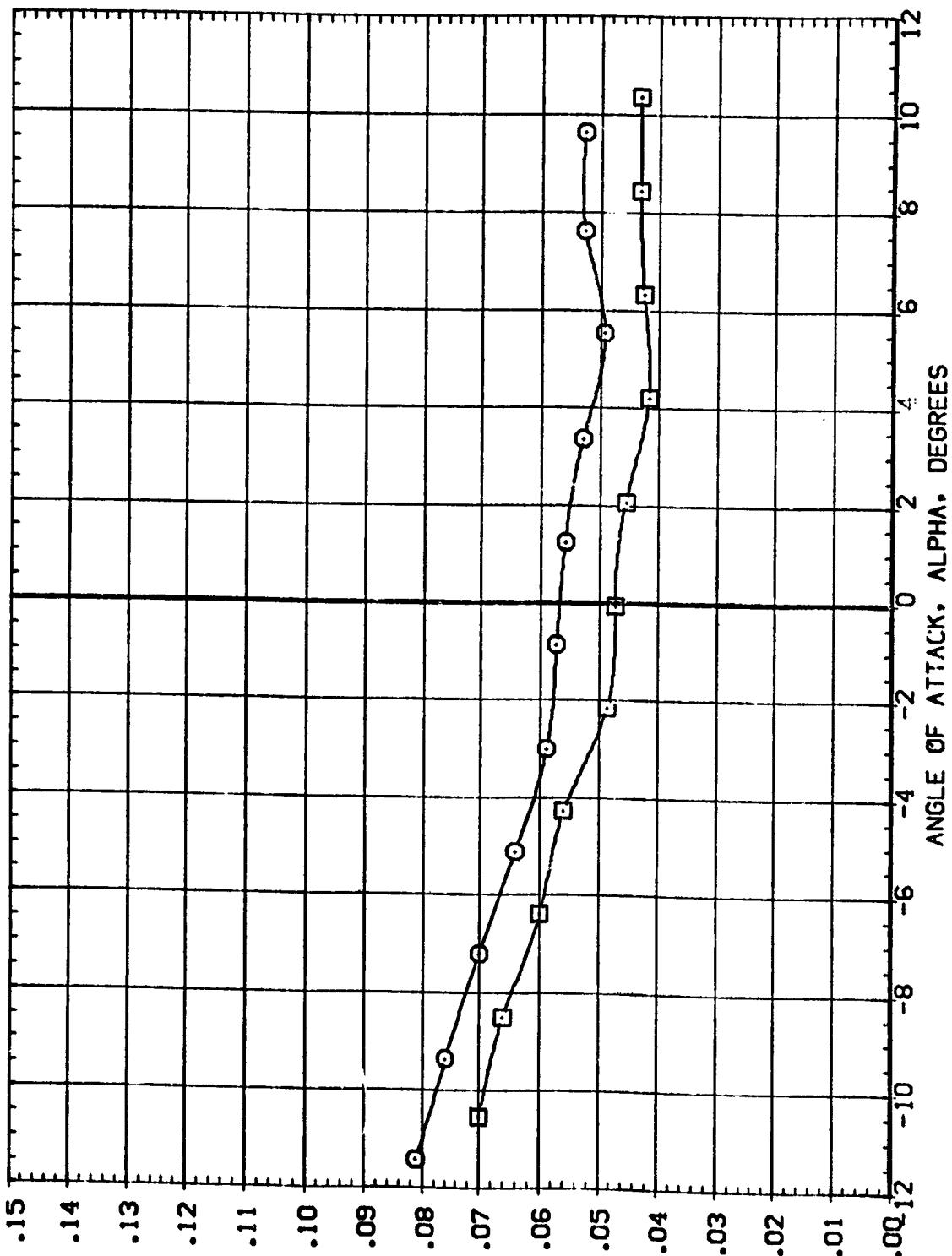
EFFECT OF GRI ON AERODYNAMIC CHARACTERISTICS

(C)<sub>MACH</sub> = 1.25

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 (B8C0100) NSFC 573(IA3)FC (G3)(T9)(S3) G

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 (.500) .140 .140 .500 .0000 .0000 .0040

EXTERNAL TANK AXIAL FORCE COEFFICIENT, CABE

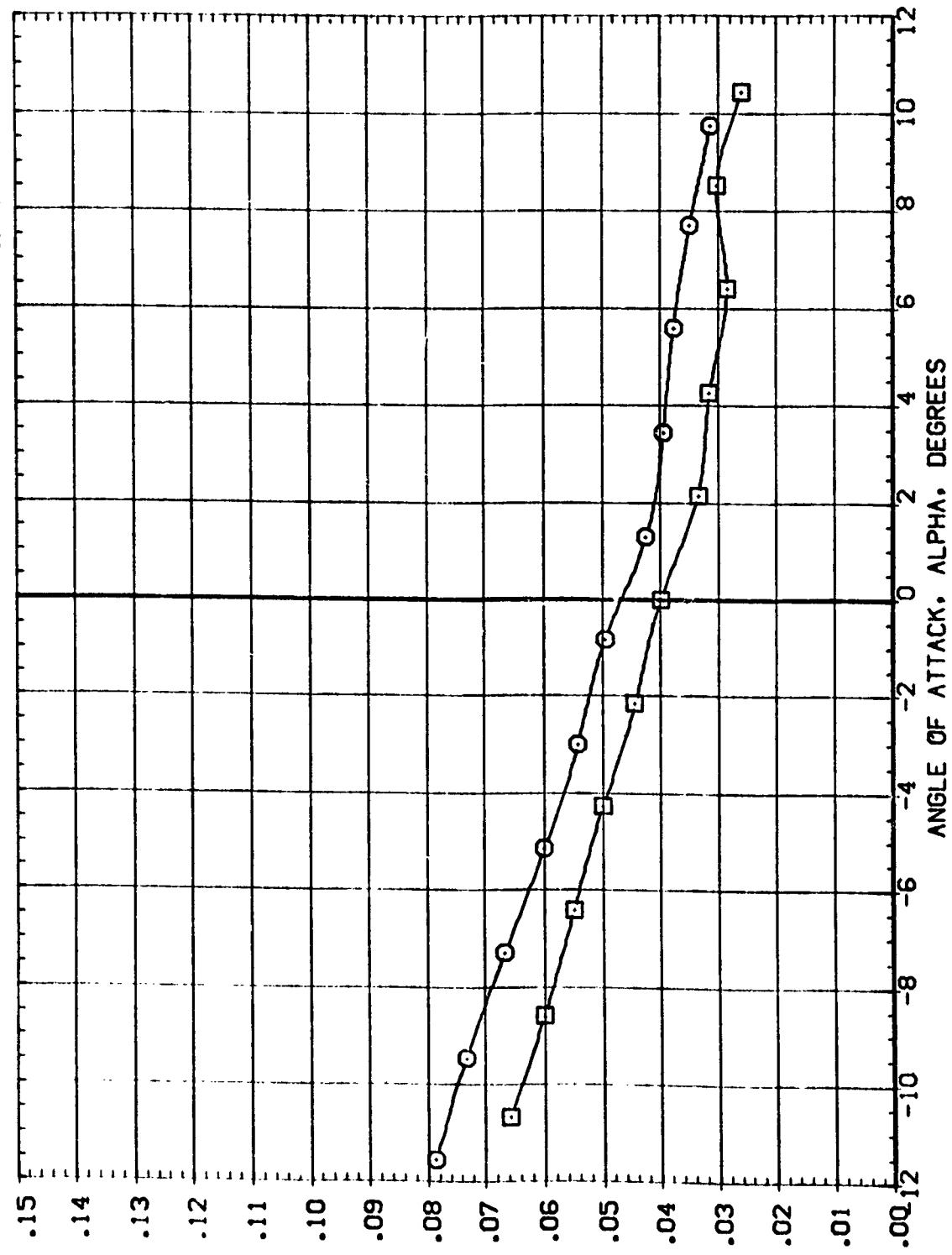


EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS  
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(SEC:CD) **□** MSEC 573(1A3FC) (03)(19)(53) 6

REFERENCE INFORMATION  
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LREF 5.3130 IN.  
BREF 5.3130 IN.  
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YRP .0000 IN.  
ZRP .0000 IN.  
SCALE .0040

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.500 .140

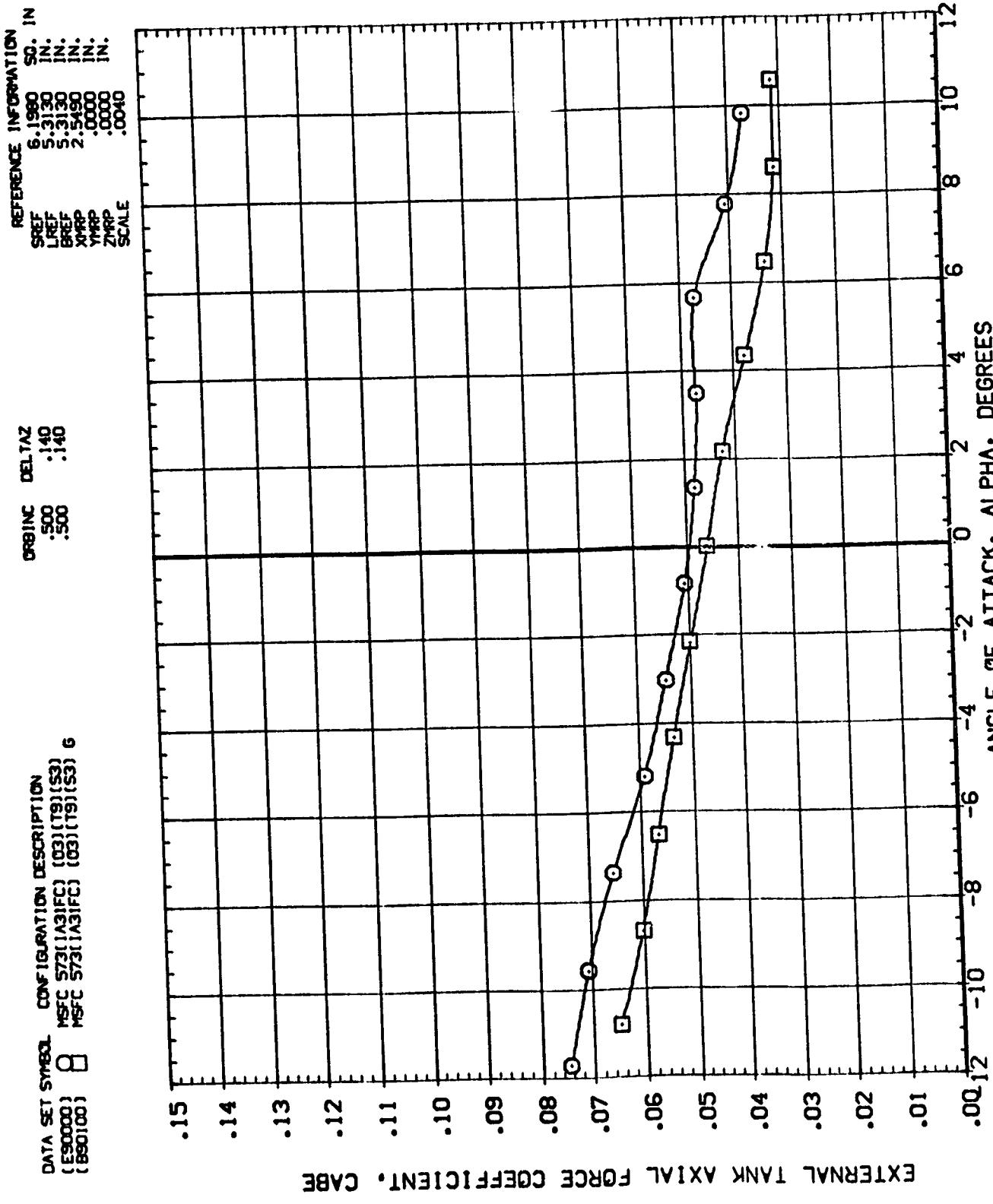


EXTERNAL TANK AXIAL FORCE COEFFICIENT, CABE

EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

PAGE 14

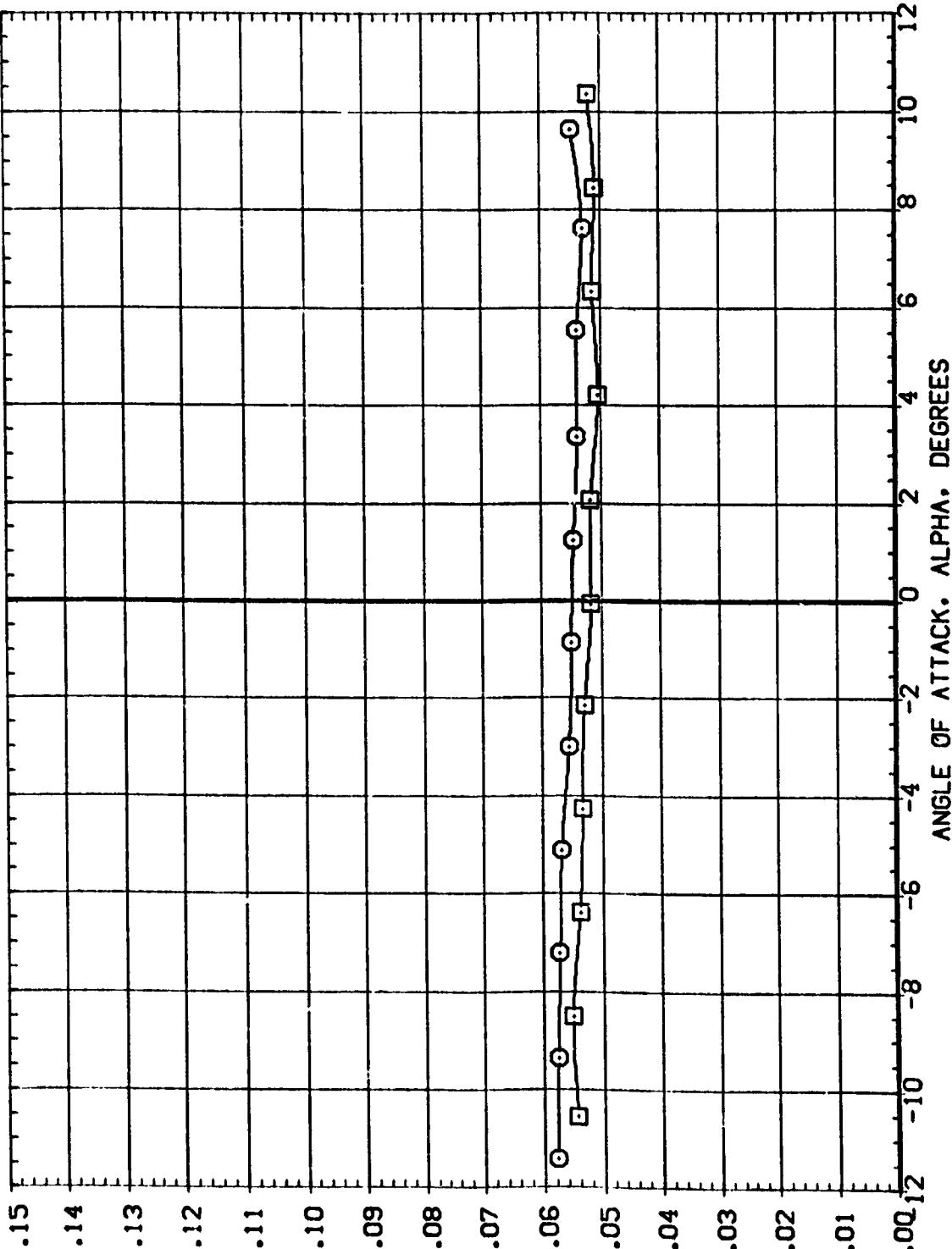


EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS  
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ORB INC DELTAZ  
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REFERENCE INFORMATION  
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PREF 5.3130 IN.  
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YMRP .0000 IN.  
ZMRP .0000 IN.  
SCALE .0040



ORBITER AXIAL FORCE COEFFICIENT, CAB0

### EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS

( $\Delta$ )MACH = .90

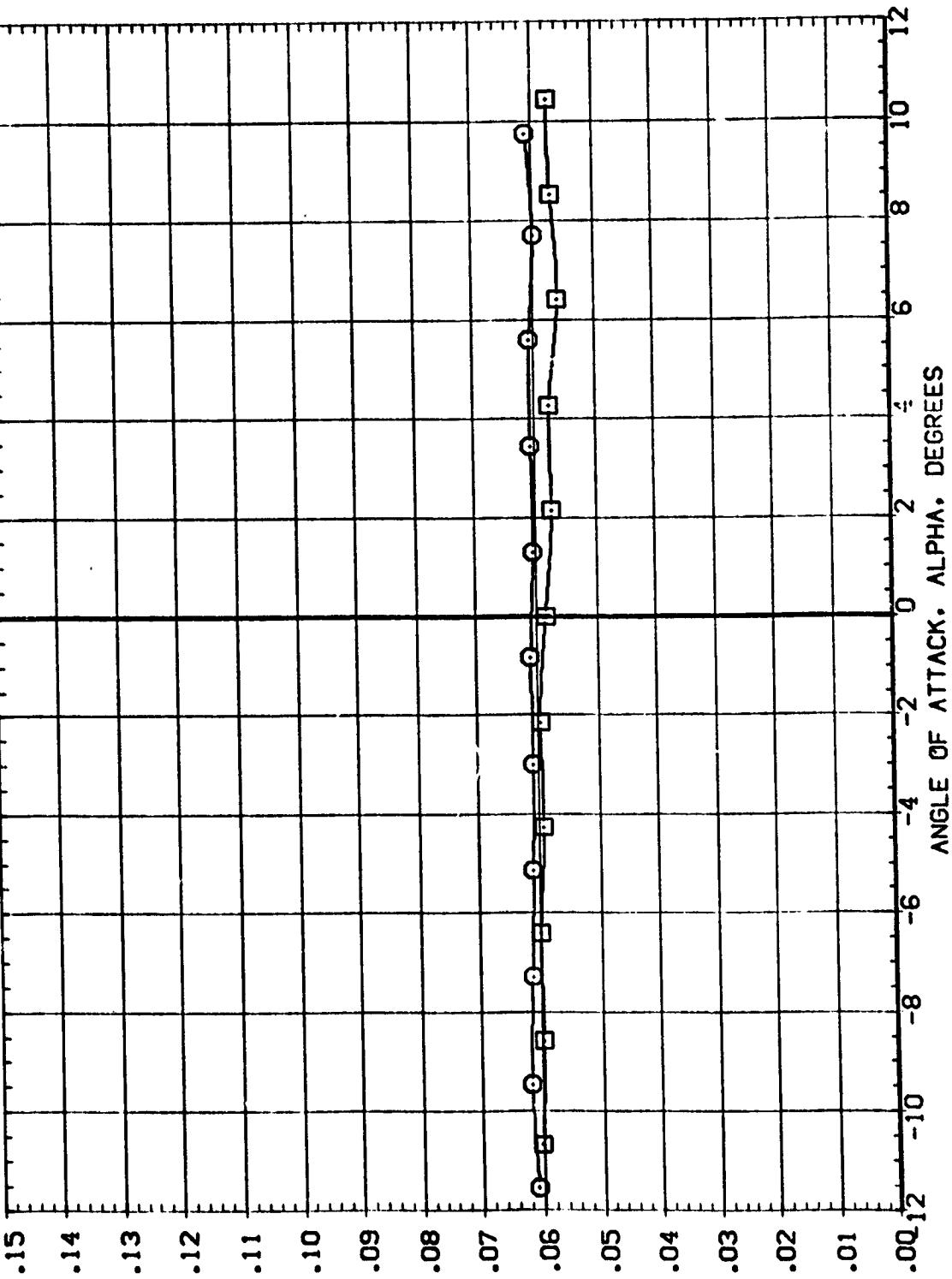
PAGE 16

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.500      .140

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SCALE      .0010

ORBITER AXIAL FORCE COEFFICIENT, CABO



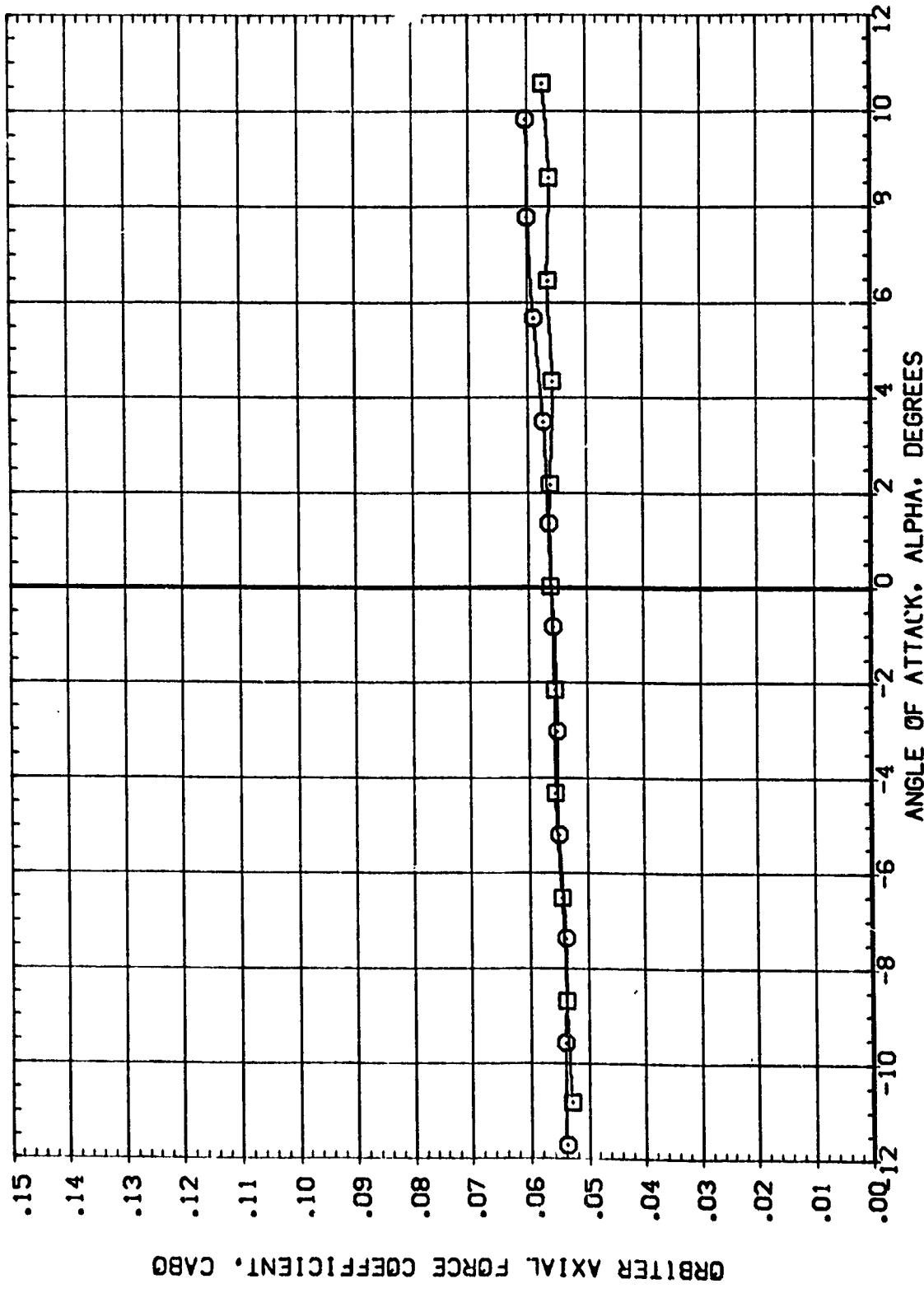
EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS

(BJMACH = 1.05

PAGE 17

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
(ES0000) NSFC 573(A3)FC (03)(T9)(S2)  
(BS0000) NSFC 573(A3)FC (03)(T9)(S3)

REFERENCE INFORMATION  
SREF 6.1980 SD. IN  
LREF 5.3130 IN.  
BREF 2.5490 IN.  
XMRP .0000 IN.  
YMRP .0000 IN.  
ZMRP .0040 IN.  
SCALE



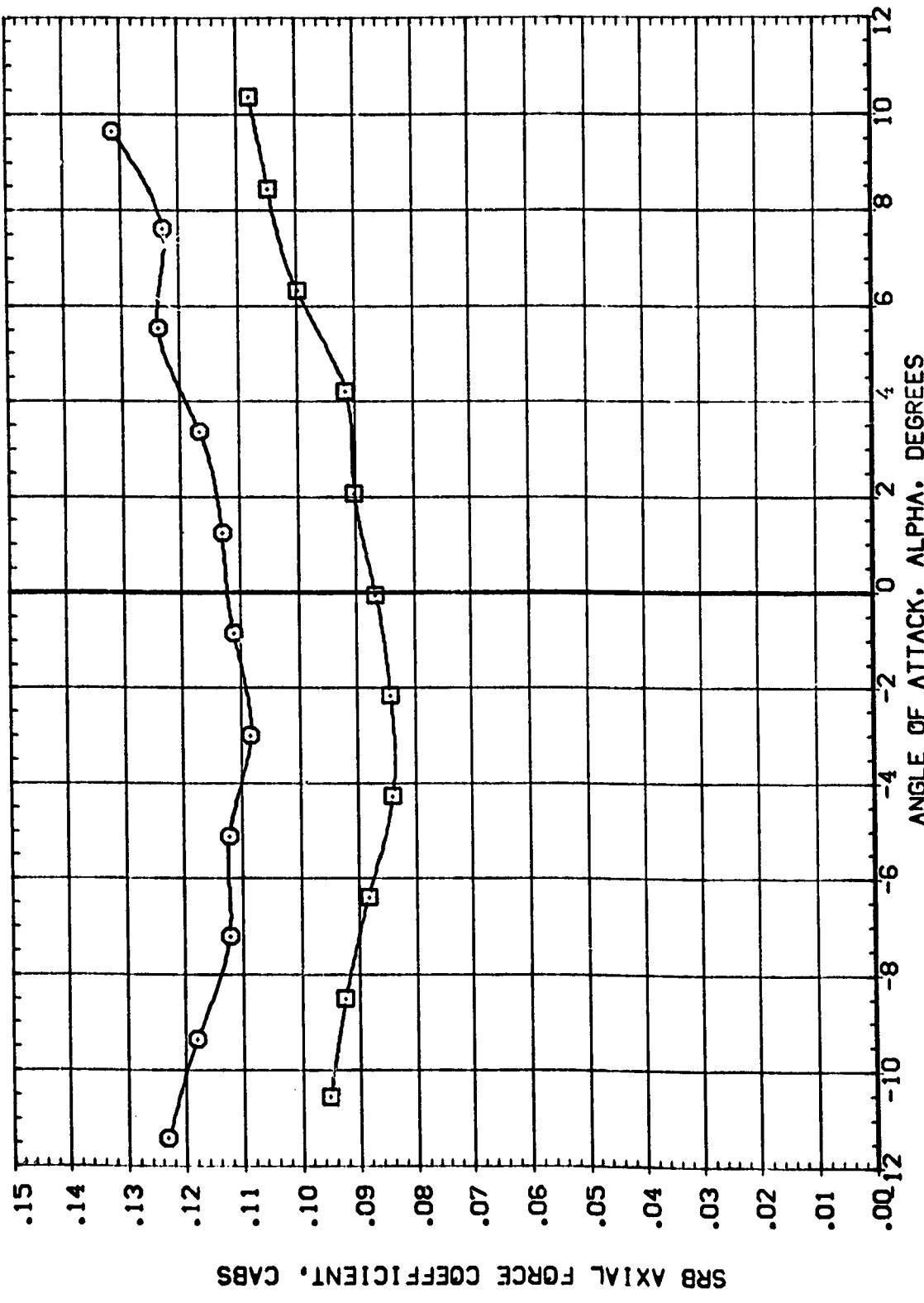
EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

PAGE 18

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
[E90000] 8 MSFC S73(A3)FC [G3][T9][S3] 6  
[E90100] 0 MSFC S73(A3)FC [G3][T9][S3] G

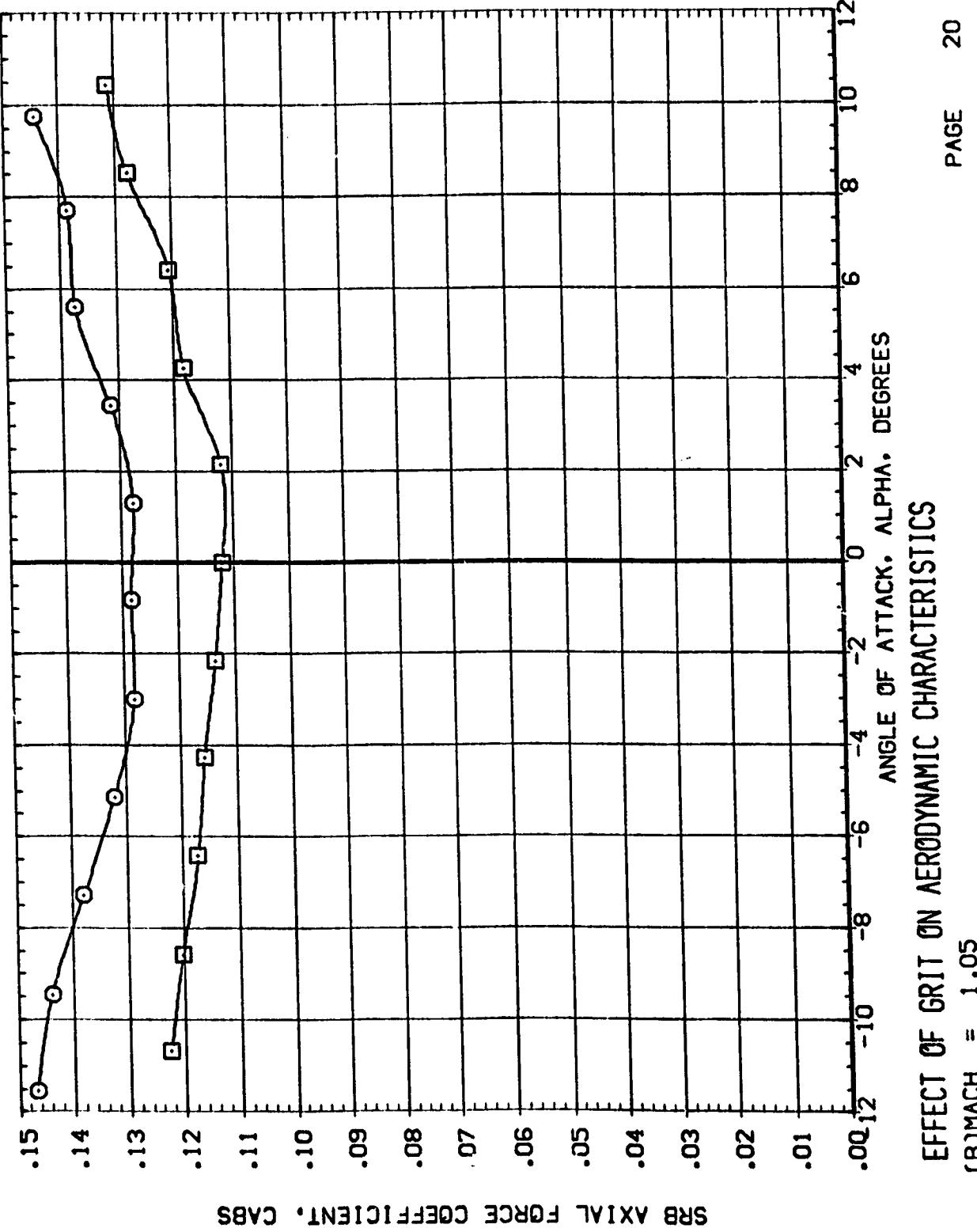
REFERENCE INFORMATION  
SREF 6.1980 SQ. IN.  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
XRP 2.5450 IN.  
YRP .0000 IN.  
ZRP .0040 IN.  
SCALE



EFFECT OF GRIT ON AERODYNAMIC CHARACTERISTICS  
(MACH = .90)

DATA SET SMCB  
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(E90100) [ ] MSFC 573(I A31FC) (03)(T9)(S3) 6

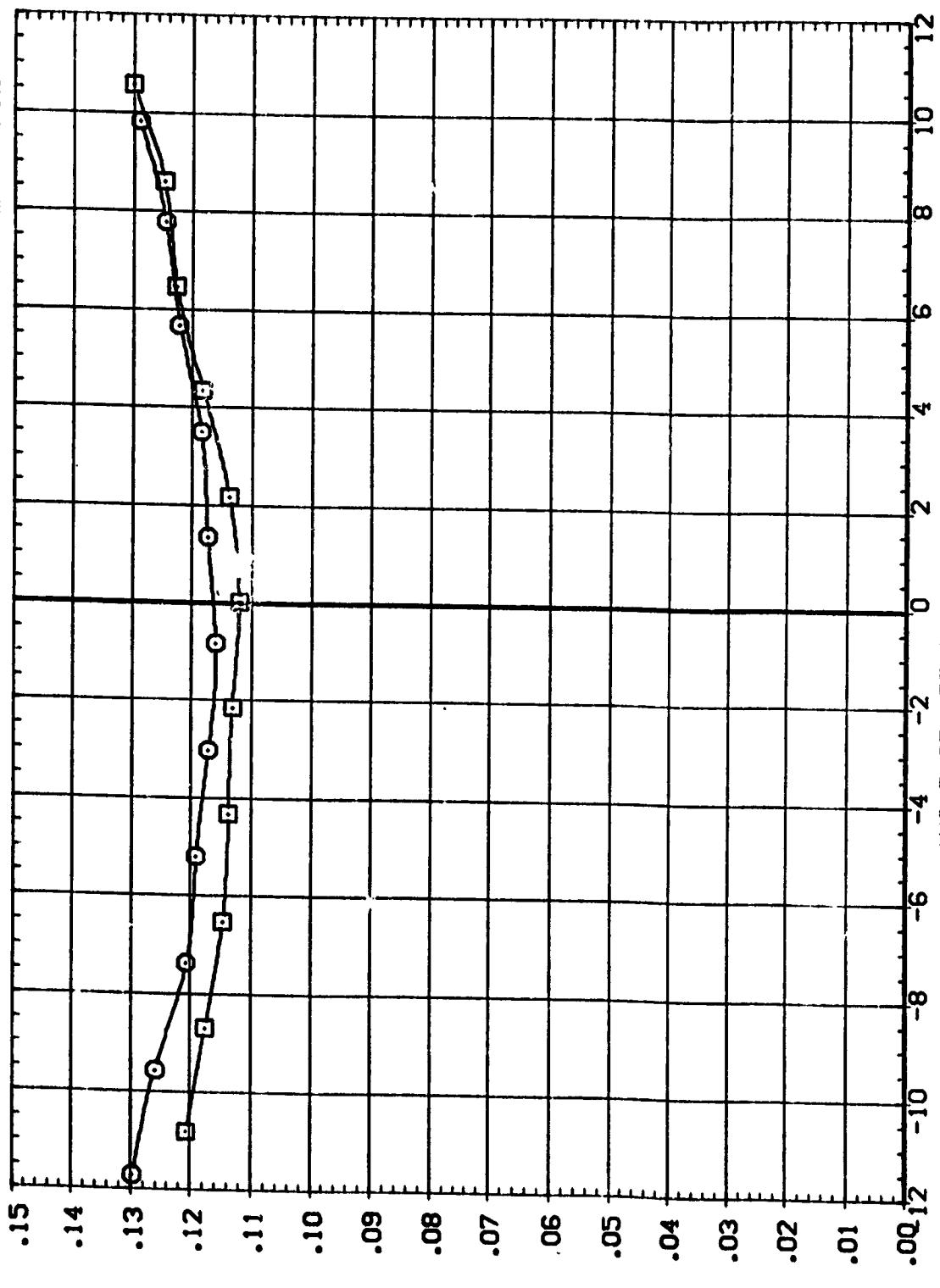
REFERENCE INFORMATION  
SREF 6.1980 SQ. IN.  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
XRP 2.5150 IN.  
YRP .0000 IN.  
ZRP .0000 IN.  
SCALE .0010



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
(E90000) □ MSFC 573(1A3)FC (G3)(19)(S3) G  
(E90100) ○ MSFC 573(1A3)FC (G3)(19)(S3)

ORB INC DELTAZ  
.500 .140  
.500 .140

REFERENCE INFORMATION  
SREF 6.190 SD. IN  
LREF 5.310 IN.  
BREF 5.330 IN.  
XMRP 2.5490 IN.  
YMRP 0.0000 IN.  
ZMRP 0.0040 IN.  
SCALE



SRB AXIAL FORCE COEFFICIENT, CABs

EFFECT OF GRII ON AERODYNAMIC CHARACTERISTICS  
(C)MACH = 1.25

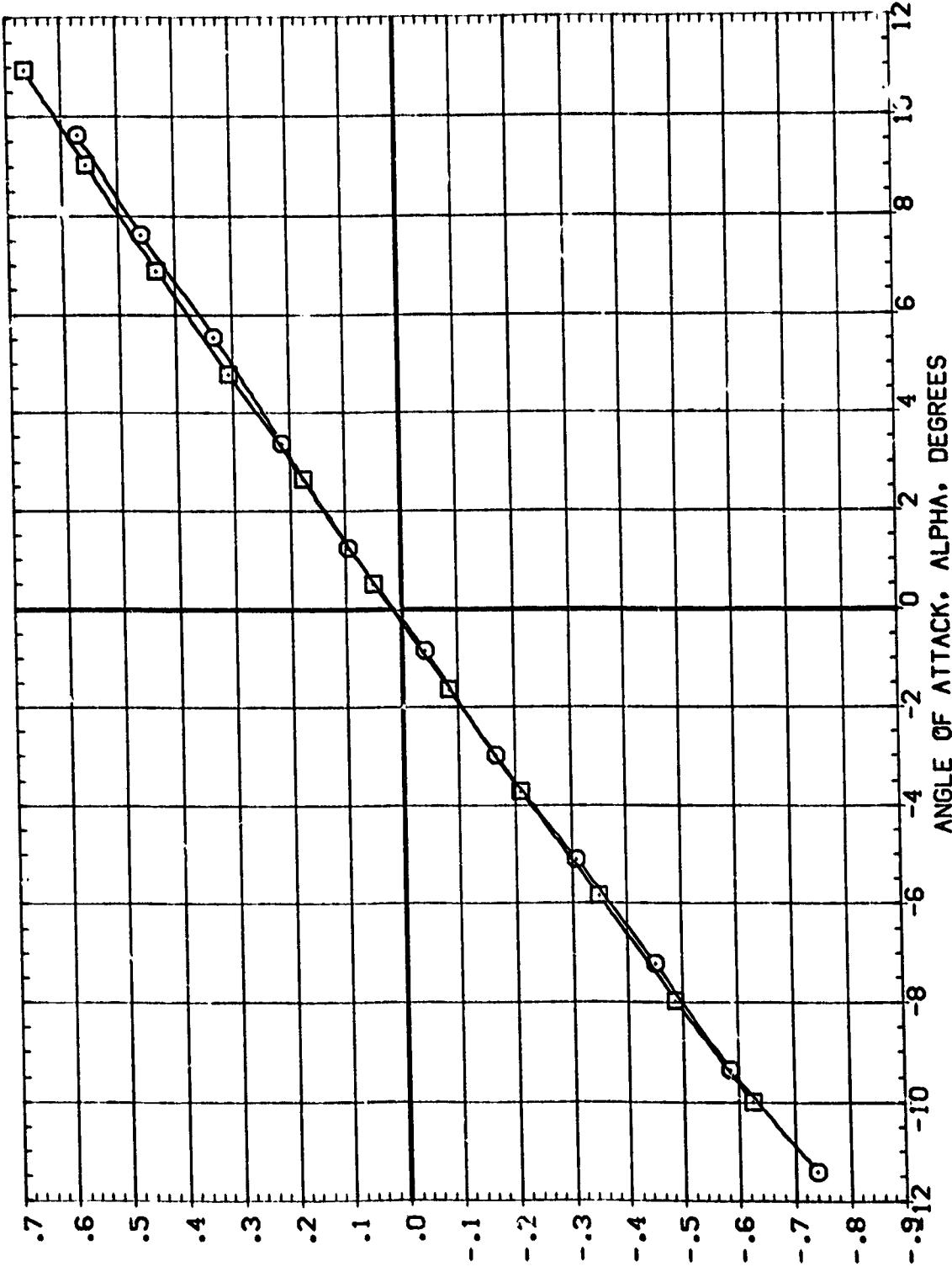
REF ID: A690000

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
8500000 MSFC 5731(131FC) (03)(19)(S3)  
(285200) MSFC 5731(131FC) (03)(19)(S3) ORB. MISLAND.  
SRFP LREF BREF XMRP YMRP ZMRP  
SCALE .0040

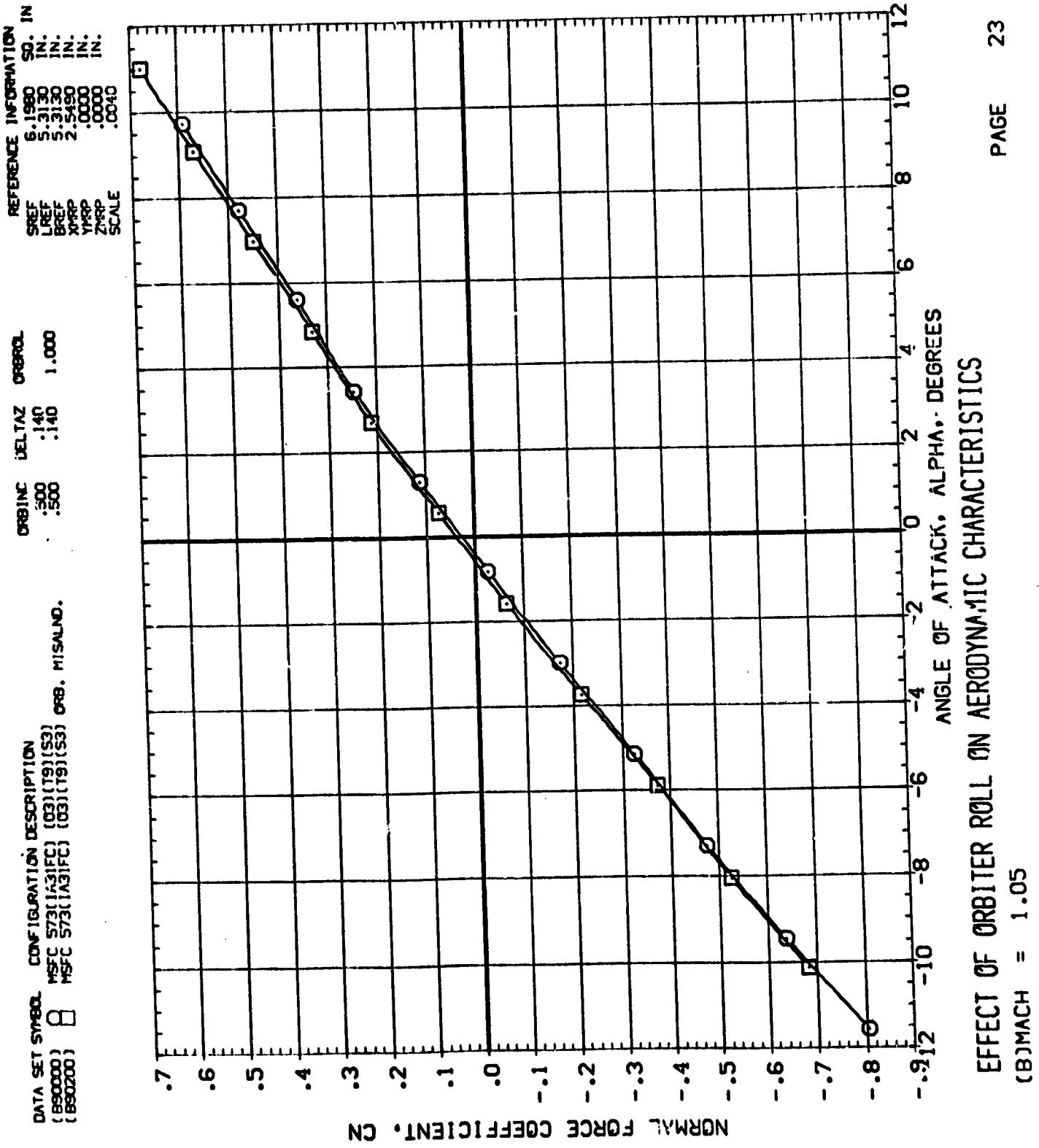
ORB INC DELTAZ GROLL  
.500 .140 1.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
8500000 MSFC 5731(131FC) (03)(19)(S3)  
(285200) MSFC 5731(131FC) (03)(19)(S3) ORB. MISLAND.

NORMAL FORCE COEFFICIENT, CN

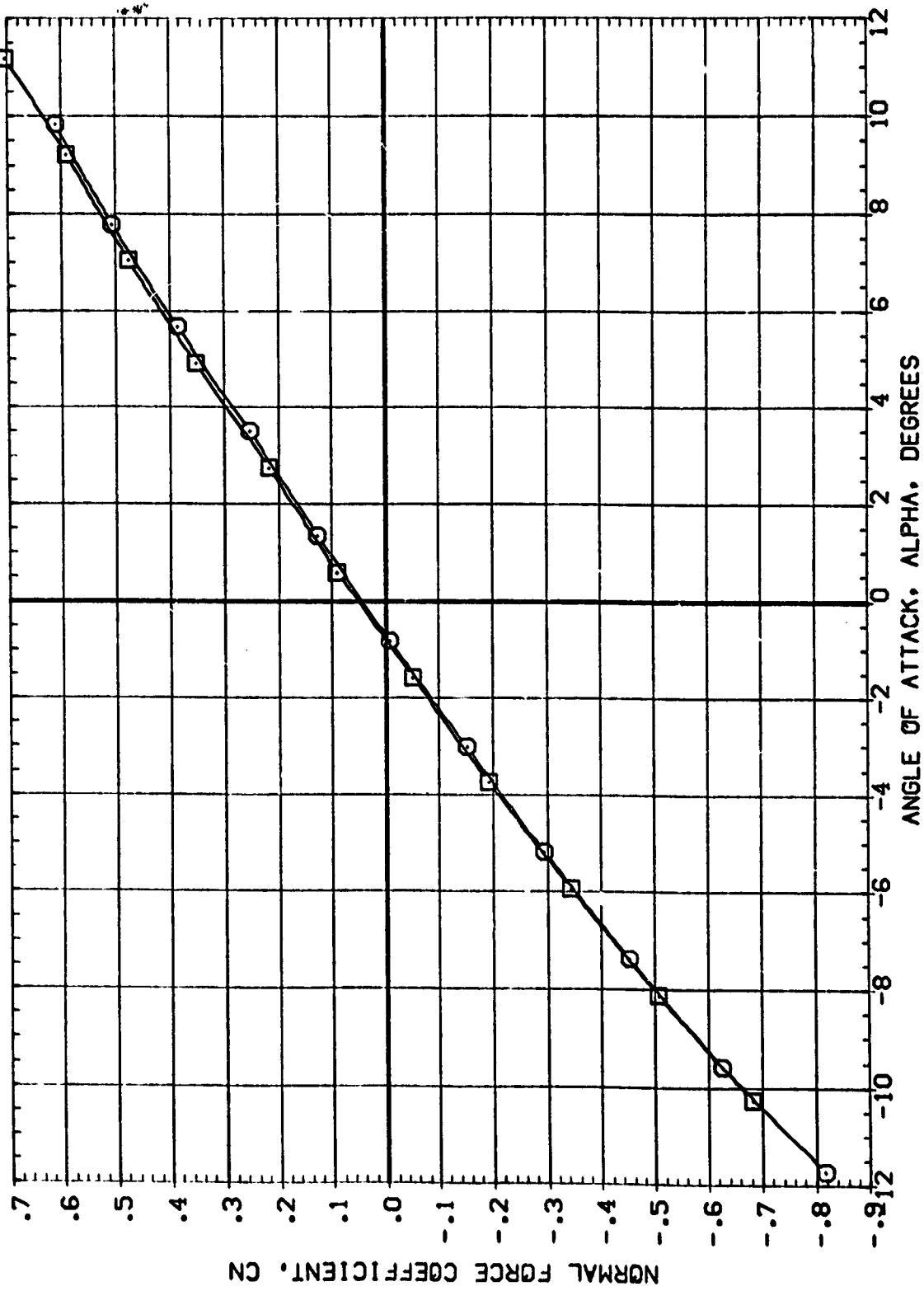


EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
(AJMACH = .90)



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
(890000) 8 NSFC 573(1A3IFC) (03)(T9)(S2)  
(890200) BEO200 MISALND.

REFERENCE INFORMATION  
SREF 6.1980 SC. IN  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
XHPP 2.5490 IN.  
YHPP .0000 IN.  
ZHPP .0000 IN.  
SCALE .0040



### EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

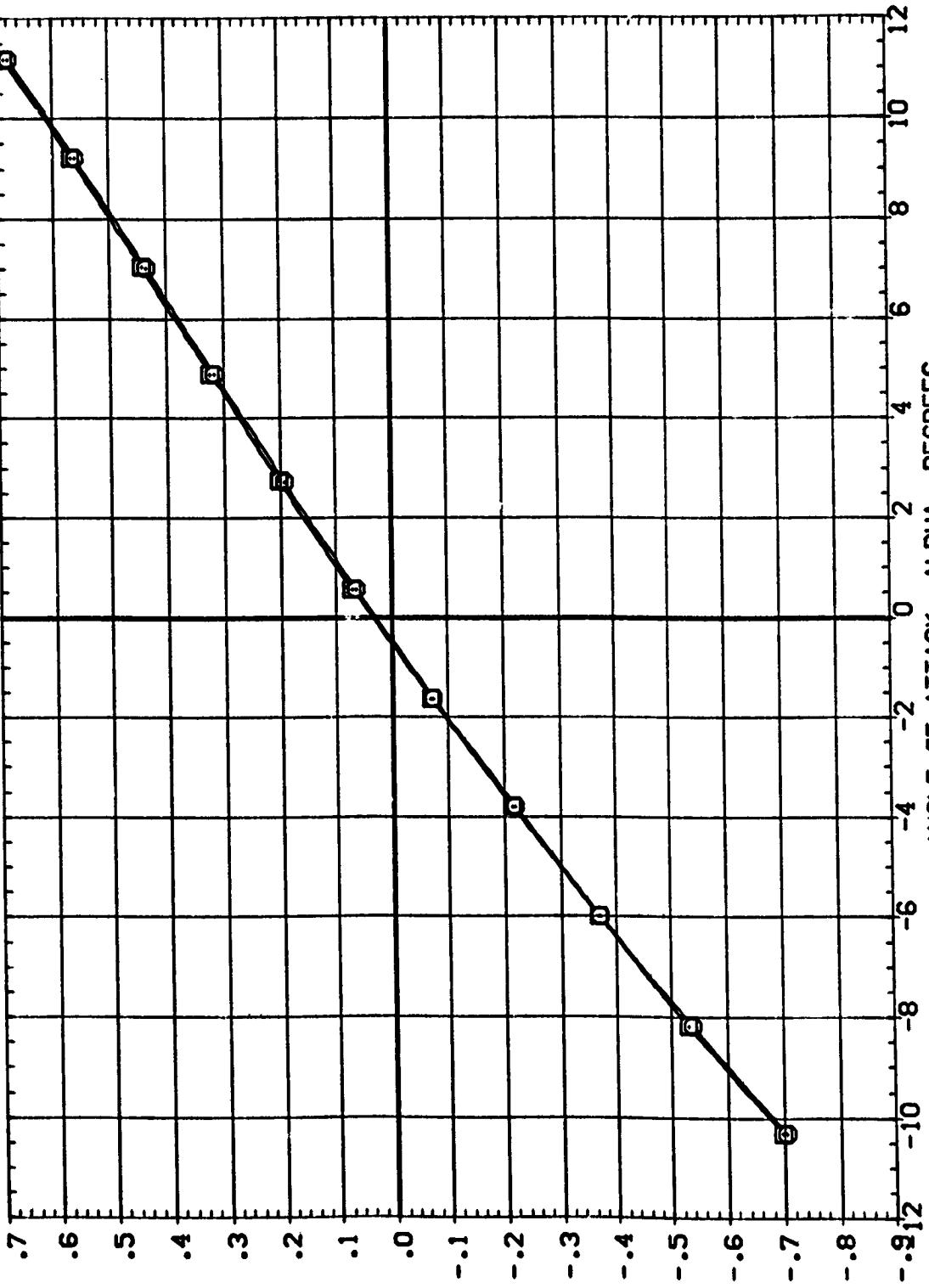
(C)MACH = 1.25

DATA SET S.190. CONFIGURATION DESCRIPTION  
 (B90000) 8 MSFC 573(A31FD) (T3)(T9)(S3)  
 (B90200) MSFC 573(A31FD) (T3)(T9)(S3) ORB. MISLAND.

ORB. INC. DELTAZ ORBZL.  
 .500 :140 1.000  
 .500 :140 1.000

REFERENCE INFORMATION  
 SPREF 6.1980 SO. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XWRF 2.5490 IN.  
 YWRF .0000 IN.  
 ZWRF .0040 IN.

NORMAL FORCE COEFFICIENT. CN



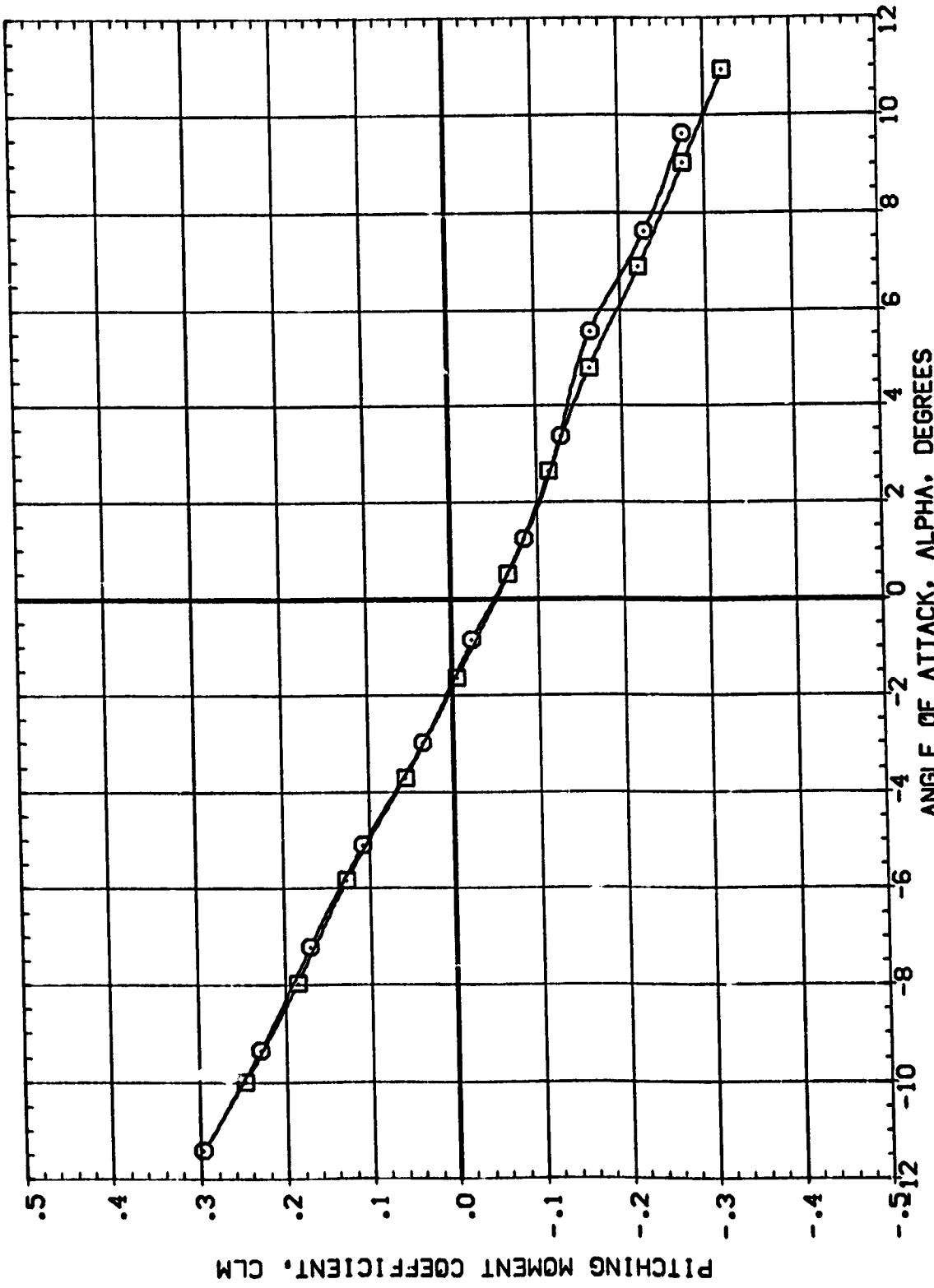
EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(CD)MACH = 1.46

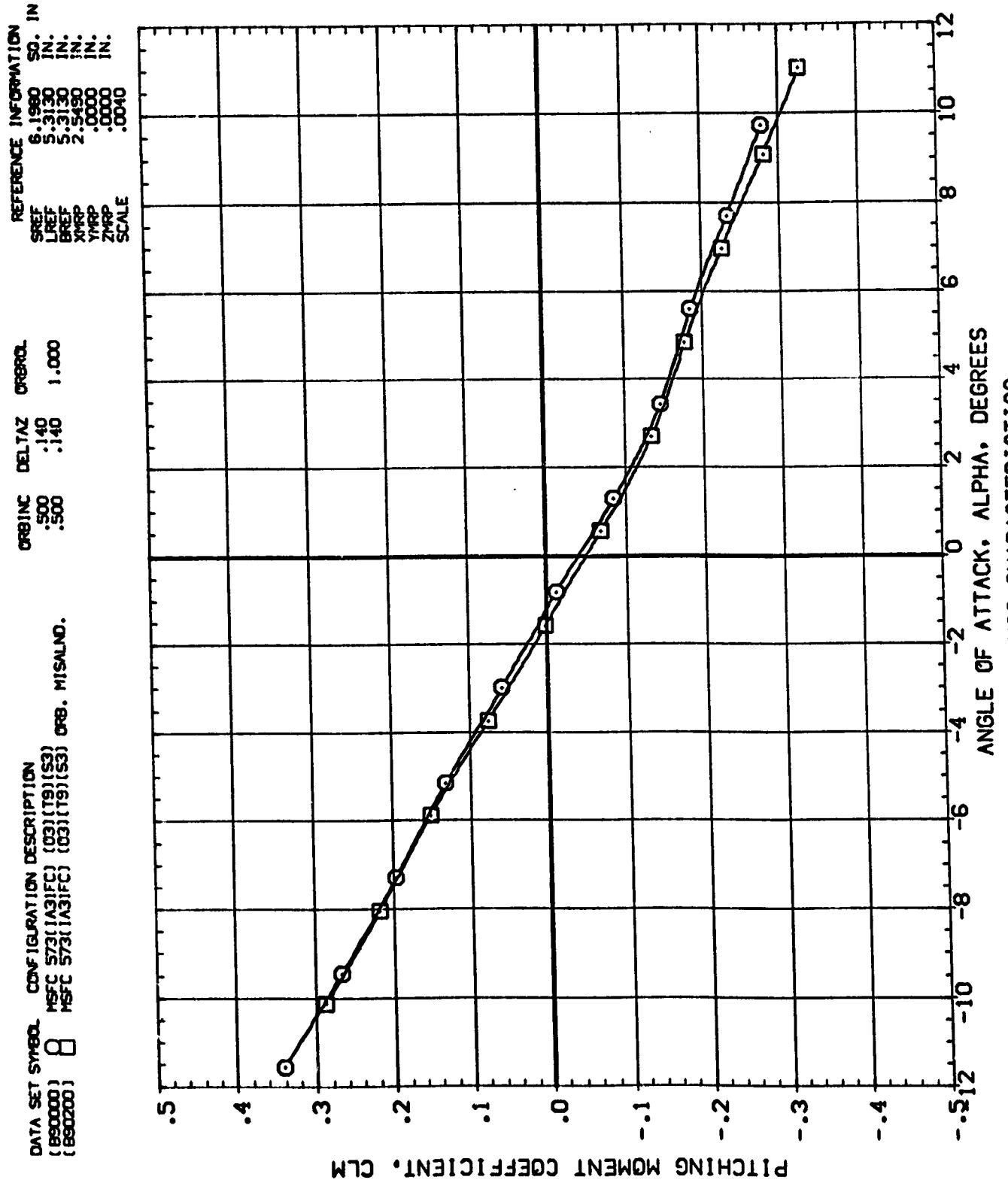
DATA SET SYMBOL. CONFIGURATION DESCRIPTION  
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(B90200)

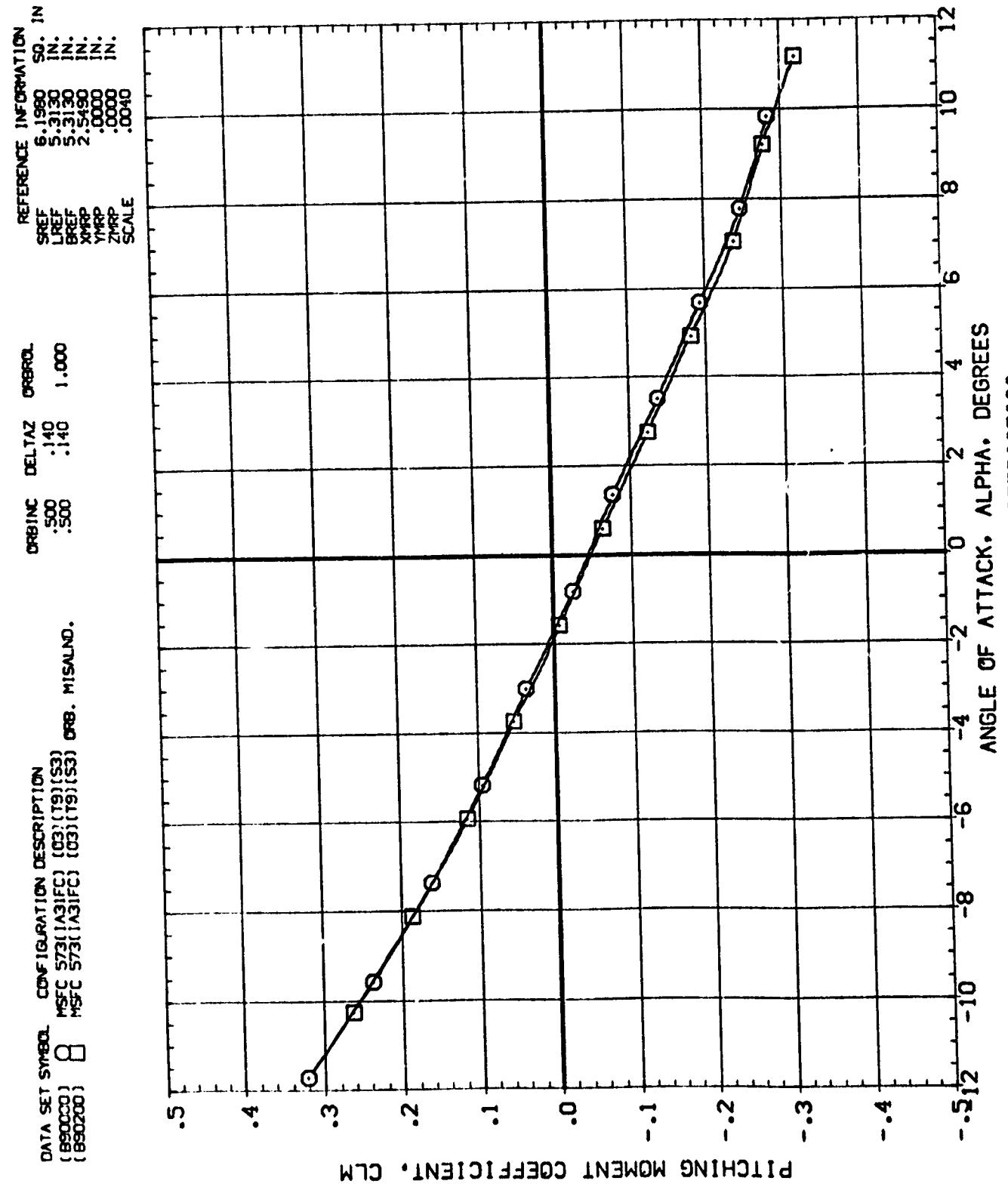
DRBLNC DELTAZ GROLL  
.500 .140 1.000  
.500 .140

REFERENCE INFORMATION  
SREF 6.1980 SD. IN  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
XMRP 2.5490 IN.  
YMRP .0000 IN.  
ZMRP .0000 IN.  
SCALE .0040



EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
(A)MACH = .90





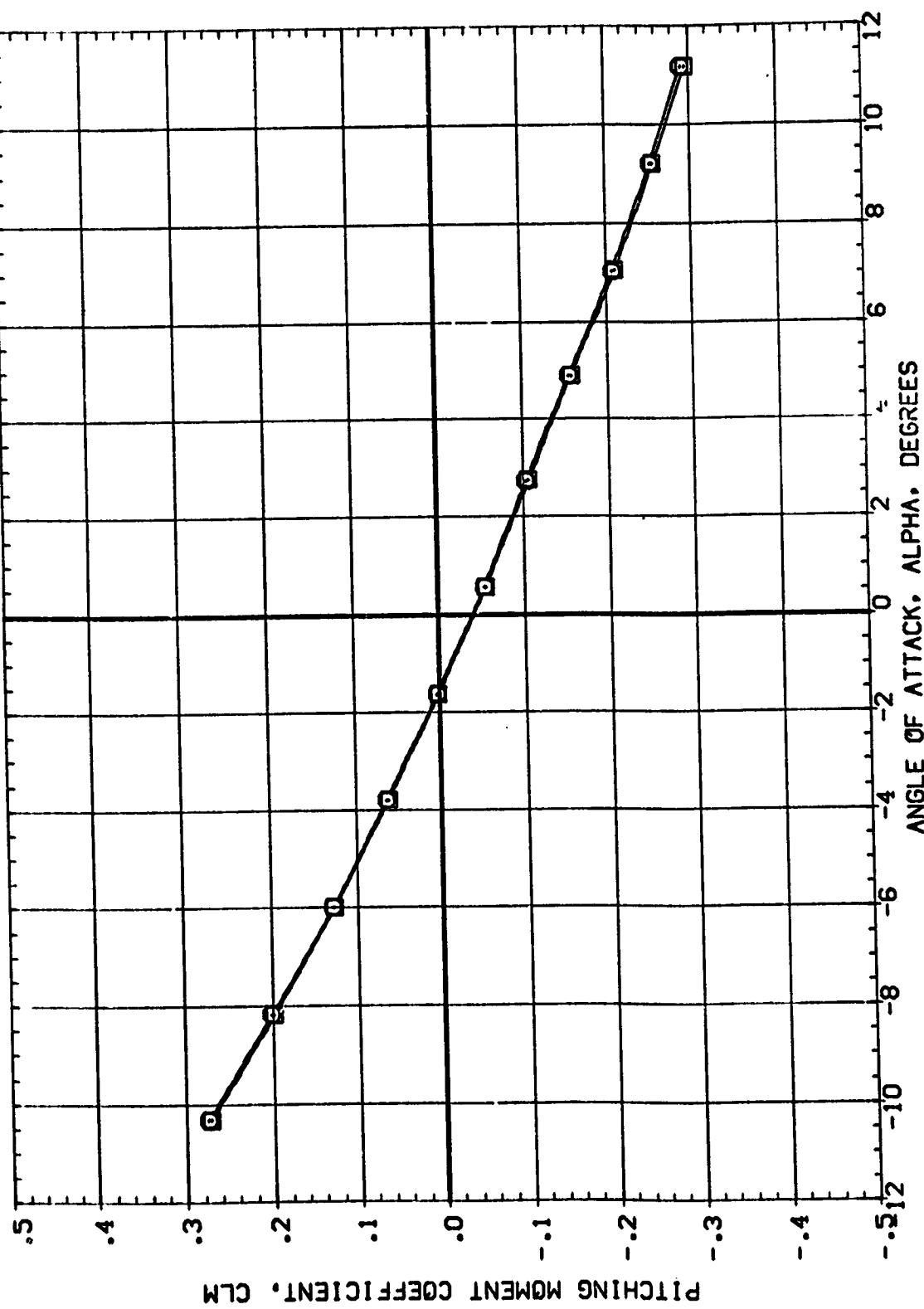
EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
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 850200) MSFC 5731A3IFC (03)(T9)(S3) ORB. MISLAND.

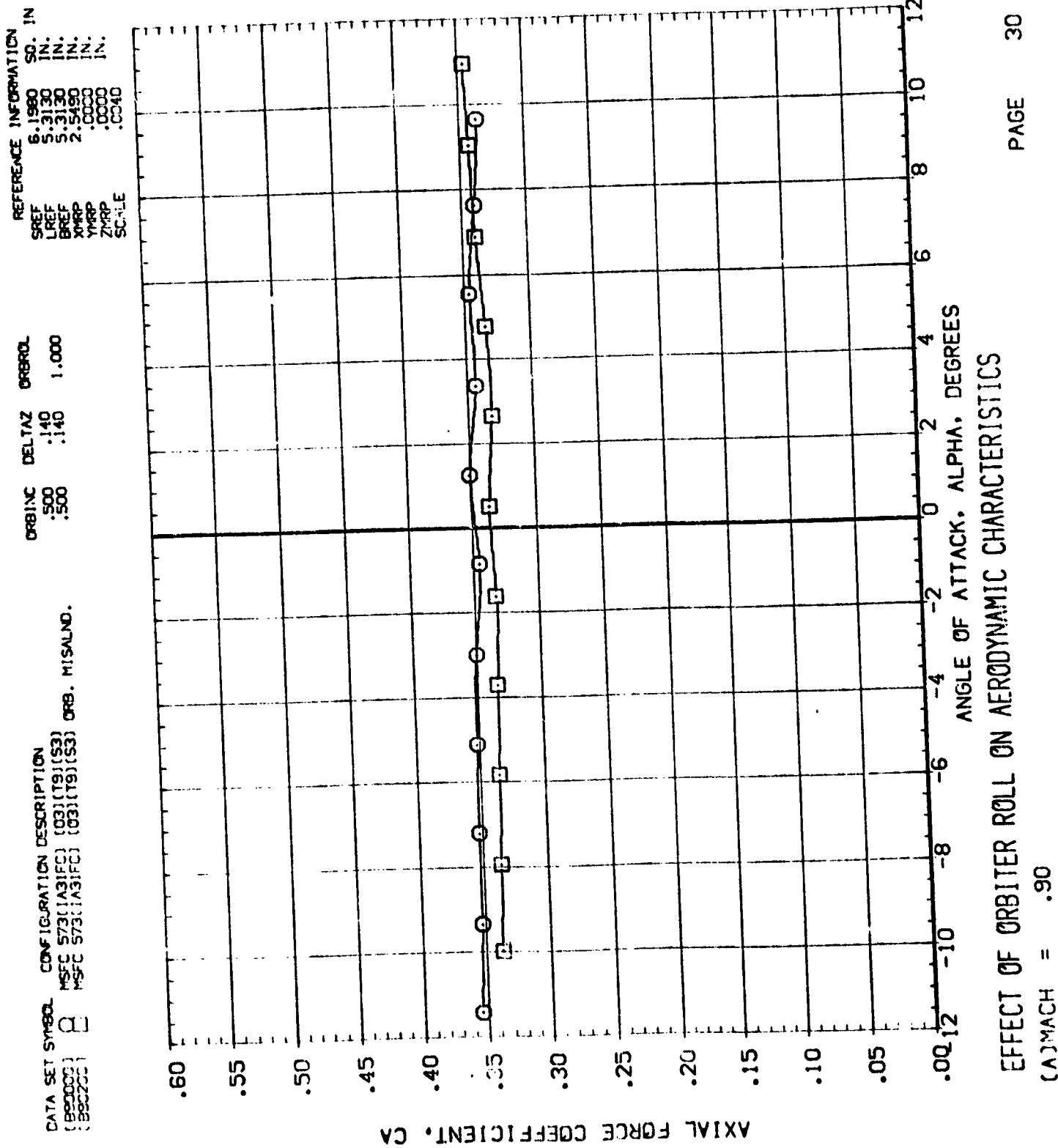
REFERENCE INFORMATION  
 SREF 61980 SD. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0000 IN.  
 SCALE .0040

DRINC DELTAZ DRROL  
 .500 .140 1.000  
 .500 .140



EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 $(\text{CD})_{\text{MACH}} = 1.46$

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B60005) SEC 573((A3)FC) (03)(T9)(S3) ORB. M/SALD.  
 (B60005) SEC 573((A3)FC) (03)(T9)(S3) ORB. M/SALD.  
 (B60005) SEC 573((A3)FC) (03)(T9)(S3) ORB. M/SALD.



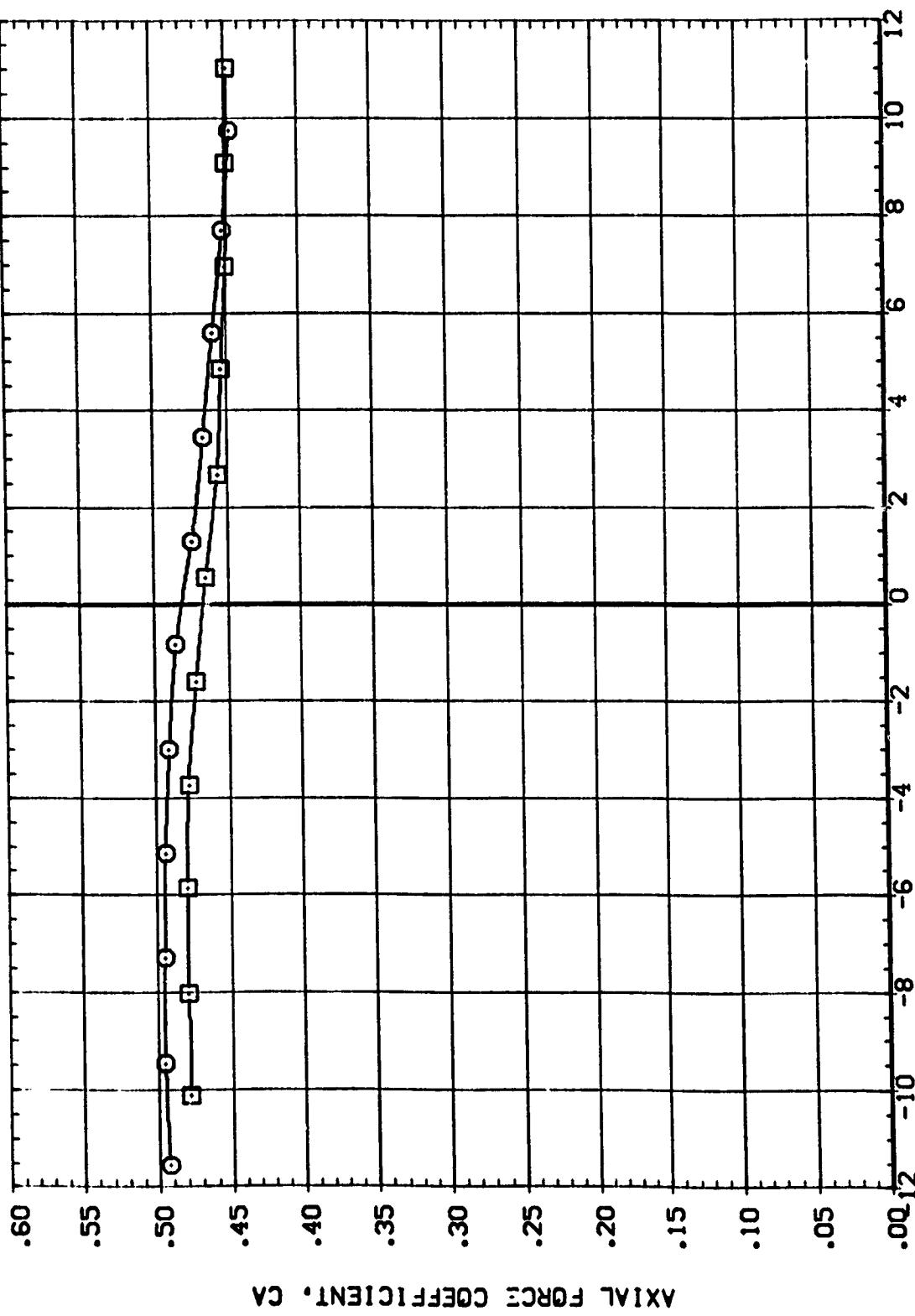
REFERENCE INFORMATION  
 SREF 6.1980 SD. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0000 IN.  
 SCLE .0040

EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 $(\Delta)MACH = .90$

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) NSFC 573(1A3)FC (03)(T9)(S3) ORB. MISLAND.  
 (B90200) NSFC 573(1A3)FC (03)(T9)(S3) ORB. MISLAND.

REFERENCE INFORMATION  
 SREF 6.1980 SO. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5450 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.

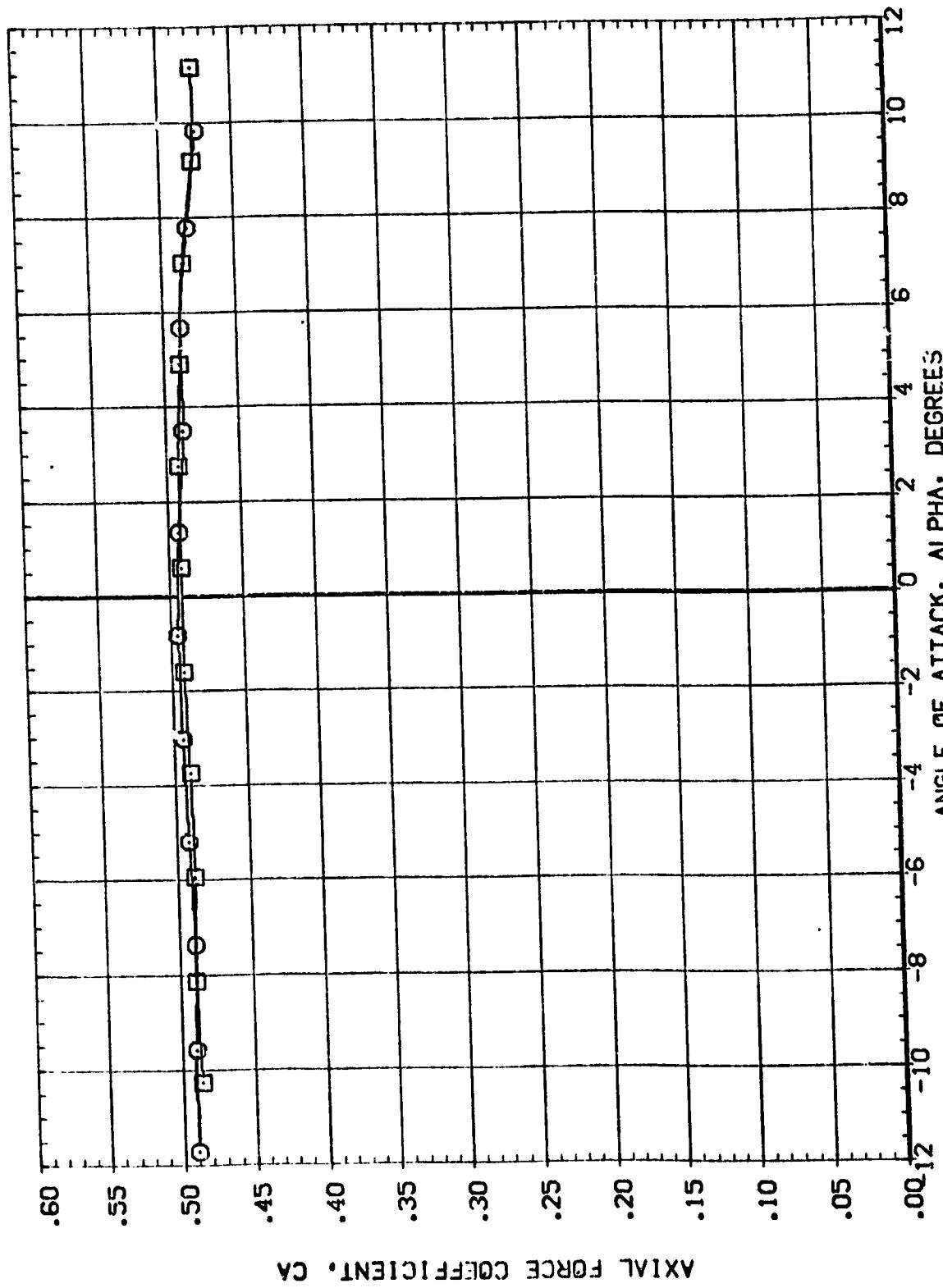
ORB INC DELTAZ ORBROL.  
 .500 .140 1.000  
 .500 .140 1.000



EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 $(B)MACH = 1.05$

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
[890000] MSFC 5731(A3)FC [03)(19)(S3)  
[890200] MS-C 5731(A3)FC [03)(19)(S3) ORB. MSLNO.

REFERENCE INFORMATION  
SREF 6.1980 SD. IN  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
XMRP 2.5490 IN.  
YMRP .0000 IN.  
ZMRP .0040 IN.  
SCALE

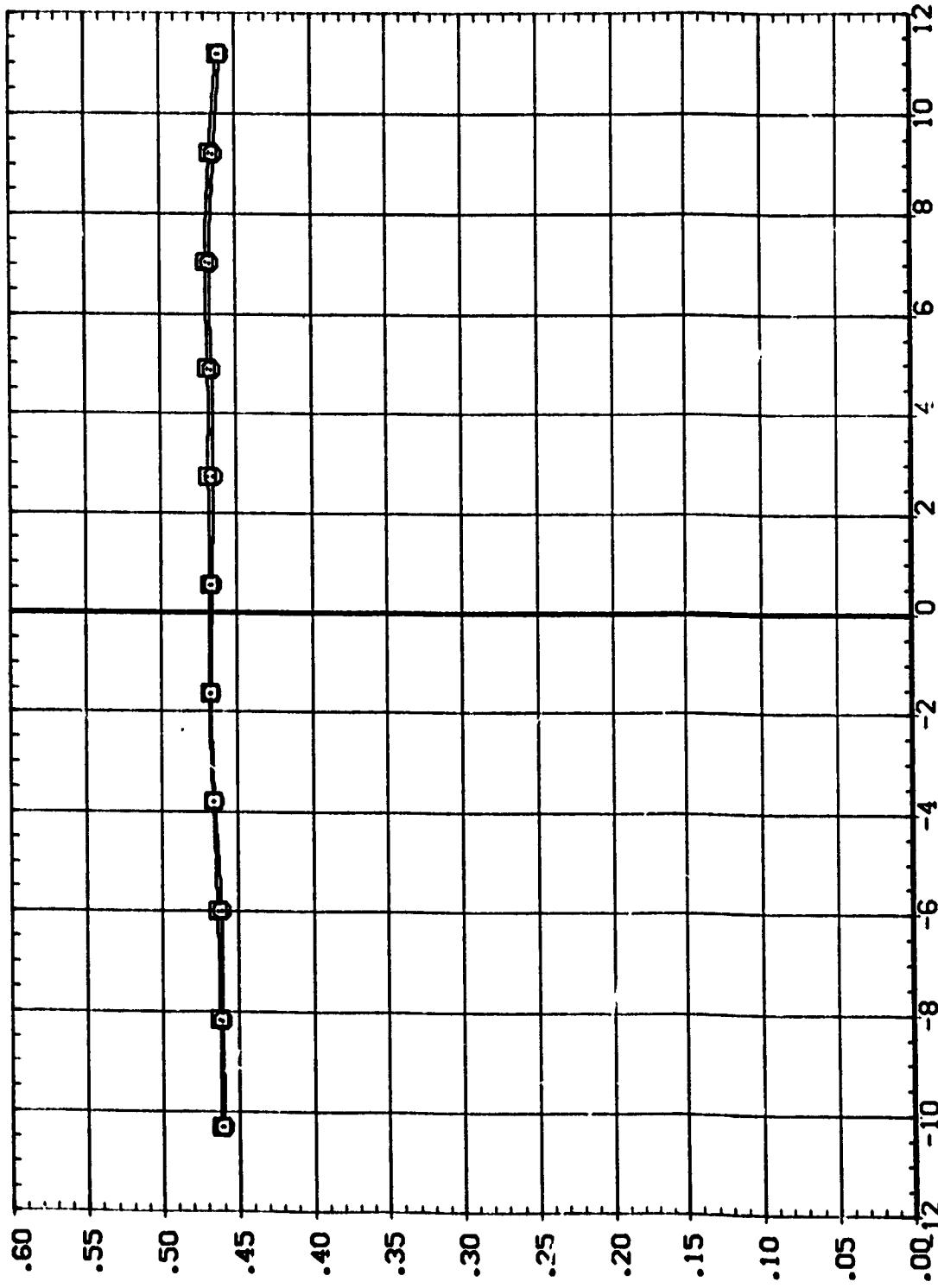


EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
(C)MACH = 1.25

DATA SET SYMBOL:  MSFC 573(1A3)FC (03)(19)(S3)  
 (690000)  MSFC 573(1A3)FC (03)(19)(S3) ORB. MISLAND.  
 (692000)

ORB INC.	DELTAZ	CRERO.
.500	.140	1.000
	.140	

REFERENCE INFORMATION  
 SREF 6.190 SO. IN  
 LREF 5.3130 IN  
 BREF 5.3130 IN  
 XMRP 2.5590 IN  
 YMRP .0000 IN  
 ZMRP .0000 IN  
 SCALE .0040



AXIAL FORCE COEFFICIENT, CA

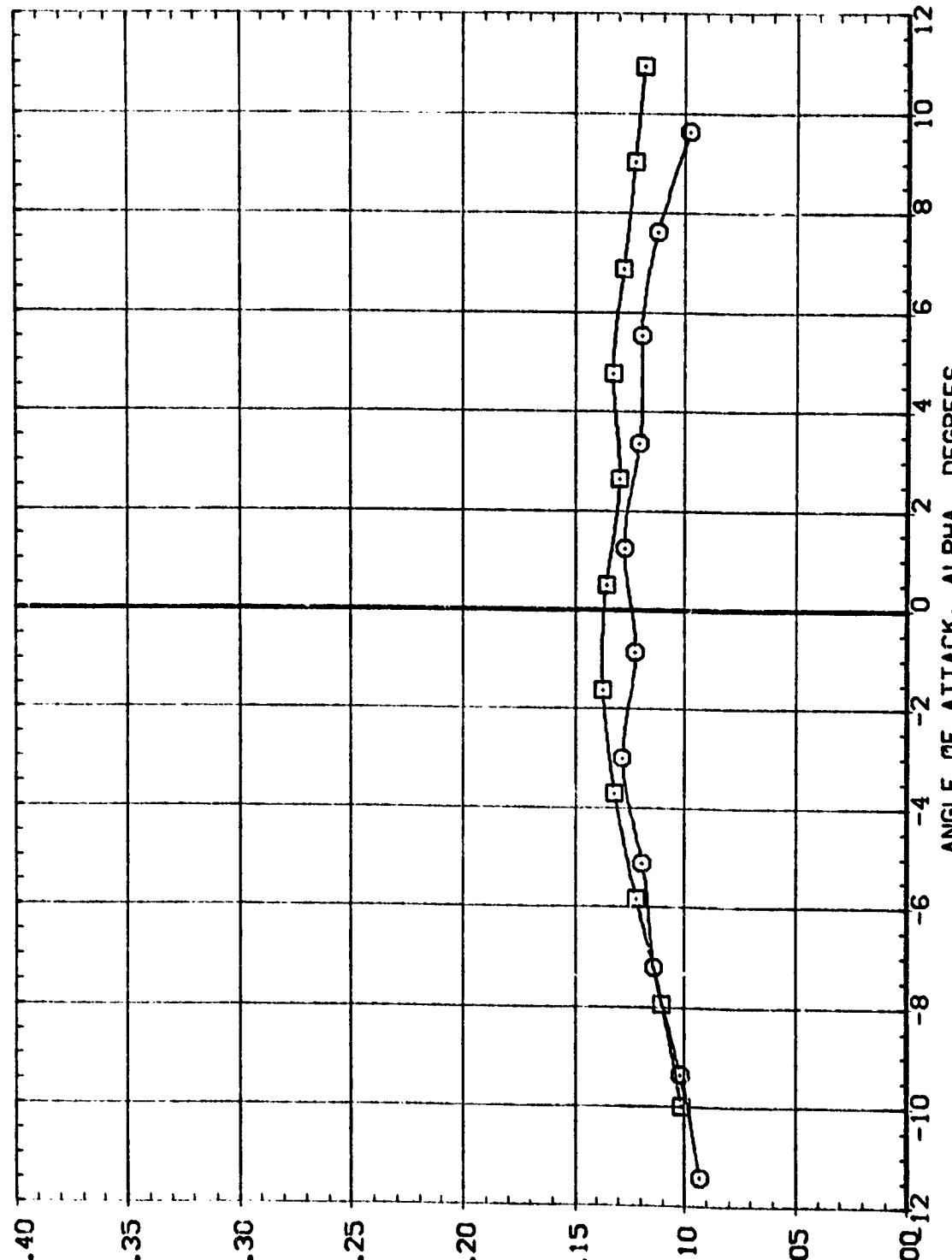
### EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(MACH = 1.46

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DATA SET SYMBOL CONFIGURATION DESCRIPTION  
[ B90000 ] MSFC 5731A31FC) (03)(T9)(S3)  
[ B90200 ] MSFC 5731A31FC) (03)(T9)(S3)

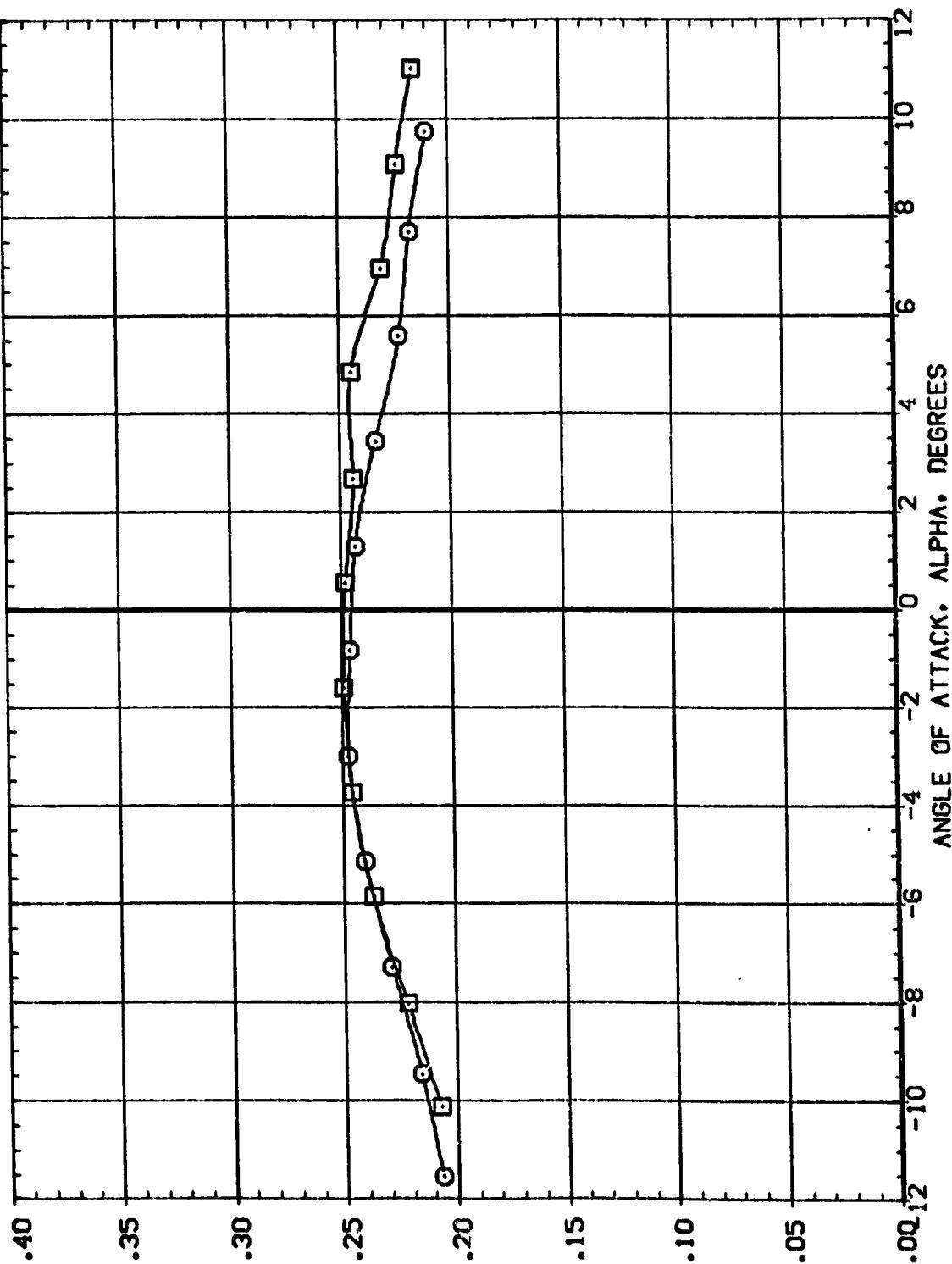
REFERENCE INFORMATION  
SREF 6.1980 SO. IN  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
X0RP 2.5450 IN.  
Y0RP .0000 IN.  
Z0RP .0000 IN.  
SCALE .0040



EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
( $\Delta$ MACH = .90)

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (890000) MSFC 573(1A3)FC (03)(19)(S3)  
 (890200) MSFC 573(1A3)FC (03)(19)(S3) ORB. MISLAND.

REFERENCE INFORMATION  
 SREF 6.1980 SD. 1.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XRP 2.5490 IN.  
 YRP .0000 IN.  
 ZRP .0040 IN.  
 SCALE



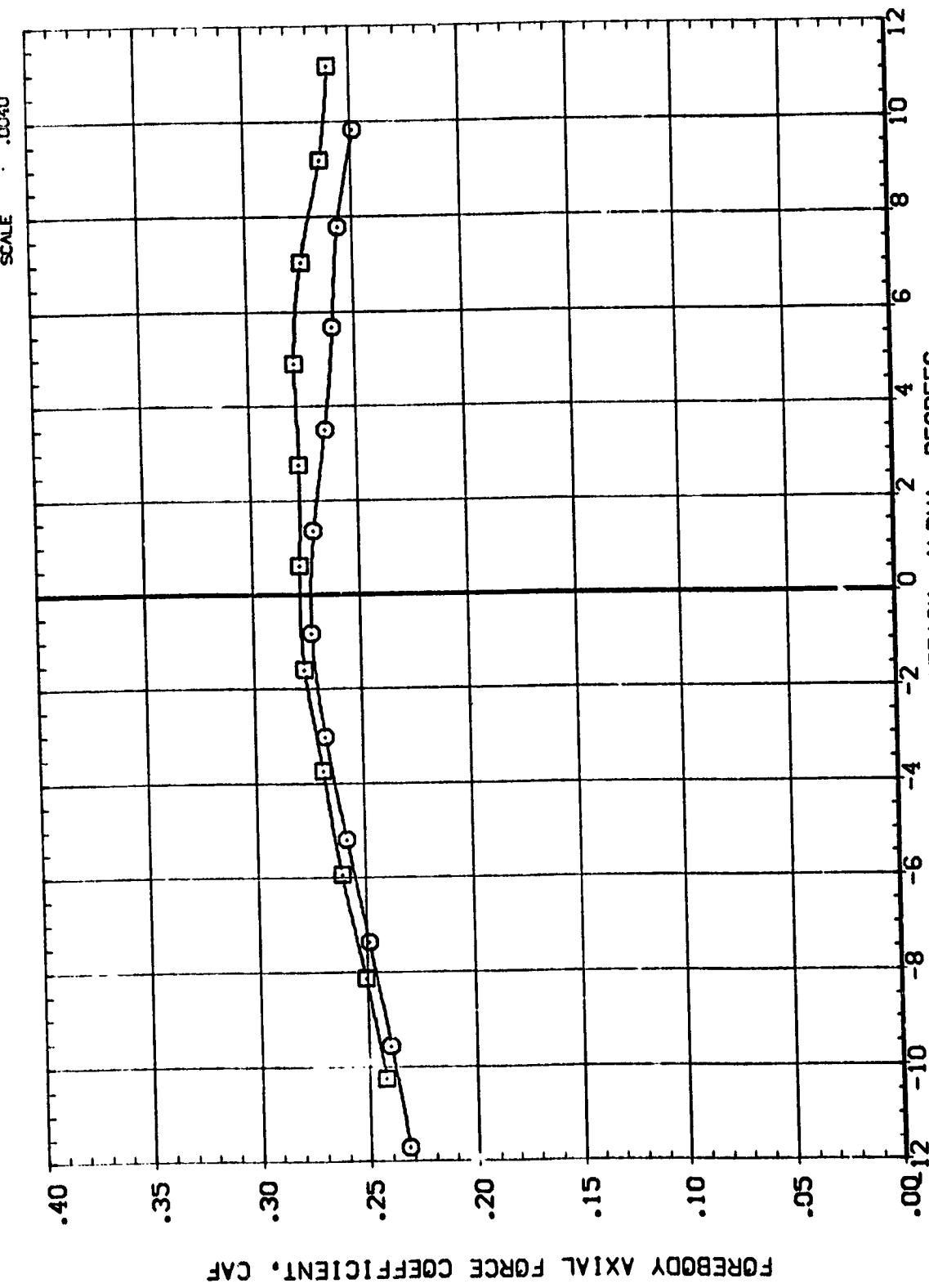
FORCEBODY AXIAL FORCE COEFFICIENT, CAF

EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(BJMACH = 1.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000)  MSFC 5731(A3)FC (03)(T9)(S3)  
 (B90200)  MSFC 5731(A3)FC (03)(T9)(S3) ORB. MSLN.D.

REFERENCE INFORMATION  
 SPEC 6.1980 SD. IN.  
 LREF 5.3130 IN.  
 RREF 5.3130 IN.  
 XRP 2.5490 IN.  
 YRP .0000 IN.  
 ZRP .0000 IN.  
 SCALE .0040



FORCEBODY AXIAL FORCE COEFFICIENT, CAF

EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 $(C)MACH = 1.25$

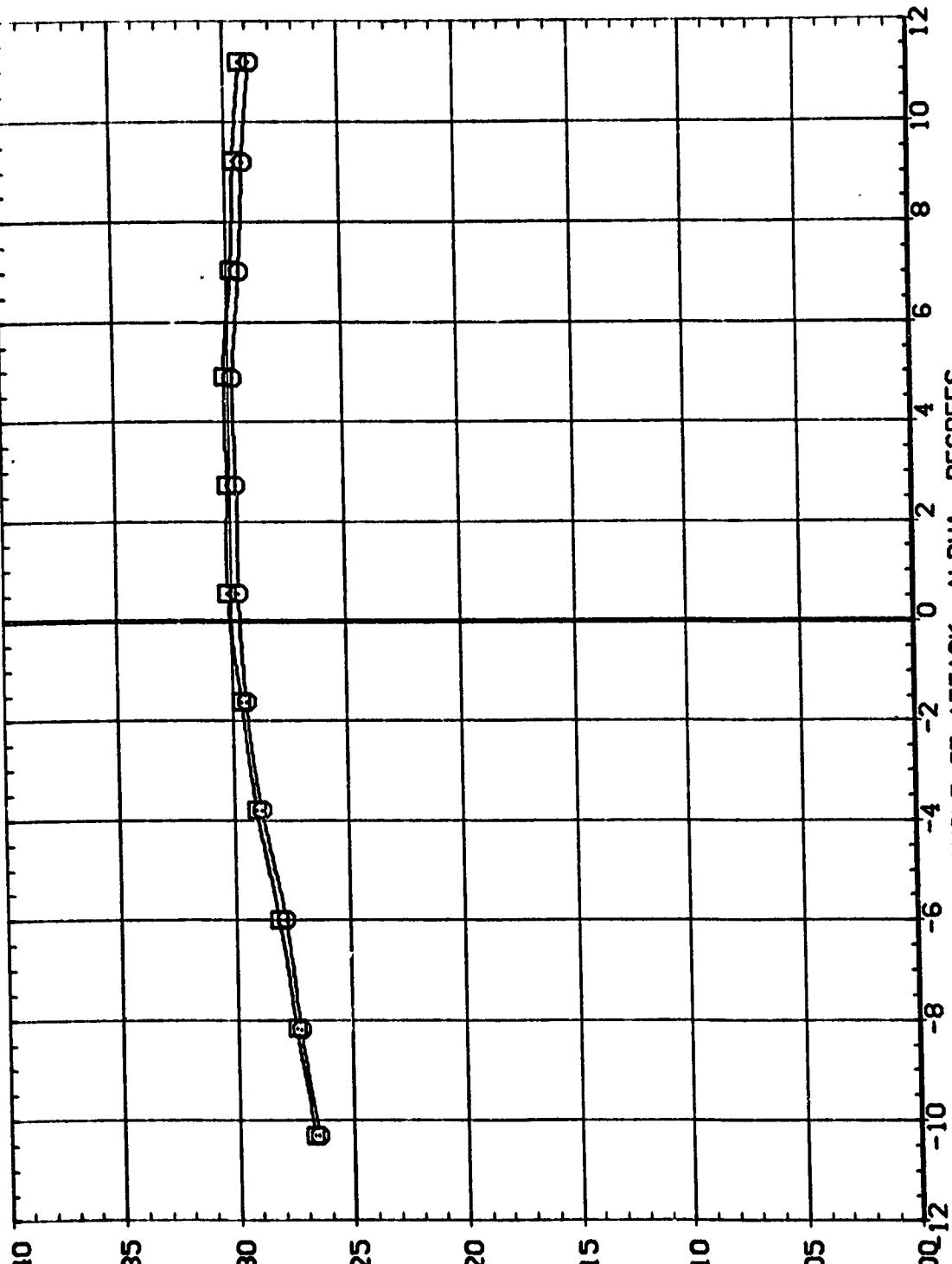
DATA SET SNAME: CONFIGURATION DESCRIPTION  
 (B90000) HSF C 573(A31FC) (63)(T9)(S3) ORB. MISALND.  
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REFERENCE	INFORMATION
SREF	6.1980 IN.
LREF	5.3130 IN.
BREF	5.3130 IN.
XMRP	2.5490 IN.
YMRP	.0000 IN.
ZMRP	.0040 IN.
SCALE	.0040 IN.

ORB INC DELTAZ ORBOL

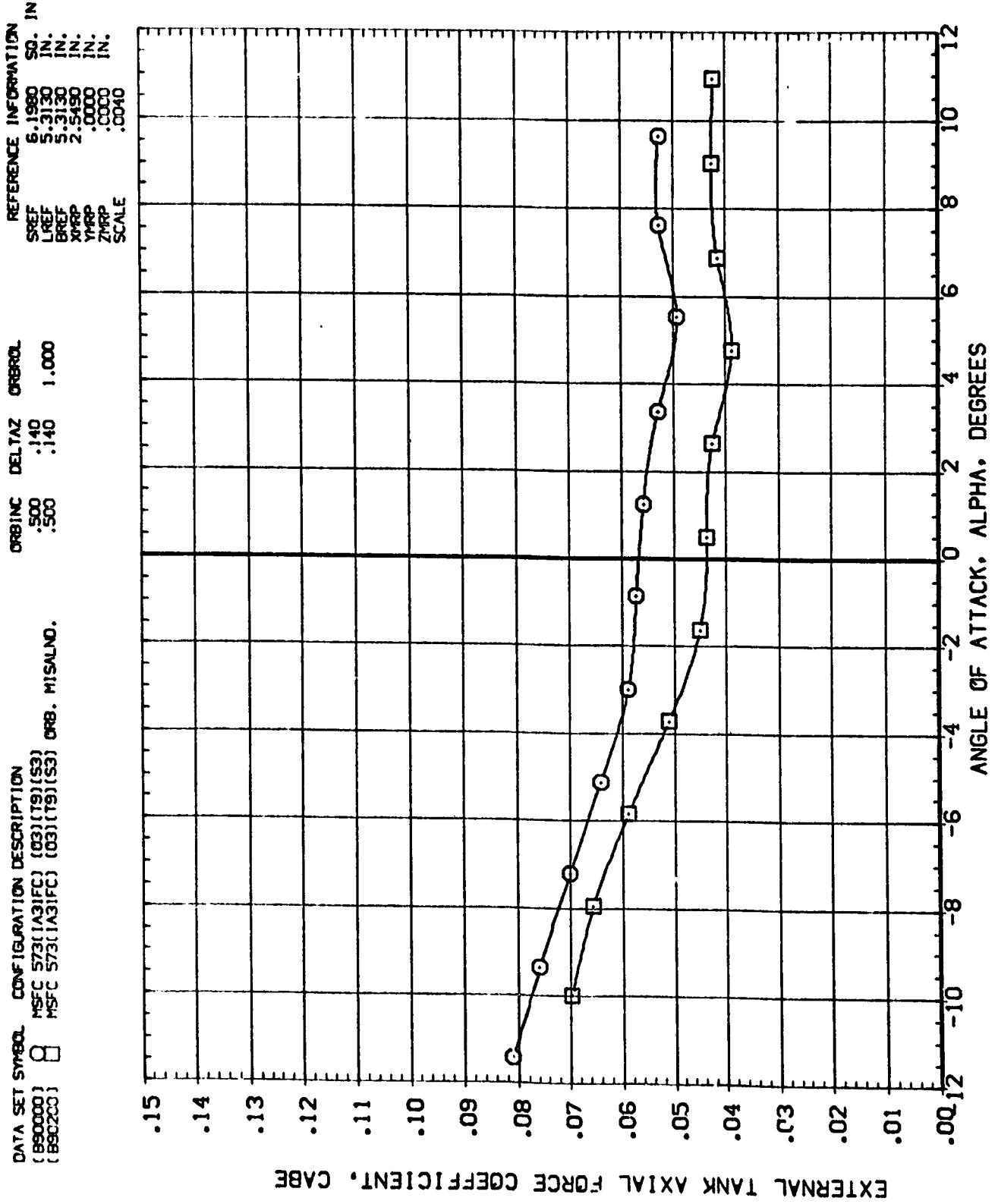
.500 .140 1.000

.500



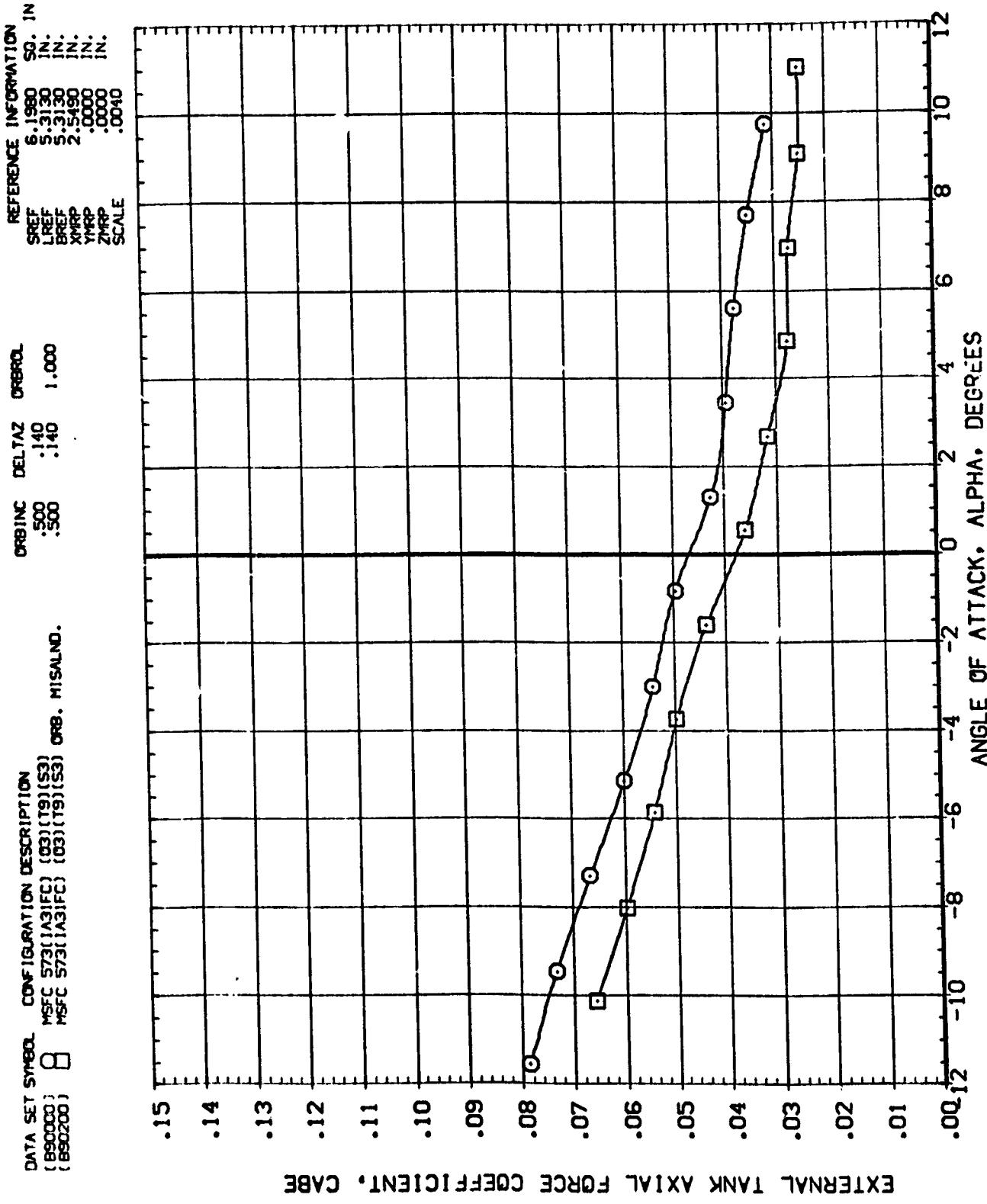
FOREBODY AXIAL FORCE COEFFICIENT, CAF

EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 $(\text{D})_{\text{MACH}} = 1.46$



EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

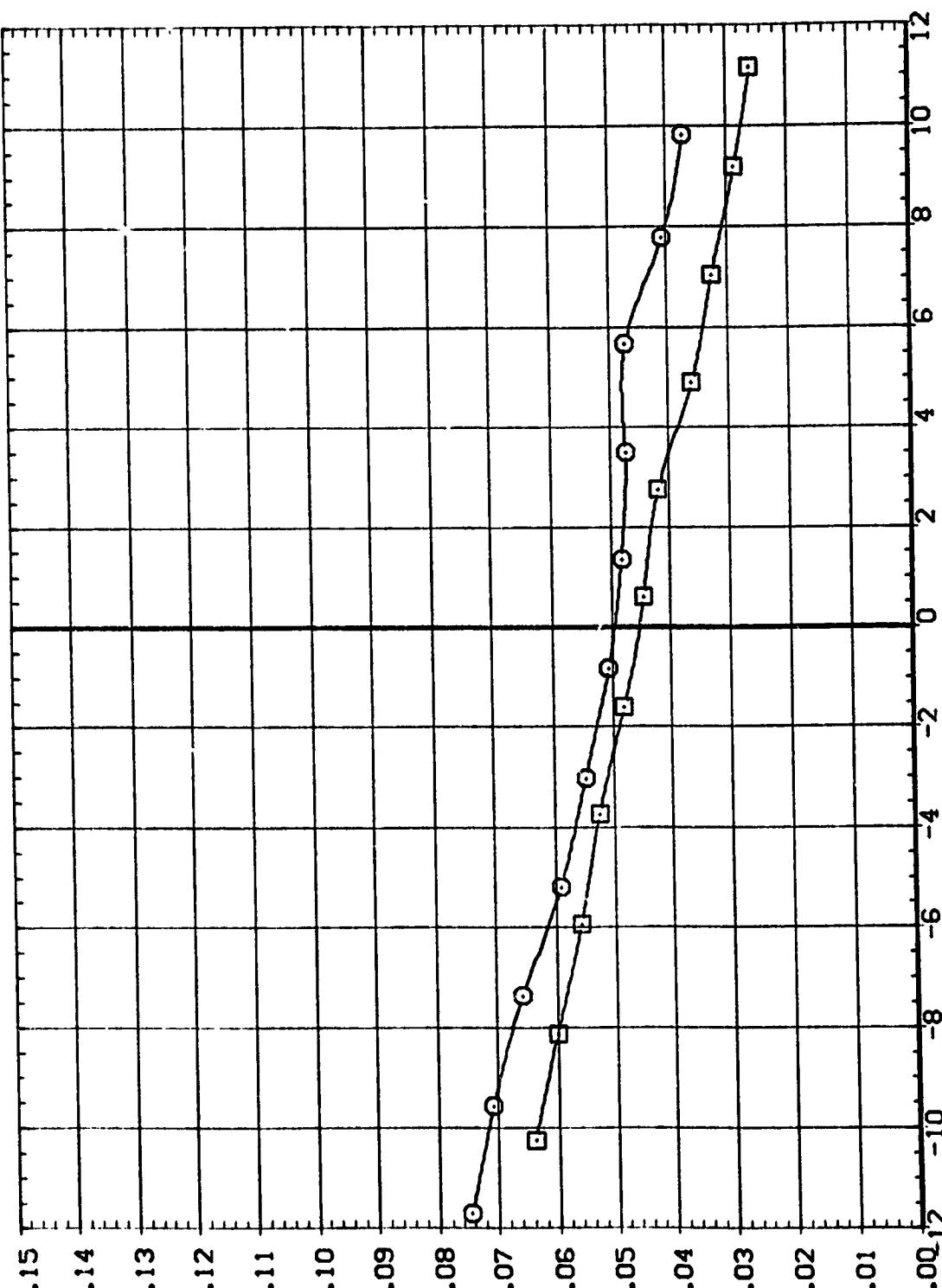
C<sub>A</sub>MACH = .90



EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

DATA SET SYMBOL. CONFIGURATION DESCRIPTION  
(890000) NSFC 5731(A3)FC (03)(T9)(S3) ORB. MISALND.  
(890200) NSFC 5731(A3)FC (03)(T9)(S3) ORB. MISALND.

REFERENCE INFORMATION  
SREF 6.1980 SQ. IN.  
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BREF 5.3130 IN.  
XMRP 2.5490 IN.  
YMRP .0000 IN.  
ZMRP .0040 IN.  
SCALE

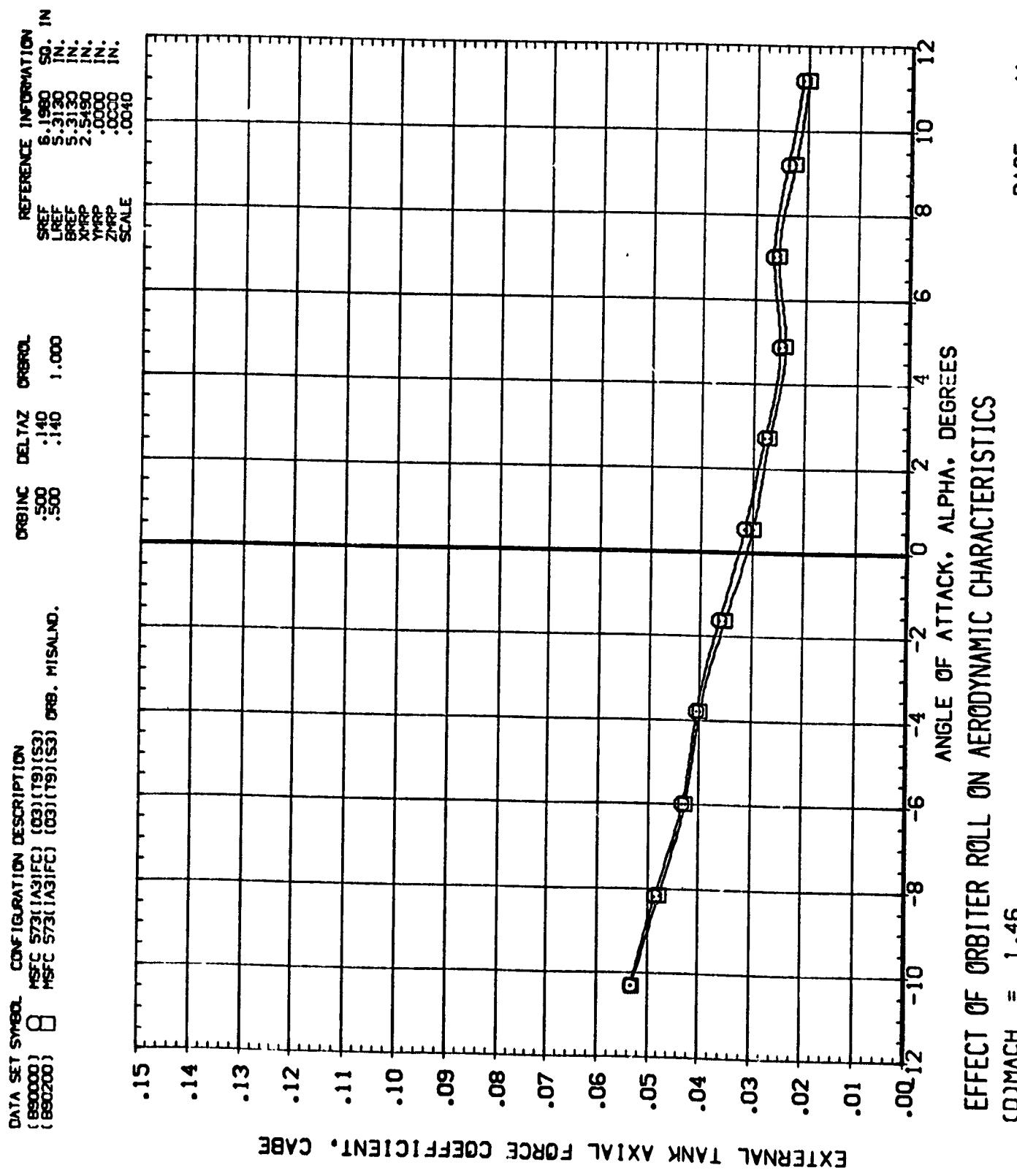


EXTERNAL TANK AXIAL FORCE COEFFICIENT, CABE

EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

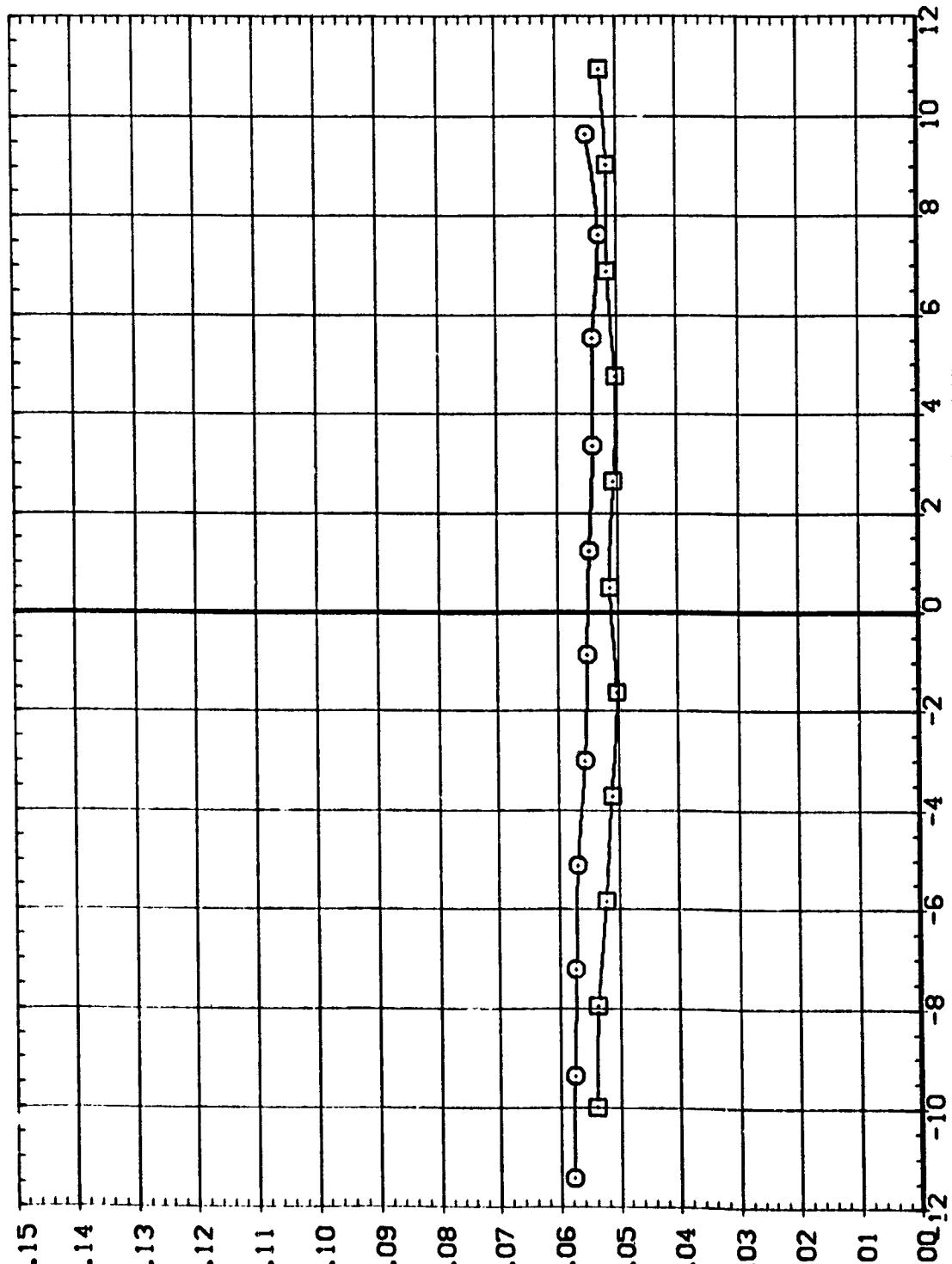
PAGE 40



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 {B90000} MSEC 573{1A31FC} {03}{T9}{S3} {03}{T9}{S3} ORB. MSLND.  
 {BEC200} MSEC 573{1A31FC} {03}{T9}{S3} {03}{T9}{S3} ORB. MSLND.

REFERENCE INFORMATION  
 SREF 6.1980 SQ. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE

ORBINC DELTAZ ORBROL  
 .500 .140 1.000  
 .500 .140

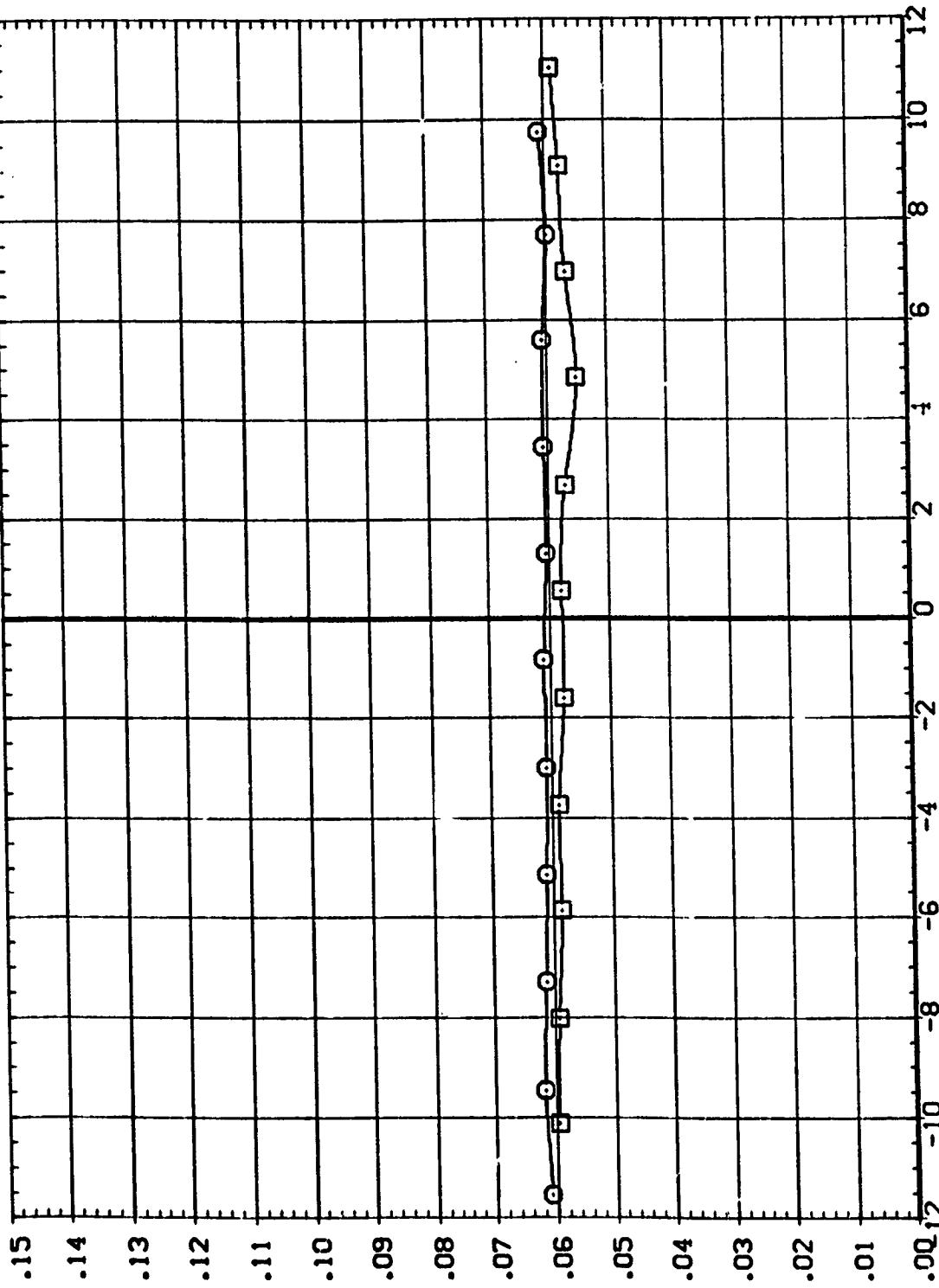


ORBITER AXIAL FORCE COEFFICIENT, CABO

EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 $(\Delta MACH = .90$

DATA SET SPEED. CONFIGURATION DESCRIPTION  
 (890000) NSFC 573(1A31FC) (G3)(19)(S3)  
 (890200) NSFC 573(1A31FC) (G3)(19)(S3) ORB. M/S/NO.

REFERENCE INFORMATION  
 SREF 6.1980 2. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5590 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.



ORBITER AXIAL FORCE COEFFICIENT, CABO

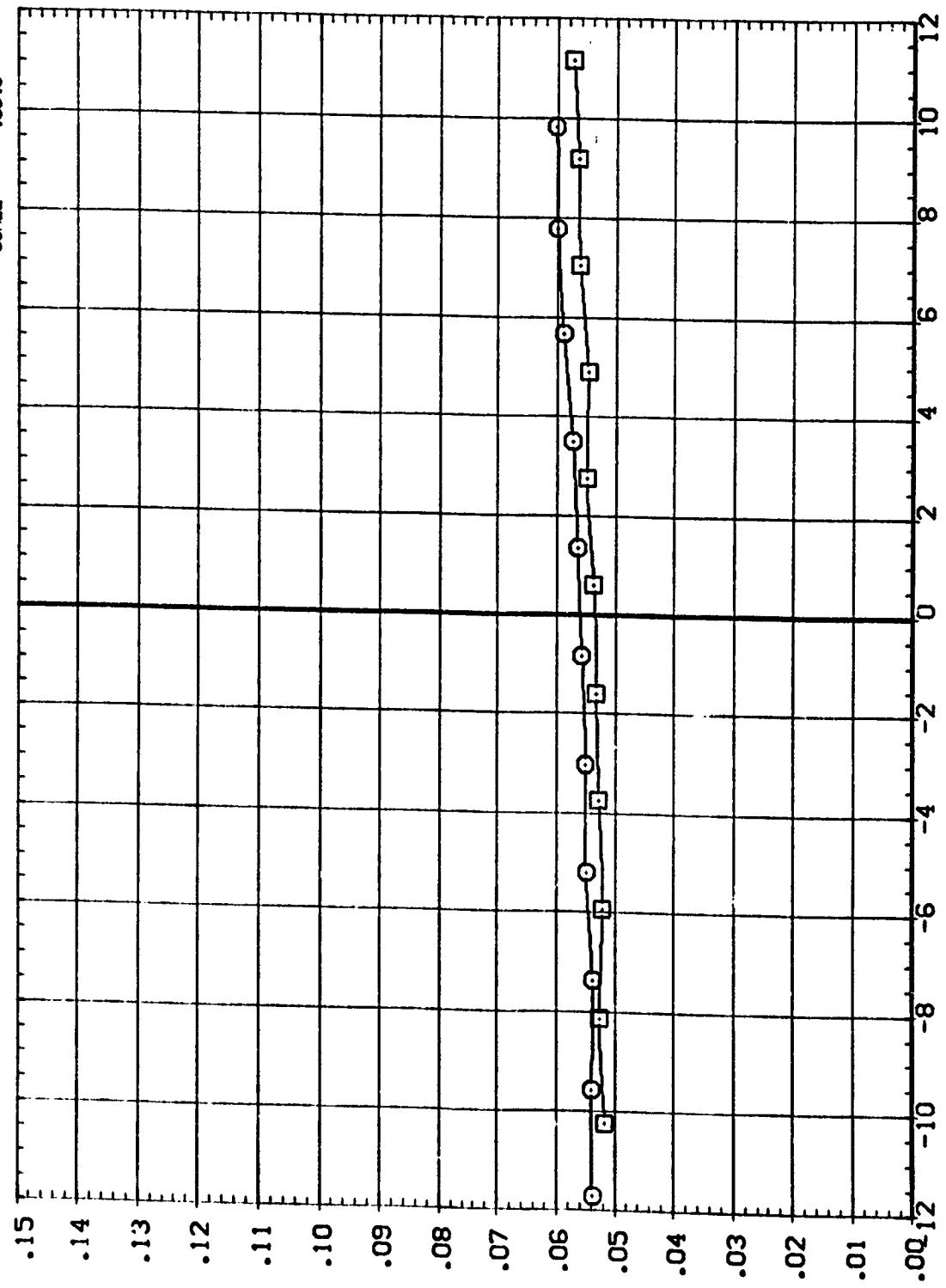
### EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
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 (890200) MSFC S73(1A31FC) (03)(19)(S3) ORB. MISALN.

ORBINC DELTAZ ORBRO.  
 .500 .140 1.000  
 .500 .140

REFERENCE INFORMATION  
 SRFF 6.1980 SG. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRF 2.5450 IN.  
 YMRF .0000 IN.  
 ZMRF .0000 IN.  
 SCALE .0040



ORBITER AXIAL FORCE COEFFICIENT, CABG

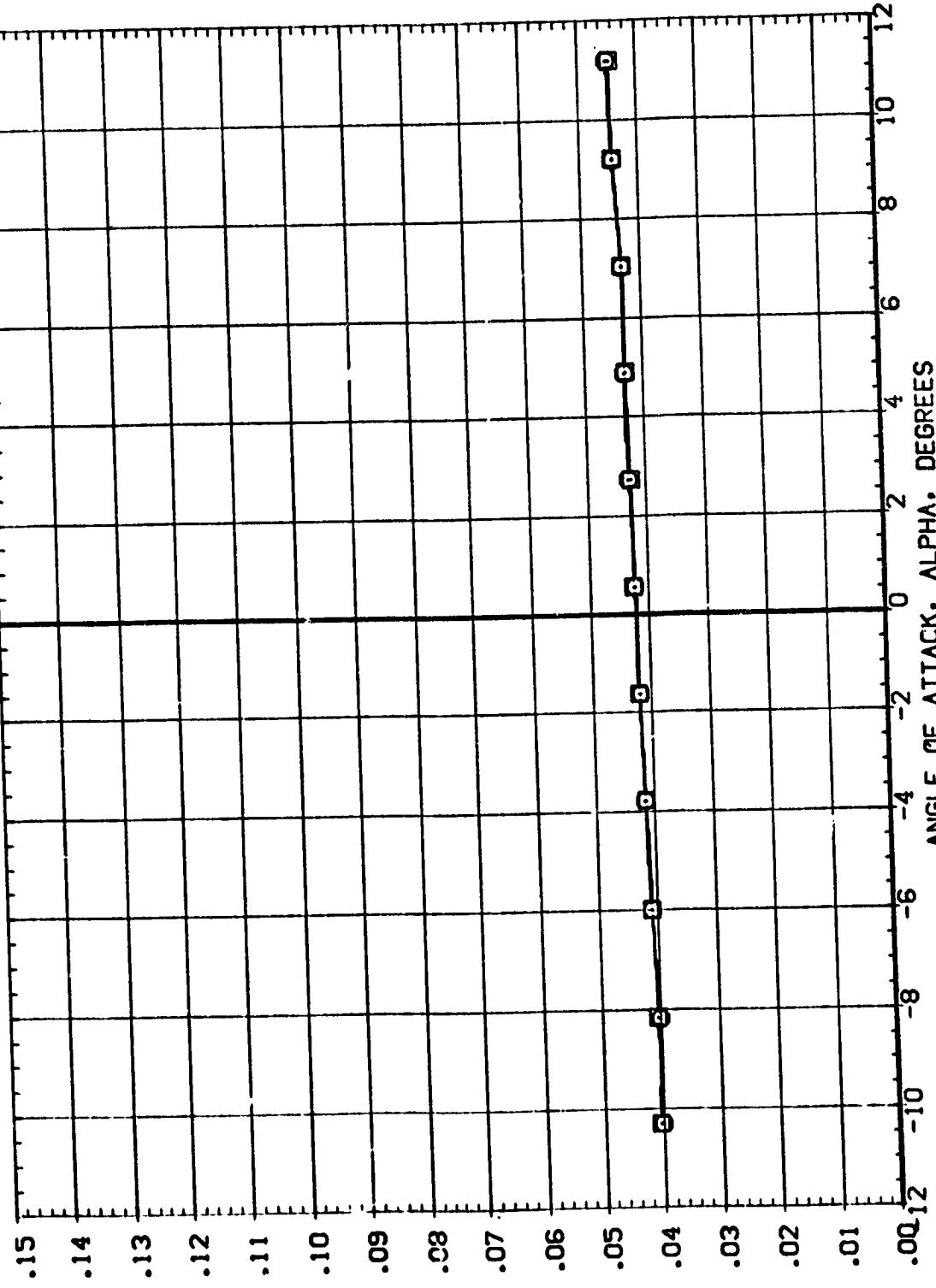
EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

DATA SET SYMBOL: **□** CONFIGURATION DESCRIPTION: MSFC ST3((A3)FC) (03)(19)(S3)  
 (B00000) MSFC ST3((A3)FC) (03)(19)(S3) ORB. MSLND.  
 (B00200)

REFERENCE INFORMATION  
 SREF 6.1980 SQ. IN  
 LREF 5.3130 IN  
 BREF 5.3130 IN  
 XHPP 2.5490 IN  
 YHPP .0000 IN  
 ZHPP .0000 IN  
 SCALE .0040

ORBITER AXIAL FORCE COEFFICIENT, CAB0

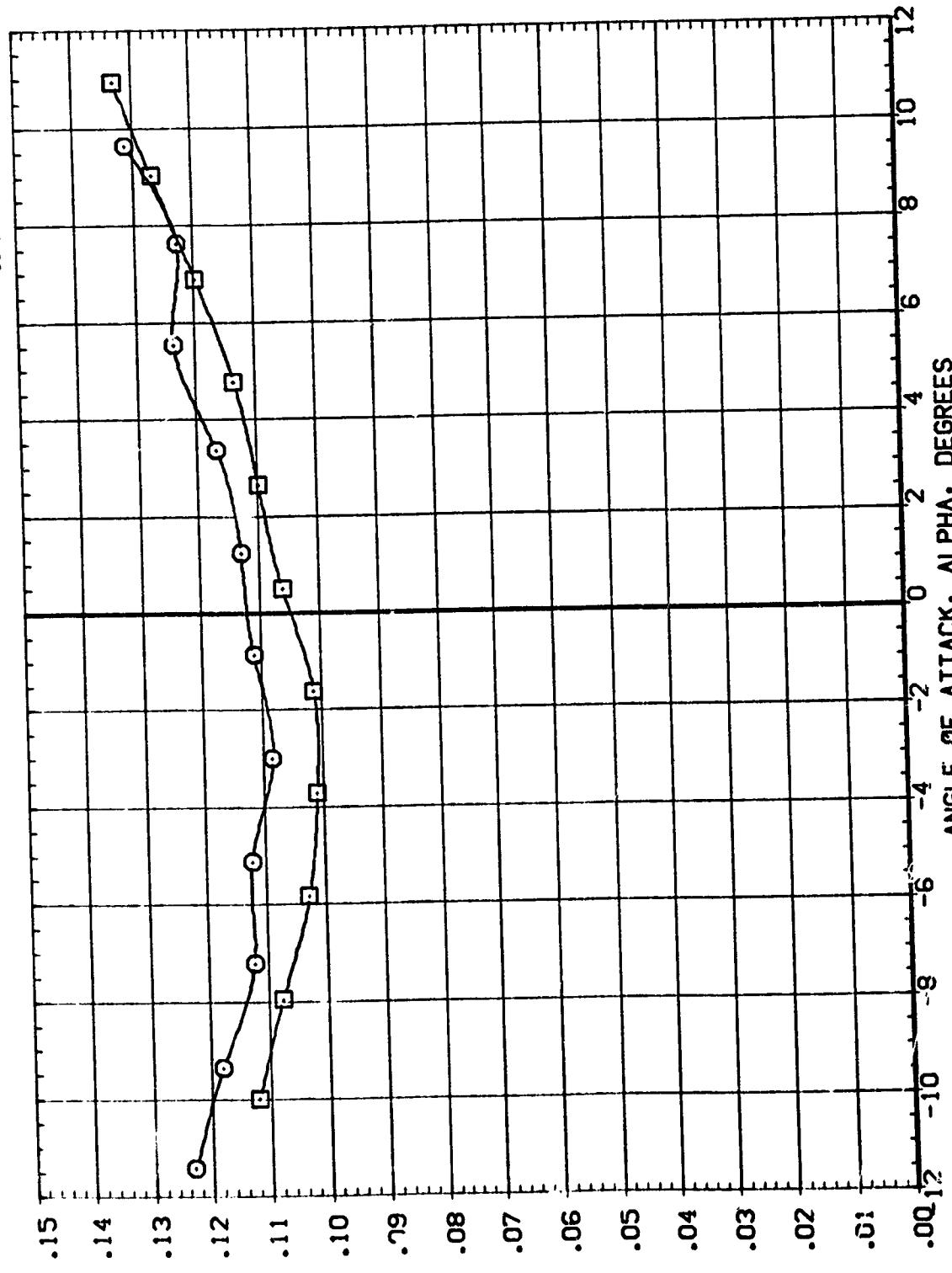


EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(COMACH = 1.46

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (890000) NSFC 573(A3)FC (03)(T9)(S3)  
 (862000) NSFC 573(A3)FC (03)(T9)(S3) ORB. MISALND.

REFERENCE INFORMATION  
 SREF 6.1980 SO. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.

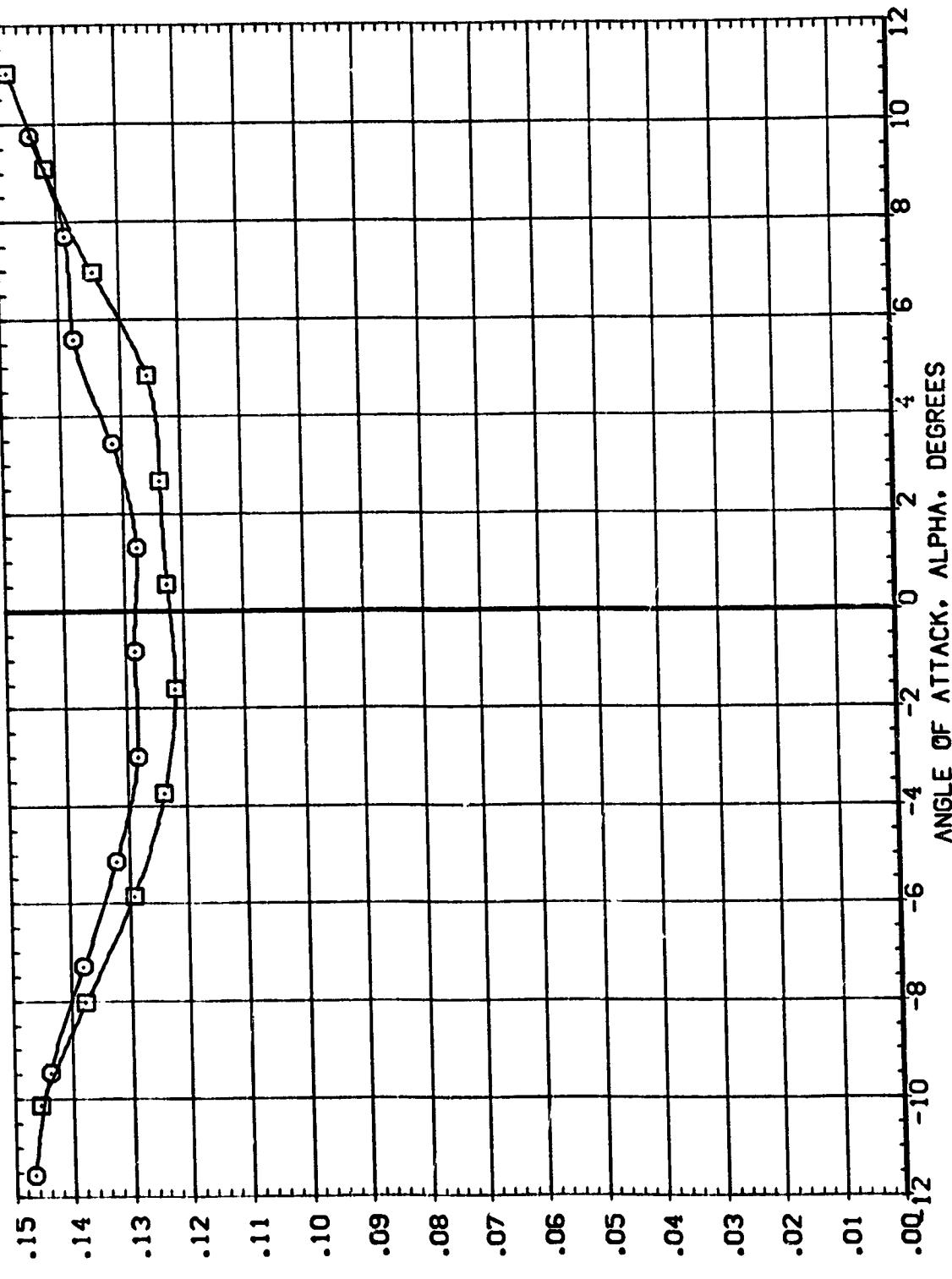


SRB AXIAL FORCE COEFFICIENT, CABs

EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 $(\Delta)MACH = .90$

DATA SET SYMBOL. CONFIGURATION DESCRIPTION  
 (890000) NSFC 573(A3IFC) (03)(T9)(S3)  
 (890200) NSFC 573(A3IFC) (03)(T9)(S3) ORB. MSL AND.

REFERENCE INFORMATION  
 SREF 6.1980 SD. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XRP 2.5450 IN.  
 YRP .0000 IN.  
 ZRP .0040 IN.  
 SCALE



SRB AXIAL FORCE COEFFICIENT. CABS

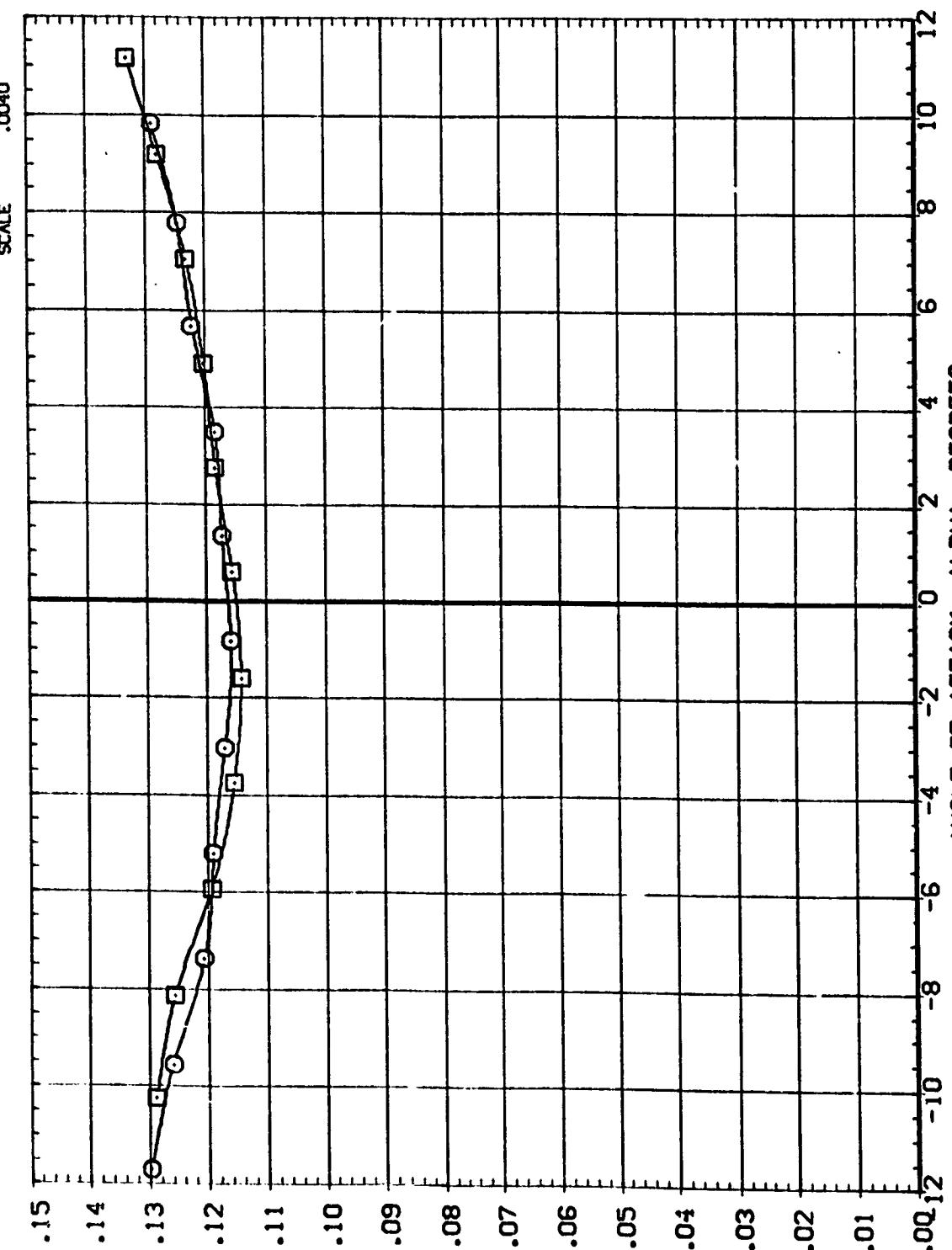
EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
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(B9D2C0) MSLC S73(1A3)FC (03)(79)(S3)

REFERENCE INFORMATION  
SREF 6.1980 SQ. IN.  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
XMRP 2.5490 IN.  
YMRP .0000 IN.  
ZMRP .0040 IN.

SCALE



SRB AXIAL FORCE COEFFICIENT, CABs

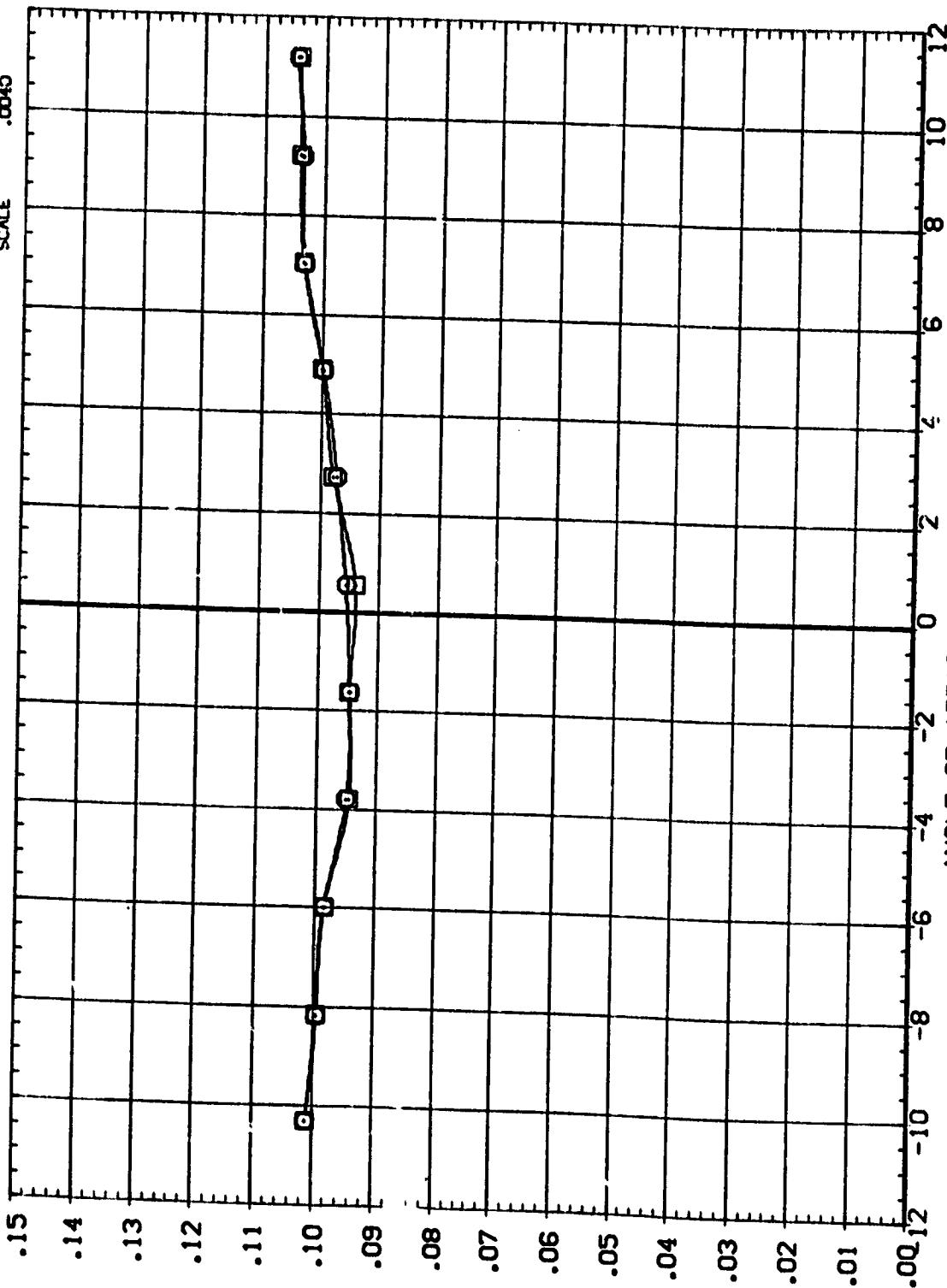
### EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

PAGE 48

DATA SET SYMBOL: **□** CONFIGURATION DESCRIPTION: MSFC 57301A3(FC) (03)(TS)(S3)  
 (890000) MSFC 57301A3(FC) (03)(TS)(S3) ORB. MSLND.  
 (890200)

REFERENCE INFORMATION  
 SREF 6.1980 SD. IN  
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 BREF 5.3130 IN.  
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 YRP .0000 IN.  
 ZRP .0000 IN.  
 SCALE .0043

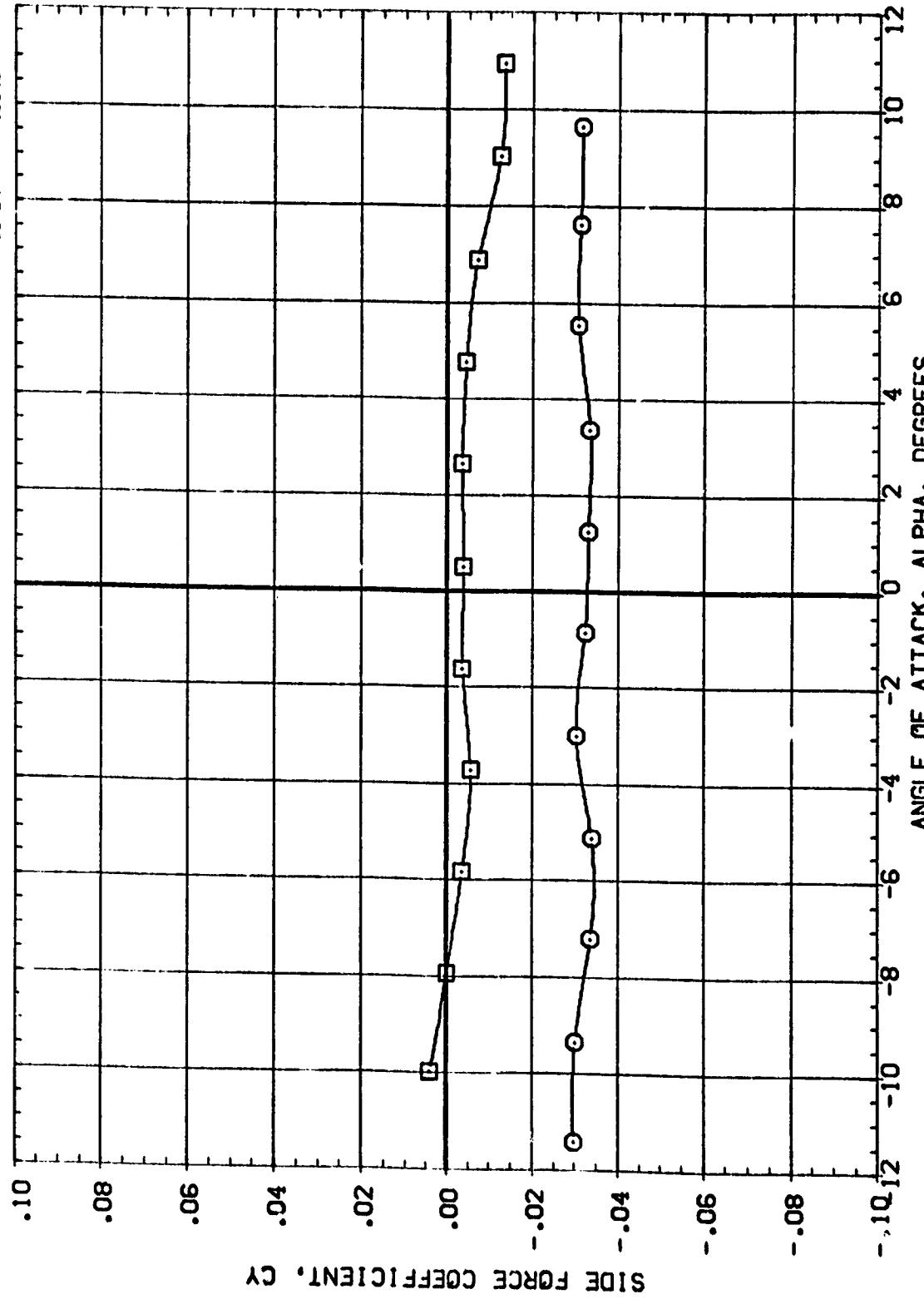


SRB AXIAL FORCE COEFFICIENT. CABs

EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 (D)MACH = 1.46

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
[890000] 8 MSFC 573(1A3)FC (03)(19)(S3)  
[890200] 8 MSFC 573(1A3)FC (03)(19)(S3) ORB. MISALNO.

REFERENCE INFORMATION  
SC. IN  
SREF 6.1980  
LREF 5.3130  
BREF 5.3130  
XRP 2.5450  
YRP .0000  
ZRP .0000  
SCALF .0010

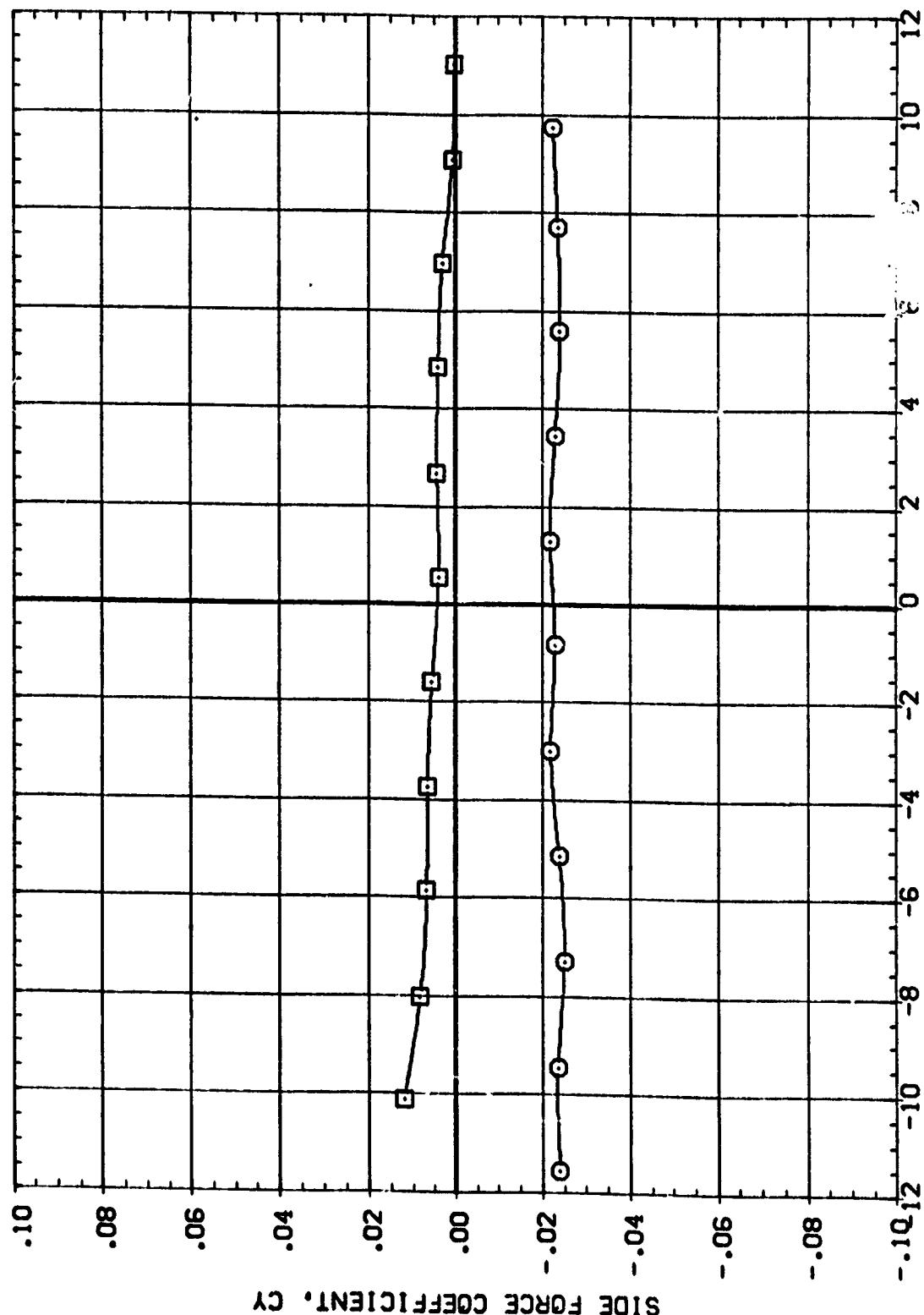


EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(A)MACH = .90

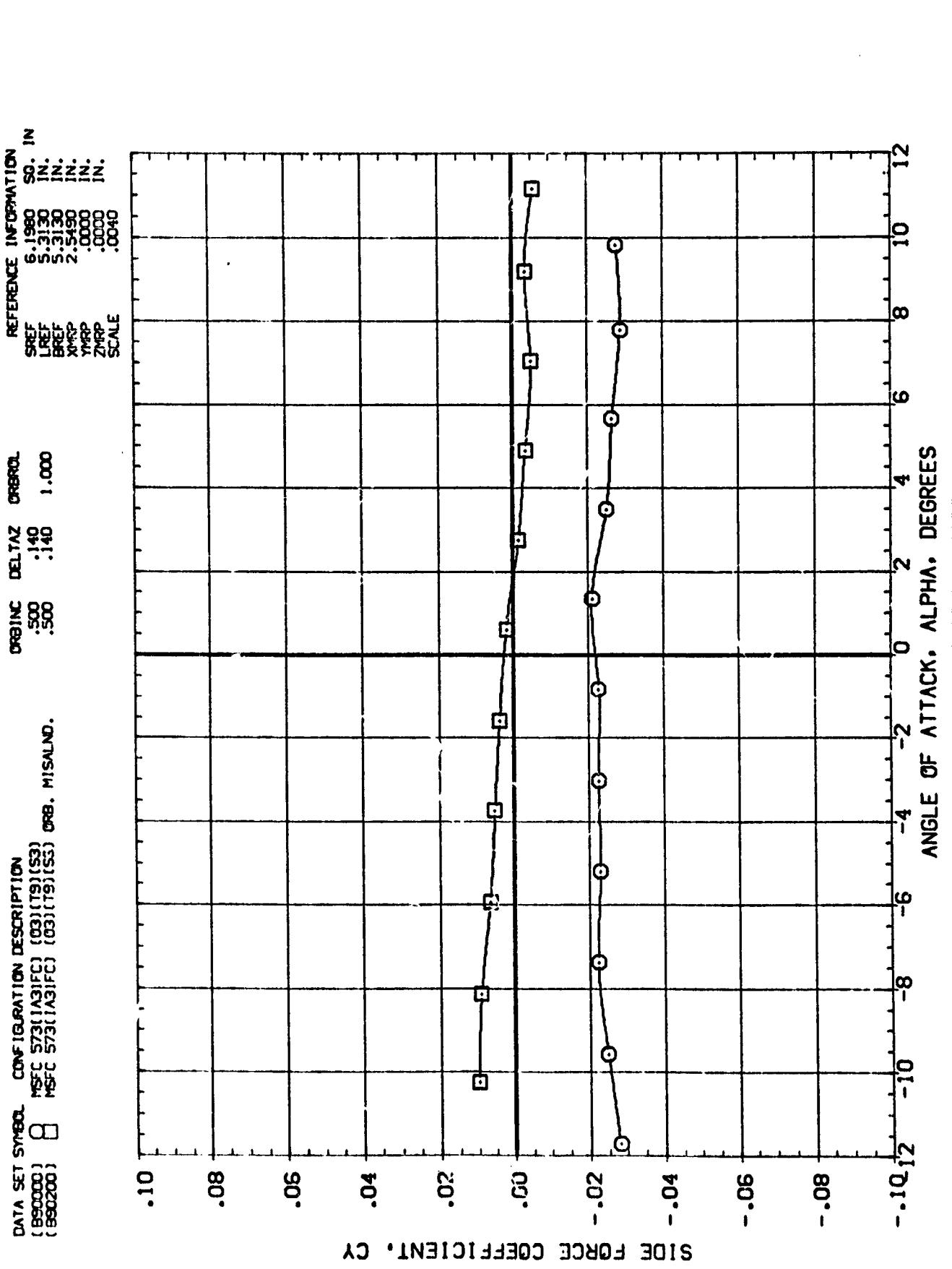
DATA SET SYMBOL      CONFIGURATION DESCRIPTION  
 (B800000)      NSFC 573(I)A3(IFC) (03)(T9)(S3)  
 (B80200)      NSFC 573(I)A3(IFC) (03)(T9)(S3)

REFERENCE INFORMATION  
 ORBINC .500      DELTAZ .140      GROL 1.000  
 (B800000)      SREF 6.1980 SD. IN.  
 (B80200)      LREF 5.3130 IN.  
 SREF 5.3130 IN.  
 XTRP 2.5590 IN.  
 YTRP .0000 IN.  
 ZTRP .0000 IN.  
 SCALE .0040



### EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

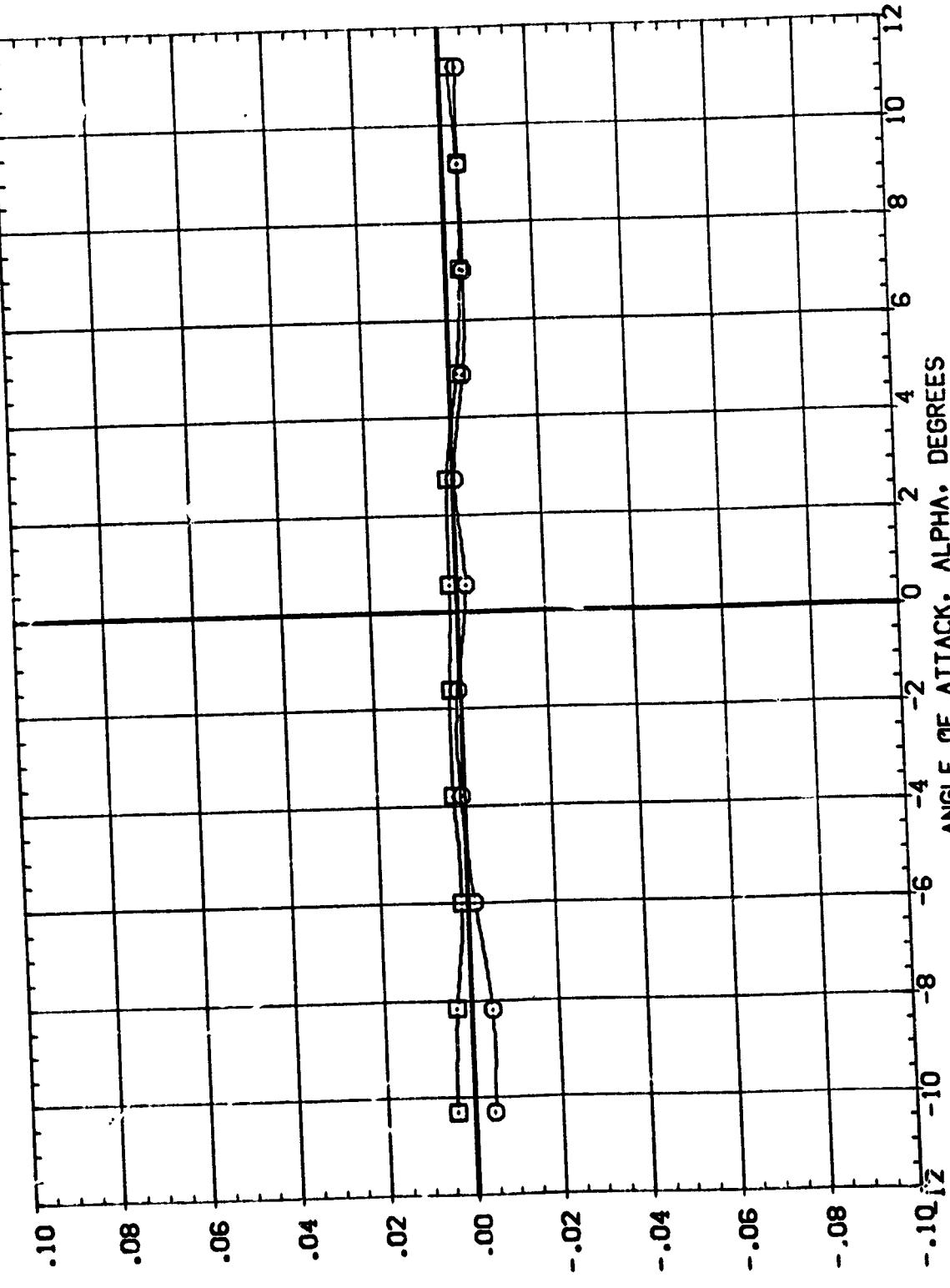


EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(C)<sub>MACH</sub> = 1.25

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
MSFC 573(1A31FC) (63)(T9)(53) ORB. MISALND.  
(B80000) (B80200) MSFC 573(1A31FC) (63)(T9)(53) ORB. MISALND.

REFERENCE INFORMATION  
SREF 6.1980 SQ. IN.  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
XREF 2.2890 IN.  
YREF .0000 IN.  
ZREF .0000 IN.  
SCALE .0040

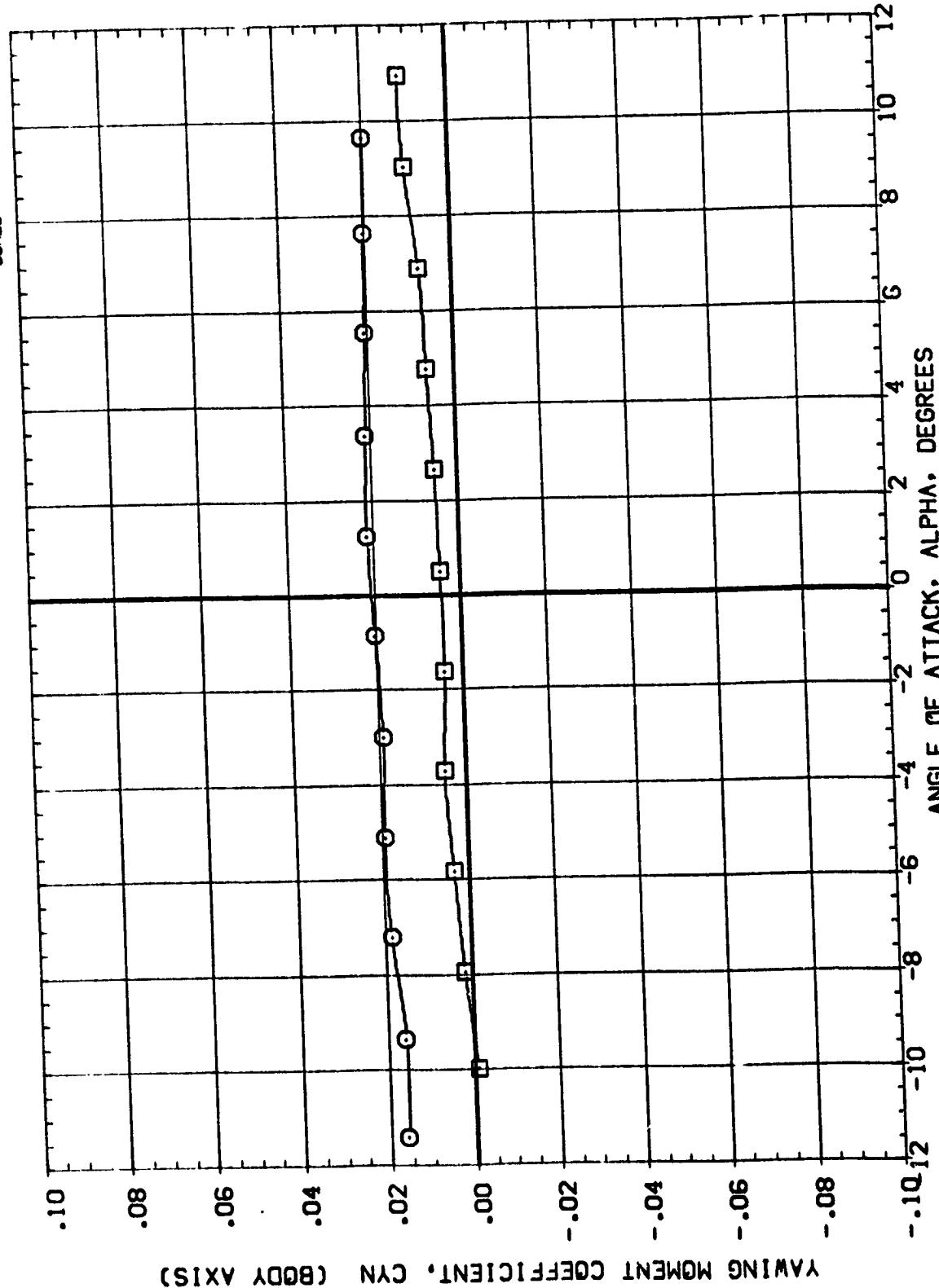


EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 $(CD)_MACH = 1.46$

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DATA SET S730. CONFIGURATION DESCRIPTION  
(B90000) 8 MSC 5731(A3IFC) (03)(19)(S3)  
(B90200) 8 MSC 5731(A3IFC) (03)(19)(S3) ORB. MISALN.

REFERENCE INFORMATION  
ORBINC DELTAZ ORBROL  
.500 .140 1.000  
.500 .140 1.000  
SREF LREF  
XMRP YMRP  
ZMRP  
SCALE

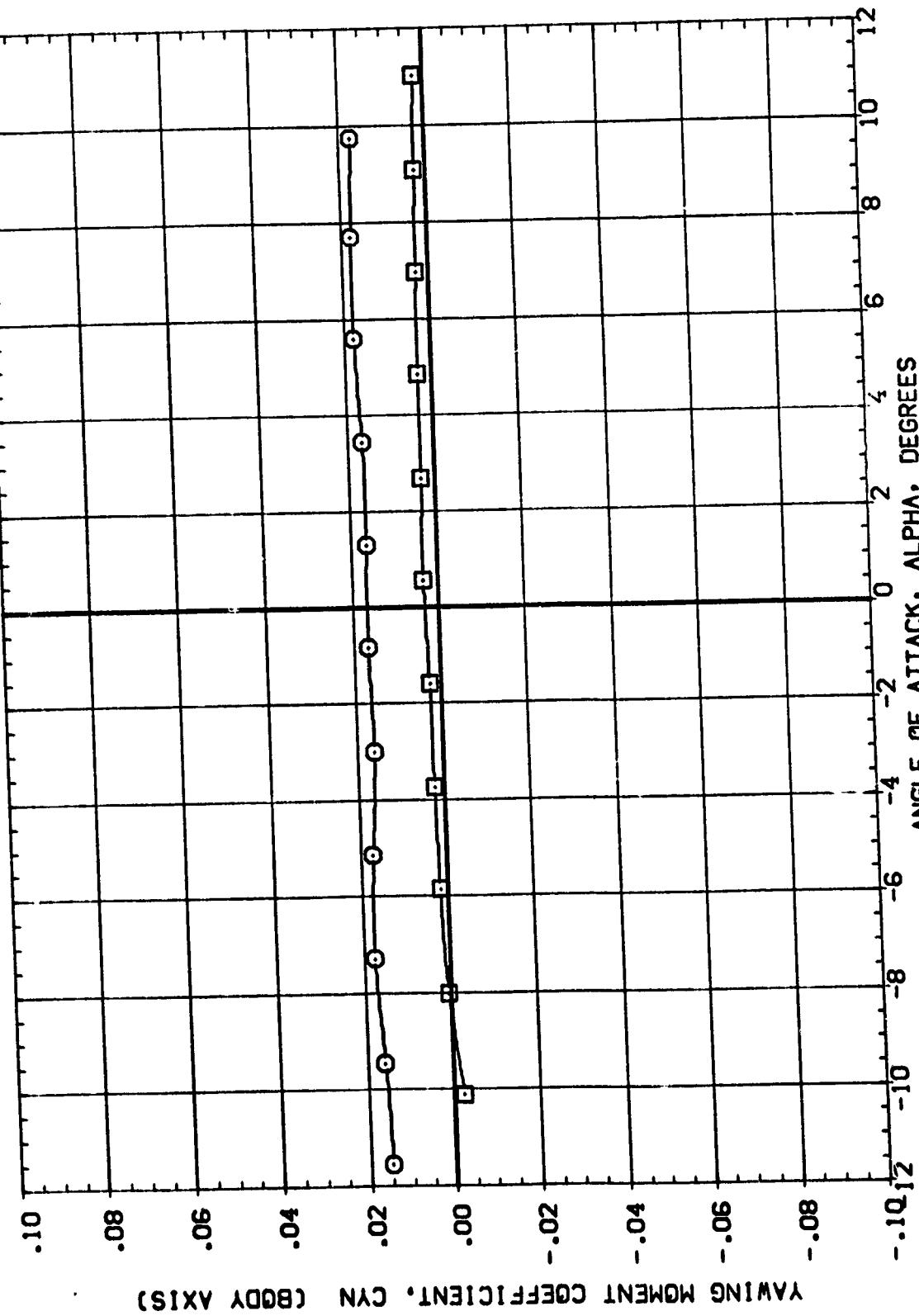


EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(A)MACH = .90

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) 8 MSGC 573(I)A3(FC) (03)(T9)(S3) ORB. MISALND.  
 (B90200)

ORB INC DELTAZ ORBRL  
 .500 .140  
 .500 1.000  
 REFERENCE INFORMATION  
 SREF 6.1980 SQ. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XRP 2.5490 IN.  
 YRP .0000 IN.  
 ZRP .0040 IN.

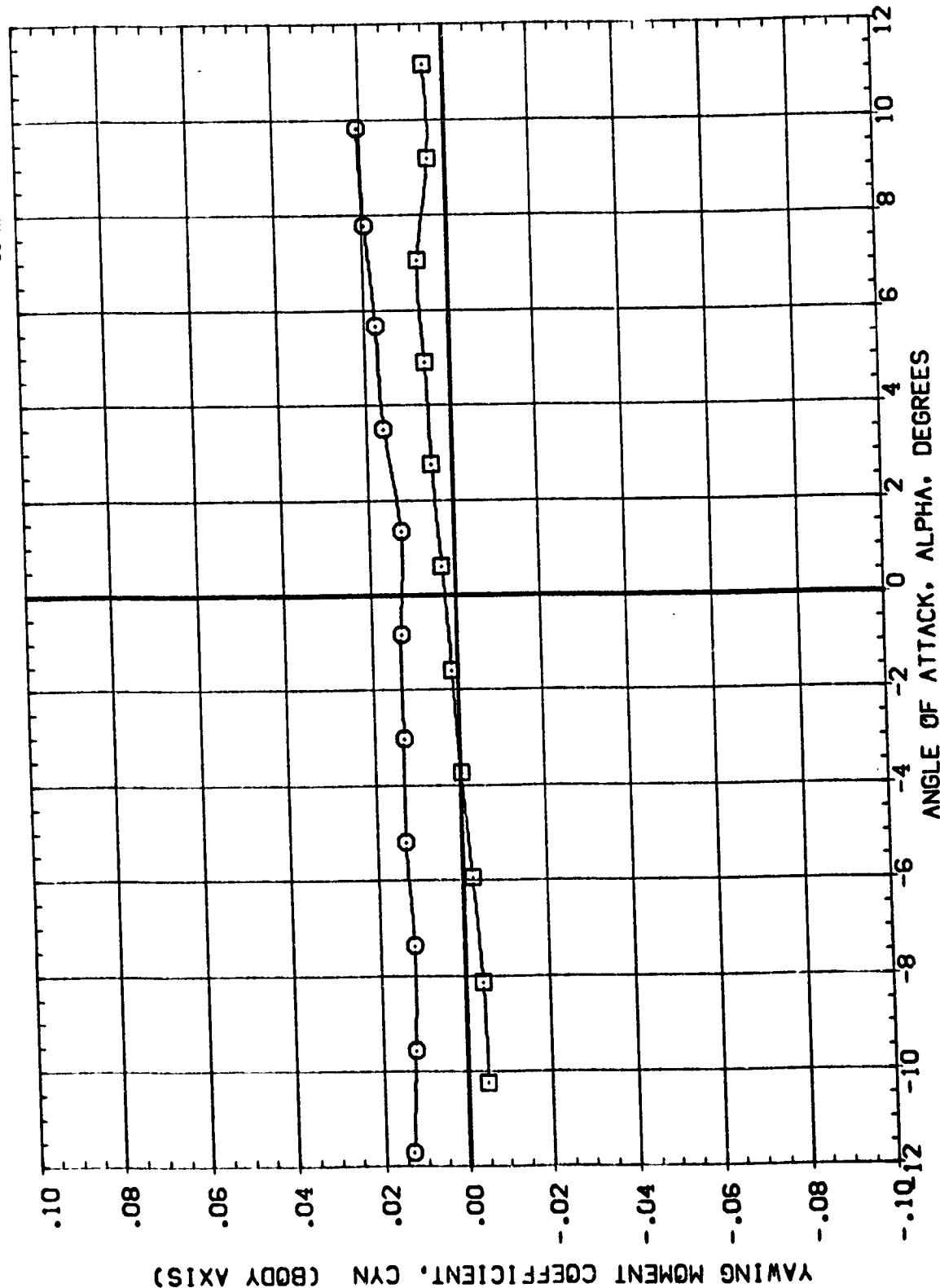


EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
[ESC0200] MSFC 5731(A3)FC [03](T9)(S3)  
[ESC0200] MSFC 5731(A3)FC [03](T9)(S3) ORB. MISALD.

REFERENCE INFORMATION  
ORBITC DELTAZ ORBRD.  
.500 .140 1.000  
.500 .140 1.000  
SREF LREF DREF  
XMRP YMRP ZMRP  
SCALE

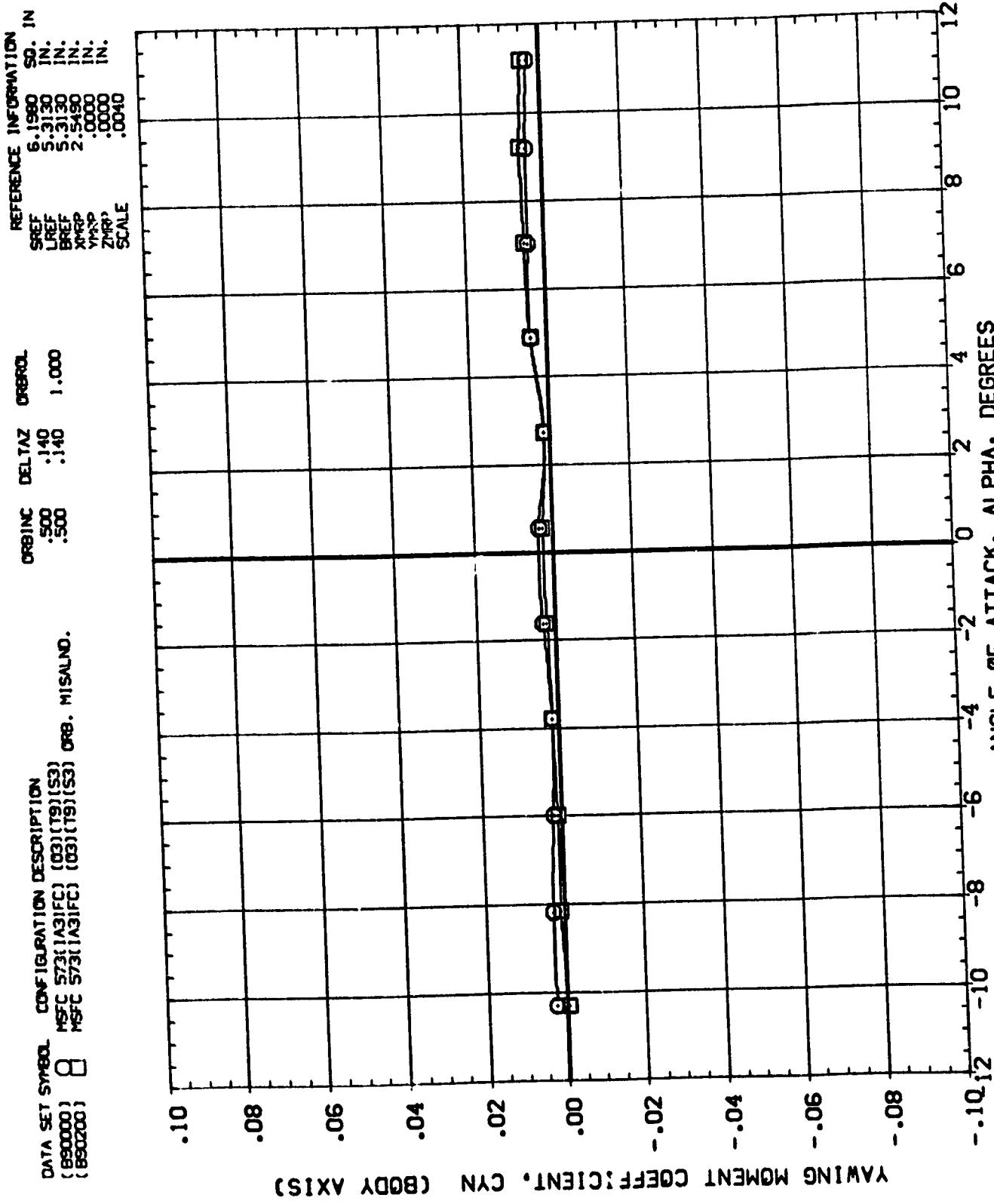


EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
(C)MACH = 1.25

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EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

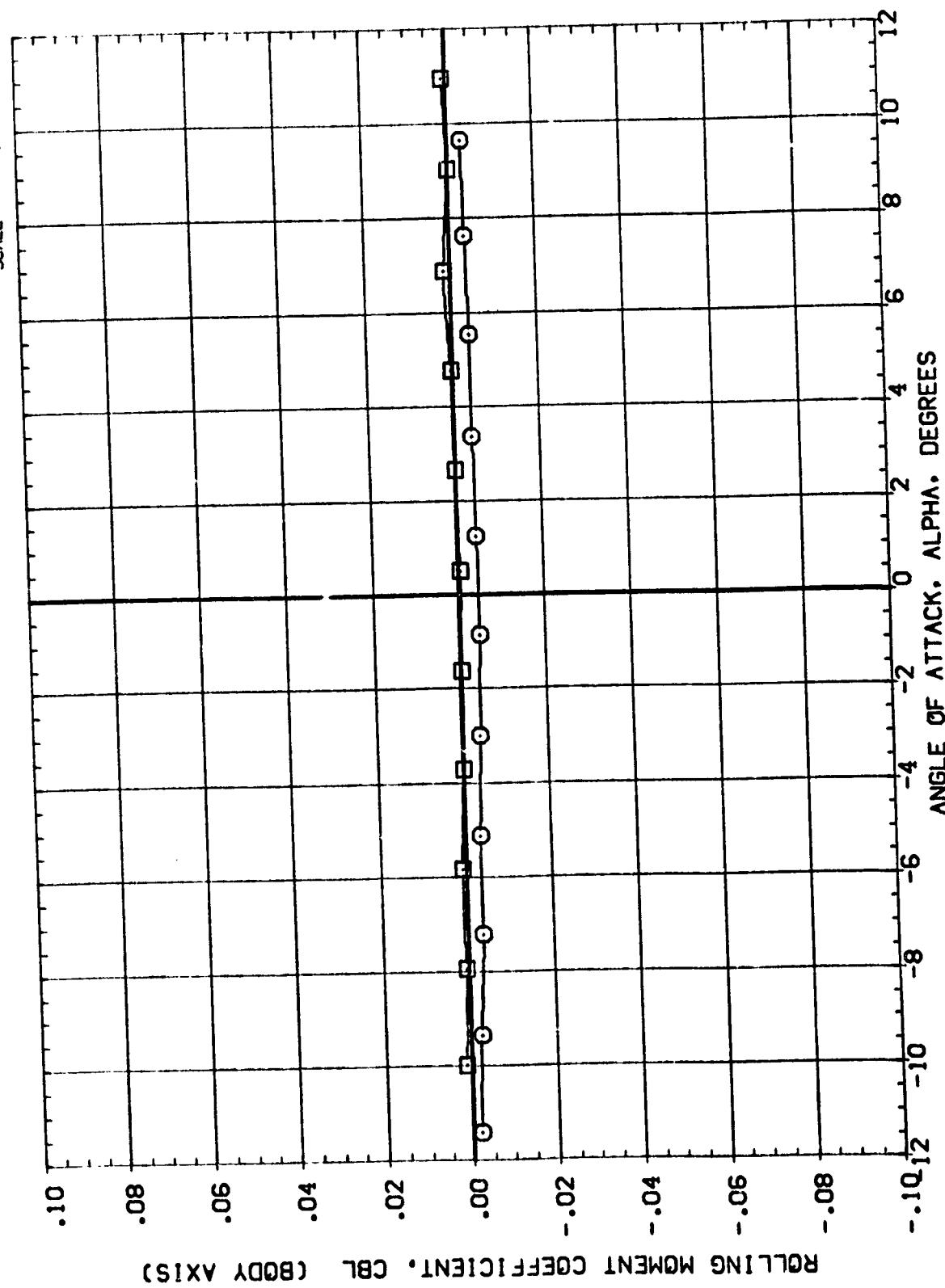
(MACH = 1.46



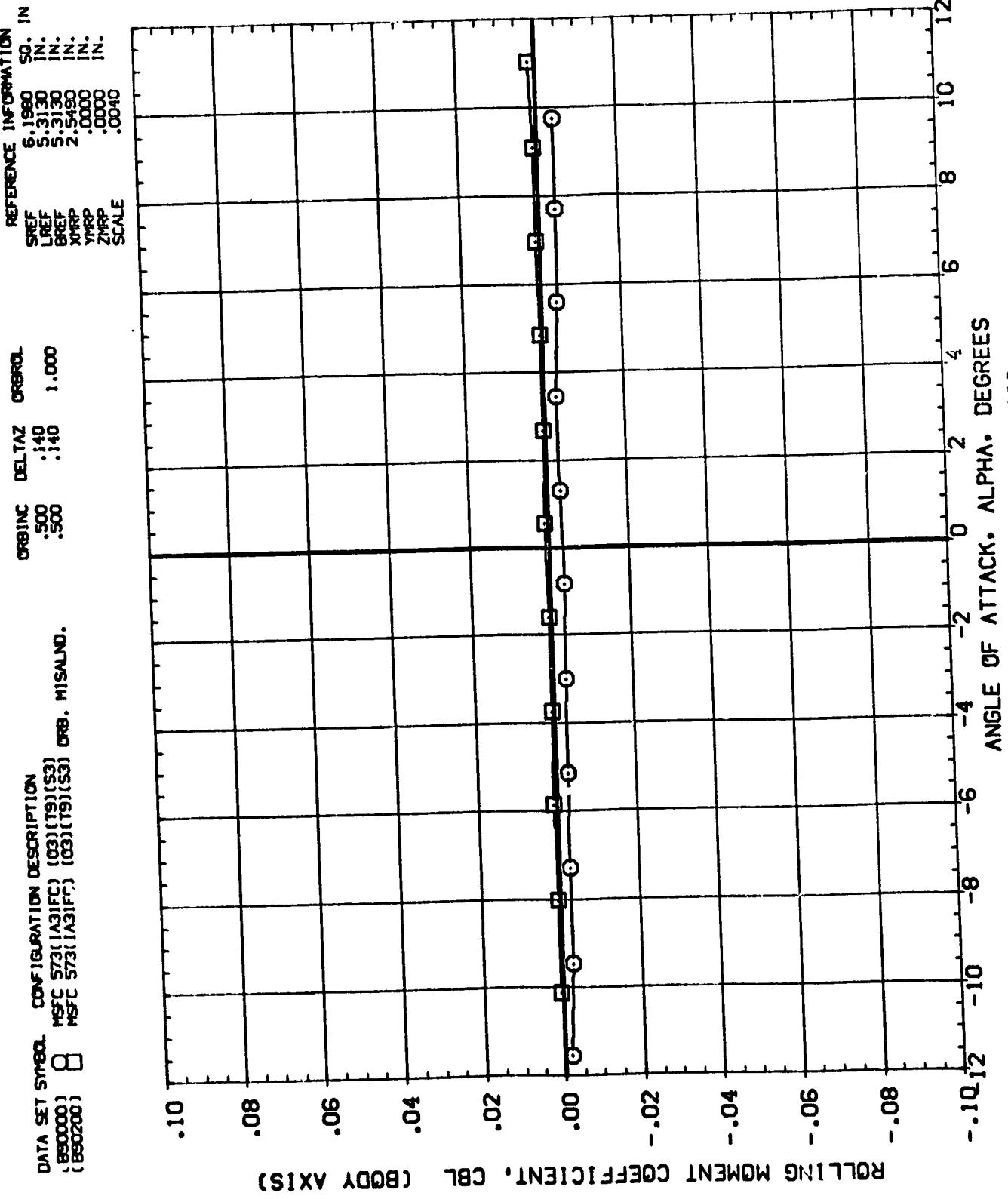
REFERENCE INFORMATION

SREF	6.1980	SO. IN.
LREF	5.3130	IN.
BREF	5.3130	IN.
XMRP	2.5490	IN.
YMRP	.0000	IN.
ZMRP	.0040	IN.
SCALE		

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 MSFC 5731(A3)FC (03)(19)(S3)  
 (850000) (850200)  
 MSFC 5731(A3)FC (03)(19)(S3) ORB. MISALD.



EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 $(\Delta MACH = .90)$



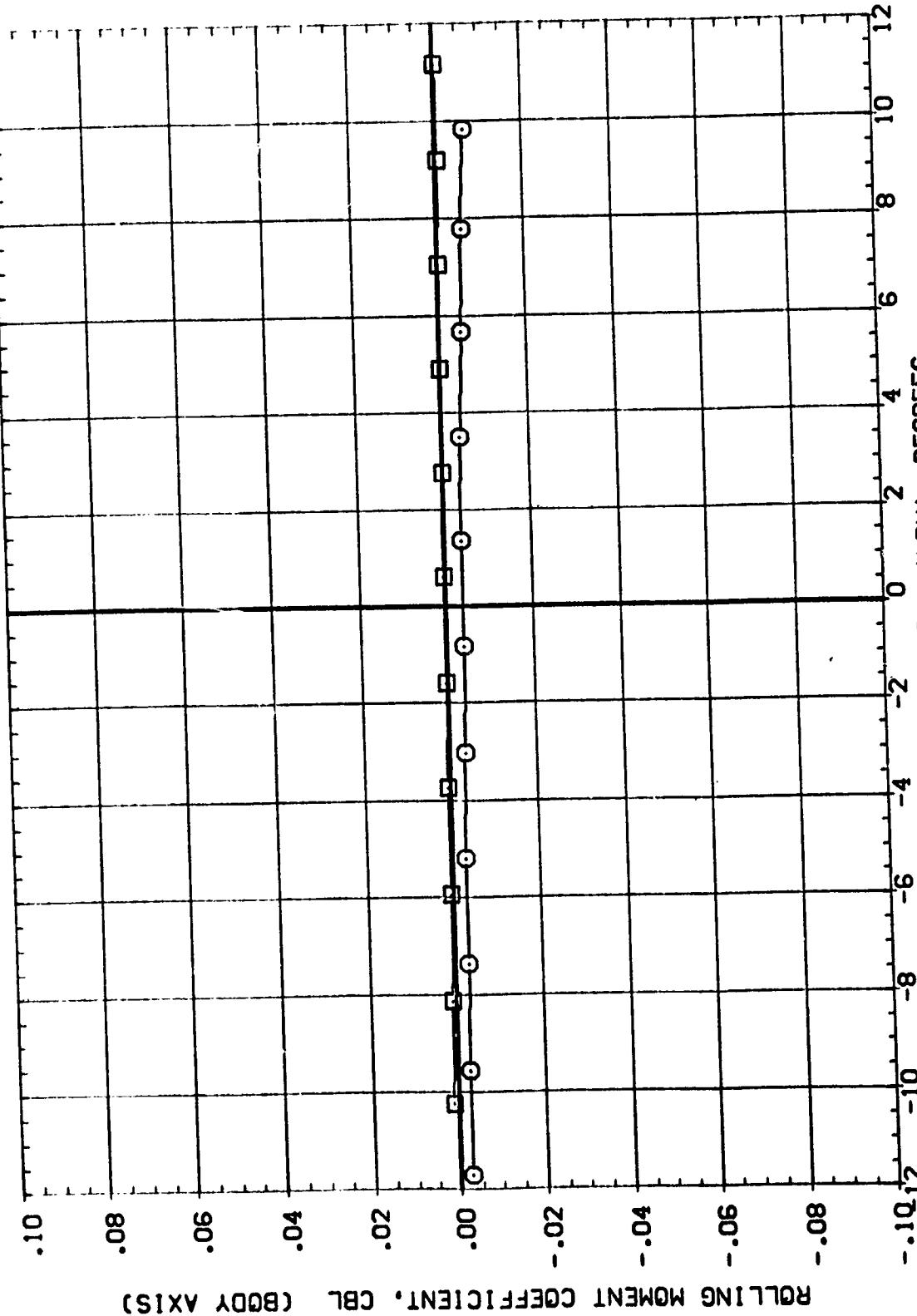
EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 $(B)MACH = 1.05$

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B6000C) NSFC 573([A3]FC) [03][T9][S3] ORB. MISALND.  
 (B6220C) NSFC 573([A3]FC) [03][T9][S3] ORB. MISALND.

ORB INC.	DELTAZ	CRBL
.500	.140	1.000
.500	.140	

REFERENCE INFORMATION

REF	6.1980	SC. IN
LREF	5.3130	IN.
BREF	5.3130	IN.
XMRP	2.5490	IN.
YMRP	.0000	IN.
ZMRP	.0000	IN.
SCALE	.0040	



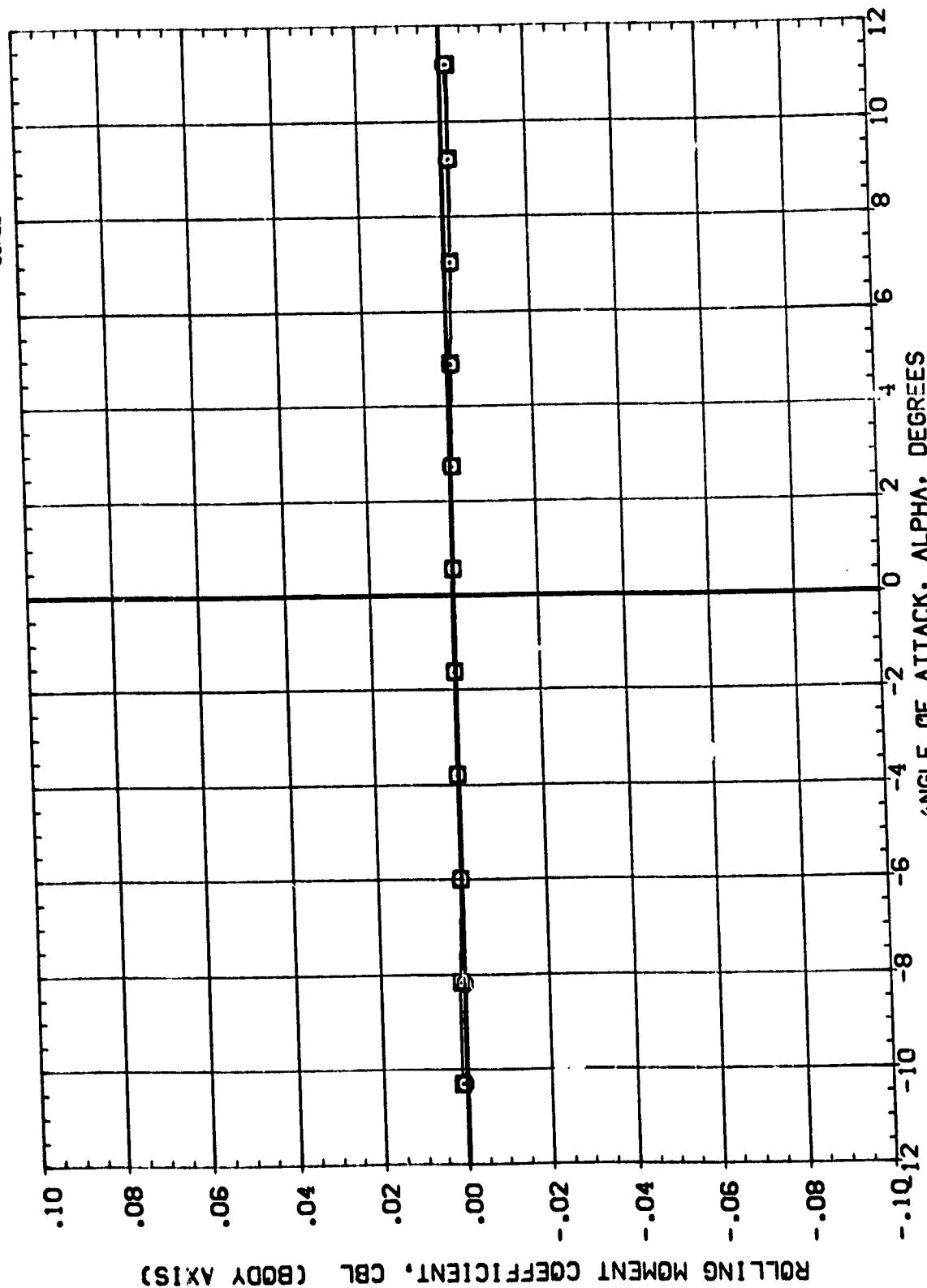
### EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS

$(C)MACH = 1.25$

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DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) B NSFC 5731(A3)FC (03) (T9)(S3) ORB. MISALND.  
 (B90200)

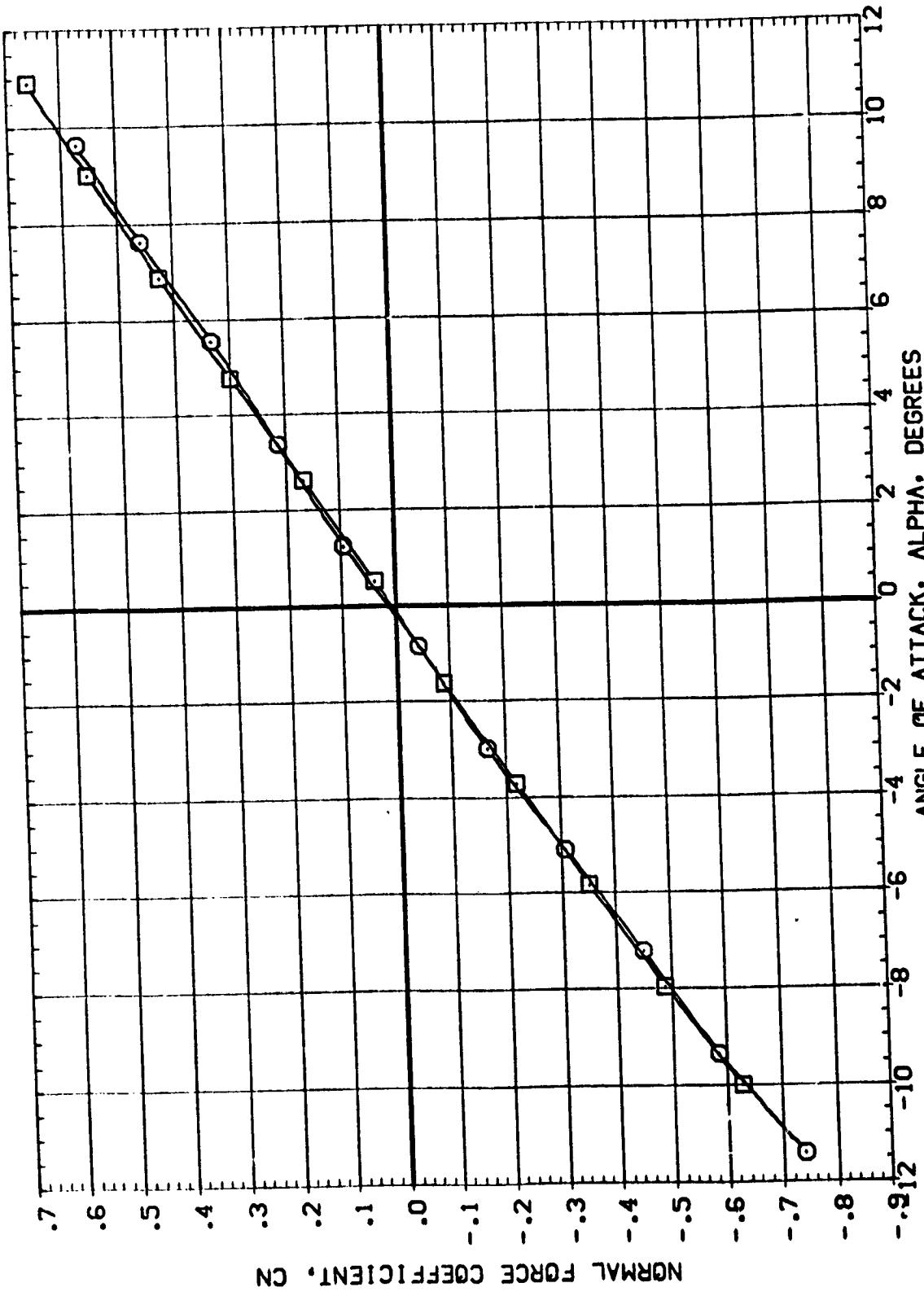
REFERENCE INFORMATION IN  
 SPREF 6.1980 SD. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE



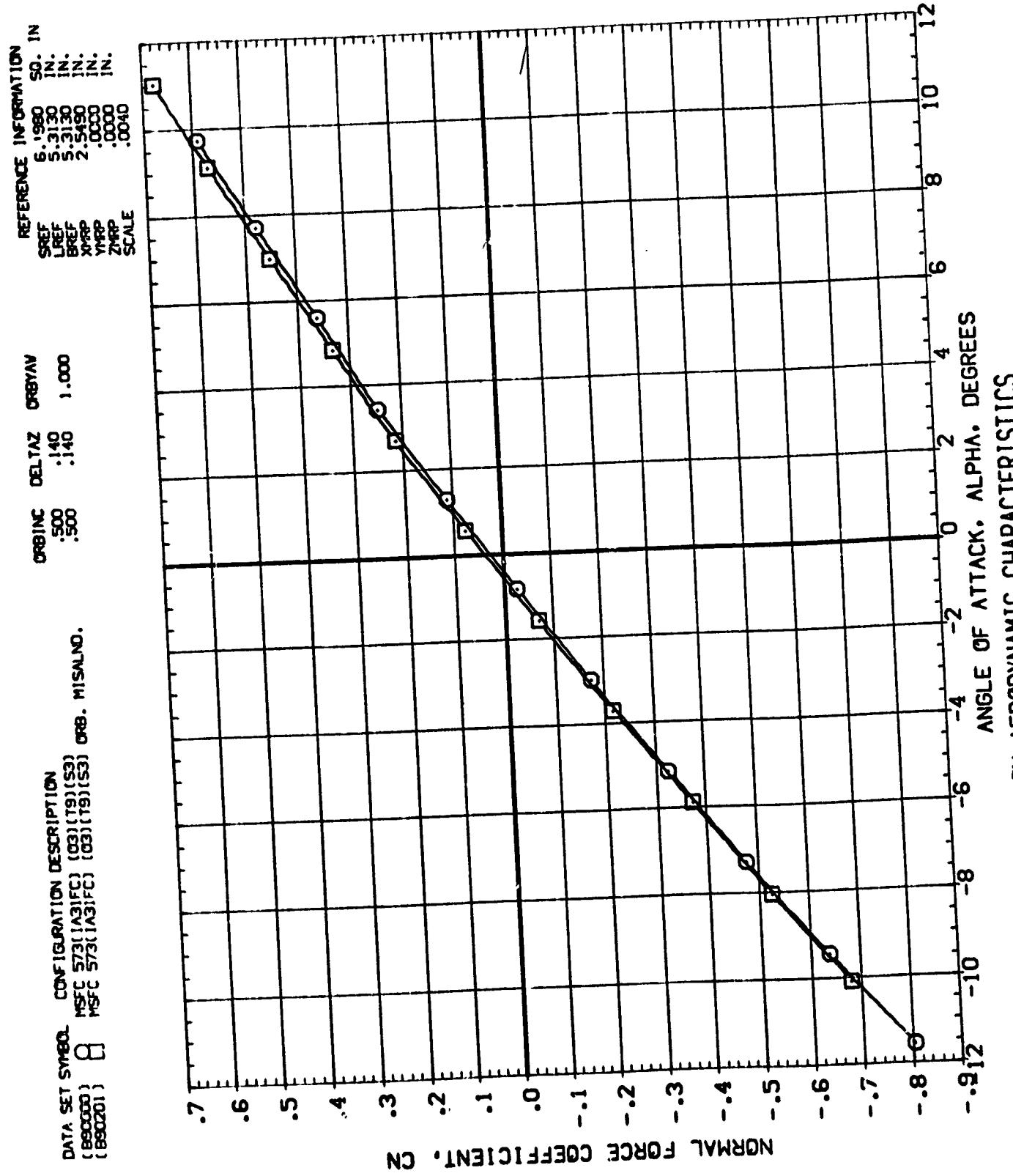
EFFECT OF ORBITER ROLL ON AERODYNAMIC CHARACTERISTICS  
 $(C)_MACH = 1.46$

DATA SET SYMBOL      CONFIGURATION DESCRIPTION  
 (B90000)      NSFC 573(A3)FC (03)(19)(S3)  
 (B90201)      NSFC 573(A3)FC (03)(19)(S3)

REFERENCE INFORMATION  
 SREF 6.1980 SD. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE



EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS  
 $(\Delta MACH = .90)$

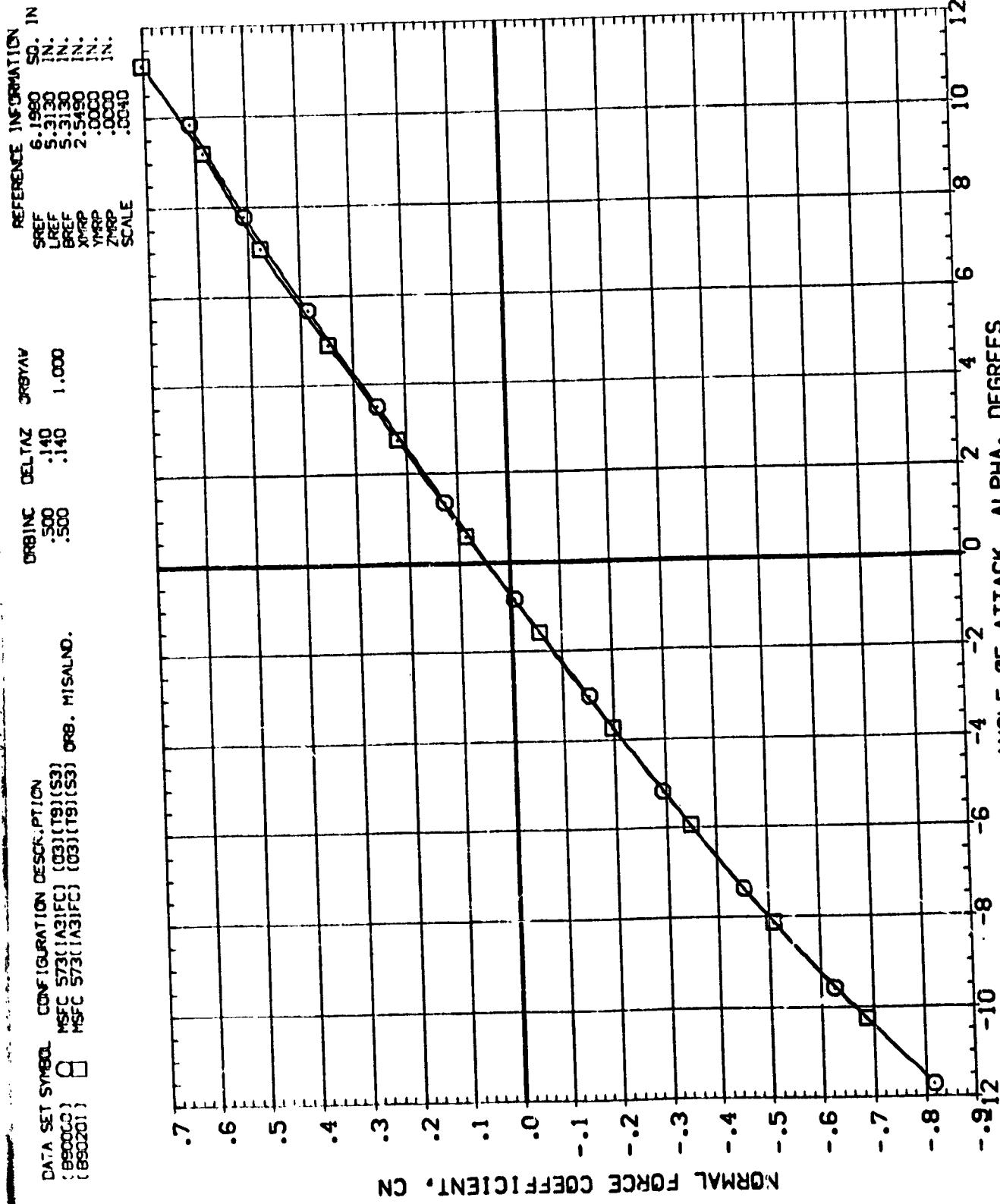


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EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(E)MACH = 1.05

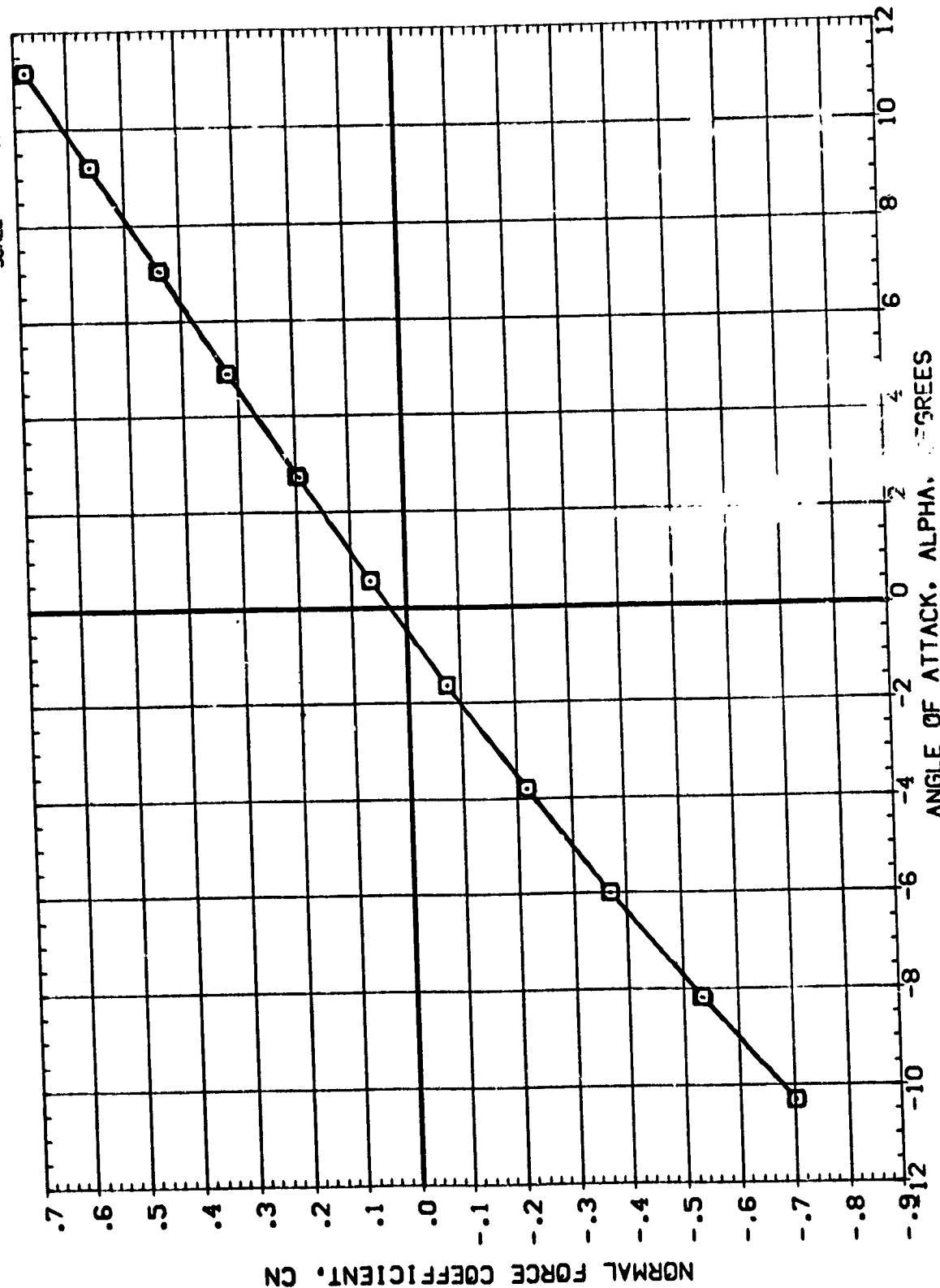
DATA SET SYMBOL CONF/CONFIGURATION DESCRIPT/ION  
 (B800C) NSFC 573(1A3)FC (03)(19)(S3)  
 (850201) NSFC 573(1A3)FC (03)(19)(S3) MIS/AL NO.



### EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS (C<sub>MACH</sub> = 1.25)

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DATA SET SYMBOL      CONFIGURATION DESCRIPTION  
 (B90000)      RSEFC ST73(1A3)IFC (03)(T9)(S3) ORB. MISALND.  
 (B90251)      RSEFC ST73(1A3)IFC (03)(T9)(S3) ORB. MISALND.  
 REFERENCE INFORMATION  
 SREF      6.1980 SD. IN.  
 LREF      5.3130 IN.  
 BREF      5.3130 IN.  
 XTRP      2.5450 IN.  
 YTRP      .0000 IN.  
 ZTRP      .0000 IN.  
 SCALE      .0040



EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTIC

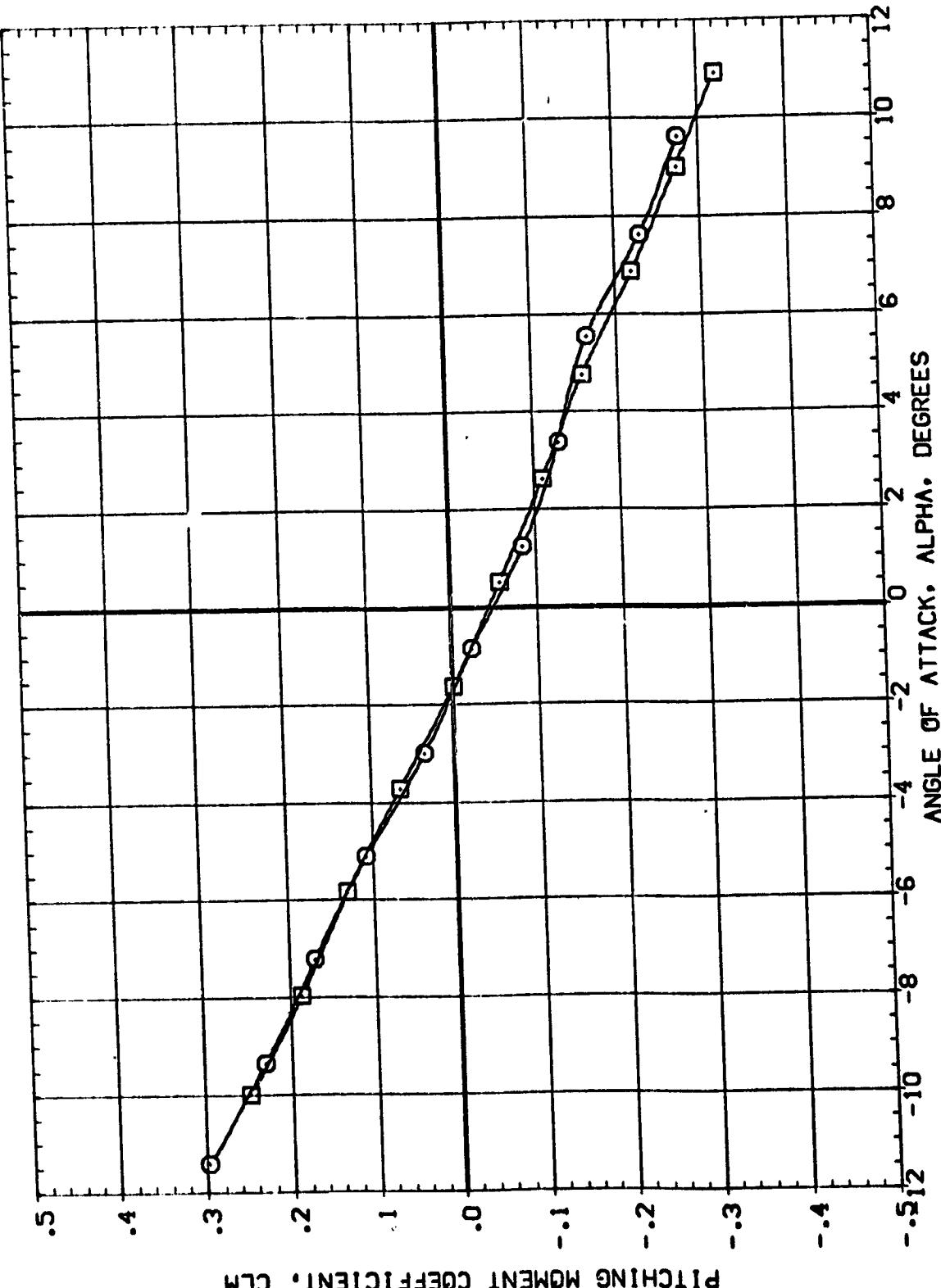
(MACH = 1.46

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REFERENCE INFORMATION  
 SO. IN.  
 SRREF 6.1980  
 LREF 5.3130  
 BREF 5.3130  
 XMRP 2.5490  
 YMRP .0000  
 ZMRP .0040  
 IN.  
 SCALE

ORB INC DELTAZ ORBYAN  
 .500 .140 .140  
 .500 1.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (6920000) MSFC S731(A3)FC (03)(T9)(S3) ORB. MISALD.  
 (690201) MSFC S731(A3)FC (03)(T9)(S3)



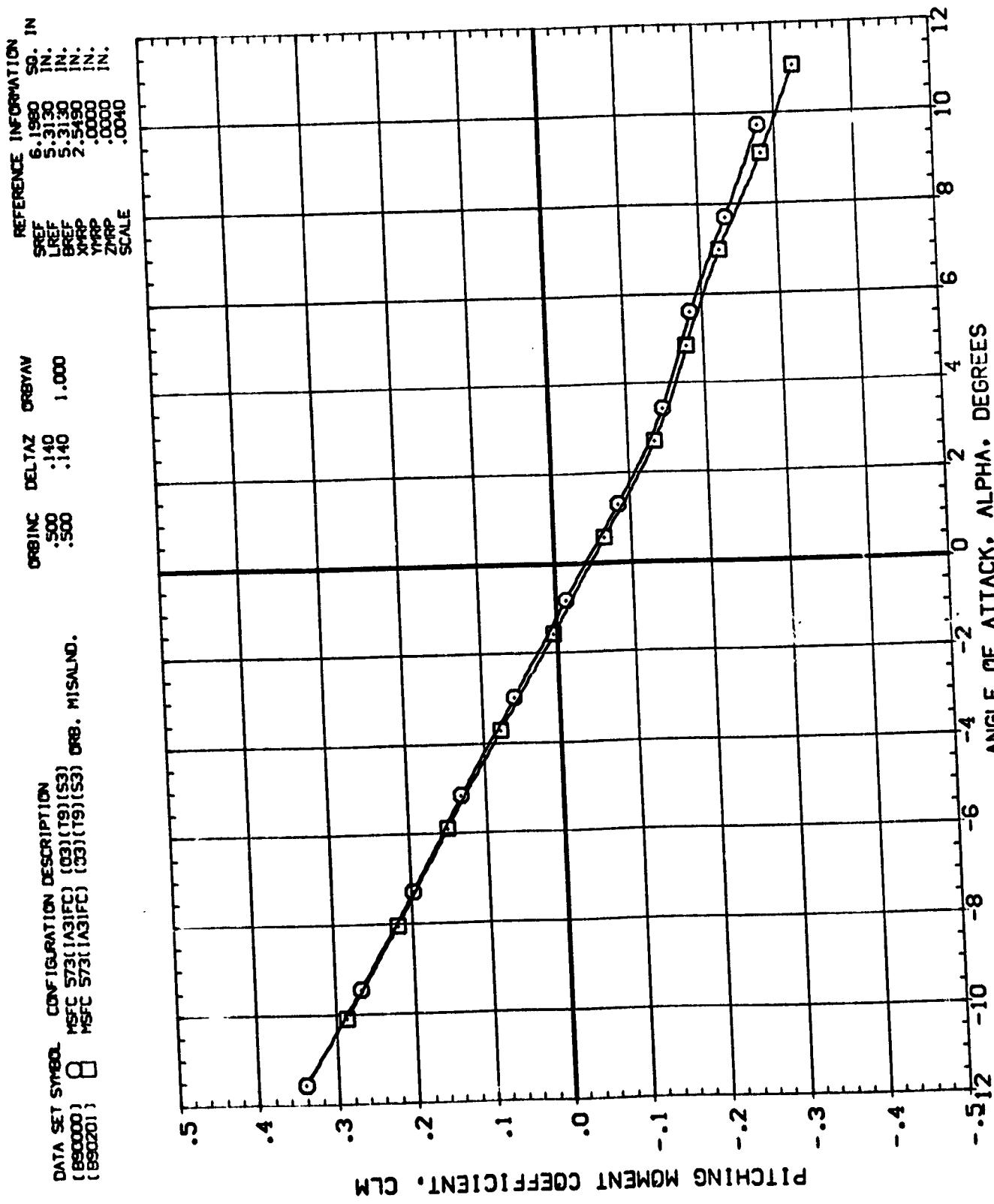
EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS  
 (MACH = .90)

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EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

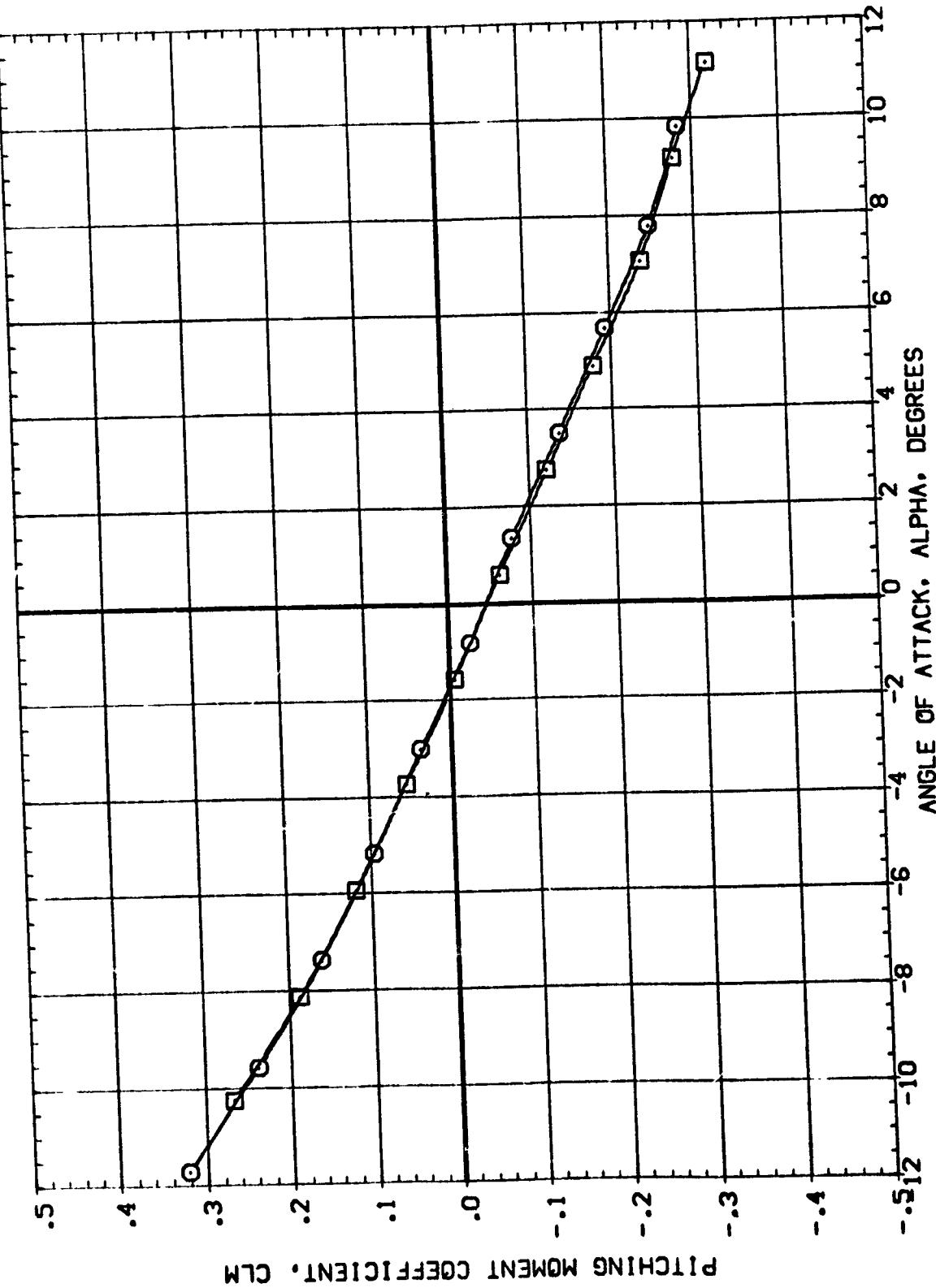
(B)MACH = 1.05



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (S50000) MSFC 573(1A3)FC (G3)(19)(S3)  
 (880201) MSFC 573(1A3)FC (G3)(19)(S3) ORB. MISALNO.

ORB INC	DELTAZ	ORB YAW
.500	:140	1.000
.500	:140	

REFERENCE INFORMATION  
 SREF 6.1980 SQ. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE

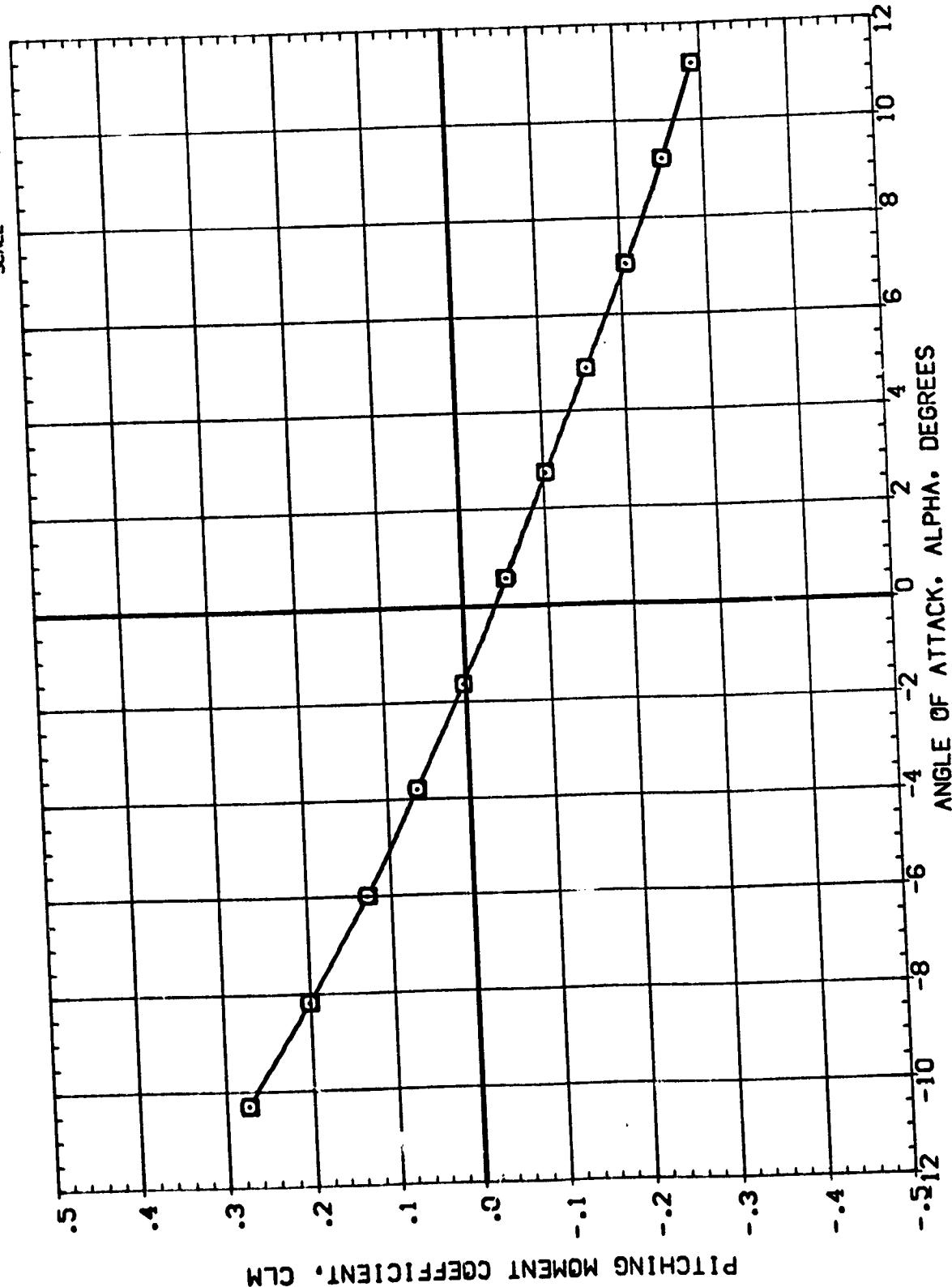


EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS  
 $(C)_MACH = 1.25$

REF ID: A6910

REFERENCE INFORMATION  
SREF 6.1980 SQ. IN.  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
XMRP 2.5490 IN.  
YMRP .0000 IN.  
ZMRP .0000 IN.  
SCALE .0040

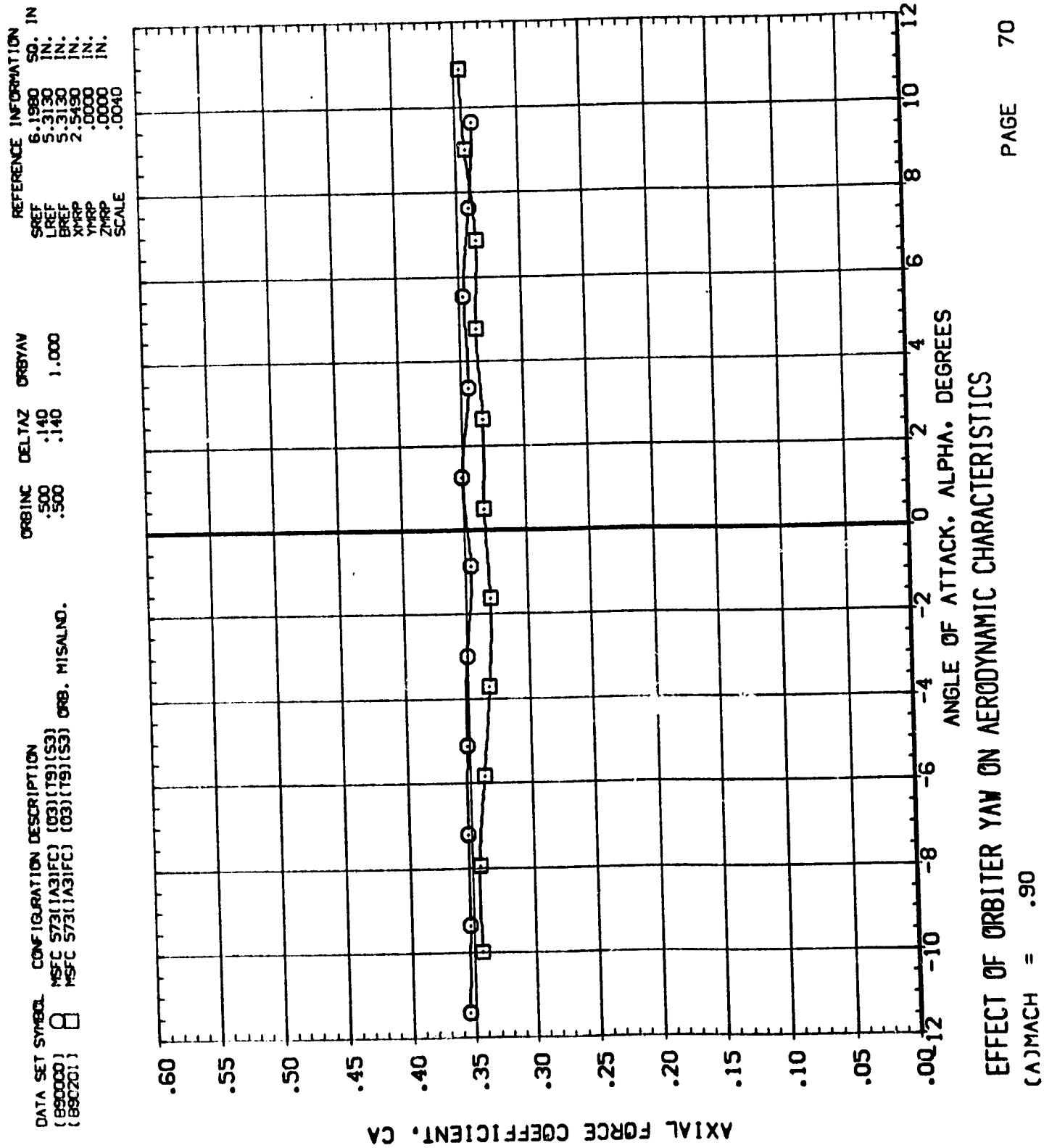
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
(B00000) 8 HSC S73(I3)FC (03)(19)(53) ORB. MISALNO.  
(B00001)



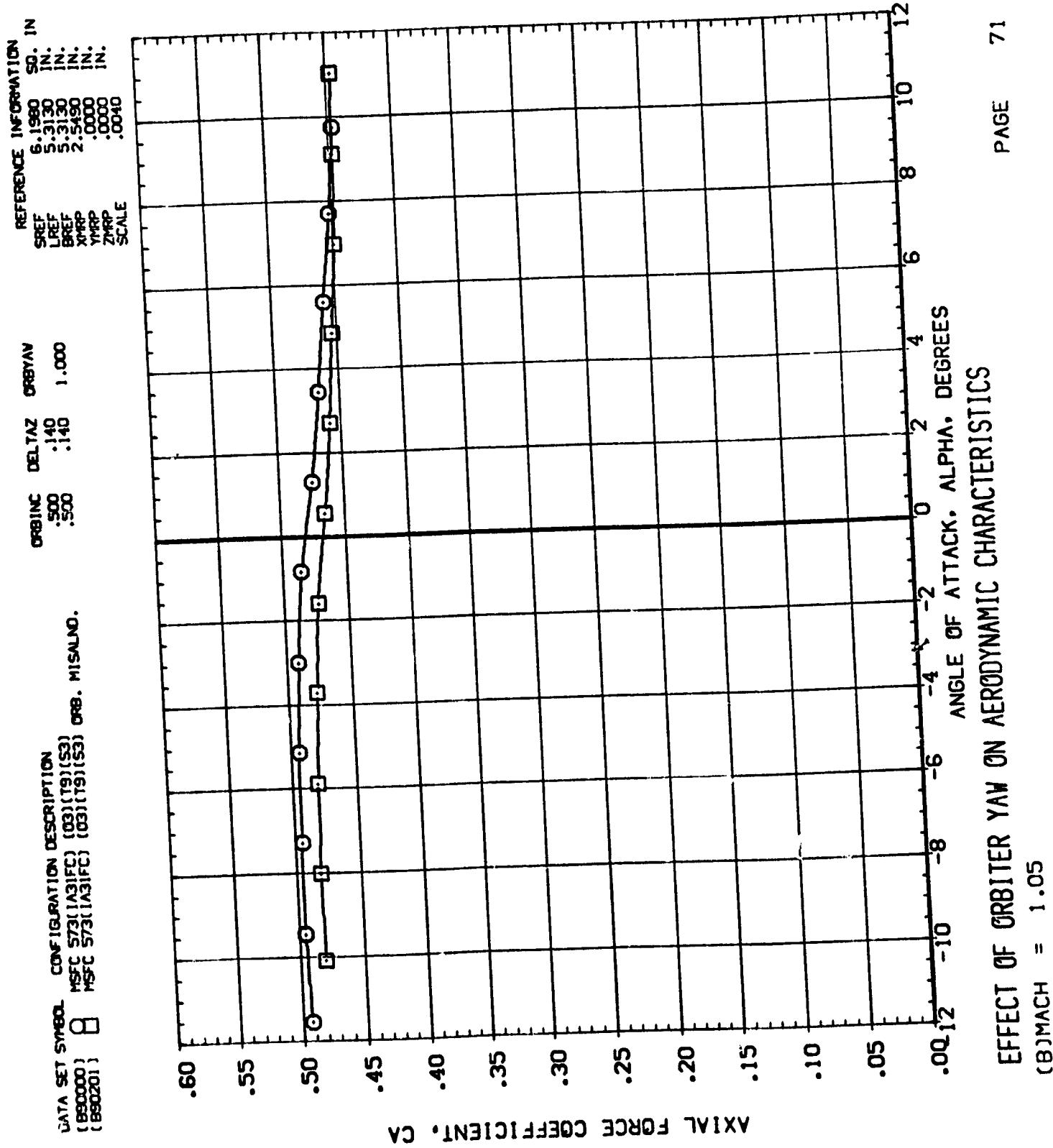
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EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(D)MACH = 1.46

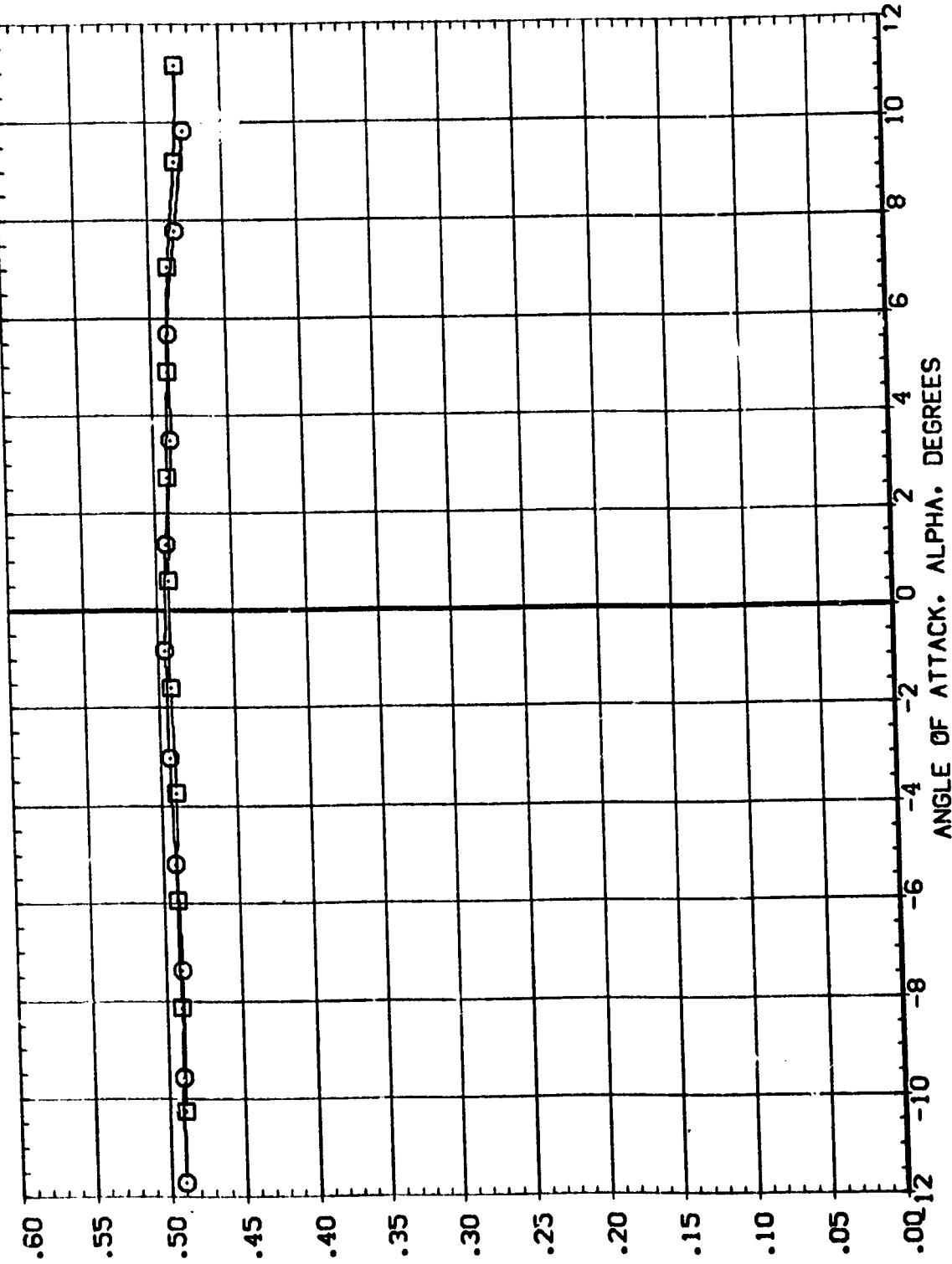


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DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (S600000)  NSFC 573(A3)FC (03)(T9)(S3) ORB. MISN. NO.  
 (S60201)

ORB INC	DELTAZ	CRBYAV	REFERENCE INFORMATION
.500	.140	1.000	SREF 6.1980 SQ. IN
	.140		LREF 5.3130 IN.
			BREF 5.3130 IN.
			XMRP 2.5590 IN.
			YMRP .0000 IN.
			ZMRP .0000 IN.
			SCALE .0040



AXIAL FORCE COEFFICIENT, CA

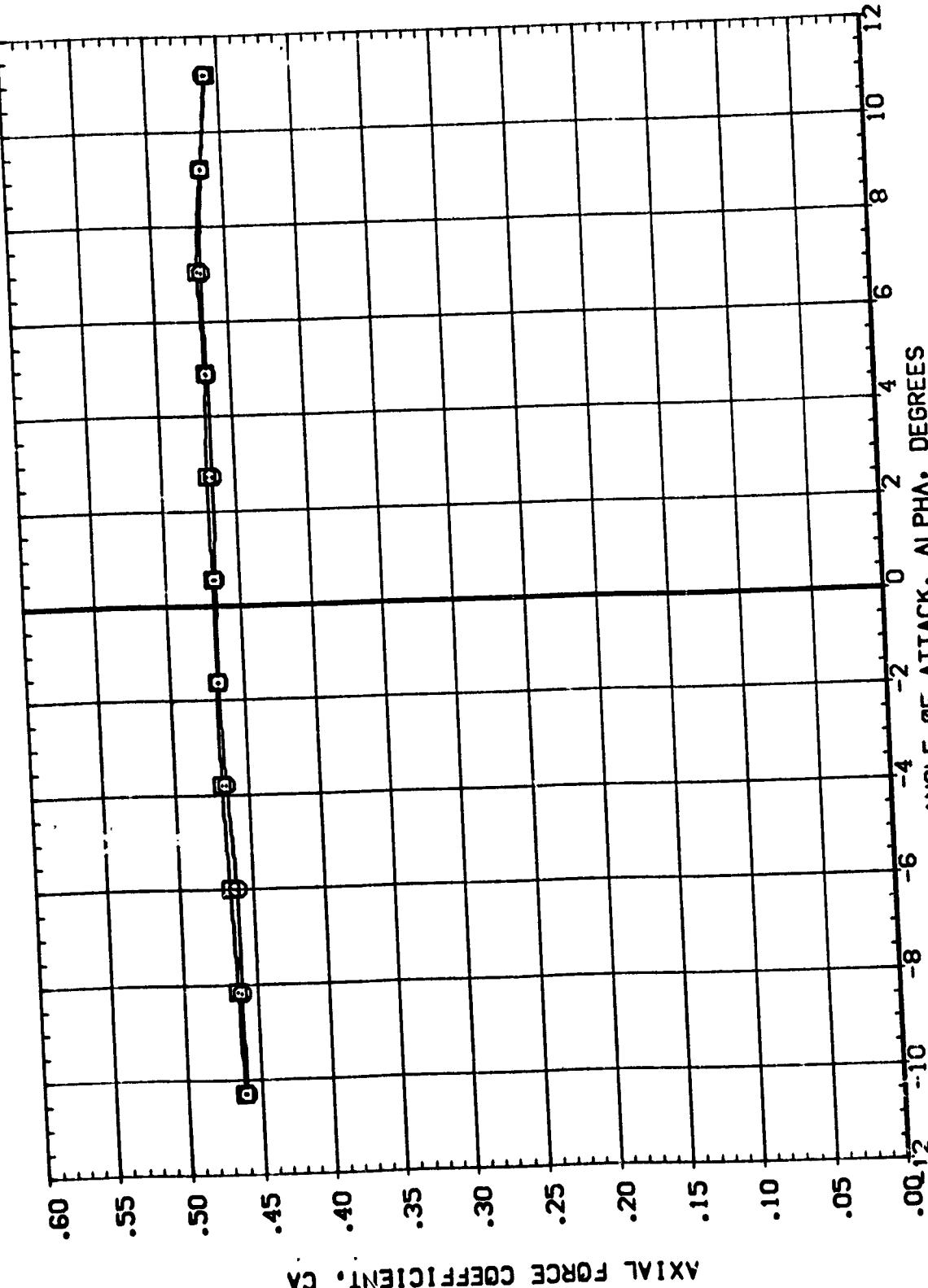
EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

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DATA SET SYMBOL:  HSF C 573(1A3)FC (S3)(T9)(S3) ORB. MISALNO.  
 (BB0000) (BB0001) (BB0020)

ORB INC	DELTAZ	ORBYAW	REFERENCE INFORMATION
.500	.140	1.000	SREF 6.1980 SO. IN LREF 5.3130 IN. BREF 5.3130 IN. XMRP 2.5490 IN. YMRP .0000 IN. ZMRP .0040 IN.
			SCALE

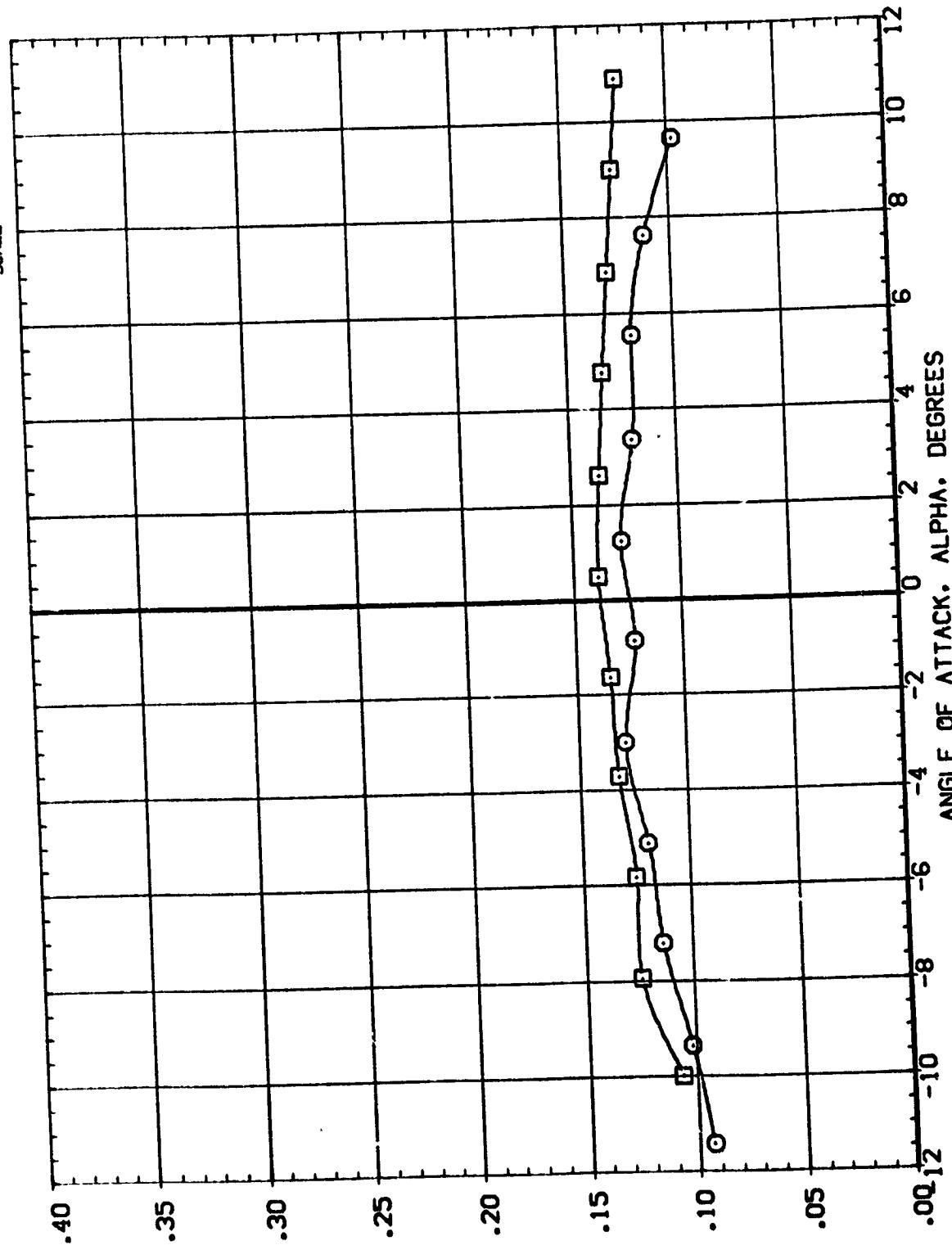


EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(D)<sub>MACH</sub> = 1.46

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 MSFC 573(1A3)FC (03)(T9)(S3)  
 MSFC 573(1A3)FC (03)(T9)(S3) ORB. MISALNO.

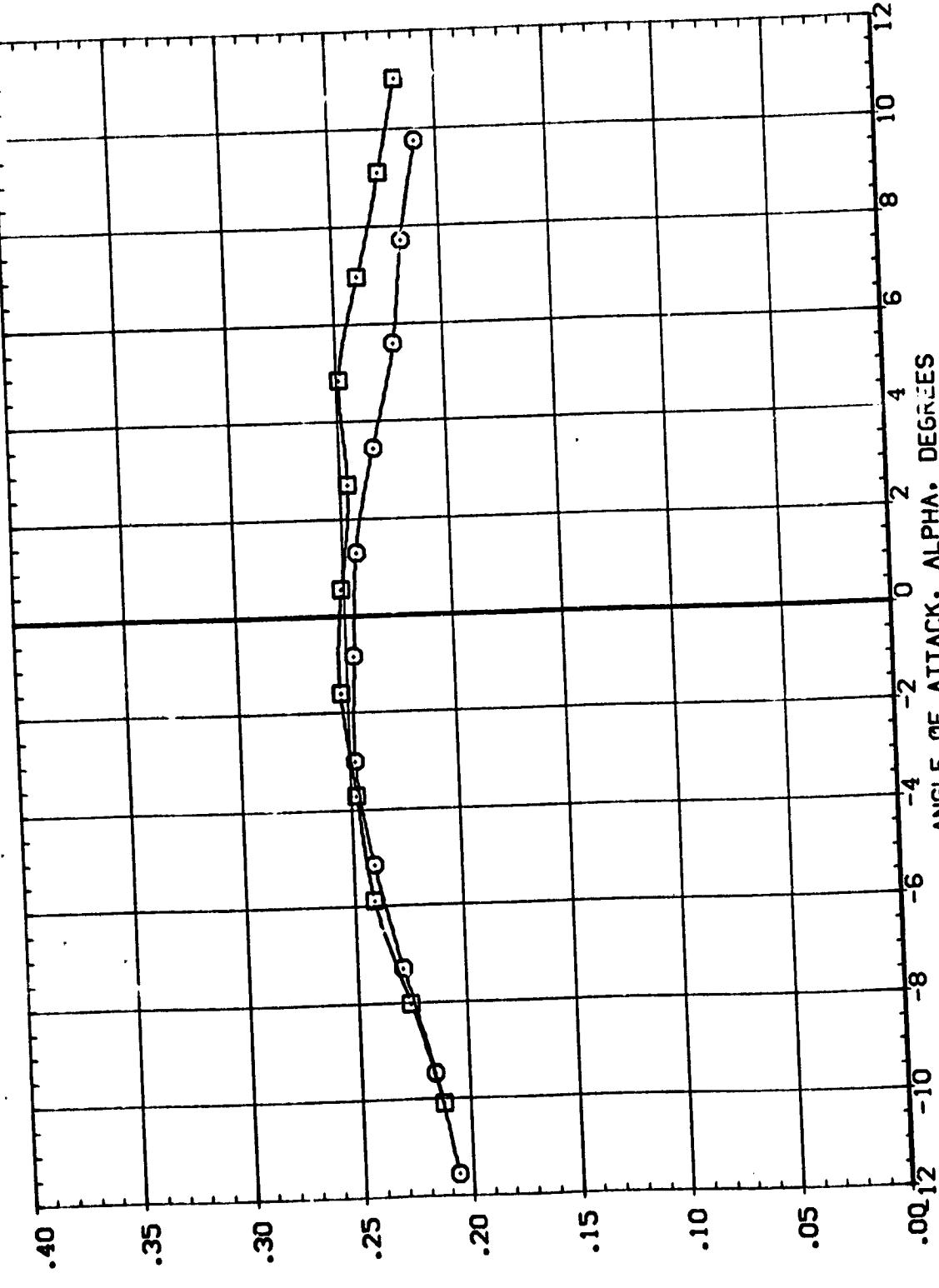
REFERENCE INFORMATION  
 SREF 6:1980 SC. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE



GREBEBODY AXIAL FORCE COEFFICIENT. CAF

EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS  
 $(\Delta)MACH = .90$

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B500000) NSFC 573(I)A3(IFC) (03)(T9)(S3) ORB. M/S NO.  
 (B50201) 
 REFERENCE INFORMATION  
 SREF 6,1980 SQ. IN.  
 LREF 5,3130 IN.  
 BREF 5,3130 IN.  
 XMRP 2,5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE



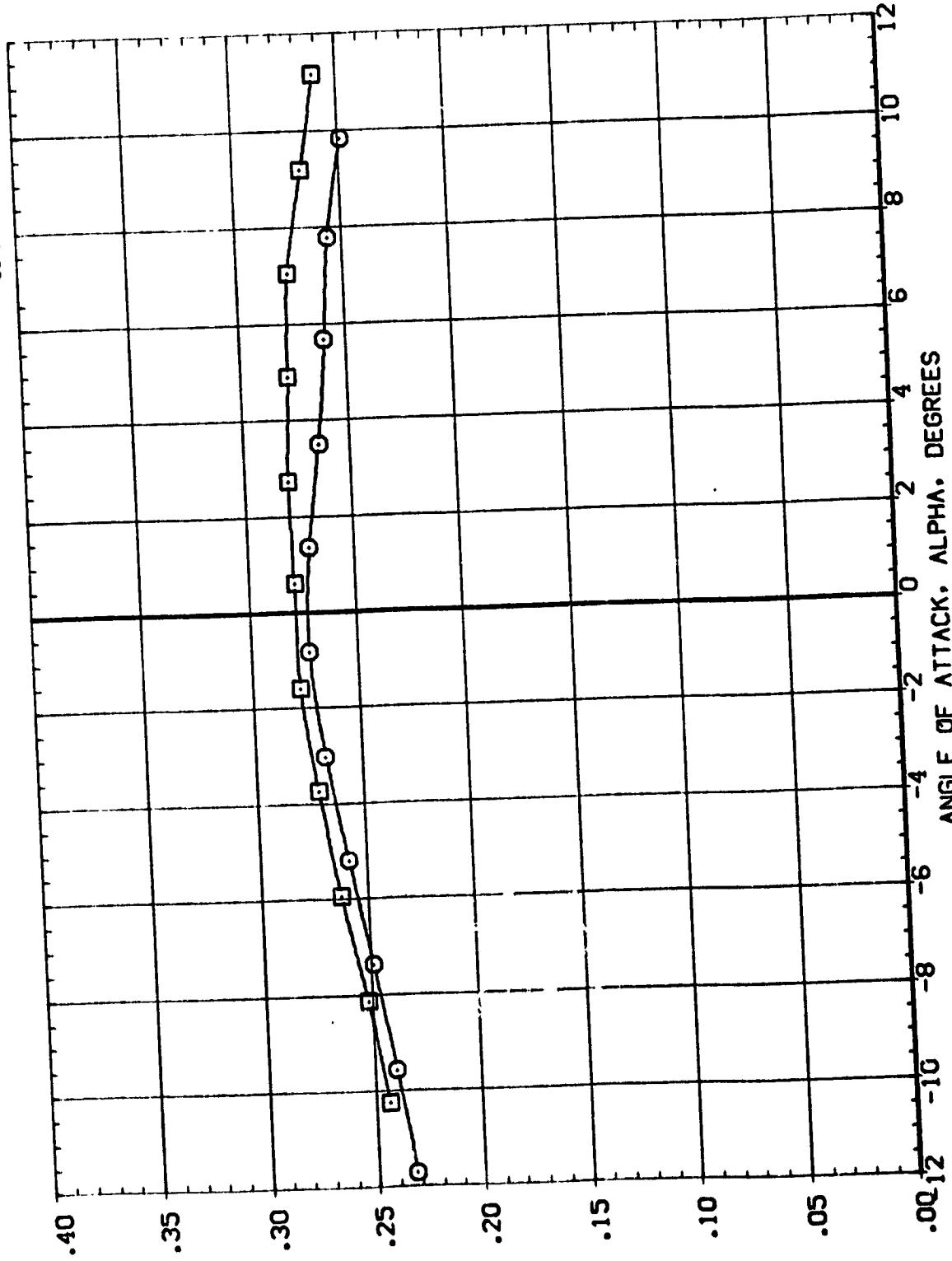
FORCEBODY AXIAL FORCE COEFFICIENT, CAF

### EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

DATA SET SYMBOL: CONFIGURATION DESCRIPTION: (03) (T9) (S3)  
 NSFC 573(A3)FC (03) (T9) (S3) ORB. MISALD.  
 (860200) (860201)  
 DATA SET SYMBOL: CONFIGURATION DESCRIPTION: (03) (T9) (S3)  
 NSFC 573(A3)FC (03) (T9) (S3) ORB. MISALD.  
 (860200) (860201)

REFERENCE INFORMATION  
 SREF 6.1980 SQ. IN.  
 LREF .3130 IN.  
 BREF 5.3130 IN.  
 XRP 2.5490 IN.  
 YRP .0000 IN.  
 ZRP .0040 IN.  
 SCALE



FORCEBODY AXIAL FORCE COEFFICIENT. CAF

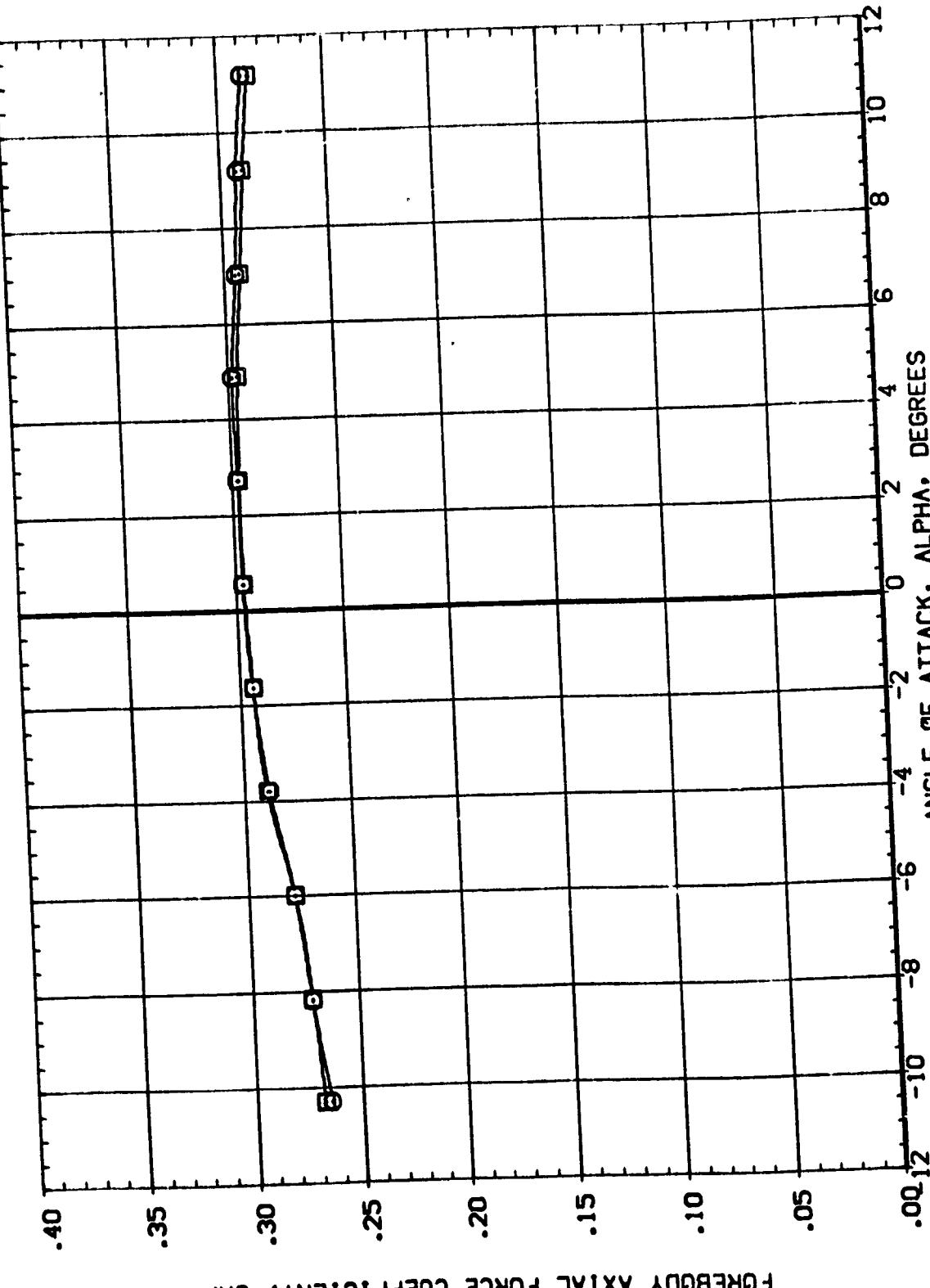
EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS  
 (C)MACH = 1.25

DATA SET NAME: CONFIGURATION DESCRIPTION  
 (B60200) 8 MSC 573(1A3)FC (03)(19)(S3) ORB. MISALD.  
 (B60201) 8 MSC 573(1A3)FC (03)(19)(S3) ORB. MISALD.

ORB INC	DELTAZ	ORBYAW
.500	.140	1.000
.500	.140	

REFERENCE INFORMATION

REF	5.1980 SQ. IN.
LREF	5.3130 IN.
BREF	5.3130 IN.
XRP	2.5450 IN.
YRP	.0000 IN.
ZRP	.0000 IN.
SCALE	.0040



FORCEBODY AXIAL FORCE COEFFICIENT, CAF

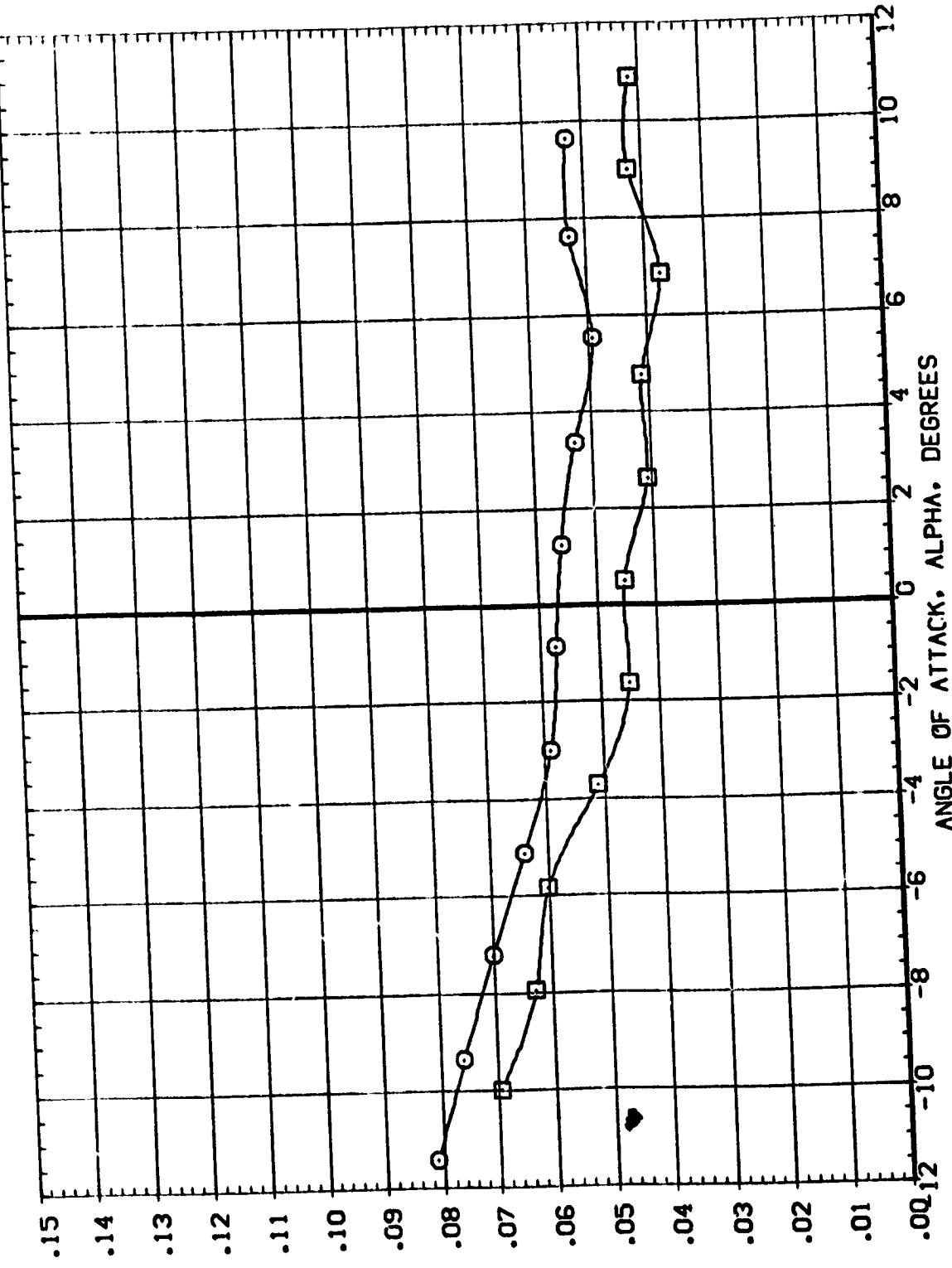
EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

$$CD MACH = 1.46$$

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) MSFC 573(1A3)FC (G3)(19)(S3)  
 (B92201) MSFC 573(1A3)FC (G3)(19)(S3) ORB. MISALND.

ORBINC	DELTAY	GRBYAV
.500	.140	1.000
.500	.140	

REFERENCE INFORMATION  
 SREF 6.1980 SC. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE



EXTERNAL TANK AXIAL FORCE COEFFICIENT, CABE

### EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

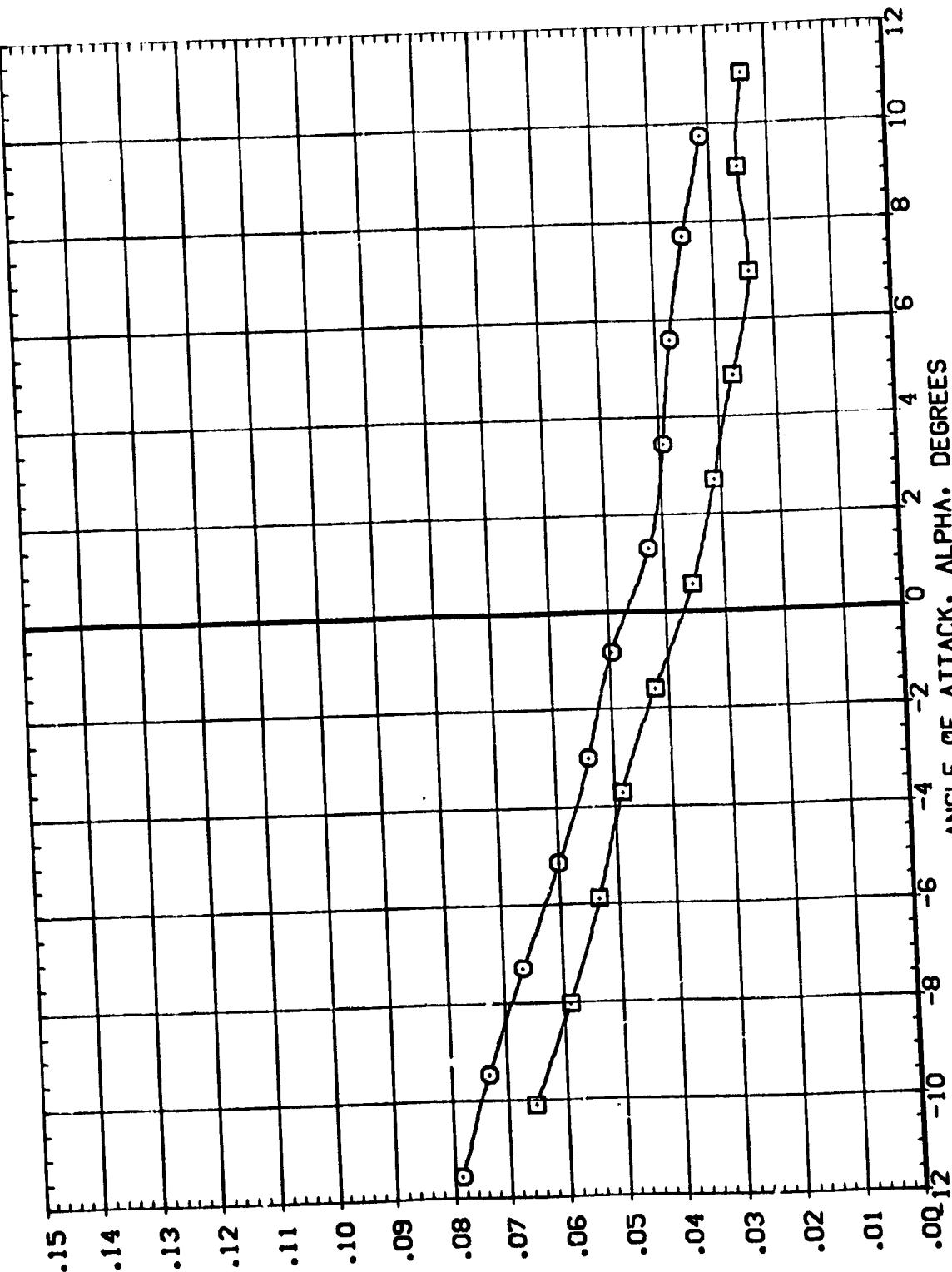
(V)MACH = .90

DATA SET SYMBOL: 8      CONFIGURATION DESCRIPTION: MSFC 573(1A3)FC (03)(19)(S3) ORB. MISALNO.  
 (B80000)      (B80201)      (B80201)

ORB INC	DELTAZ	ORBYAW
.500	.140	1.000
.500	.140	

REFERENCE INFORMATION IN  
 SREF 6.1980 SD. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XREF 2.5490 IN.  
 YREF .0000 IN.  
 ZREF .0040 IN.

SCALE



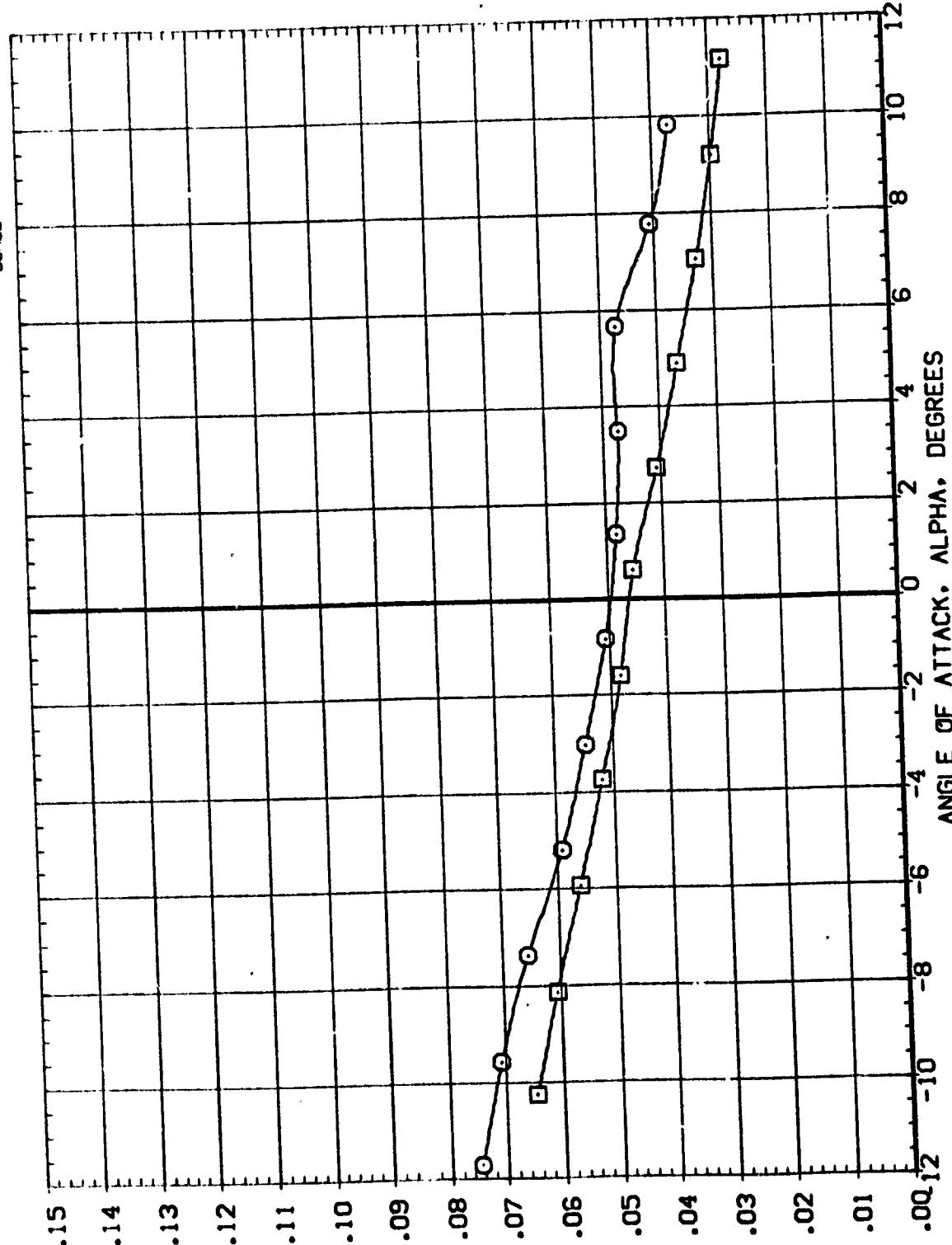
EXTERNAL TANK AXIAL FORCE COEFFICIENT, CABE

EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS  
 (B)MACH = 1.05

REFERENCE INFORMATION

SREF	6.1980	SO.	IN
LREF	5.3130	IN.	
BREF	5.3130	IN.	
XMRP	2.5490	IN.	
YMRP	.0000	IN.	
ZMRP	.0000	IN.	
SCALE	.0040		

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (691900) □ MSFC 5731A31FC (03)(19)(S3) ORB. MISALD.  
 (690201) □ MSFC 5731A31FC (03)(19)(S3) ORB. MISALD.



EXTERNAL TANK AXIAL FORCE COEFFICIENT, CABE

### EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(C)<sub>MACH</sub> = 1.25

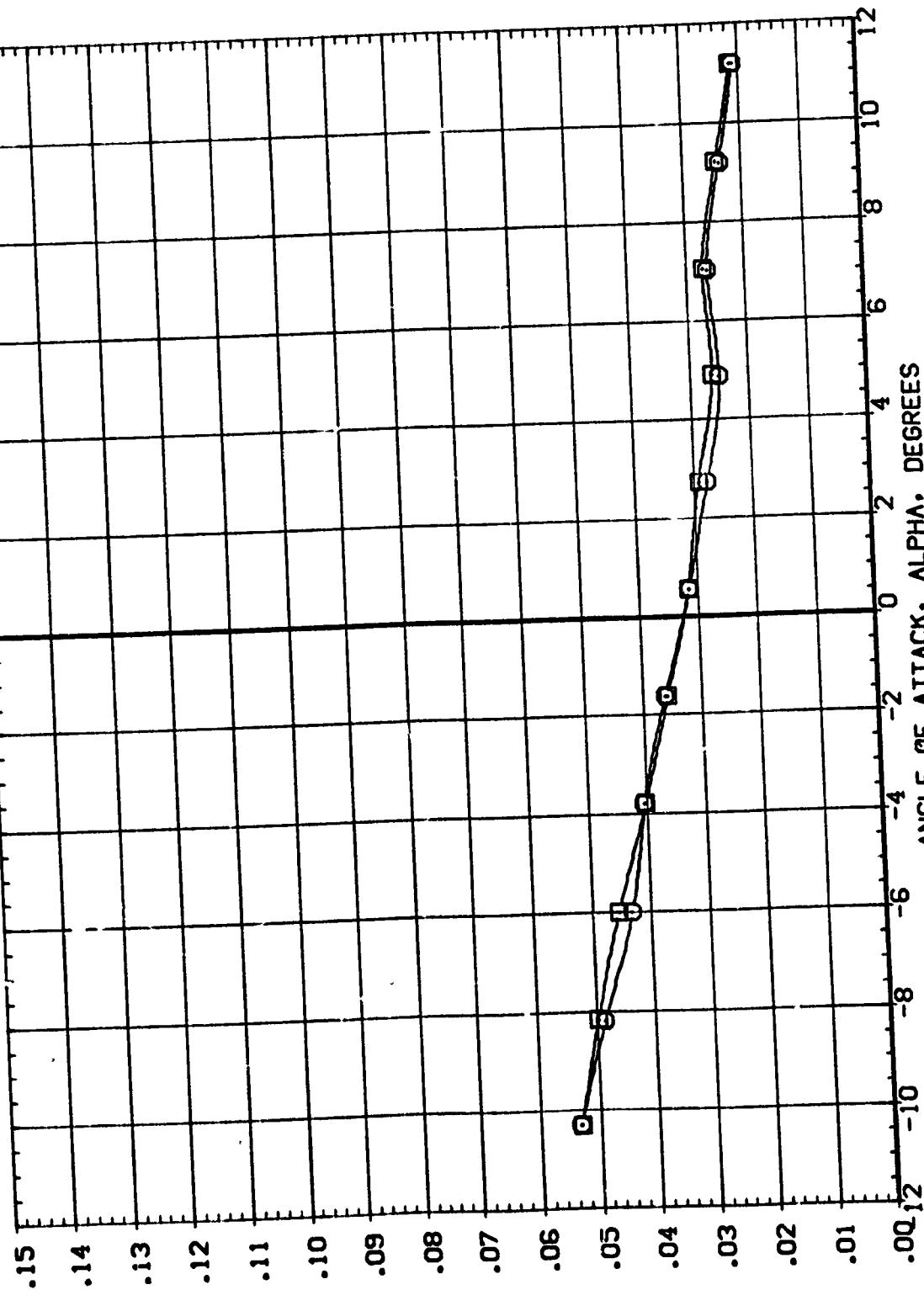
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DATA SET SYMBOL CONFIGURATION DESCRIPTION  
[B90000] MSEC ST21(ASIFC) [S3] ORB. MISALN.  
[B90201] MSEC S73(ASIFC) [S3] ORB. MISALN.

ORB INC. 0.500 0.500  
DELTAZ .140 .140  
ORBYAW 1.000 1.000

REFERENCE INFORMATION  
SREF 6.1980 SQ. IN.  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
XMP 2.5490 IN.  
YMP .0000 IN.  
ZMP .0040 IN.

EXTERNAL TANK AXIAL FORCE COEFFICIENT, CABE



EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(D)MACH = 1.46

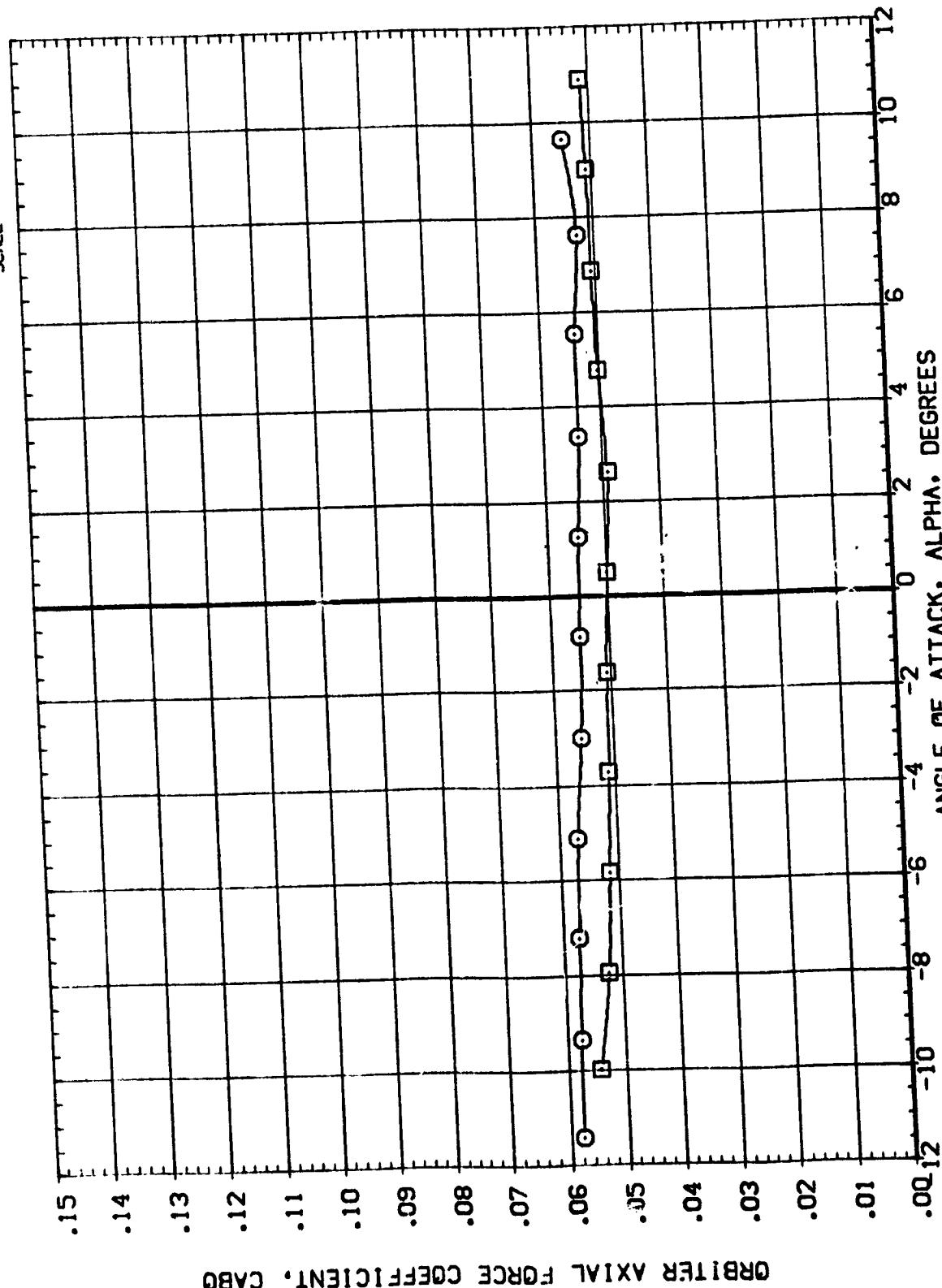
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DATA SET SYMBOL: **□** CONFIGURATION DESCRIPTION: **MSFC 573(I)A3(FC) (G1)(T9)(S3)** ORB. MISALND.  
 (BS900C0) (BS90201) **MSFC 573(I)A3(FC) (G2)(T9)(S3)** ORB. MISALND.  
 (BS90201)

ORB INC.	DELTAZ	CRBYAV
.500	.140	1.000
.500	.140	

REFERENCE INFORMATION IN SQ. IN.

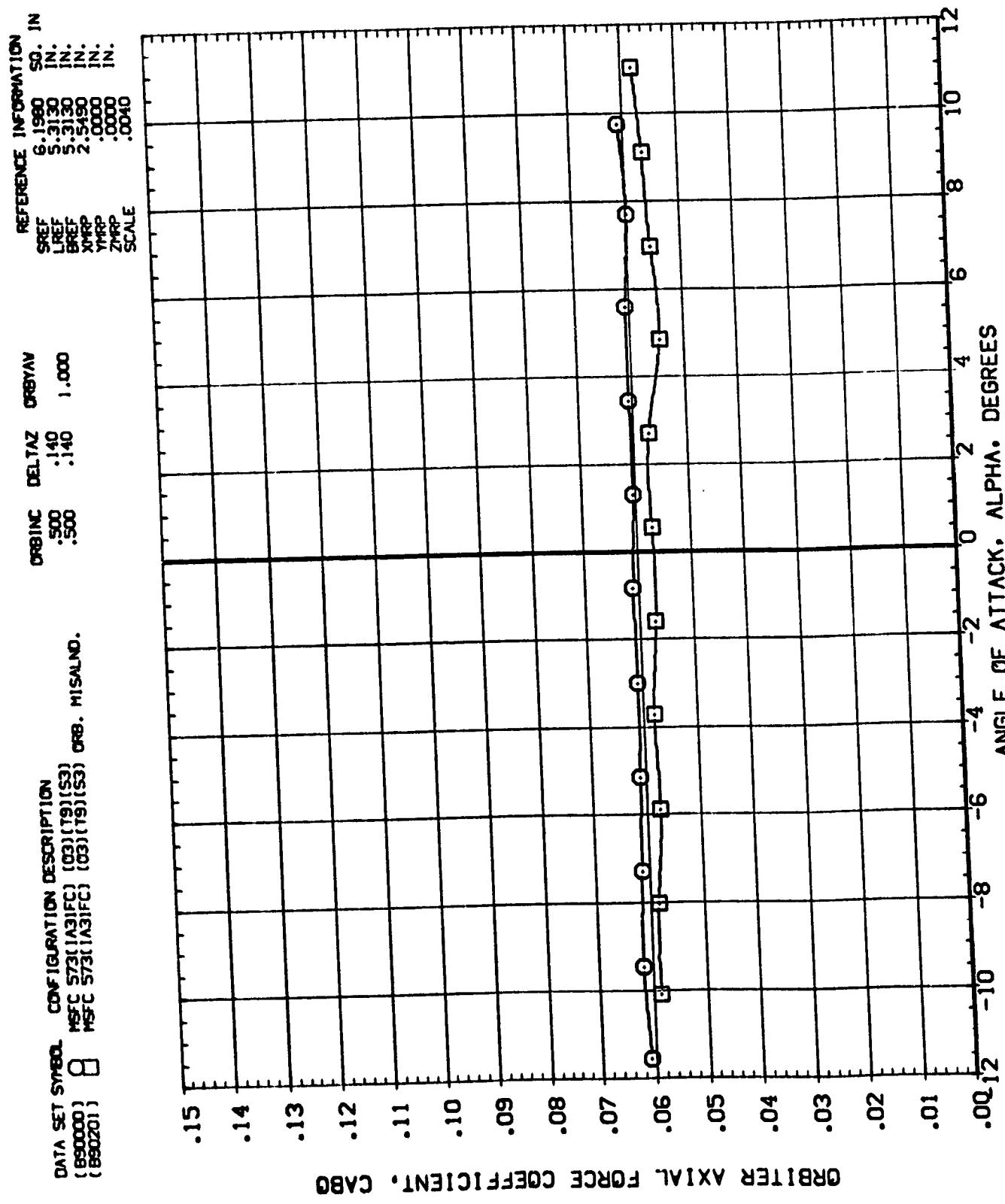
SREF	6.1980
LREF	5.3130
BREF	5.3130
XMRP	2.5490
YMRP	.0000
ZMRP	.0040
SCALE	



### EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(AJMACH = .90

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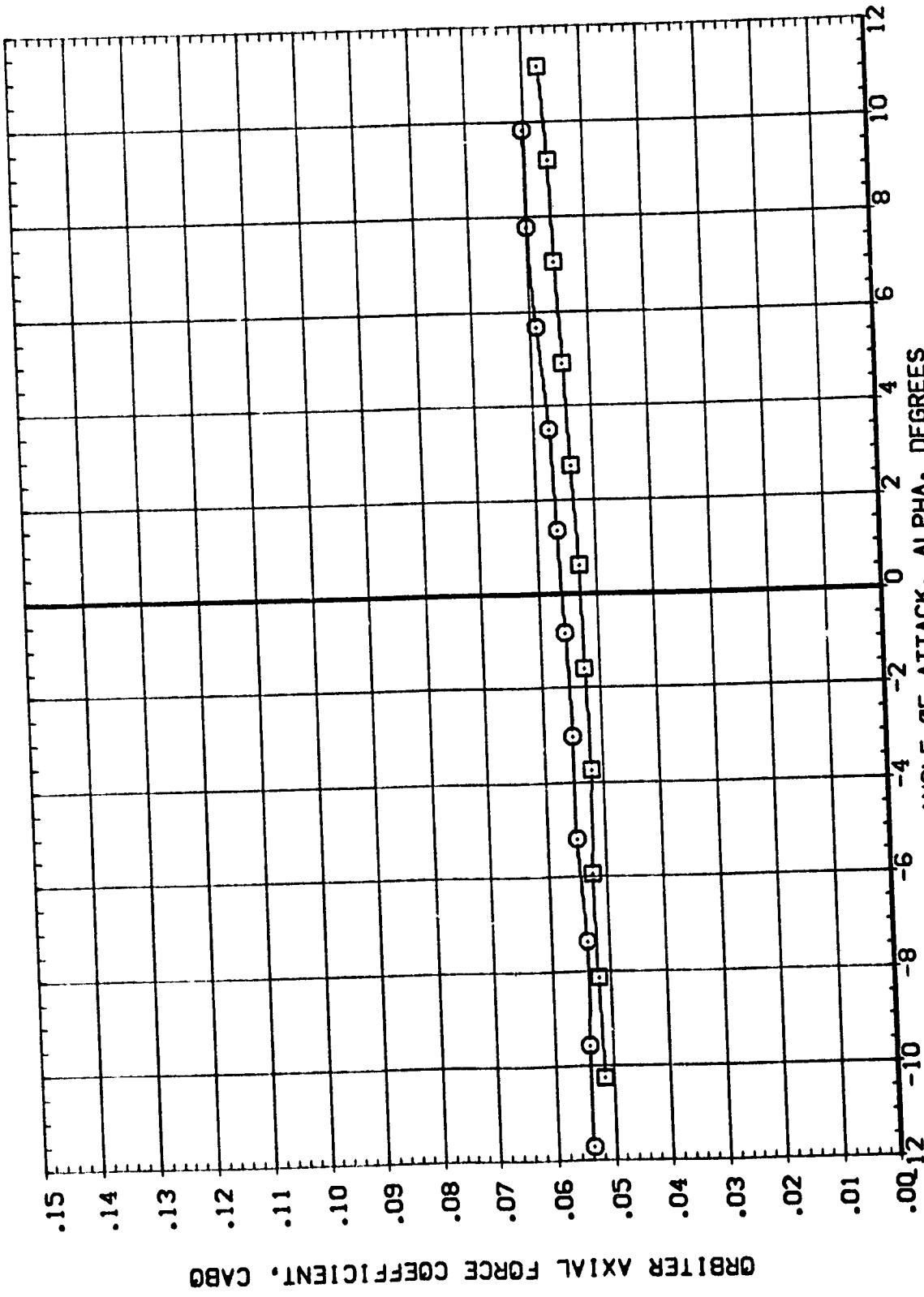


EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 G8530000 NSFC 573([A3]FC) [G3][T9][S3] ORB. MISALND.  
 (853201) 8 NSFC 573([A3]FC) [G3][T9][S3]

REFERENCE INFORMATION  
 SREF 6.1980 SQ. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.



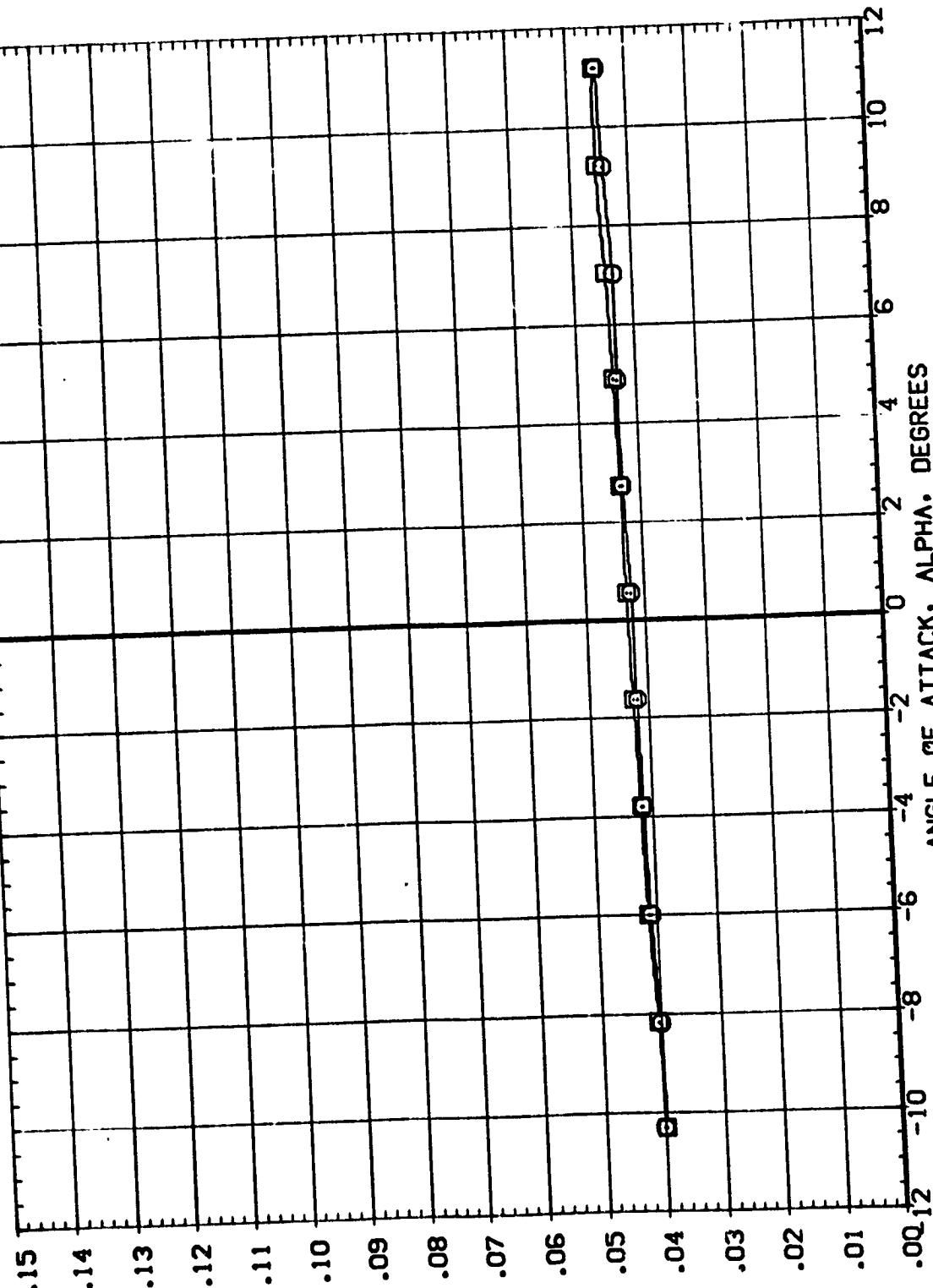
ORBITER AXIAL FORCE COEFFICIENT, CABO

EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS  
 (C)MACH = 1.25

DATA SET SYMBOl. CONFIGURATION DESCRIPTION  
 (B90000) 8 MSFC 573(LIA<sup>2</sup>-TC) (03)(19)(S3) ORB. MISALD.  
 (B90201)

REFERENCE INFORMATION  
 SPREF 6.1980 SQ. IN.  
 LREF 5.3130 IN.  
 RREF 5.3130 IN.  
 BREF 2.5490 IN.  
 XTRP .0000 IN.  
 YTRP .0000 IN.  
 ZTRP .0040 IN.  
 SCALE

ORBITER AXIAL FORCE COEFFICIENT. CABO



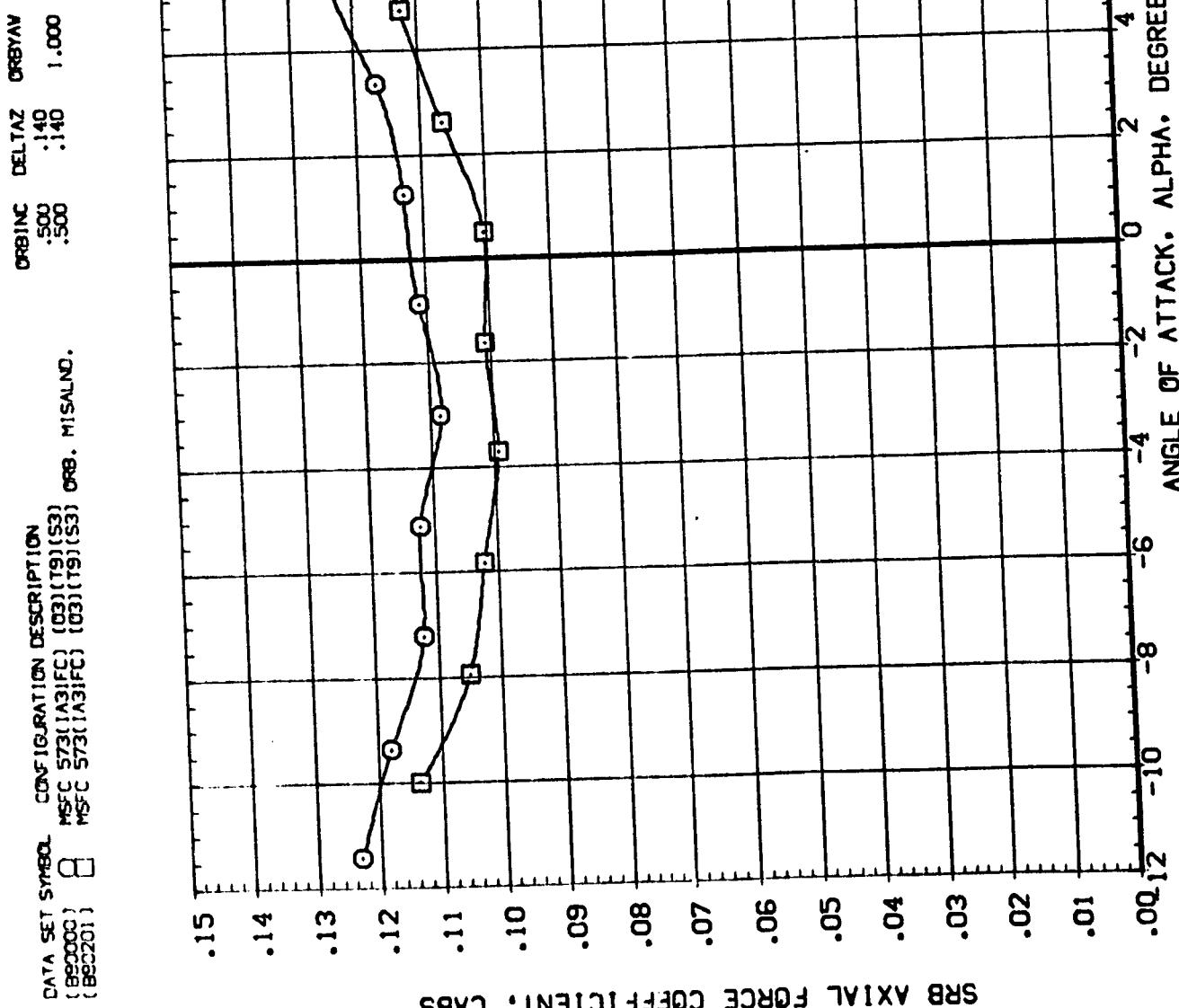
### EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(CD)MACH = 1.46

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DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (BEC0000) MSC 5731(13)FC (03)(T9)(S3) CABS. MISALND.  
 (BEC201) MSC 5731(13)FC (03)(T9)(S3)

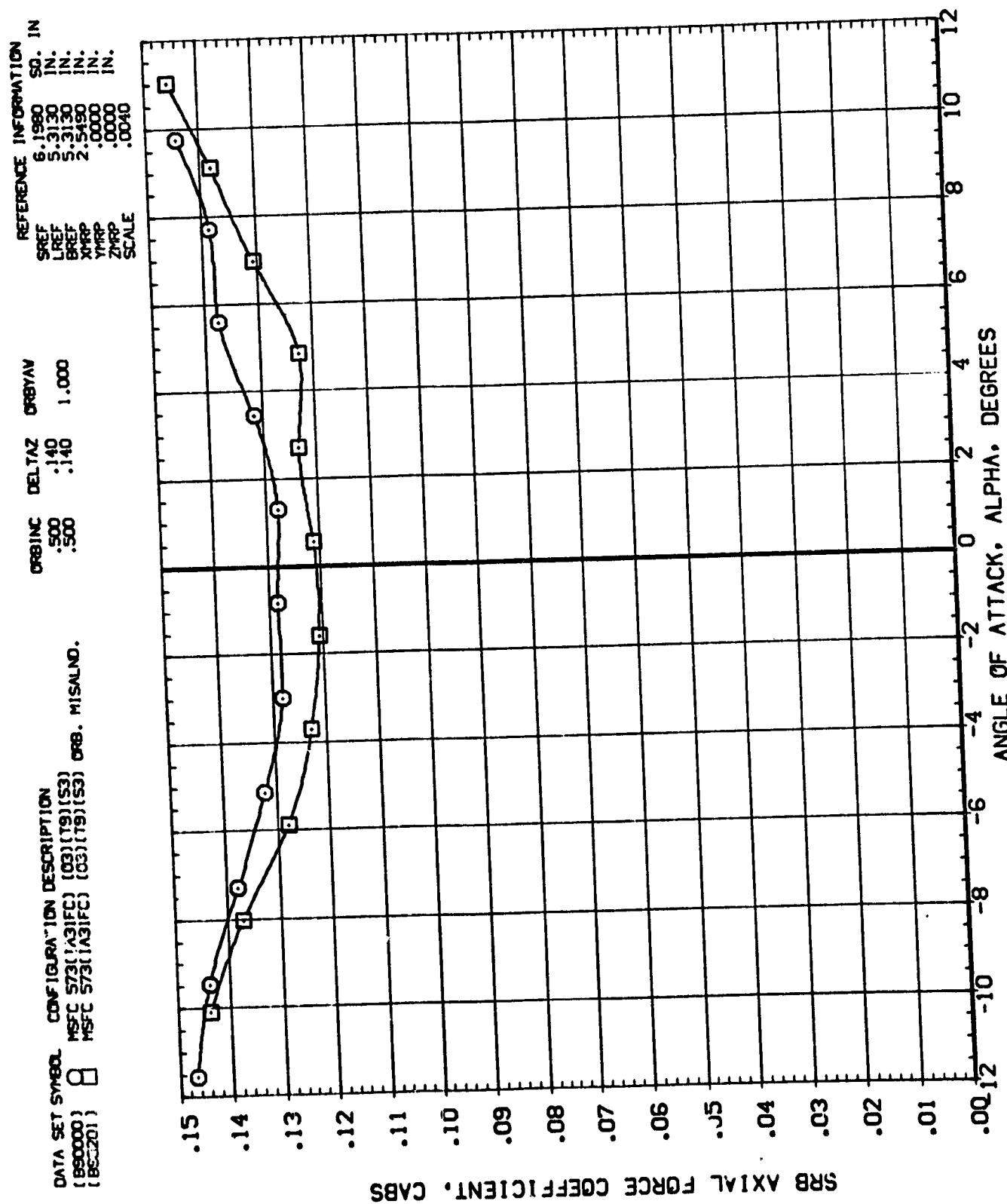
REFERENCE INFORMATION  
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 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5450 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.



### EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

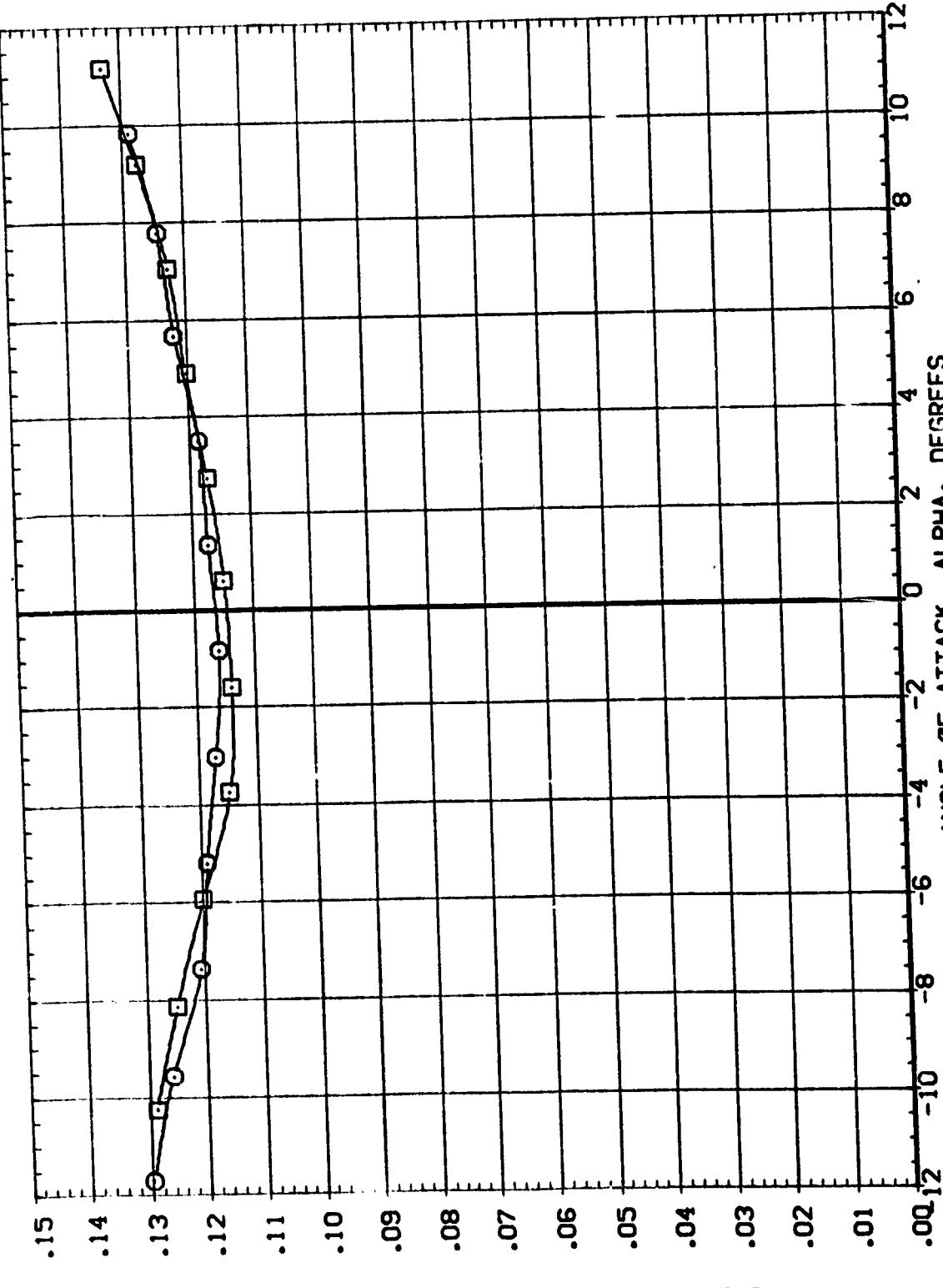
(AJMACH = .90

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DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (S) MSFC 573(1A31FC) (03)(19)(S3)  
 (S) MSFC 573(1A31FC) (03)(19)(S3) ORB. MSLN. NO.  
 (S) 5500000  
 (S) 650201

REFERENCE INFORMATION  
 SRREF 6.1980 SQ. IN.  
 LRREF 5.3130 IN.  
 BRREF 5.3130 IN.  
 XHREF 2.5490 IN.  
 YHREF .0000 IN.  
 ZHREF .0040 IN.  
 SCALE

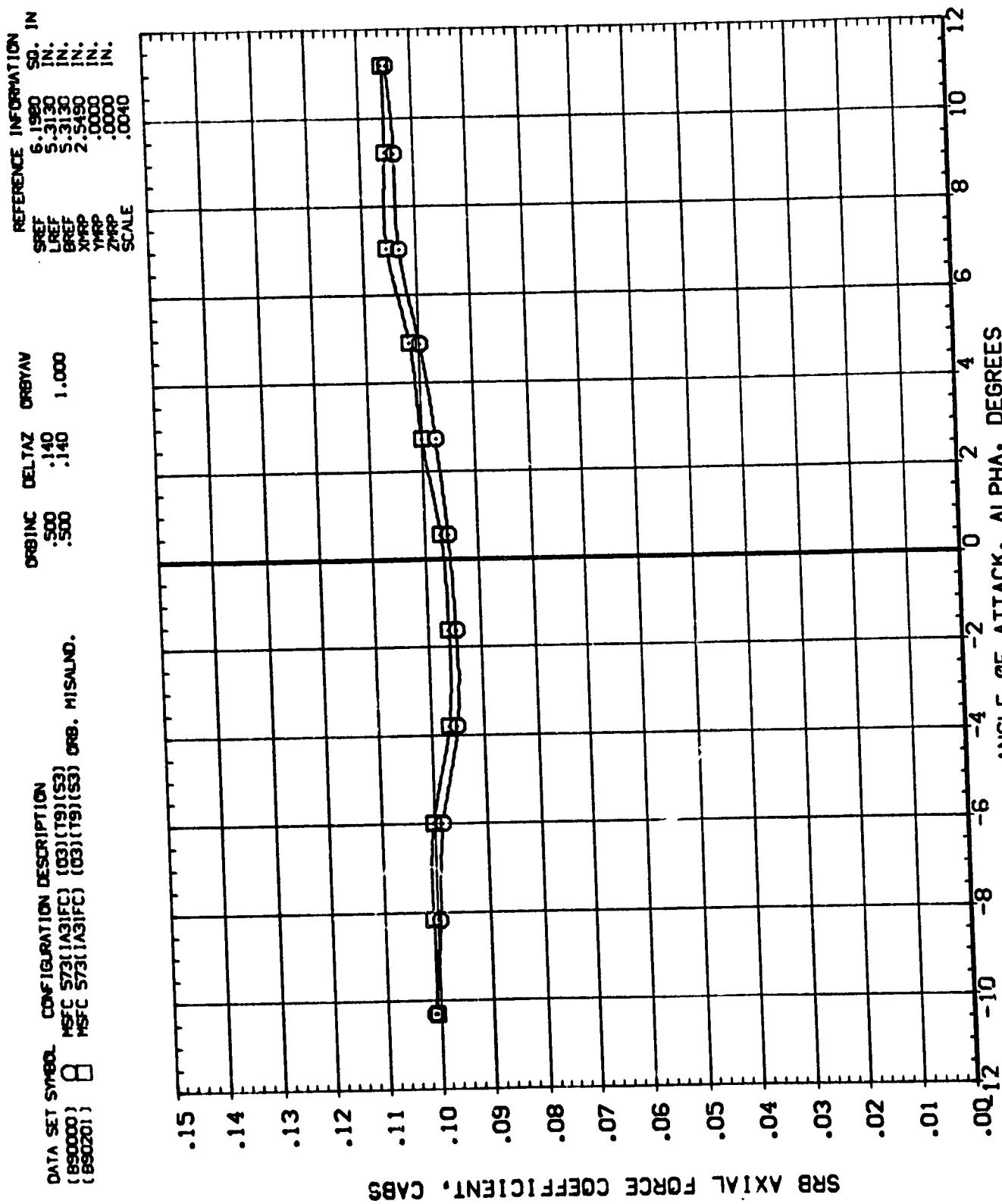


SRB AXIAL FORCE COEFFICIENT, CABs

### EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

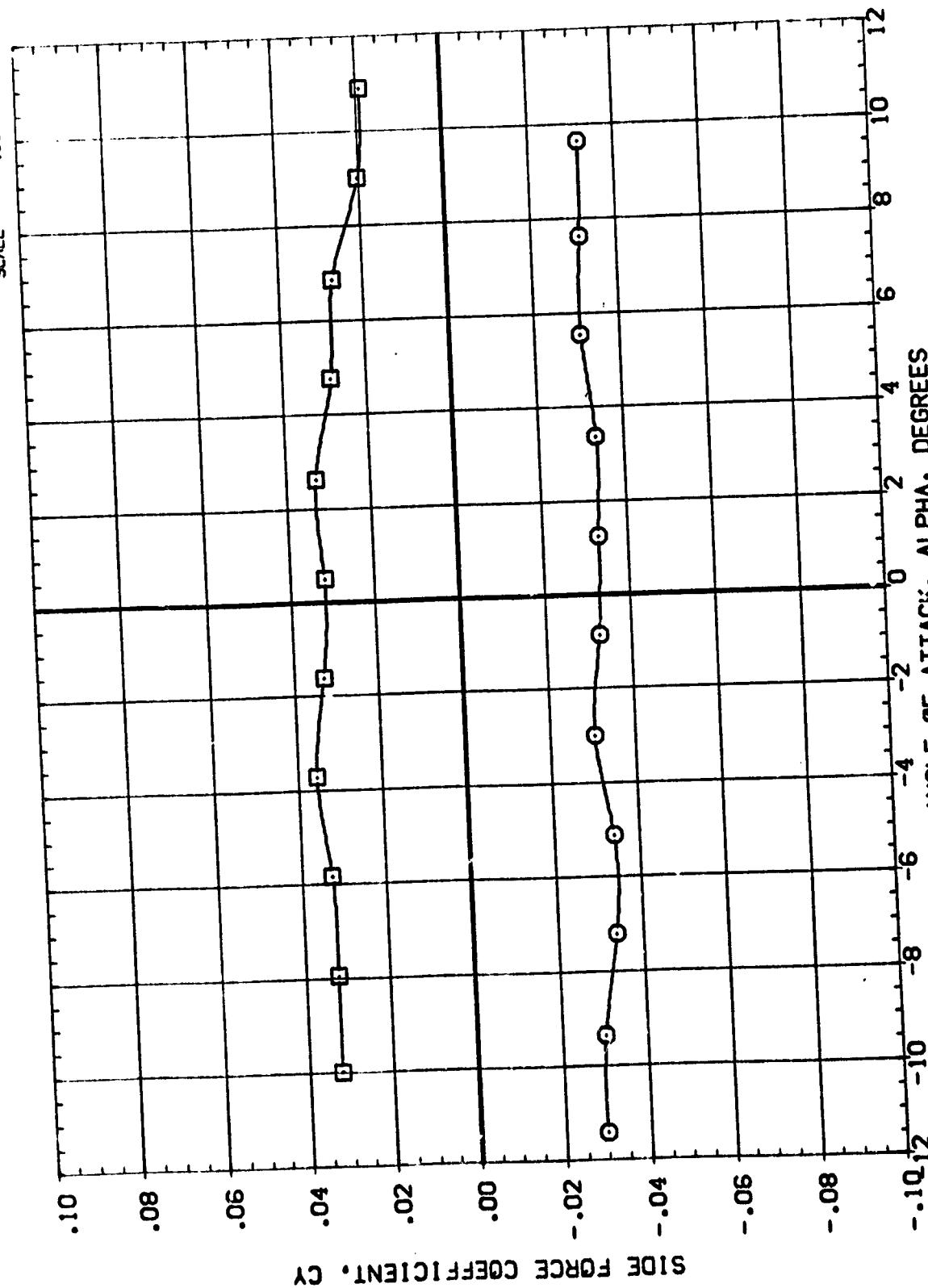
EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS  
 $(\text{CDMACH} = 1.46)$



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 NSFC S73([A3]FC) (S3)(T9)(S3) ORB. MIS AND.  
 (S90000) (S90101)

ORB INC DELTAZ ORBYAW  
 .500 .140 1.000  
 .500 .140 .000  
 IN. IN. IN.

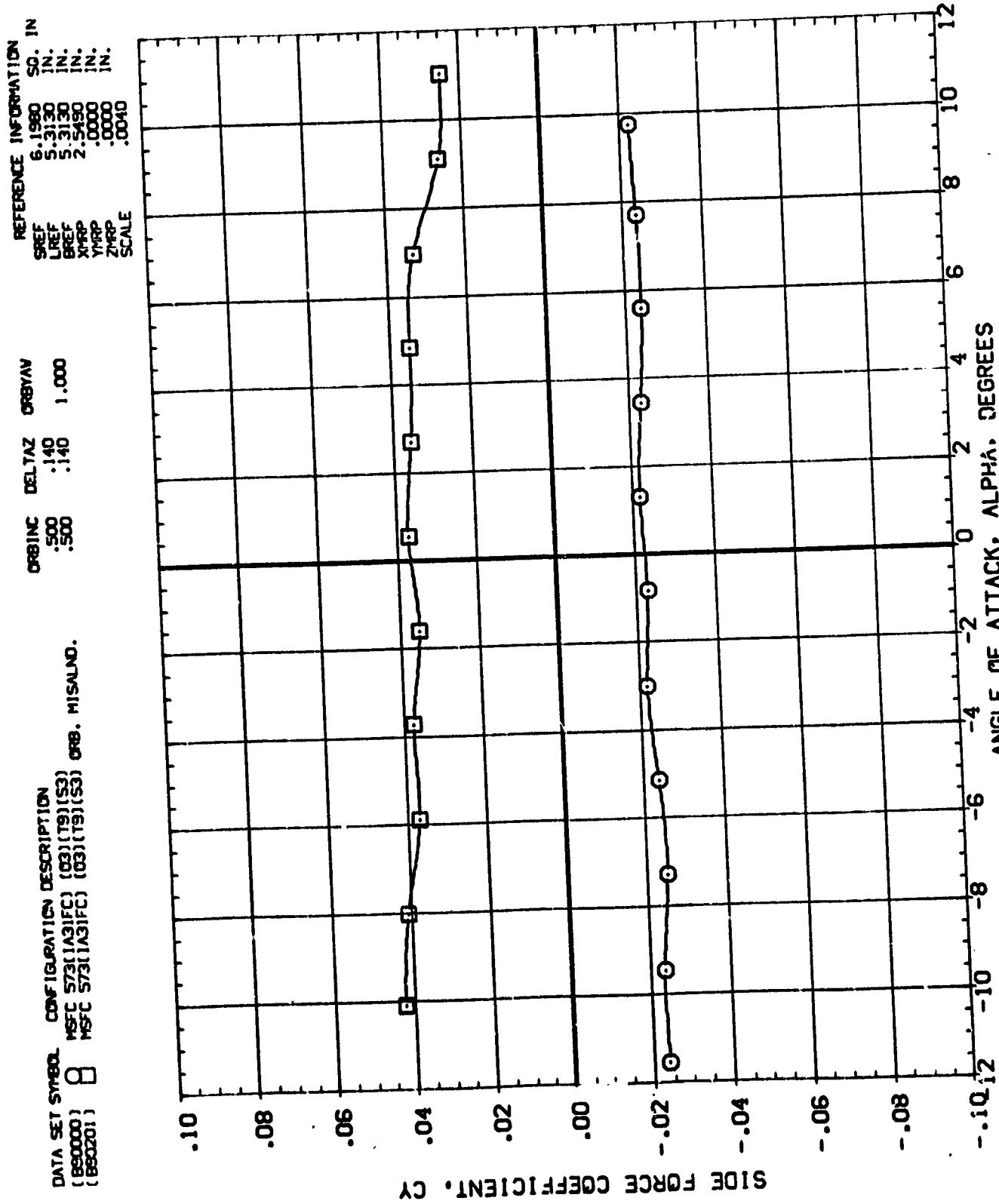
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 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE



EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(A)MACH = .90

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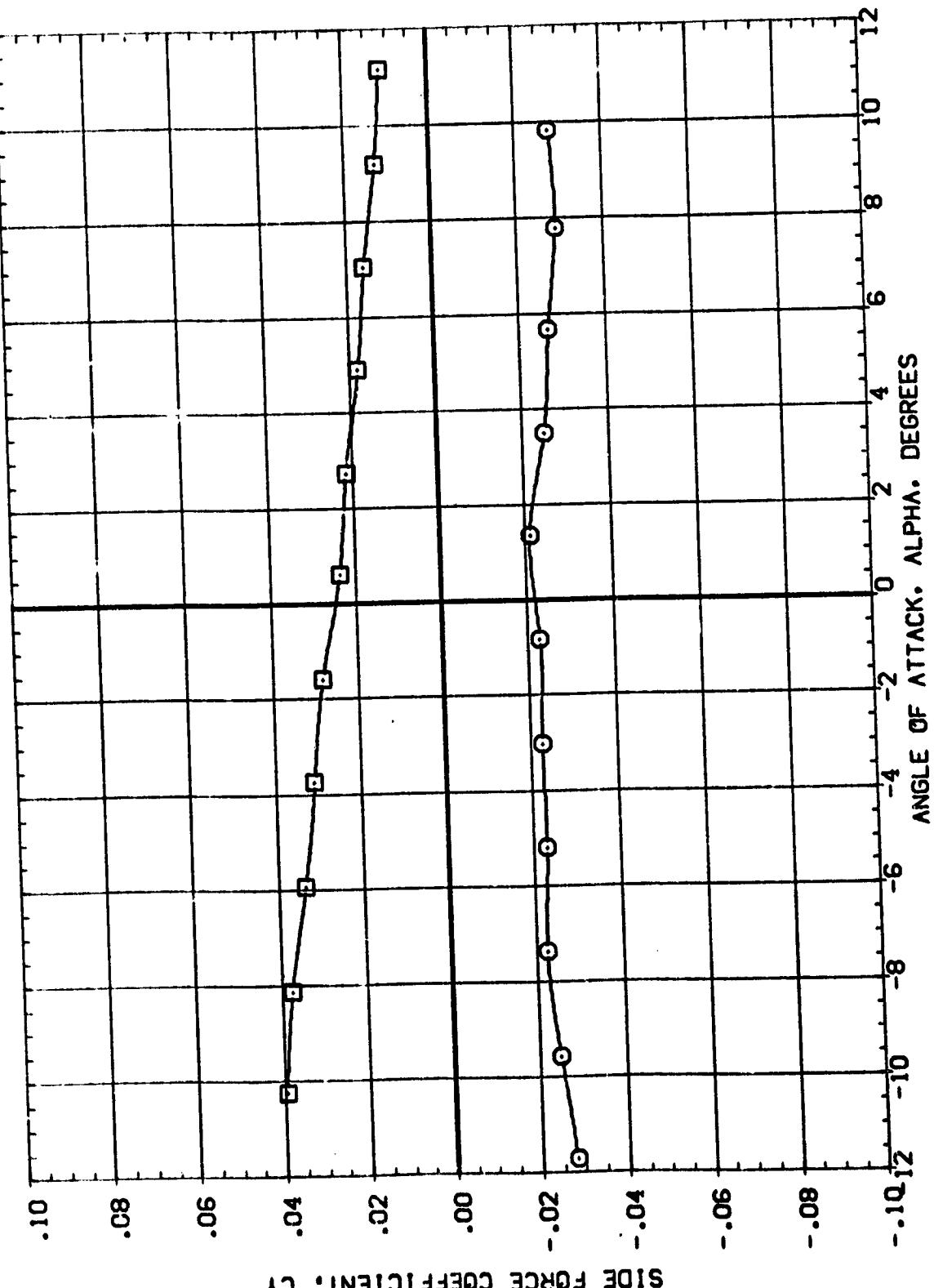
EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

PAGE 91

DATA SET SYMBOL      CONFIGURATION DESCRIPTION  
      (B90000)      MSFC 5731(A3)FC (03)(19)(S3)  
      (B90201)      MSFC 5731(A3)FC (03)(19)(S3) ORB. MISALNO.

REFERENCE INFORMATION  
SREF 6.1980 SD. IN.  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
XREFP 2.5430 IN.  
YREFP .0000 IN.  
ZREFP .0000 IN.  
SCALE .0040



EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

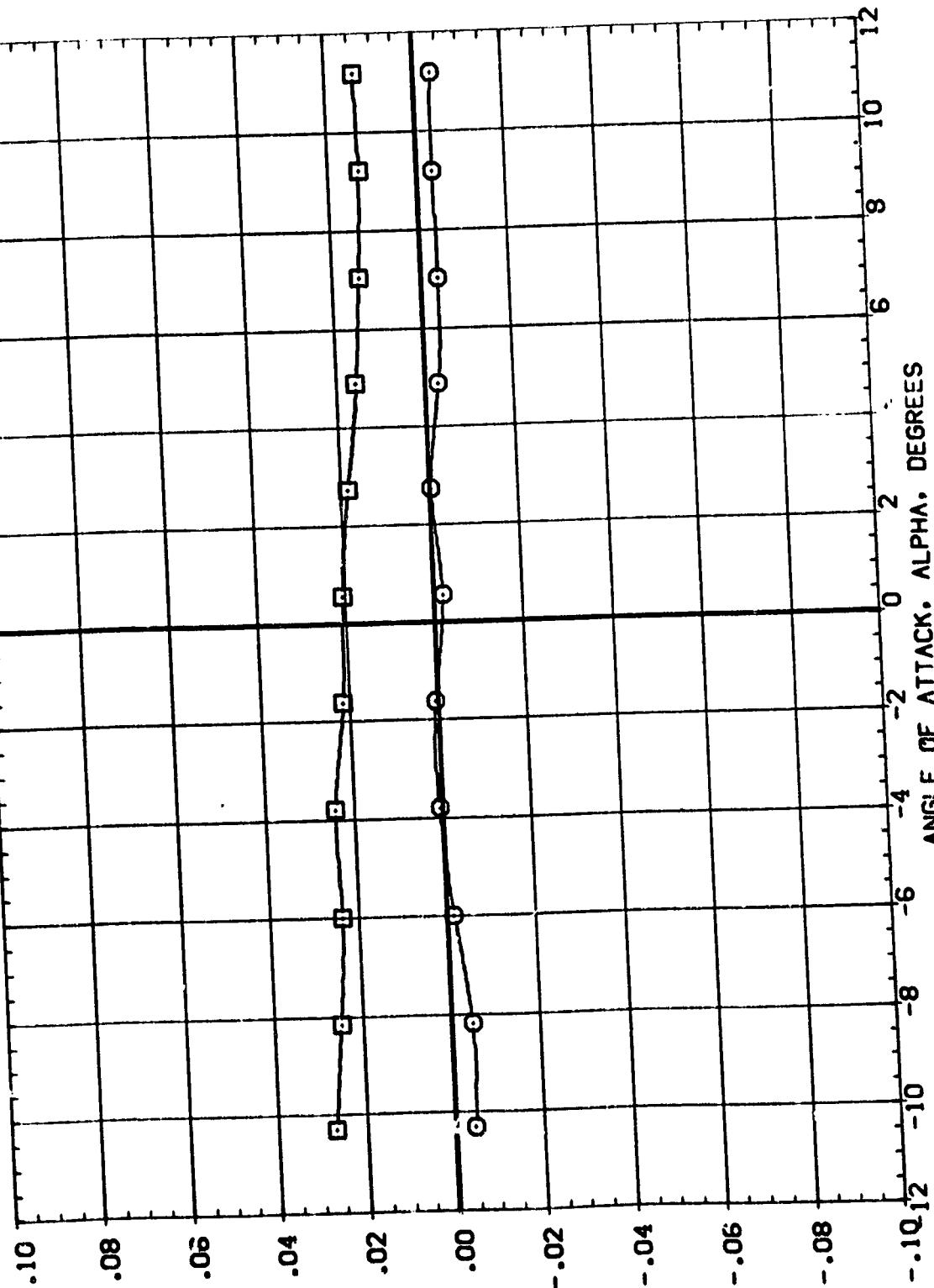
PAGE 92

DATA SET SYMBOL. CONFIGURATION DESCRIPTION  
 (B90000) 8 MSFC 573(1A3)FC (03)(19)(53)  
 (B90201) 8 MSFC 573(1A3)FC (03)(19)(53) ORB. MISALN.

REFERENCE INFORMATION IN  
 SREF 6.1980 SD. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP 0.0000 IN.  
 ZMRP .0000 IN.  
 SCALE .0040

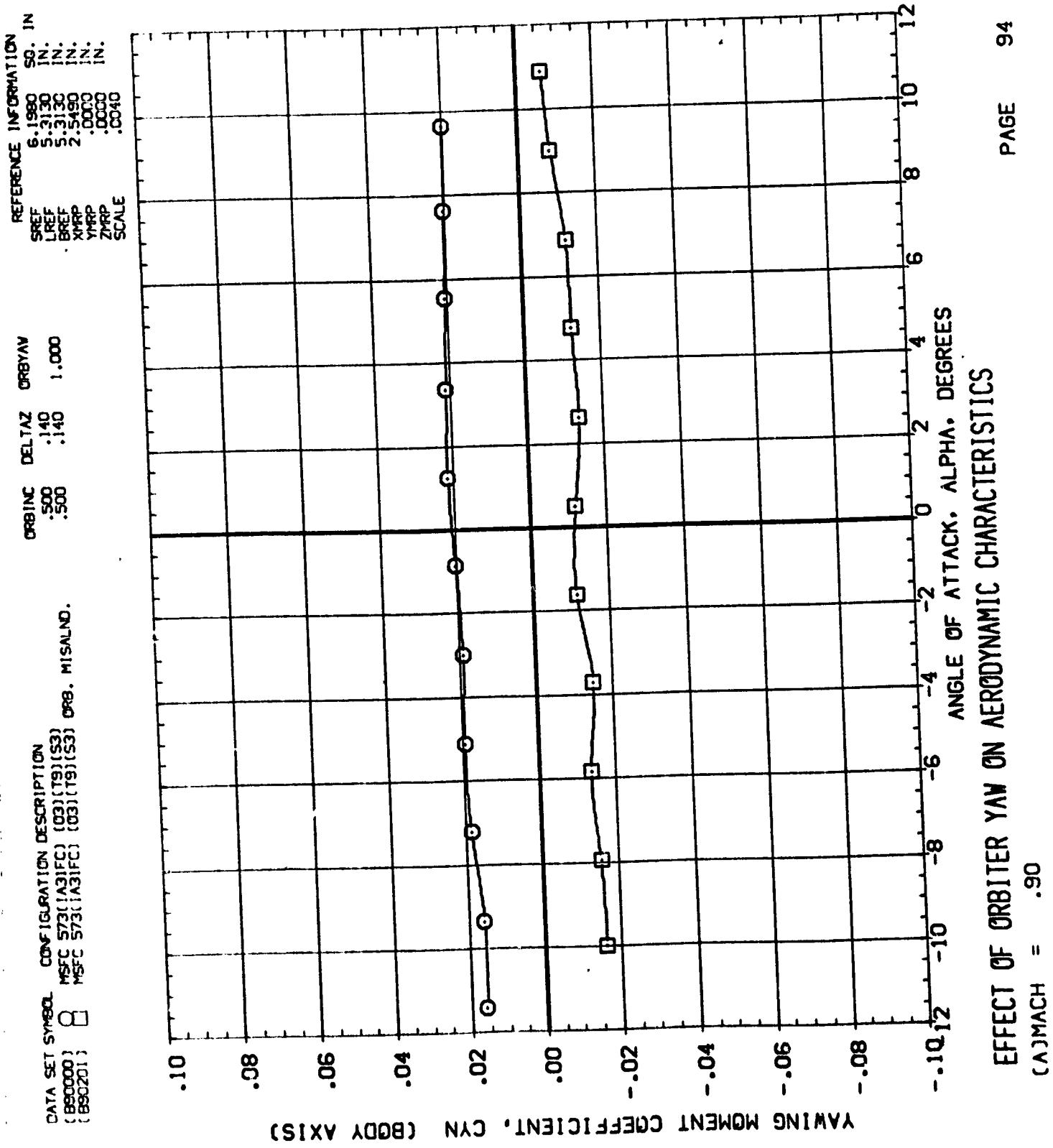
ORB INC DELTAZ GRYAV  
 :500 :140 1.000

SIDE FORCE COEFFICIENT. CY

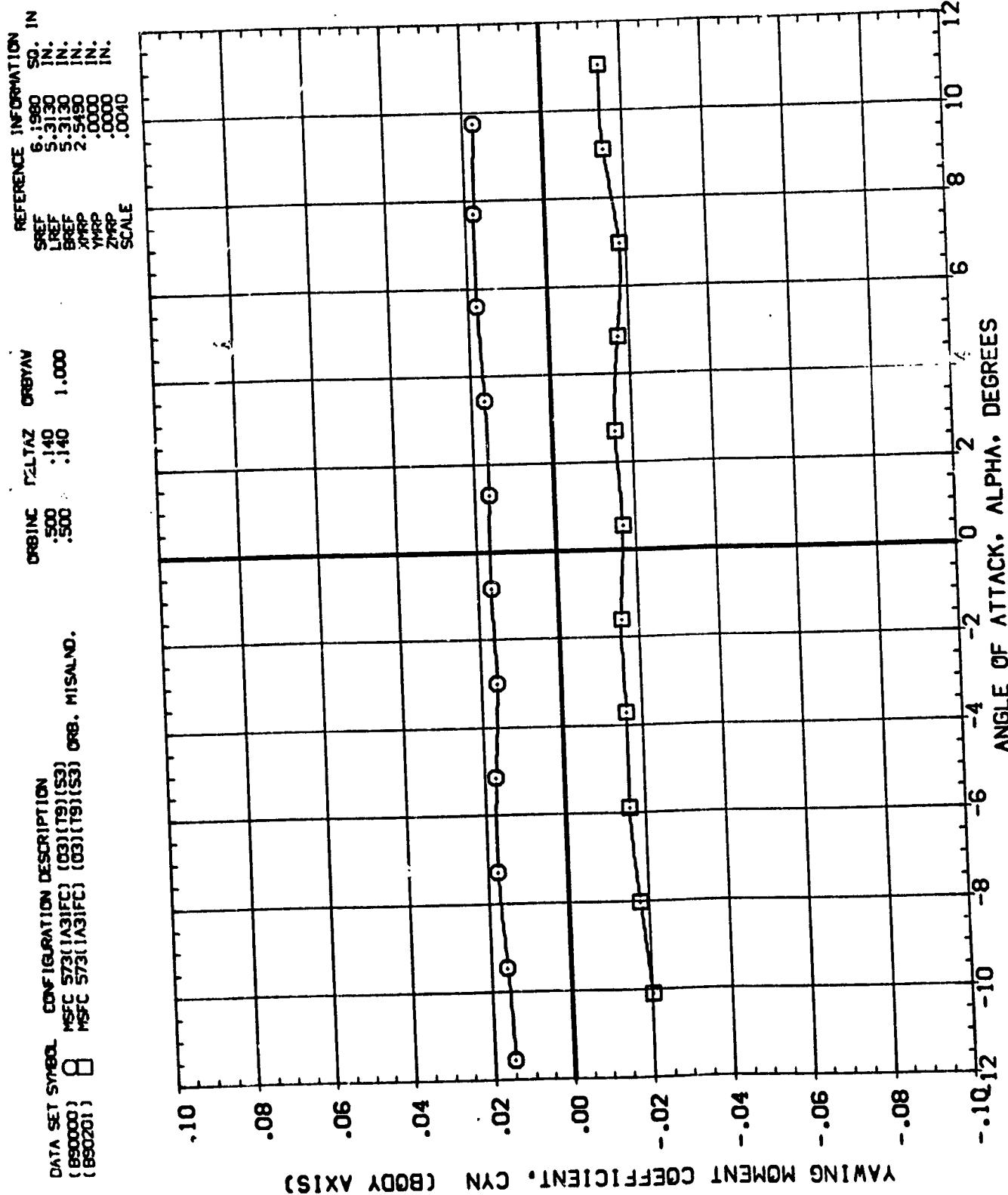


EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.46



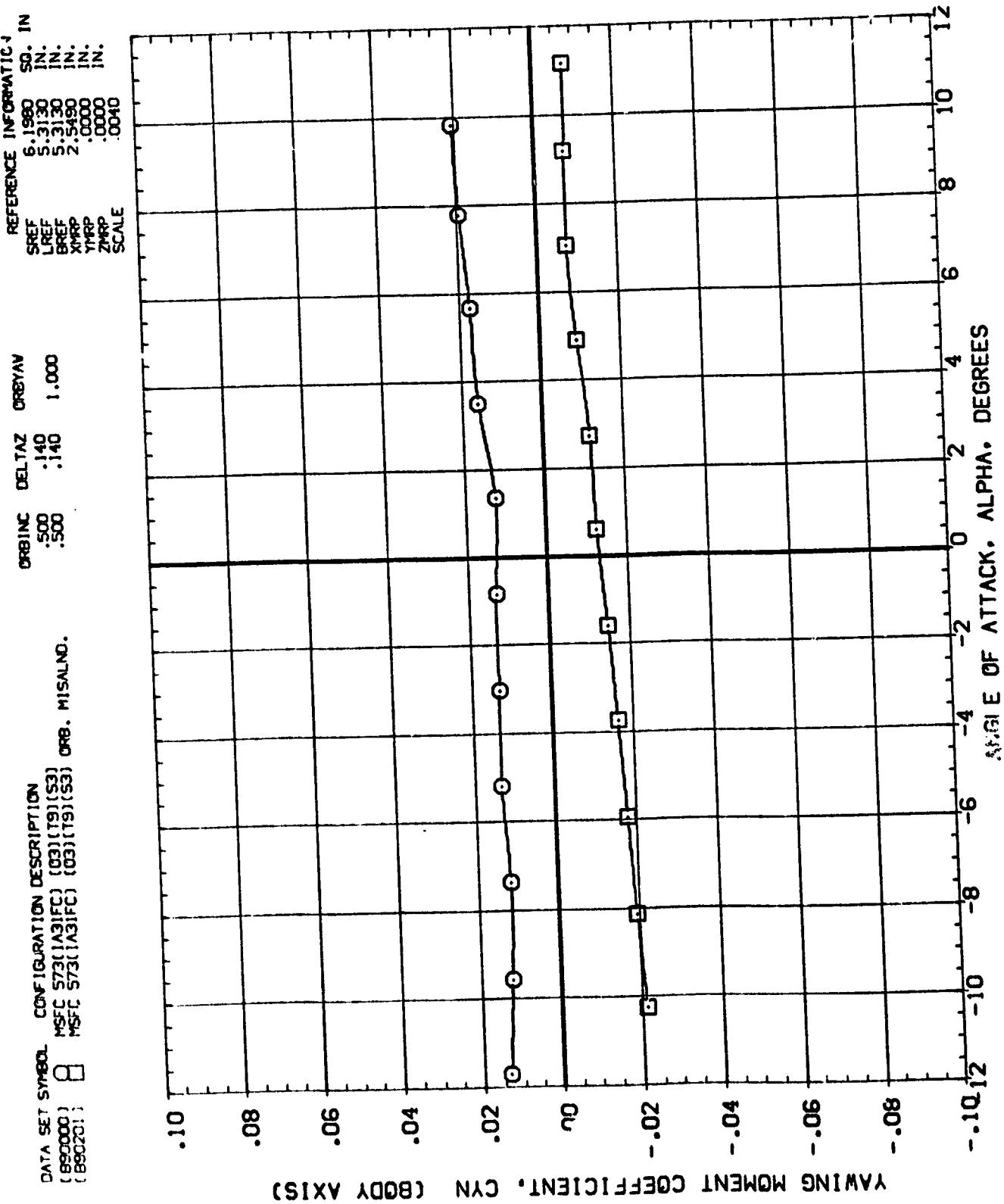
PAGE 94



EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

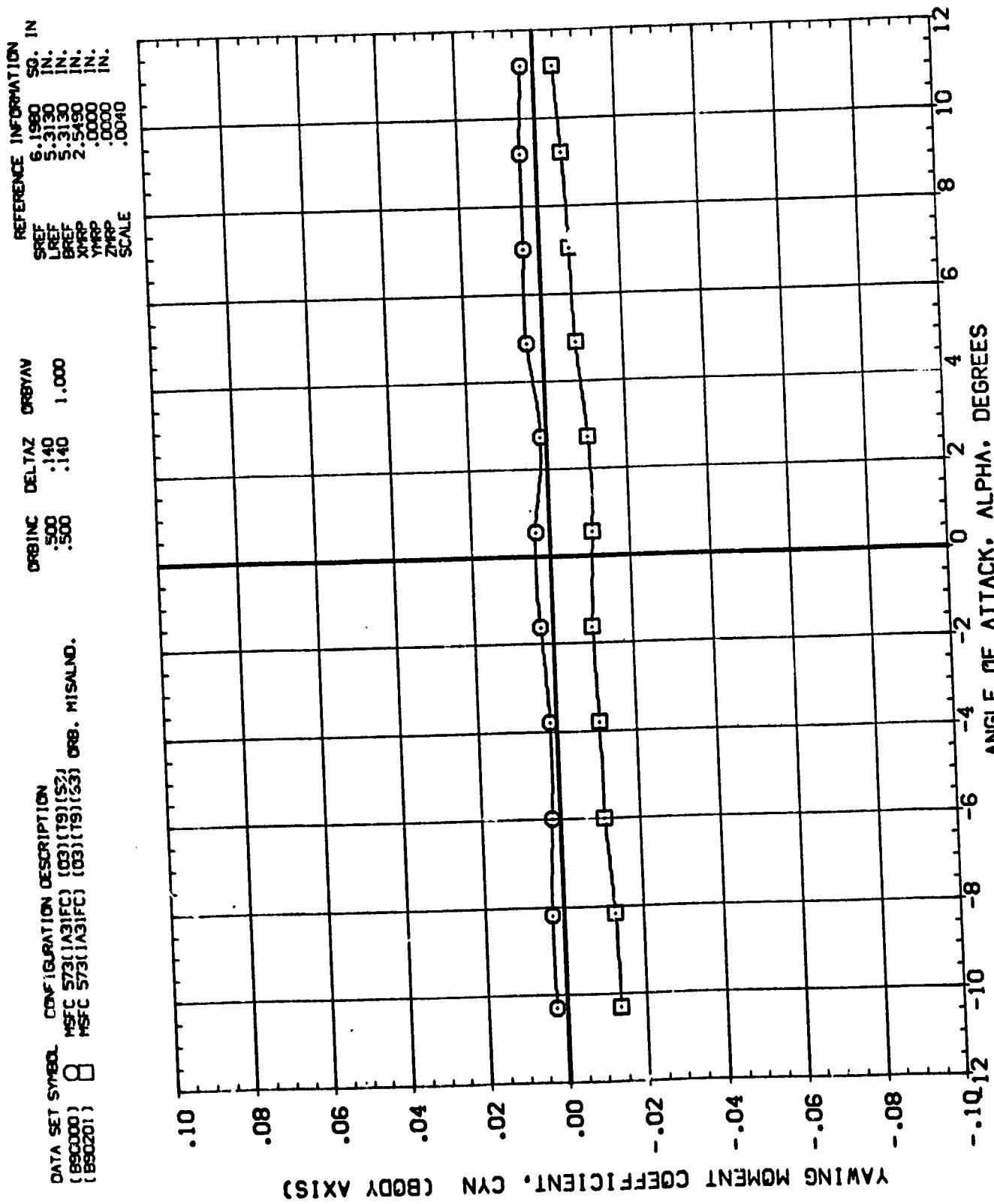
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 NSFC 573((A3)FC) (03)(T9)(S3)  
 NSFC 573((A3)FC) (03)(T9)(S3) ORB. MISALD.  
 (8890000) (8890201)



### EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

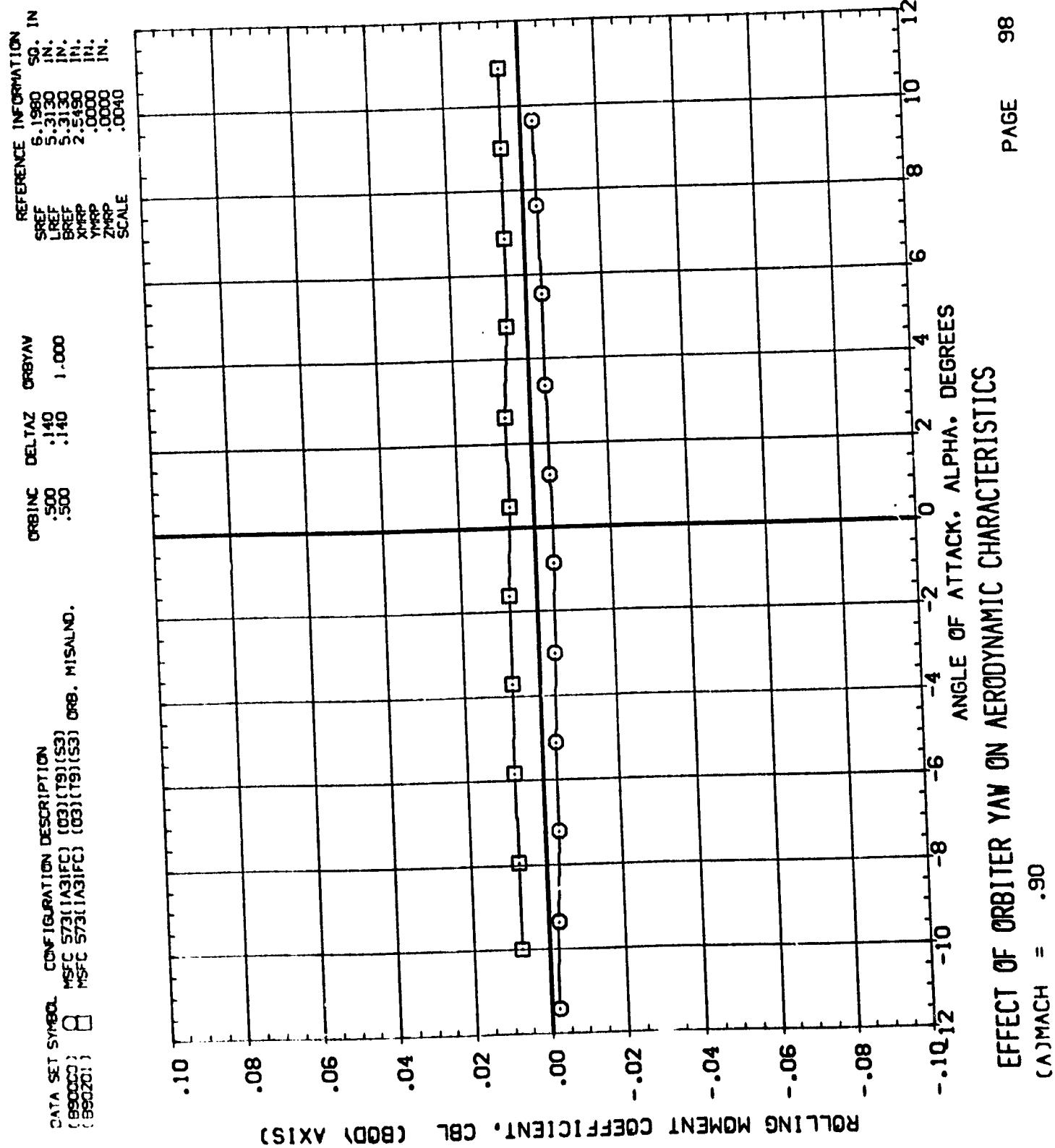
(C)MACH = 1.25

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EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

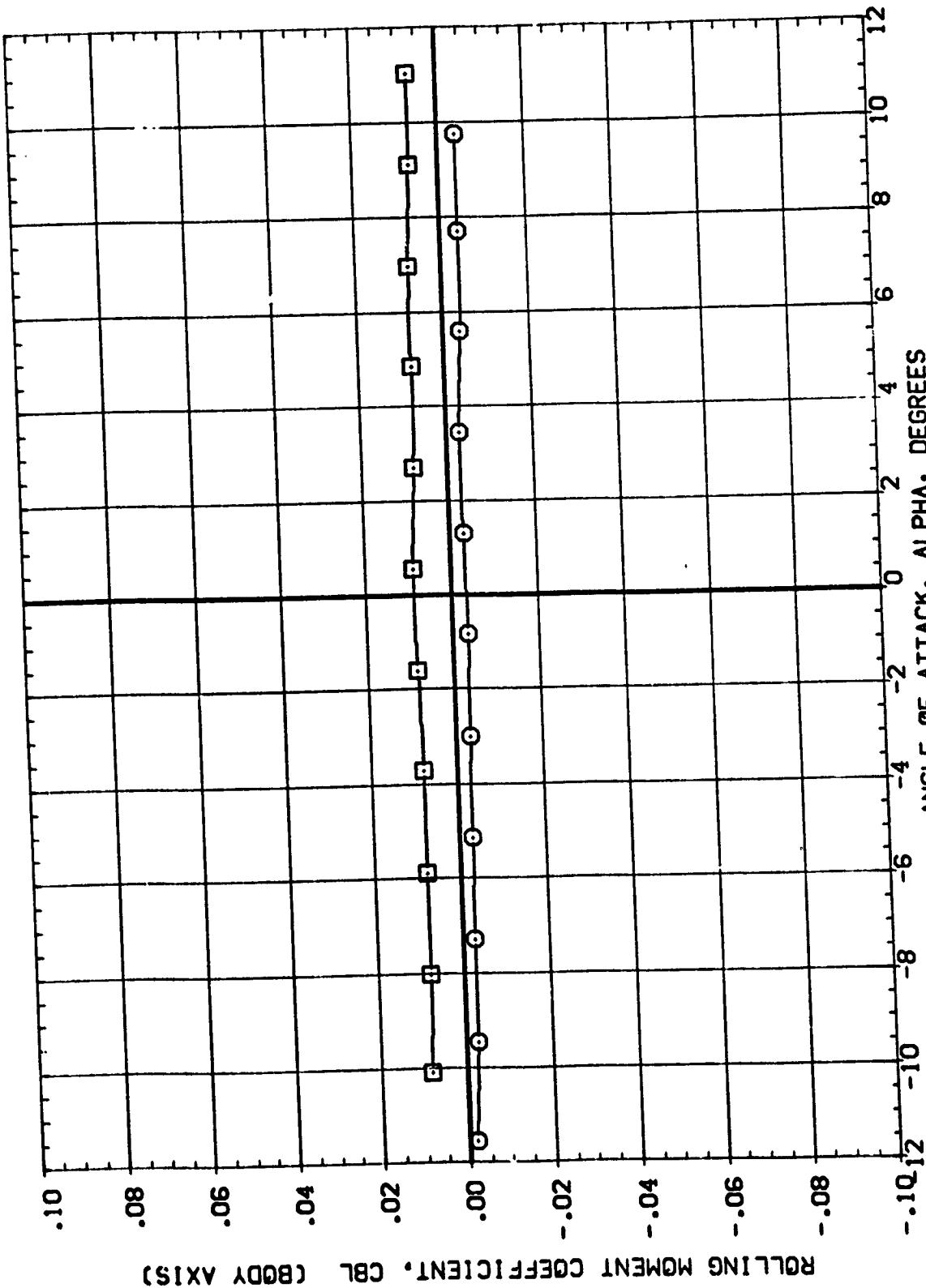
(D)<sub>MACH</sub> = 1.46



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B80000) NSFC 573(A3)FC (03)(T9)(S3)  
 (B80200) NSFC 573(A3)FC (03)(T9)(S3) ORB. MISALND.  
 ORBINC DELTAZ ORBYAW  
 .500 .140 1.000  
 .500 .140

REFERENCE INFORMATION

SPREF 6.1980 SO. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMP 2.5490 IN.  
 YMP .0000 IN.  
 ZMP .0040 IN.



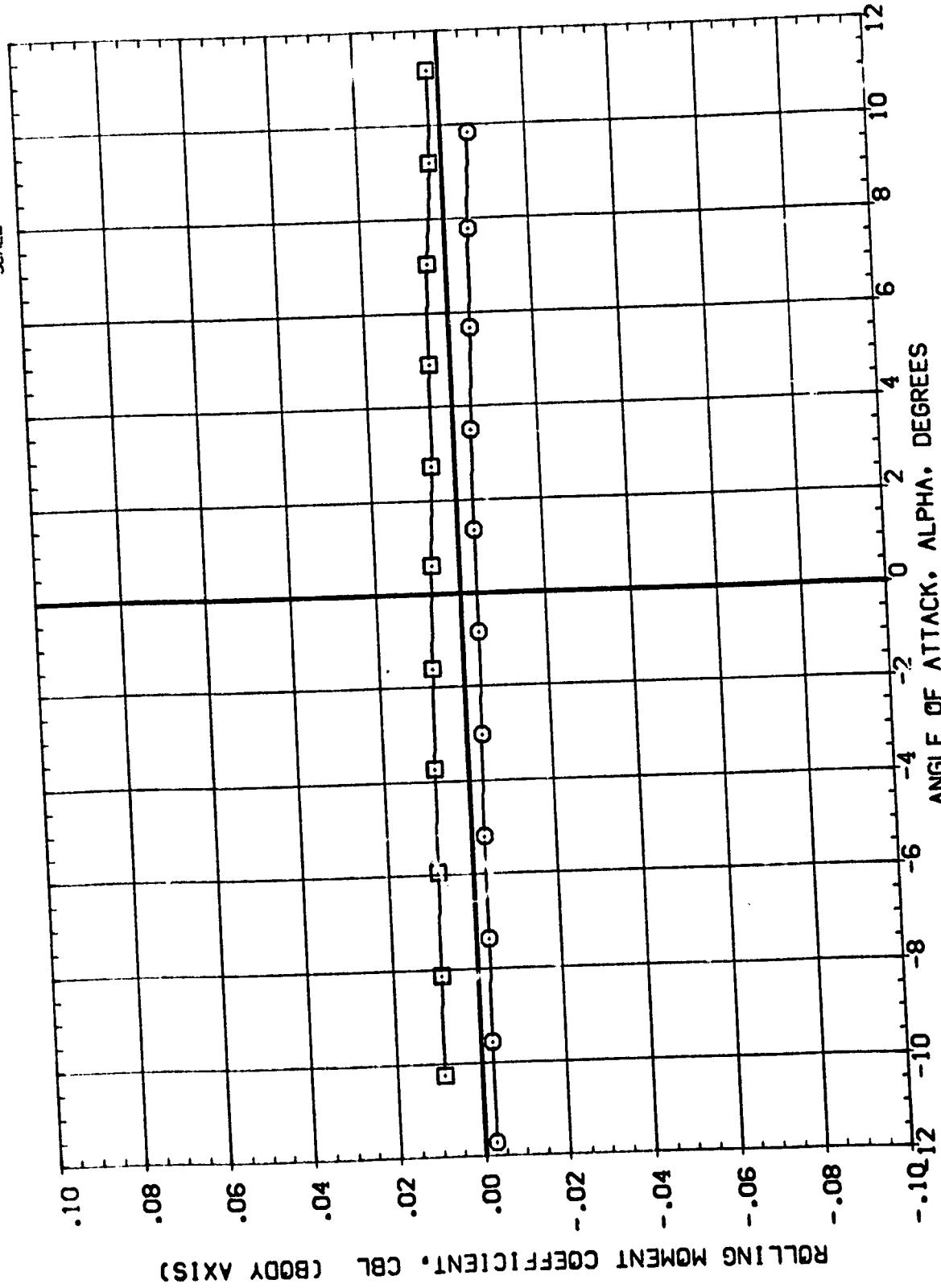
EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

REFERENCE INFORMATION  
 SREF 6.1930 SQ. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.

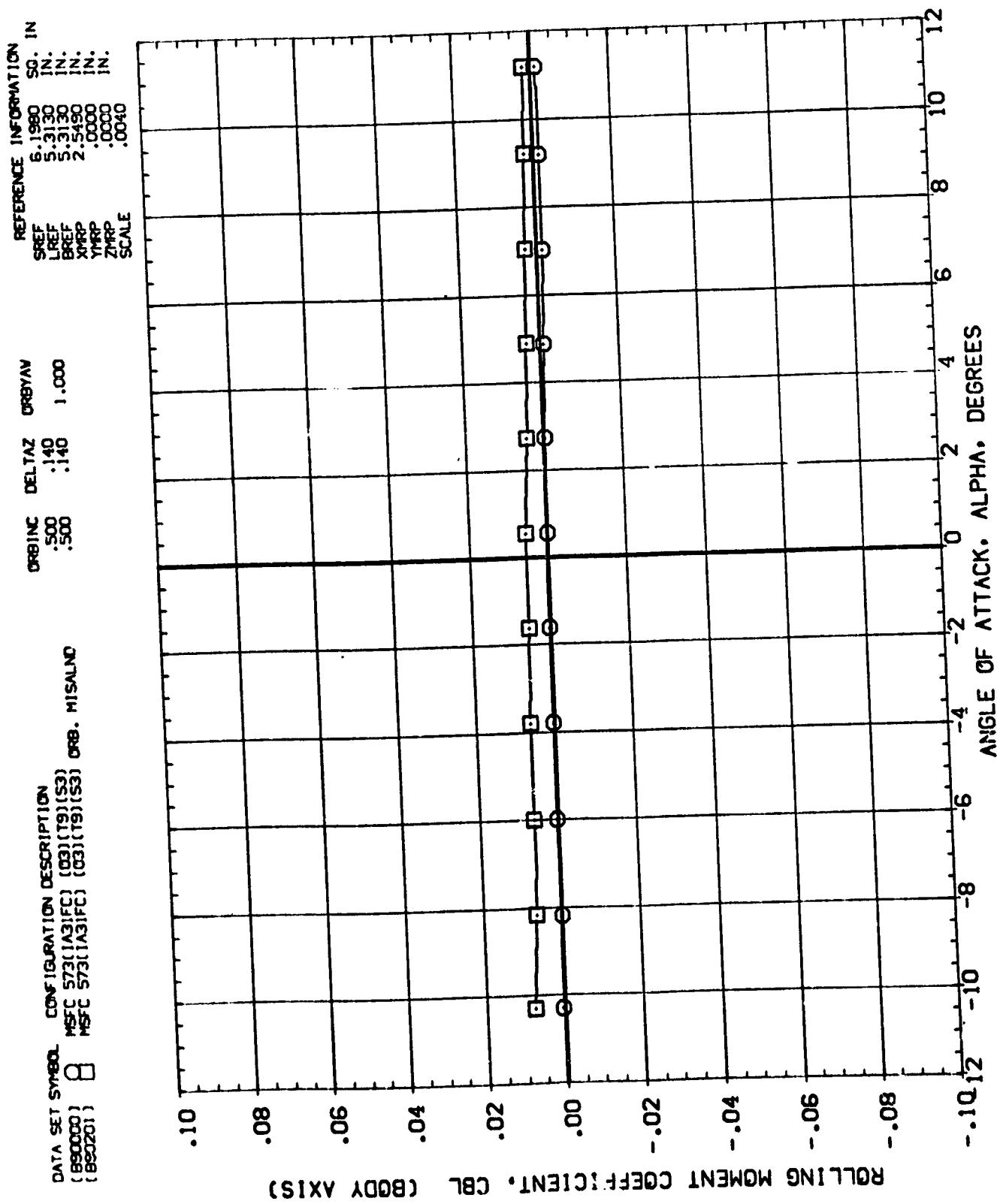
ORBINC DELTAZ ORBYAW  
 .500 .140 1.000

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
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 (B90201)



EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS  
 $(C)_{MACH} = 1.25$

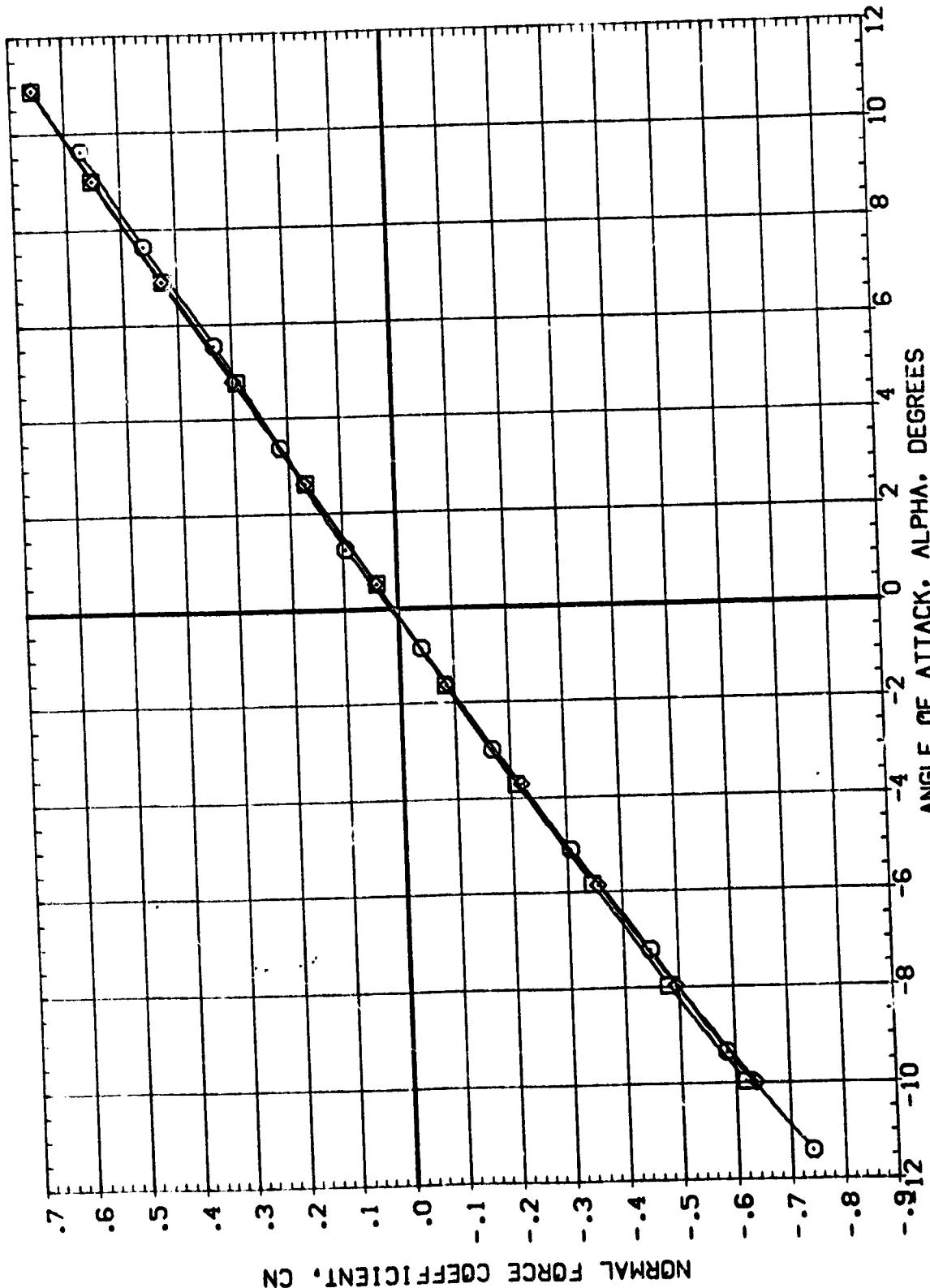
PAGE 100



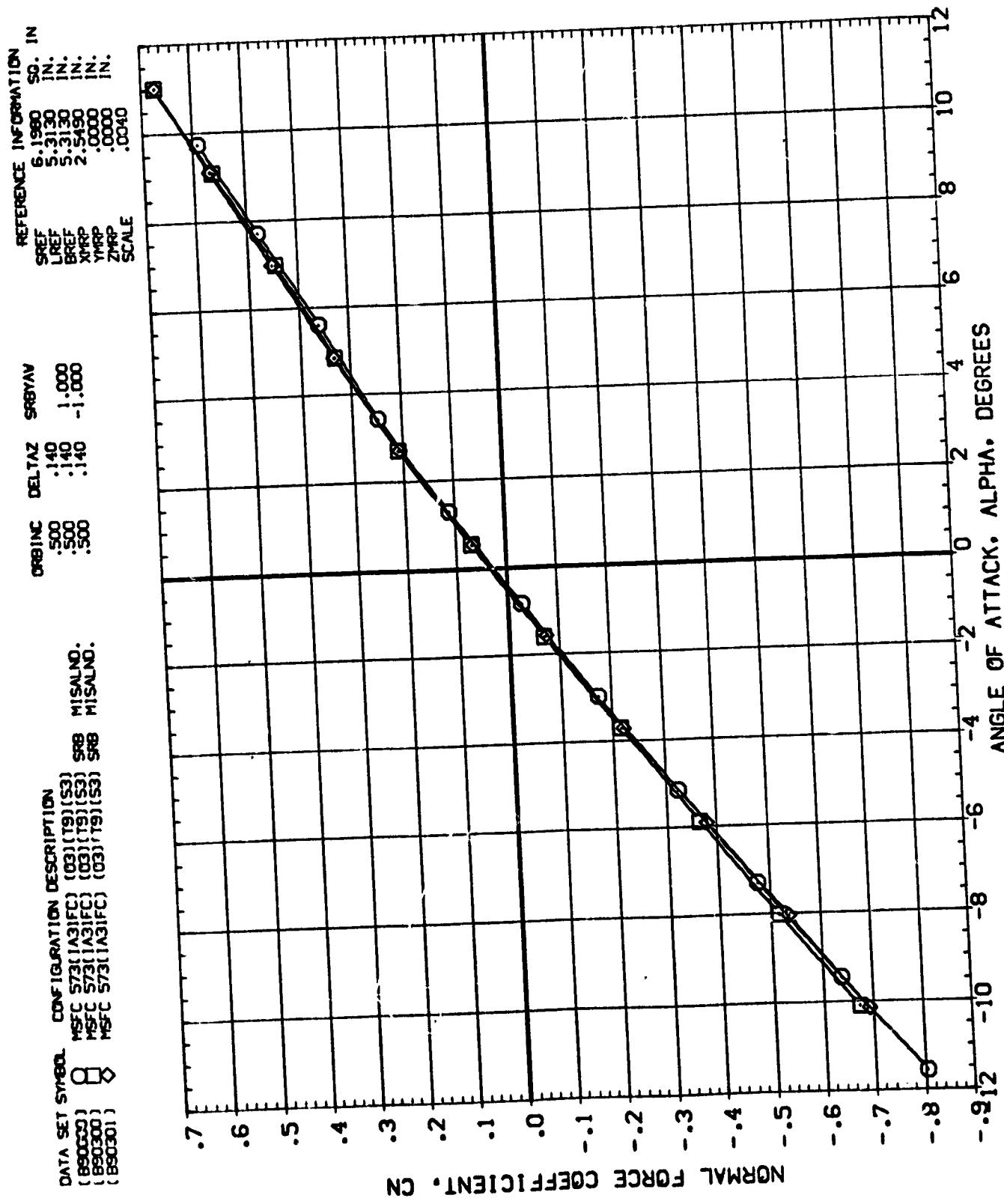
EFFECT OF ORBITER YAW ON AERODYNAMIC CHARACTERISTICS  
 $(\text{D})_{\text{MACH}} = 1.46$

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	DRAING	DELTAZ	SRYAW
(B90000)	NSFC S73[(A3)FC] (03)(T9)(S3)	.500	.140	1.000
(B90300)	NSFC S73[(A3)FC] (03)(T9)(S3)	.500	.140	-1.000
(B9C301)	NSFC S73[(A3)FC] (03)(T9)(S3)	.500	.140	MISLAND.

REFERENCE INFORMATION  
 SRREF 6.1980 SQ. IN.  
 LRREF 5.3130 IN.  
 BRREF 5.3130 IN.  
 XRP 2.5450 IN.  
 YRP .0000 IN.  
 ZRP .0040 IN.  
 SCALE



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
 $(\Delta MACH) = .90$



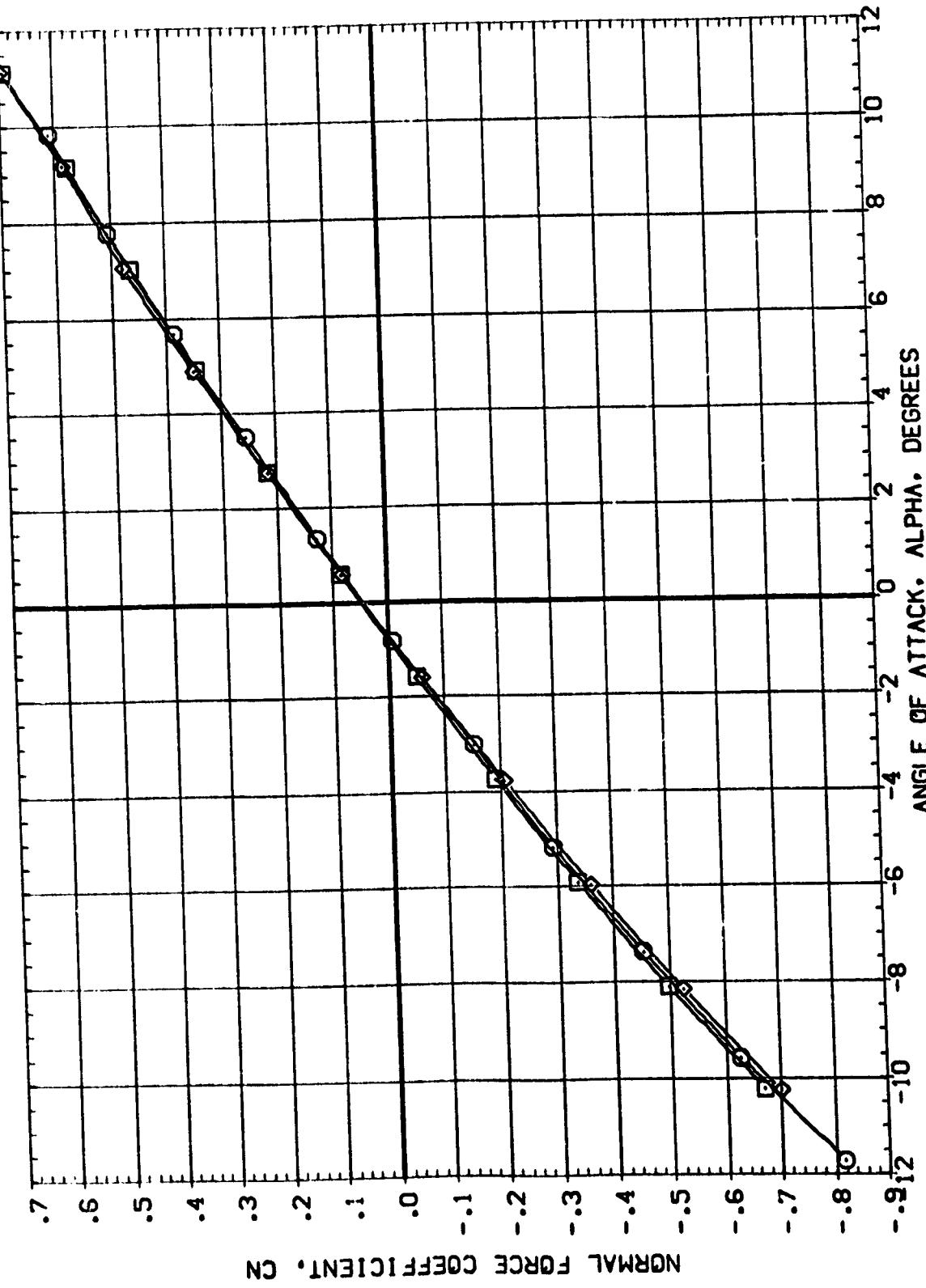
EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
(B)MACH = 1.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 NSFC S731(A3)FC (03)(T9)(S3) SRB MISALNO:  
 (880000) (03)(T9)(S3) SRB MISALNO:  
 NSFC S731(A3)FC (03)(T9)(S3) SRB MISALNO:  
 (880300) (03)(T9)(S3) SRB MISALNO:  
 (880301)

REFERENCE INFORMATION  
 SREF 6.1980 SD. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.

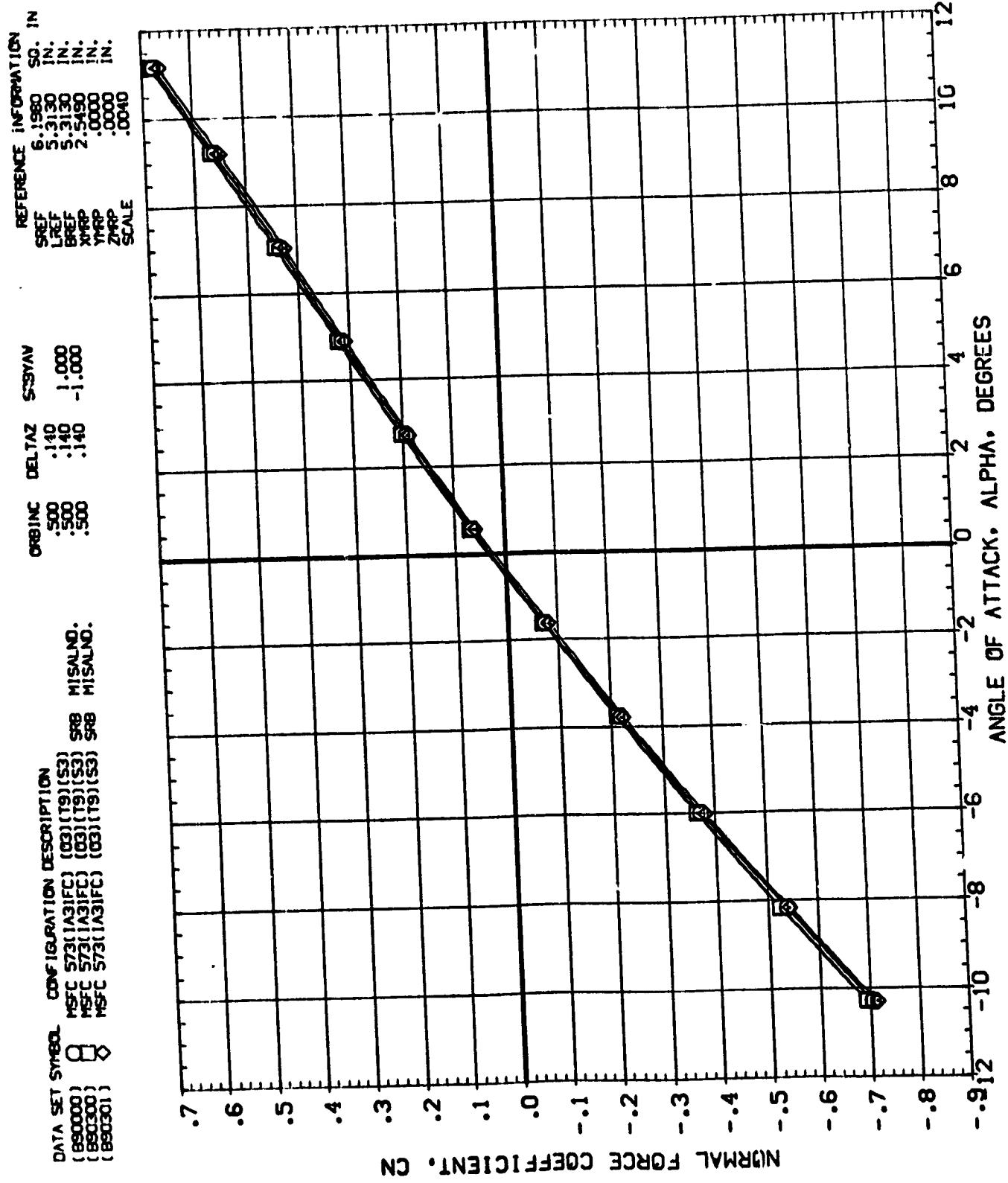
ORB INC DELTAZ SRBYAN  
 .500 .140  
 .500 1.000  
 .500 -.1000

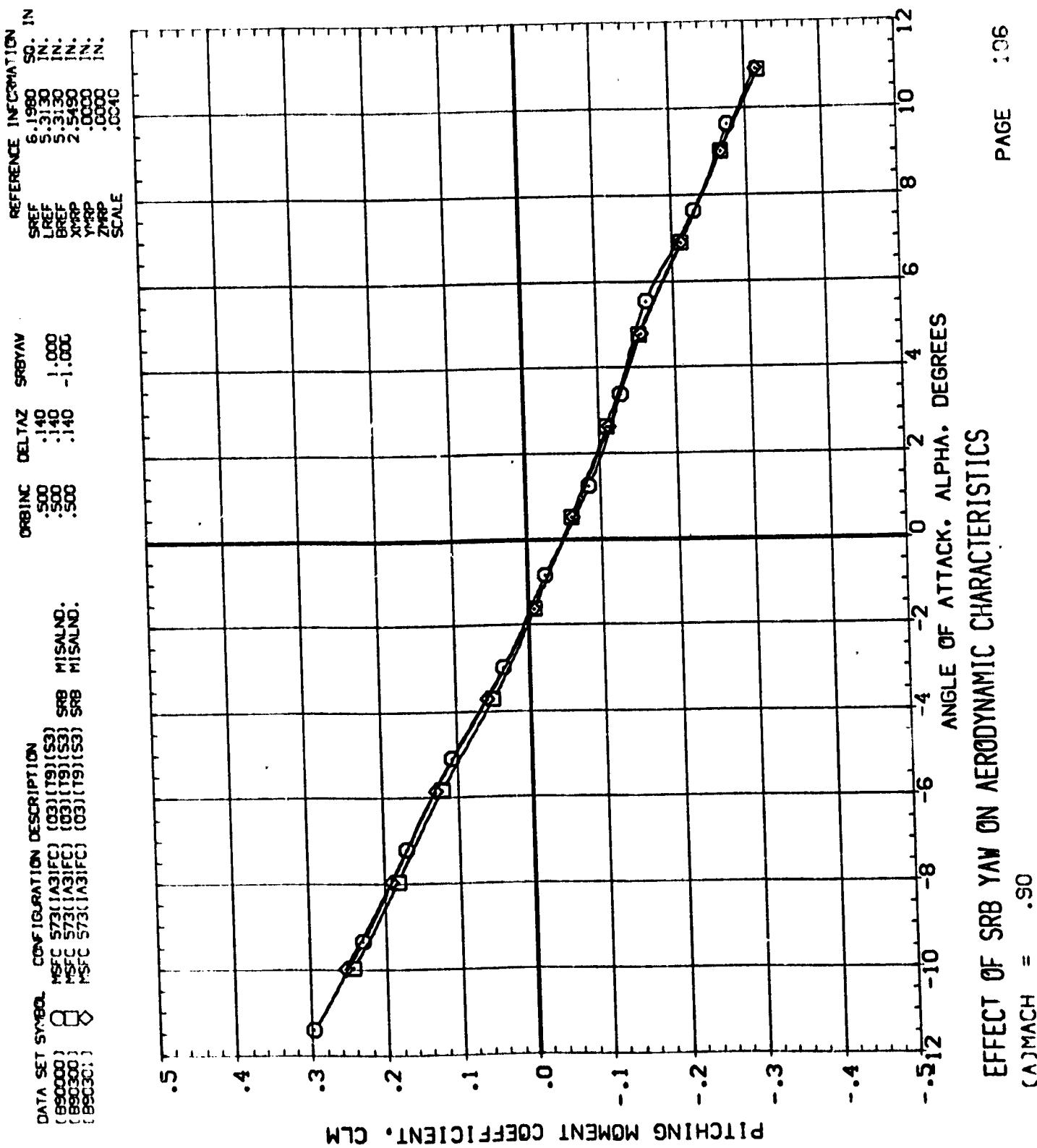
SCALE



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

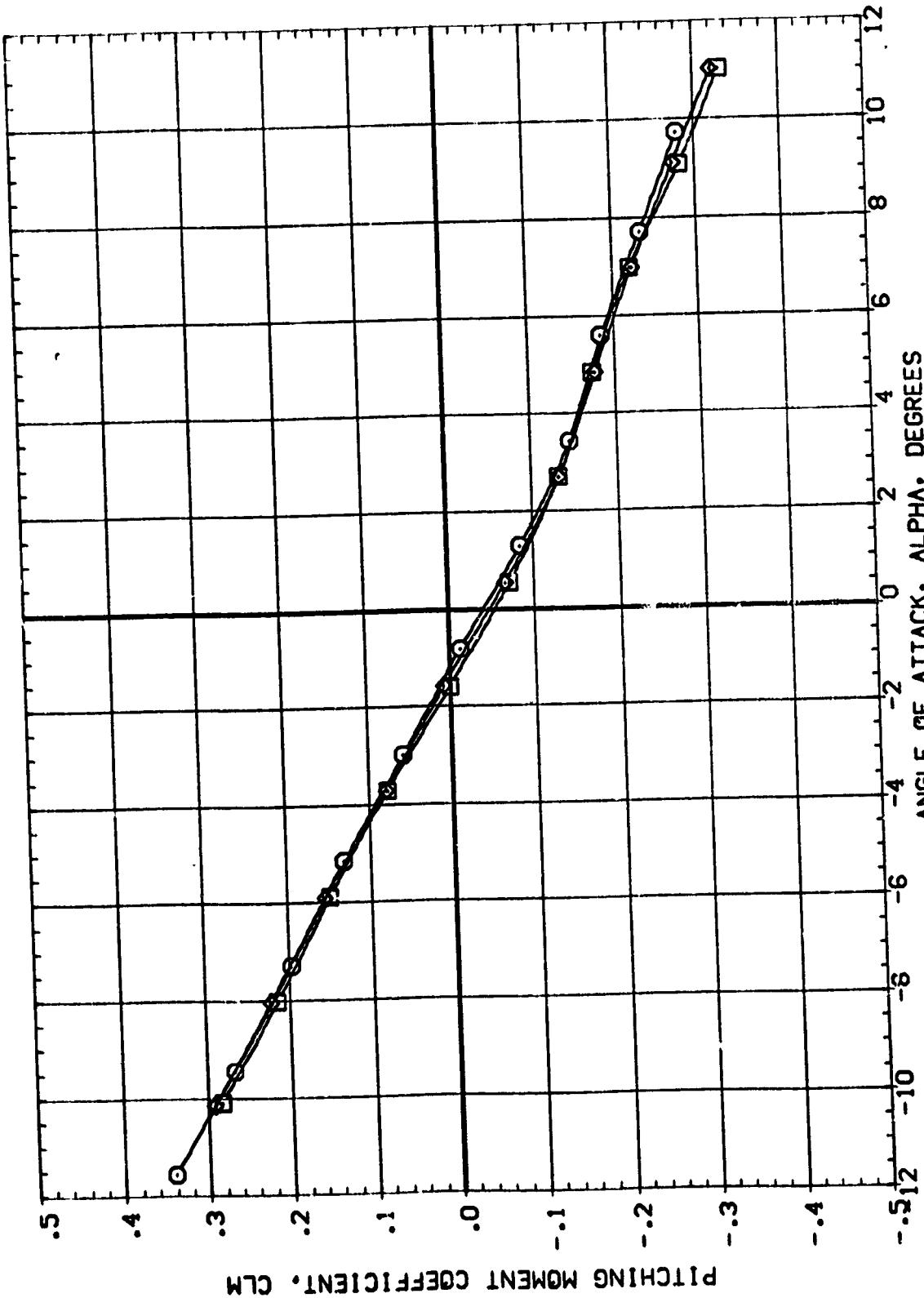
(C)MACH = 1.25





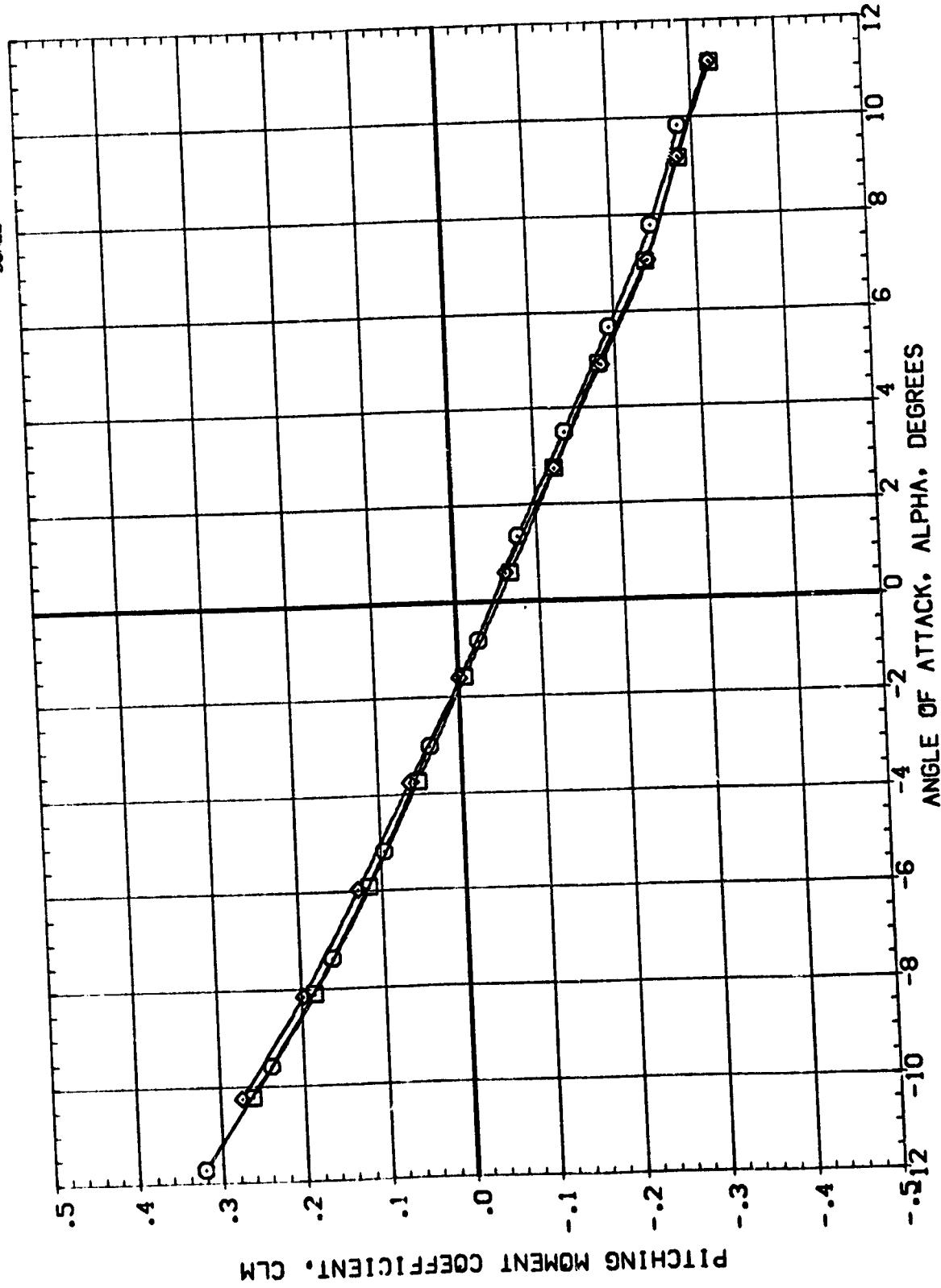
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) NSFC 573(A3)FC (03)(T9)(S3) SRB MISAL NO.  
 (B90020) NSFC 573(A3)FC (03)(T9)(S3) SRB MISAL NO.  
 (B90300) NSFC 573(A3)FC (03)(T9)(S3) SRB MISAL NO.

REFERENCE INFORMATION  
 ORB INC DELTA Z SRB YAW  
 .500 .140 1.000 SC. IN  
 .500 .140 -1.000 .3130 IN.  
 .500 .140 .0000 IN.  
 .0040



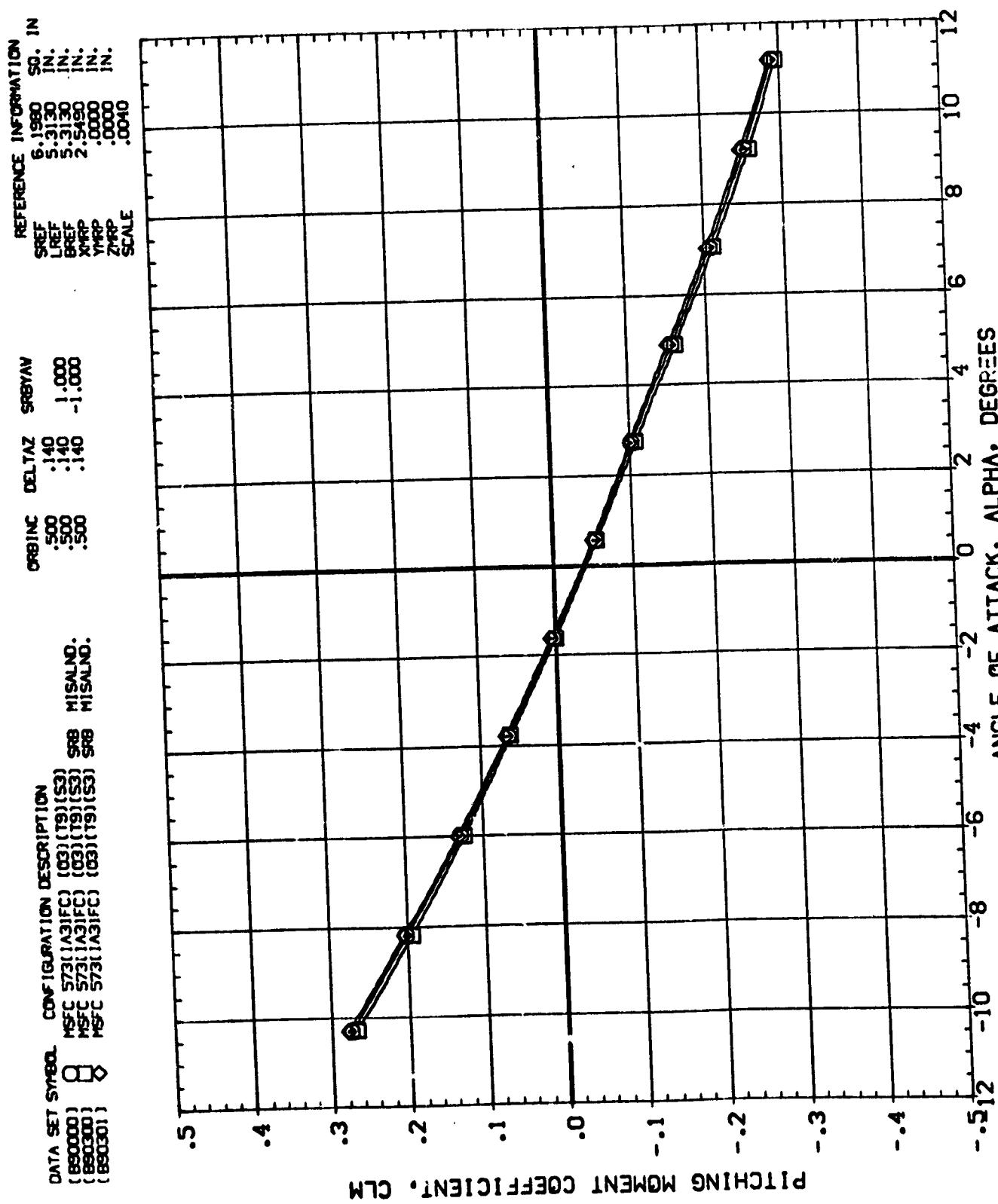
EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
 $(B)_{MACH} = 1.05$

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	CRBINC	DELTAZ	SRBYAW	REFERENCE INFORMATION
(B60000)	MSFC S731(A3IFC)	.500	.140	1.000	SREF 6.1980 SD. IN.
(B6C300)	MSFC S731(A3IFC)	.500	.140	-1.000	LREF 5.3130 IN.
(B6C301)	MSFC S731(A3IFC)	.500	.140	0.000	BREF 5.3130 IN.
					XMRP 2.5490 IN.
					YMRP .0000 IN.
					ZMRP .0040 IN.
					SCALE .0040

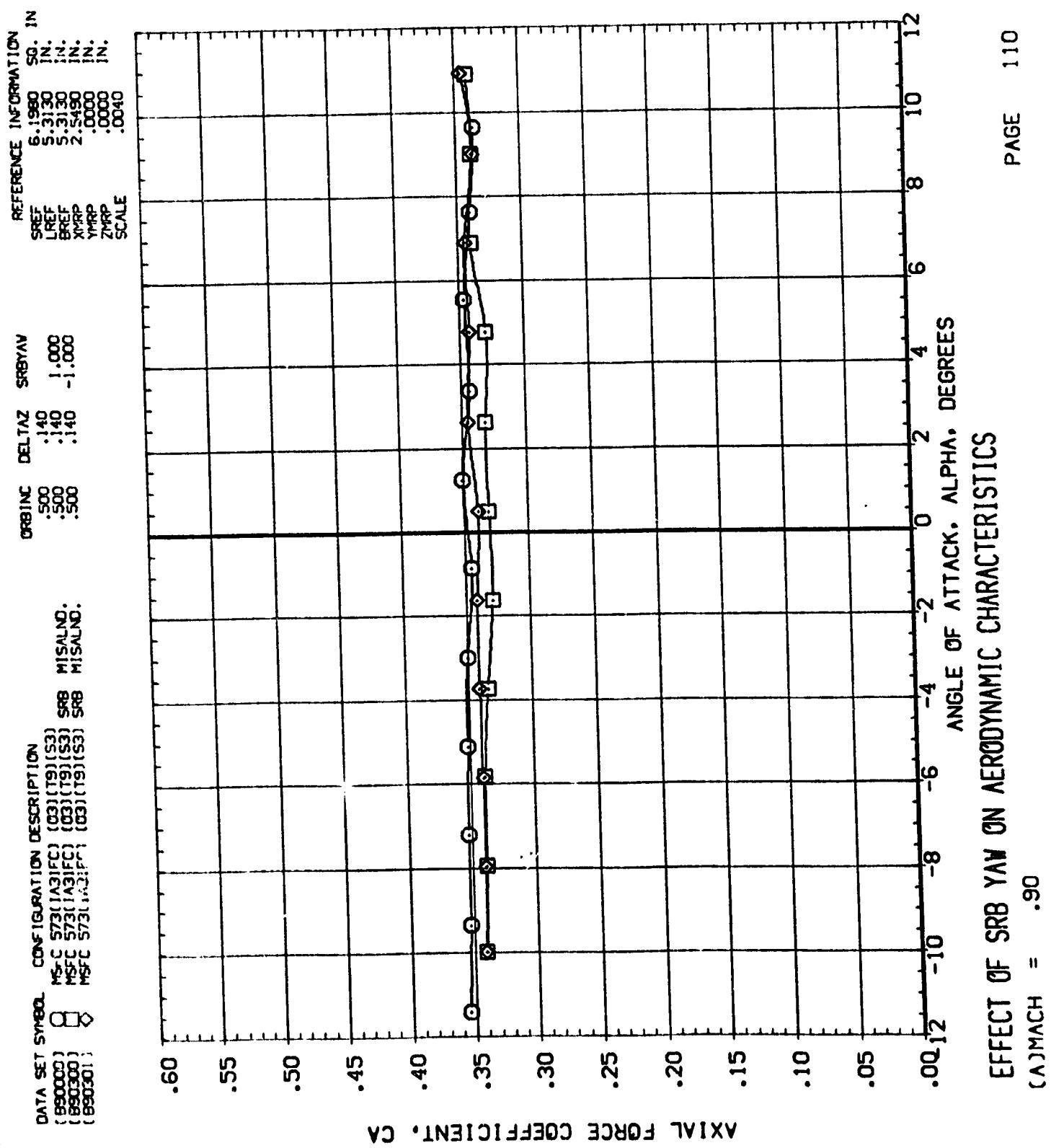


### EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(C)<sub>MACH</sub> = 1.25

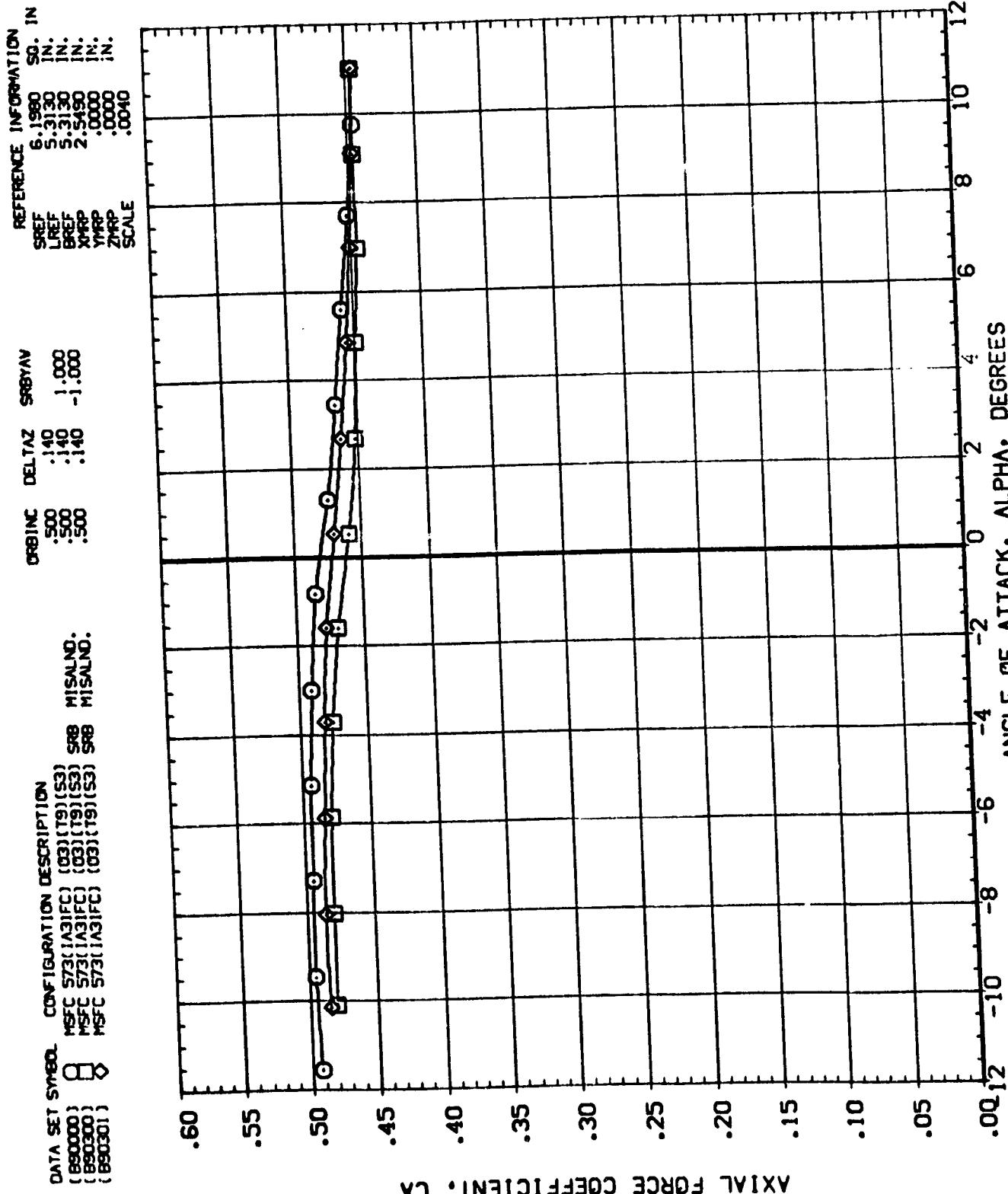


EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
 $(\text{D}\text{O}\text{M}\text{A}\text{C} \text{=} 1.46)$



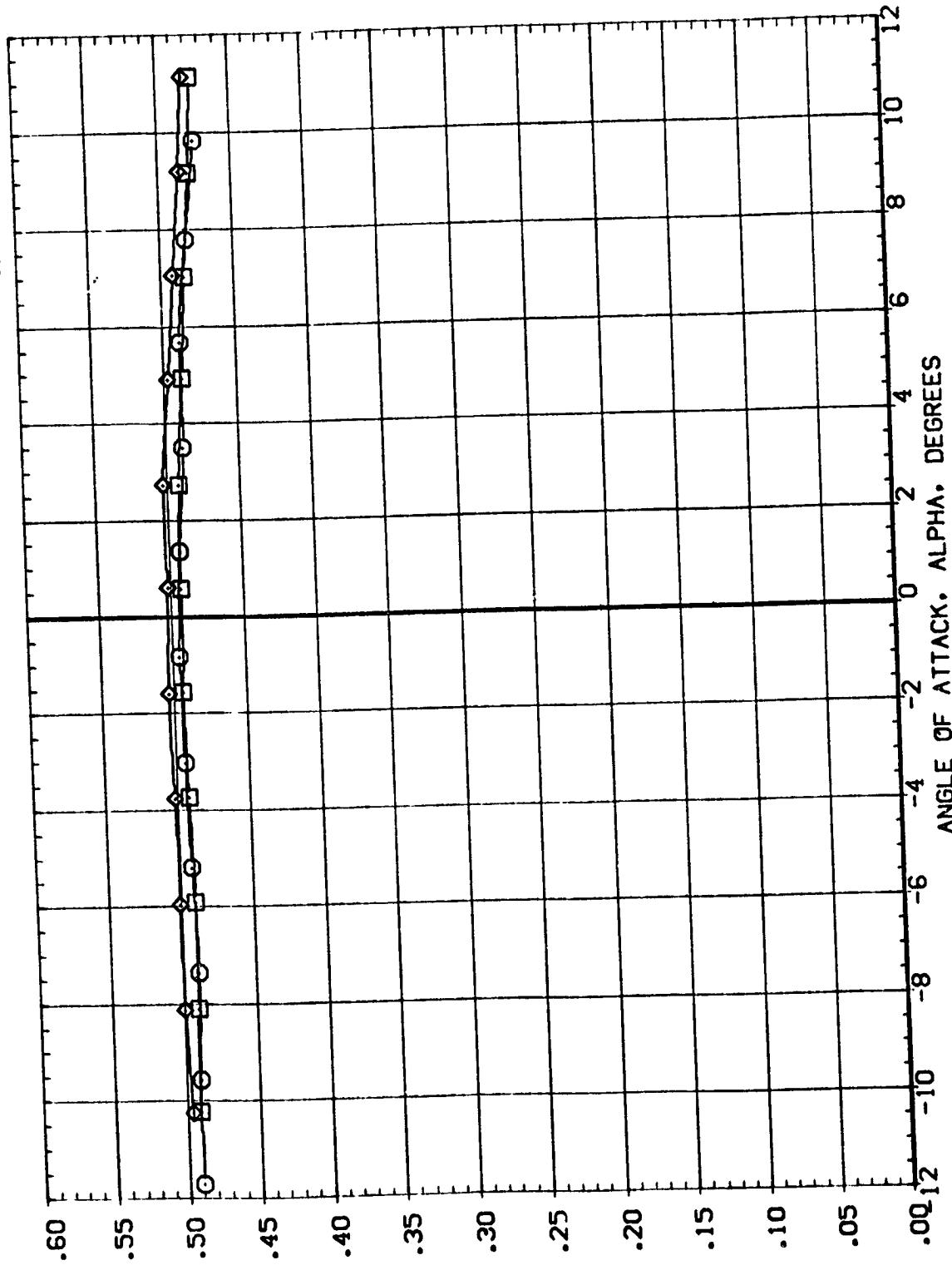
DATA SET SYMBOL CONFIGURATION DESCRIPTION

(B90000)	MSFC	573(1A3)FC	(03)(19)(S3)	SRB	MISALND.
(B90300)	MSFC	573(1A3)FC	(03)(19)(S3)	SRB	MISALND.
(B90301)	MSFC	573(1A3)FC	(03)(19)(S3)	SRB	MISALND.



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
(B)MACH = 1.05

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ORB INC	DELTAYAW	SRBYAW	REFERENCE INFORMATION
(850000)	NSFC 573([A3]FC) ([03][T9])(S3)	.500	.140	.1000	SREF 6.1980 50. IN
(850300)	NSFC 573([A3]FC) ([03][T9])(S3)	.500	.140	-1.000	LREF 5.3130 1.N.
(850301)	NSFC 573([A3]FC) ([03][T9])(S3)	.500	.140	.0000	BREF 5.3130 1.N.
					XMRP 2.5490 1.N.
					ZMRP .0000 1.N.
					SCALE .0040



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

$(C)MACH = 1.25$

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DATA SET SYMBOL CONFIGURATION DESCRIPTION

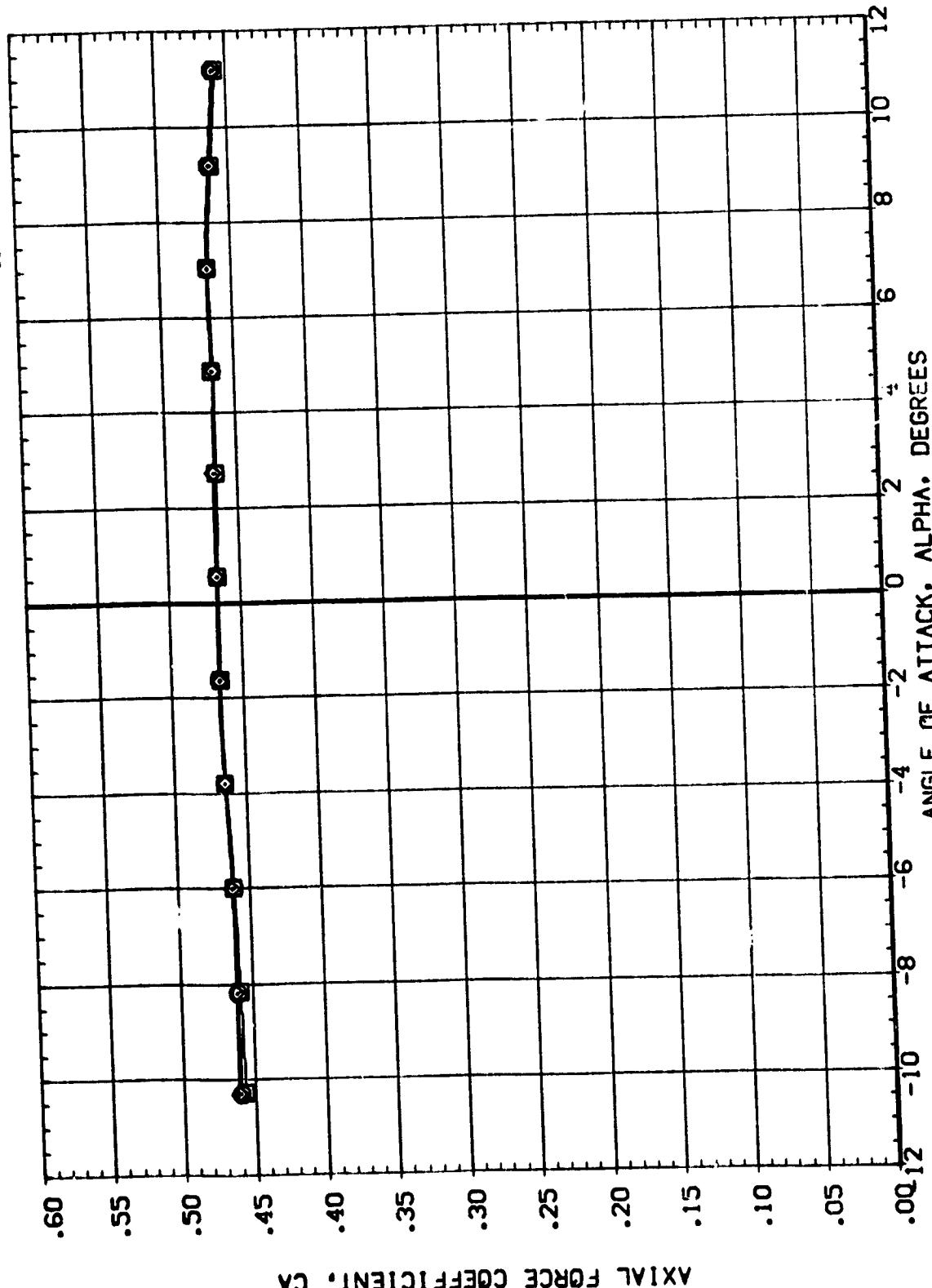
(S73000)	HSC 573(1A3)FC	(03)(T9)(S3)	SRB	MISALNO.
(S73000)	HSC 573(1A3)FC	(03)(T9)(S3)	SRB	MISALNO.
(S73000)	HSC 573(1A3)FC	(03)(T9)(S3)	SRB	MISALNO.

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(S73000)	HSC 573(1A3)FC	(03)(T9)(S3)	SRB	MISALNO.
(S73000)	HSC 573(1A3)FC	(03)(T9)(S3)	SRB	MISALNO.
(S73000)	HSC 573(1A3)FC	(03)(T9)(S3)	SRB	MISALNO.

CRBINC DELTAZ SRBYAW  
.500 .140 1.000  
.500 .140 -1.000  
.500 .140 -.0000  
SCALE .0040

REFERENCE INFORMATION  
SREF 6.1980 SQ. IN.  
LREF 5.3130 IN.  
BREF 5.3130 IN.  
XRP 2.5490 IN.  
YRP .0000 IN.  
ZMRP .0000 IN.  
SCALE .0040

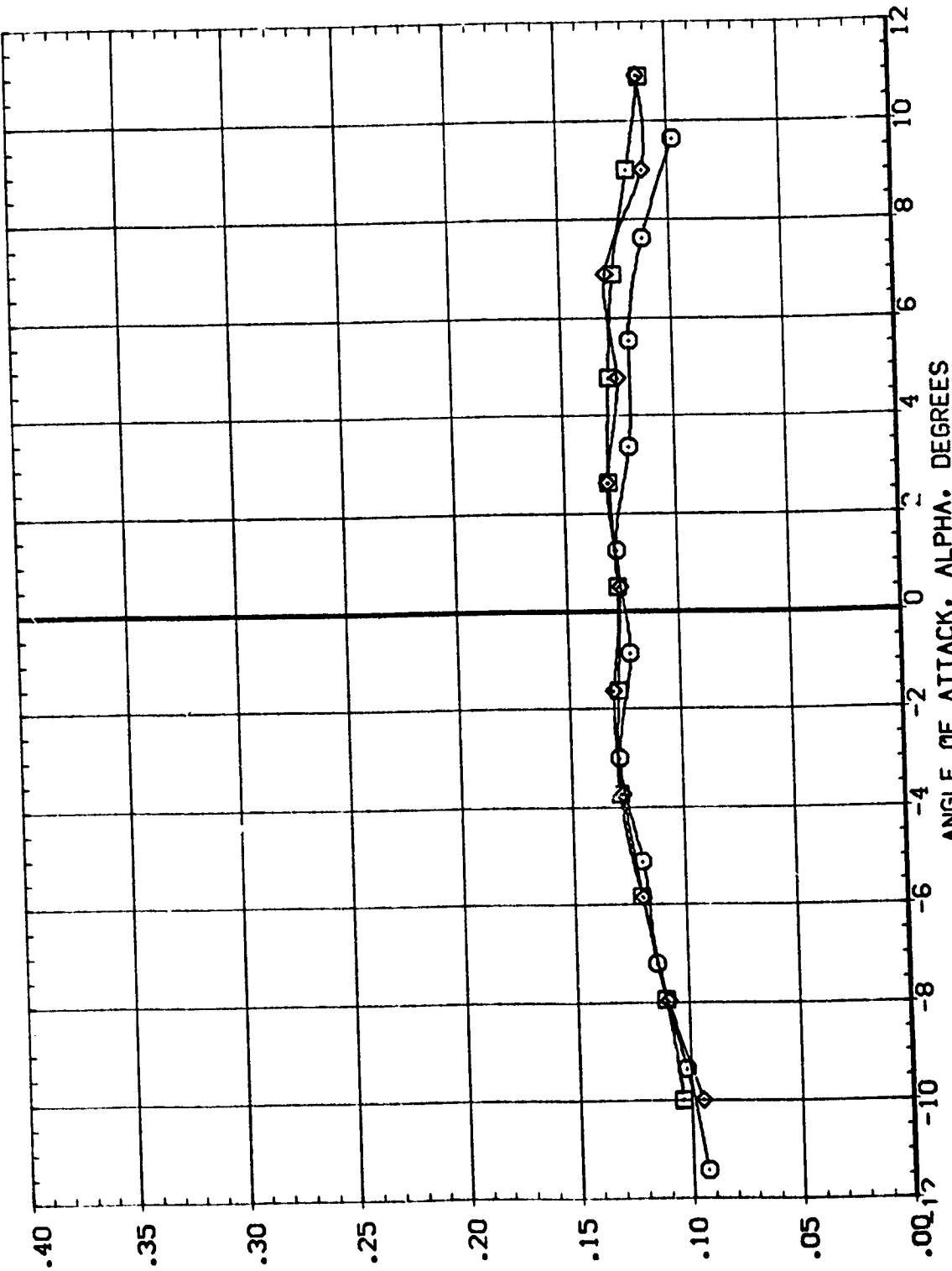


EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(D)MACH = 1.46

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (860000) NSFC S73([A3]FC) [03](T9)(S3) MISALNO.  
 (860300) NSFC S73([A3]FC) [03](T9)(S3) MISALNO.  
 (860301) NSFC S73([A3]FC) [03](T9)(S3) MISALNO.

PREFERENCE INFORMATION  
 SRREF 6.1980 SO. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XRP 2.5490 IN.  
 YRP .0000 IN.  
 ZRP .0040 IN.  
 SCALE



FORCEBODY AXIAL FORCE COEFFICIENT, CAF

### EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

( $\Delta$ MACH = .90

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B80000) NSFC 5731(A31FC) (G3)(T9)(S3) SRB MISALNO.  
 (B80000) NSFC 5731(A31FC) (G3)(T9)(S3) SRB MISALNO.  
 (B80000) NSFC 5731(A31FC) (G3)(T9)(S3) SRB MISALNO.  
 (B80000) NSFC 5731(A31FC) (G3)(T9)(S3) SRB MISALNO.

REFERENCE INFORMATION

SREF	6.1980	IN.
LREF	5.3130	IN.
BREF	5.3130	IN.
XTRP	2.5490	IN.
YTRP	.0000	IN.
ZTRP	.0040	IN.

DRBINC DELTAZ SPBYAV

.500 .140 1.000

.500 .140 -1.000

.500 .140 .000

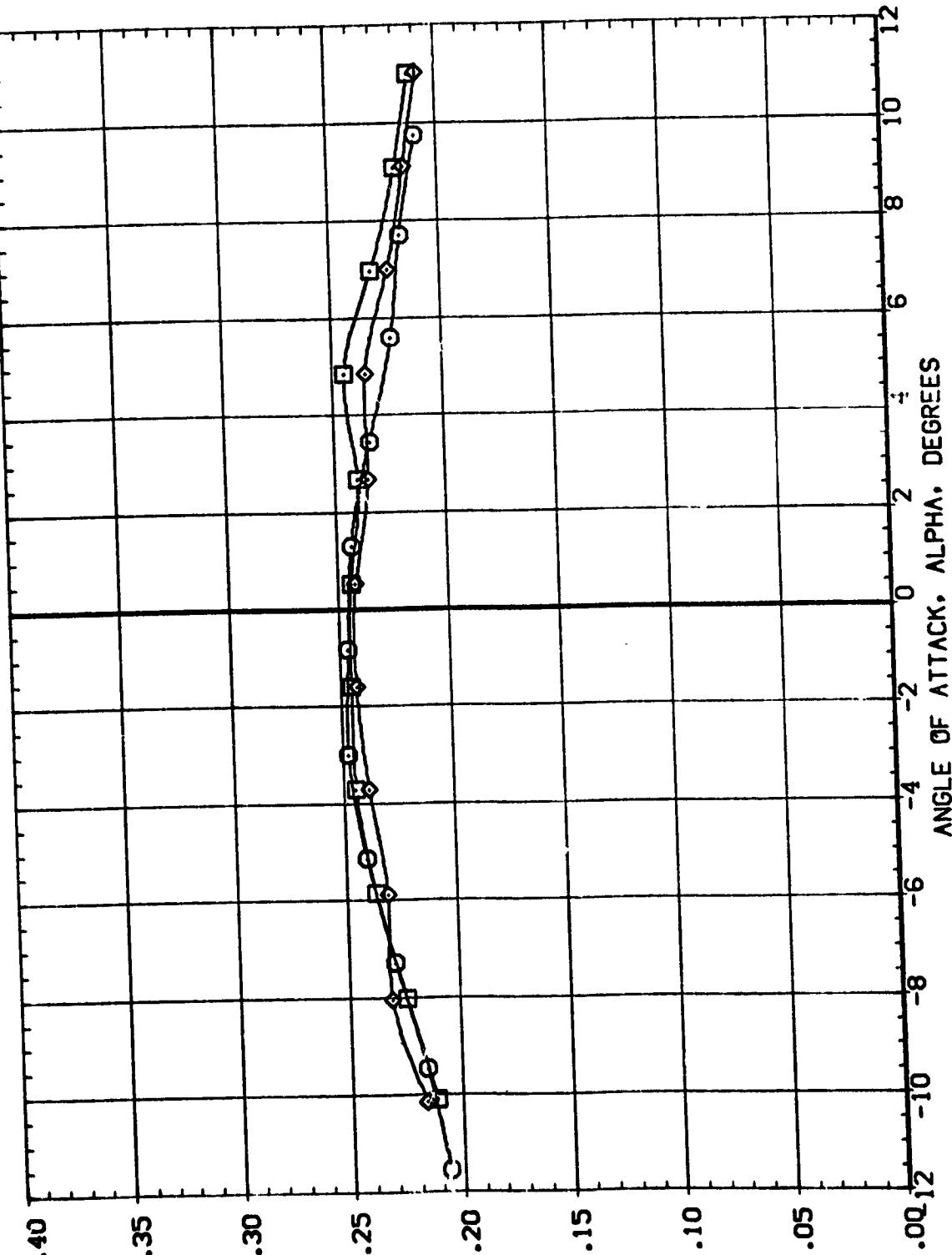
.000 .000 .000

DRBINC

DELTAZ

SPBYAV

FORBODDY AXIAL FORCE COEFFICIENT, CAF

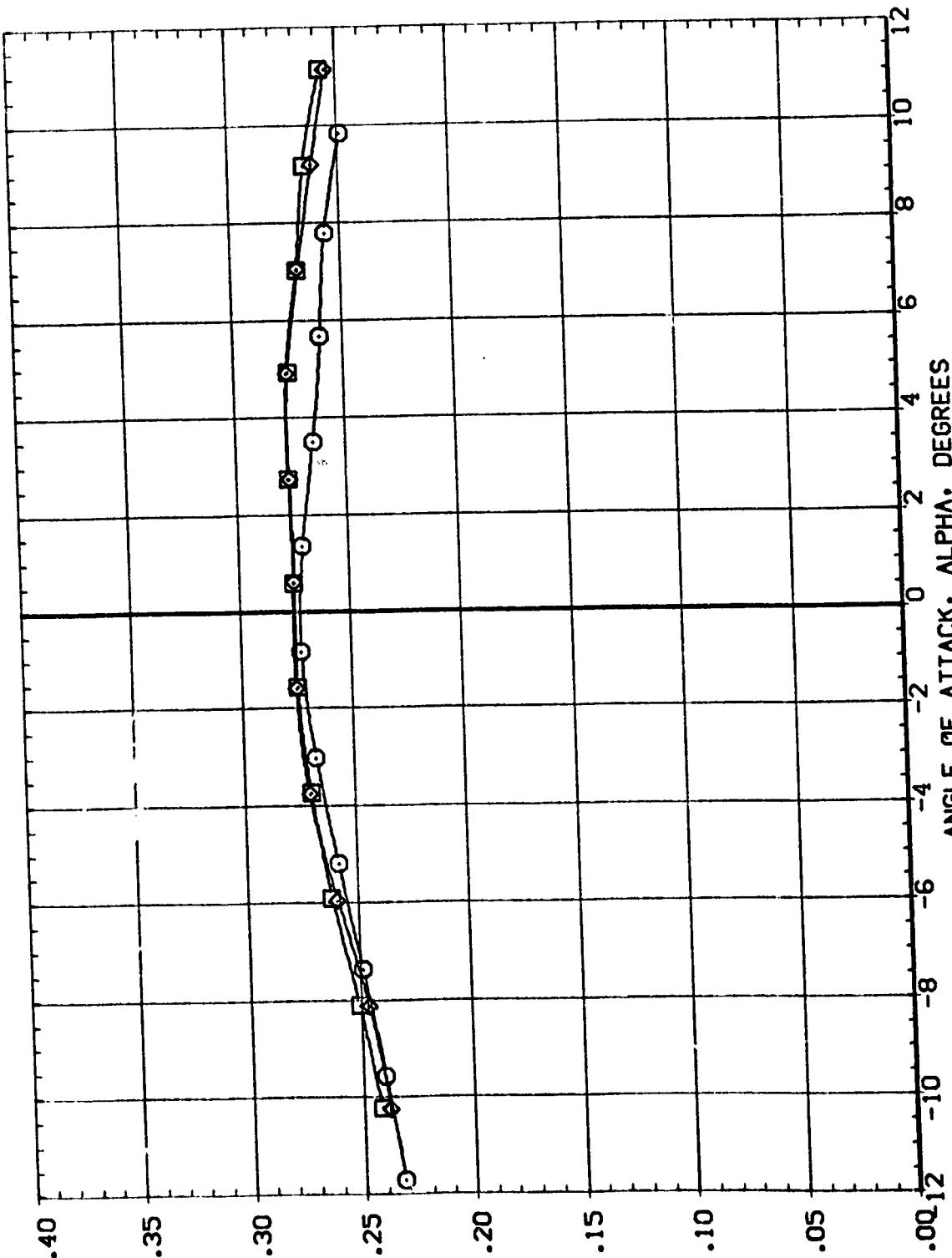


EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	DRB INC	DELTAZ	SRBYAW
(890000)	NSFC 573(I A3(FC) (03)(T9)(S3)	.500	.140	
(890300)	NSFC 573(I A3(FC) (03)(T9)(S3)	.500	.140	1.000
(893001)	NSFC 573(I A3(FC) (03)(T9)(S3)	.500	.140	-1.000

REFERENCE INFORMATION  
 SD. IN.  
 6.1980  
 5.3130  
 5.3130  
 5.3130  
 XMRP  
 YMRP  
 ZMRP  
 .0000  
 .0000  
 .0040  
 IN.  
 SREF  
 LREF  
 BREF  
 XMRP  
 YMRP  
 ZMRP  
 SCALE

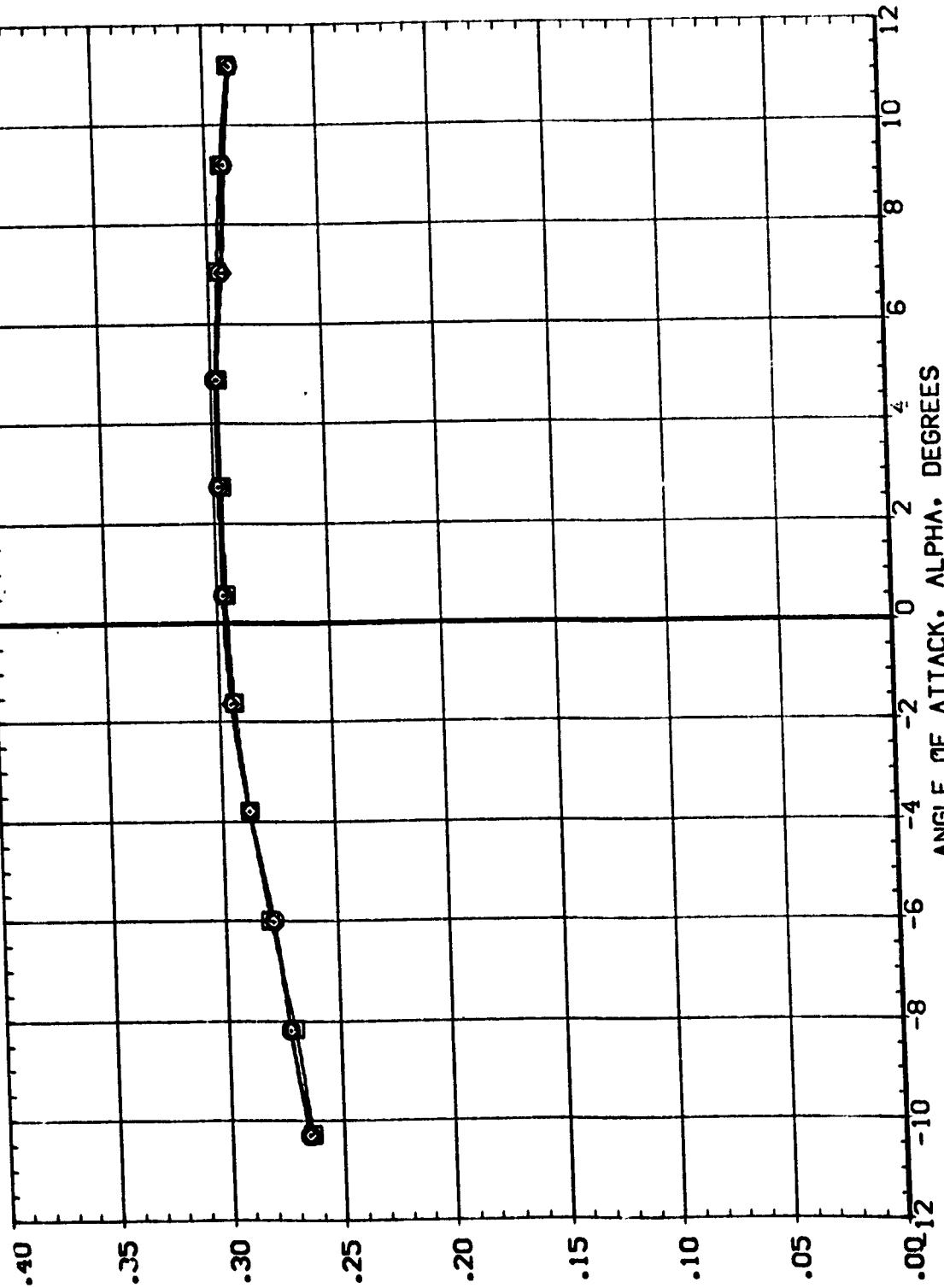


FORCEBODY AXIAL FORCE COEFFICIENT, CAF

EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
 (C)MACH = 1.25

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	REFERENCE INFORMATION
(650000)	MSFC 5731(A3)IFC	SREF 6.1980 50. N.
(650300)	MSFC 5731(A3)IFC	LREF 5.3130 1.N.
(650301)	MSFC 5731(A3)IFC	BREF 5.3130 1.N.
		XRP 2.5490 1.N.
		YRP .0000 1.N.
		ZRP .0040 1.N.

ORB INC DELTAZ SRBYAW  
.500 .140 1.000  
.500 .140 -1.000  
.500 .140

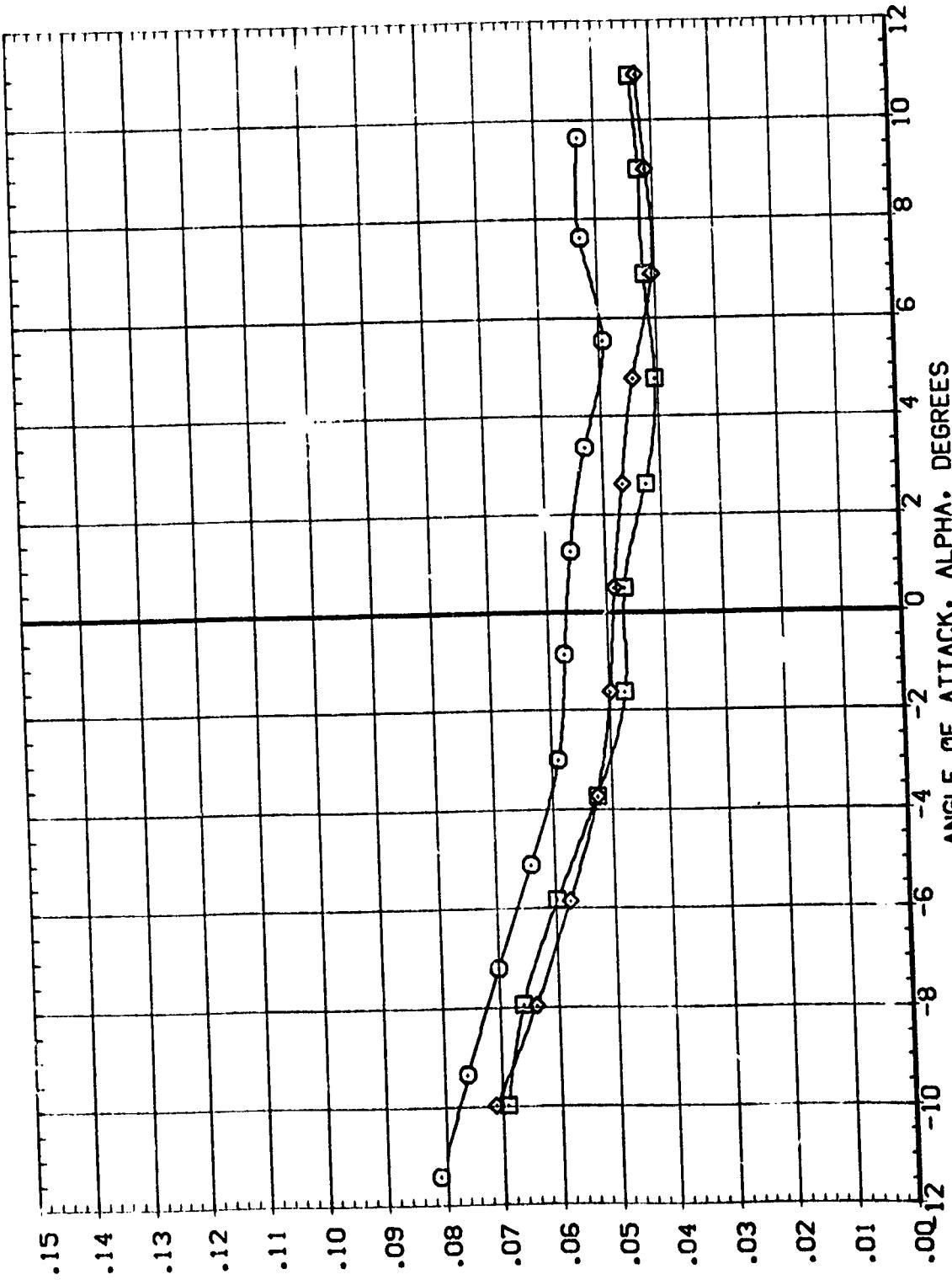


FORCEBODY AXIAL FORCE COEFFICIENT, CAF

EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(D)MACH = 1.46

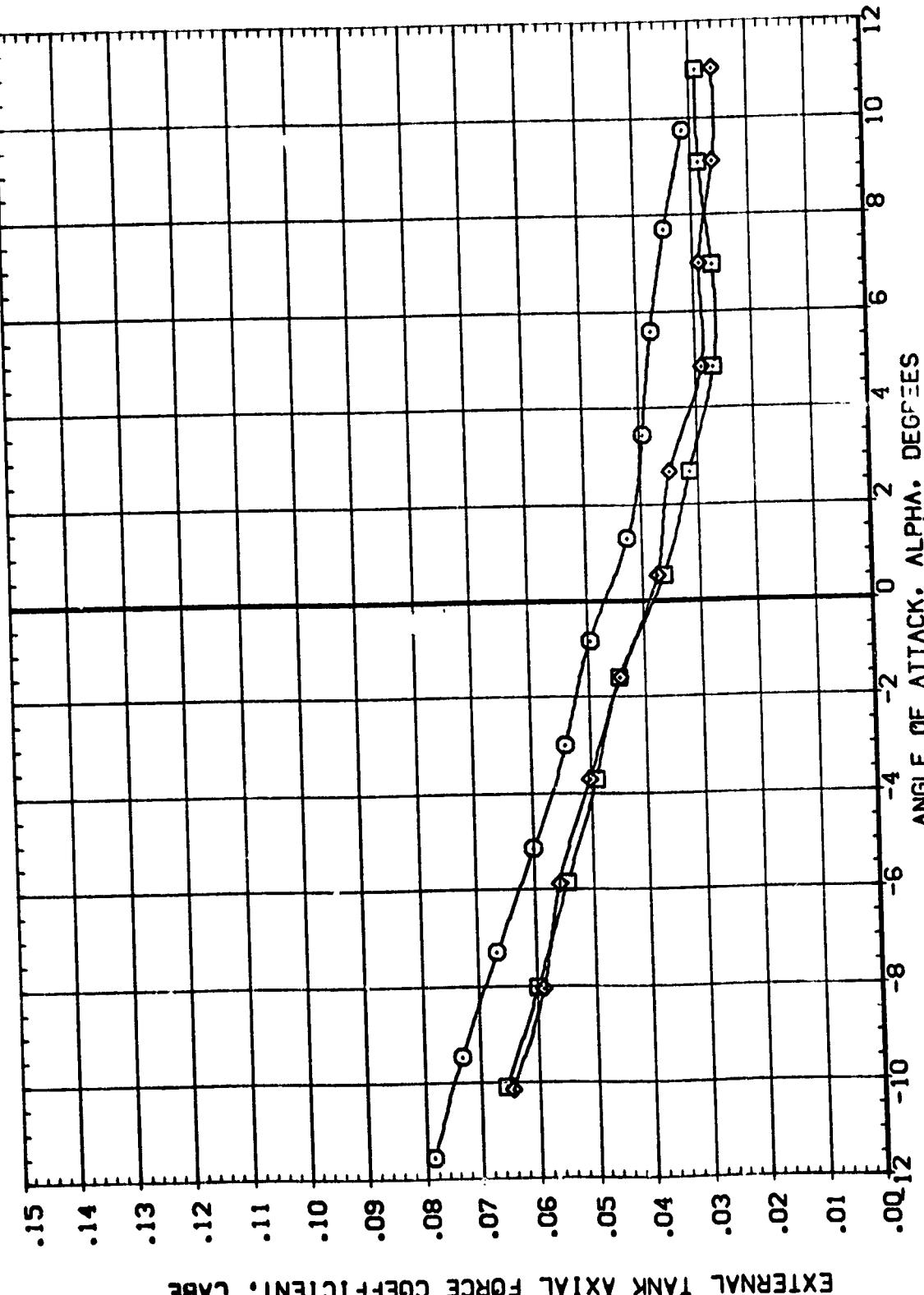
DATA SET SYMBOL	CONFIGURATION DESCRIPTION	SRB INC.	DETAZ	SRBYAW	REFERENCE INFORMATION
(890000)	MSFC 573([A3]IFC)	.500	.140	SD. IN.	
(890300)	MSFC 573([A3][S3])	.500	.140	6.1980 SD. IN.	
(890301)	MSFC 573([A3]IFC)	.500	.140	5.3130 IN.	
	MSFC 573([A3][S3])	.500	.140	5.3130 IN.	
	MSFC 573([A3]IFC)	.500	.140	2.5450 IN.	
	MSFC 573([A3][S3])	.500	.140	.0000 IN.	
				ZTRP .0040 IN.	



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
(A)MACH = .90

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B500000) NSFC 5731(A31FC) (S3) SRF MISLAND.  
 (B500300) NSFC 5731(A31FC) (S3) SRF MISLAND.  
 (B500301) NSFC 5731(A31FC) (S3) SRF MISLAND.

REFERENCE INFORMATION  
 SREF 6.1980 SD. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(B)<sub>MACH</sub> = 1.05

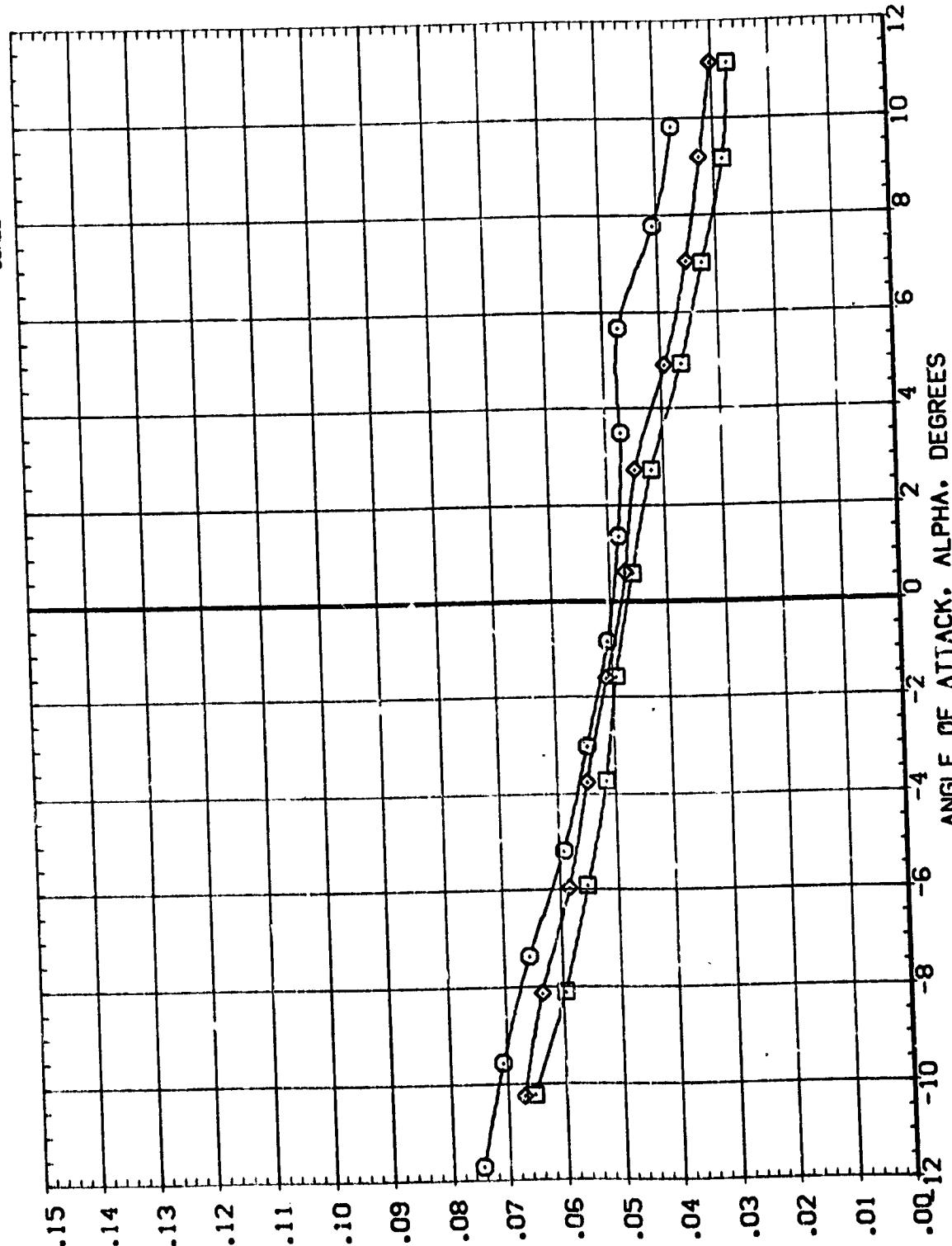
REFERENCE INFORMATION

SREF	5.1980	SD. IN
LREF	5.3130	IN.
BREF	5.3130	IN.
XMRP	2.5490	IN.
YMRP	.0000	IN.
ZMRP	.0040	IN.

SCALE

DATA SET SYMBOL CONFIGURATION DESCRIPTION

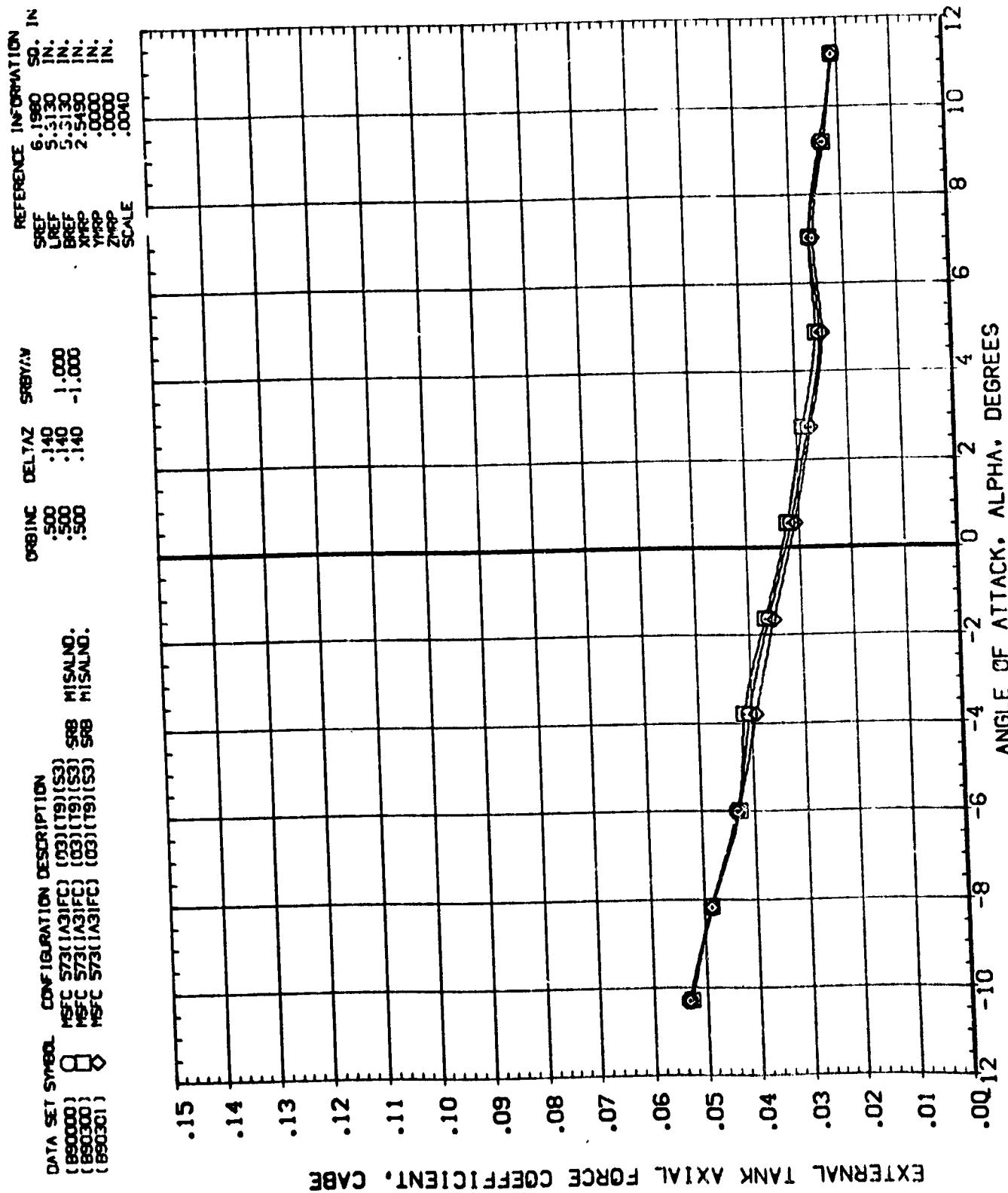
(B00000)	MSFC S73[(A3)FC]	[03][19][S3] SRB MISALND.
(B0C300)	MSFC S73[(A3)FC]	[03][19][S3] SRB MISALND.
(B0301)	MSFC S73[(A3)FC]	[03][19][S3] SRB MISALND.



### EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

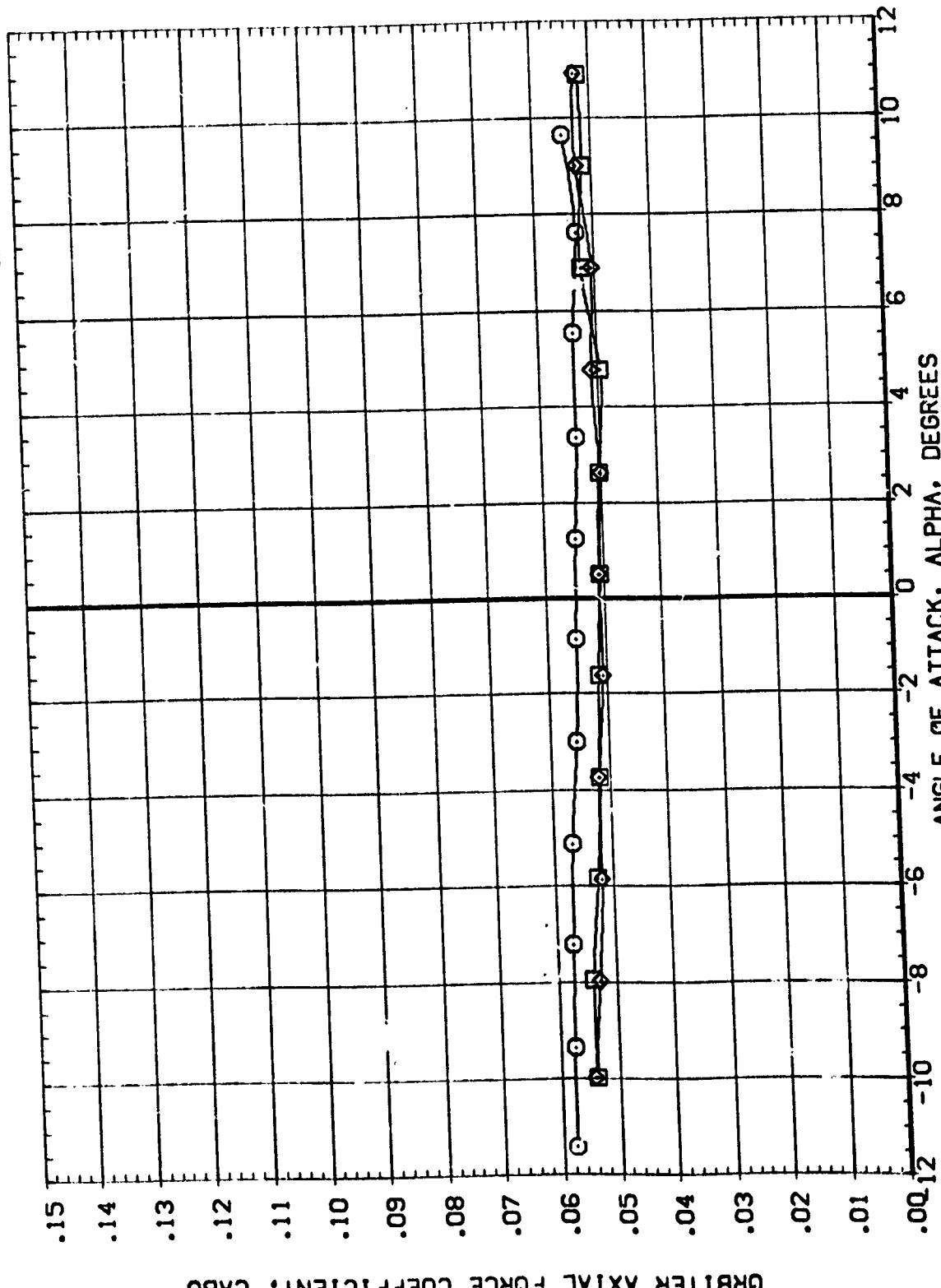
(C)MACH = 1.25

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EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

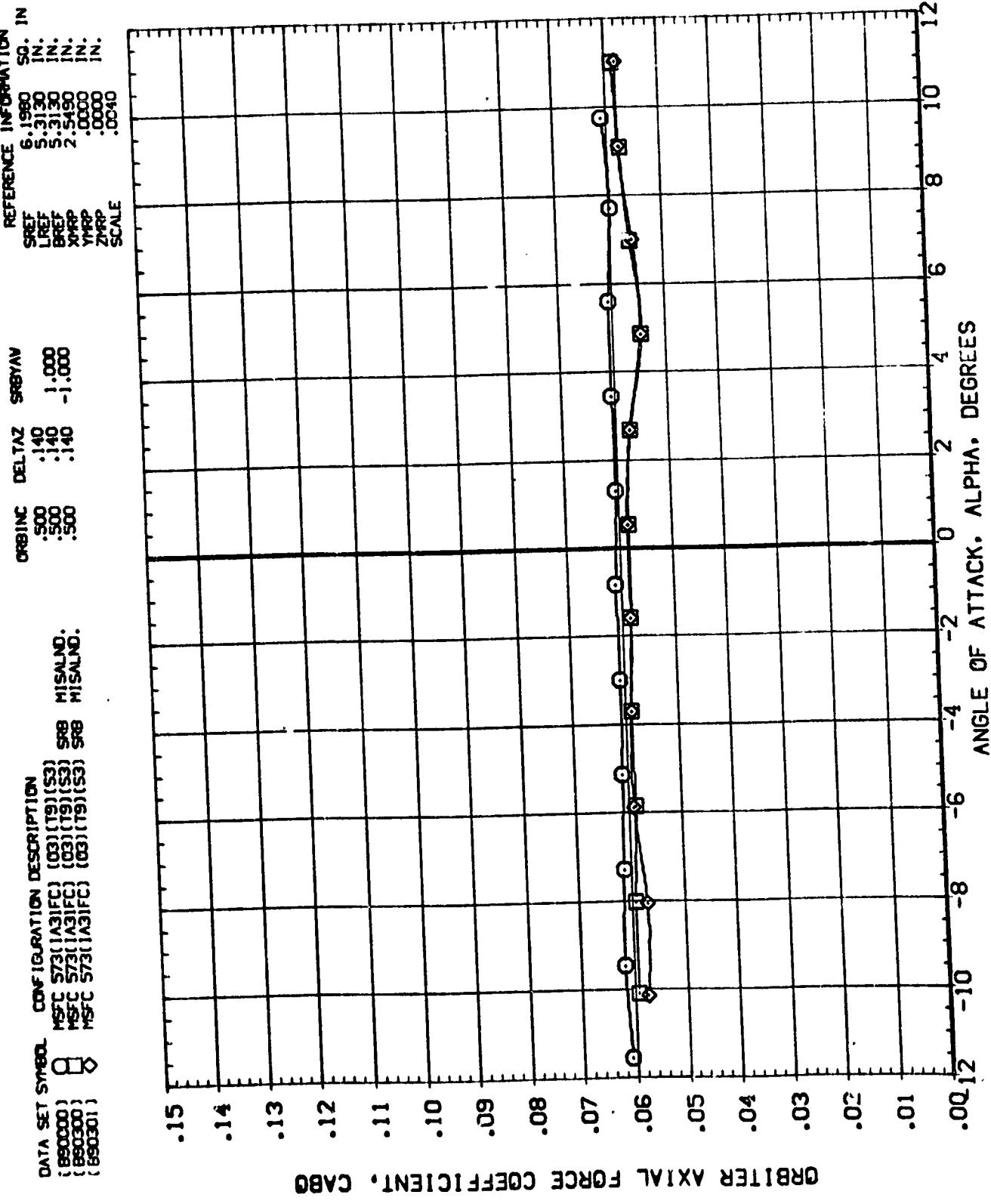
REFERENCE INFORMATION  
 DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) MSFC 5731(A3)FC (03)(19)(S3) SRB MISALND.  
 (B90300) MSFC 5731(A3)FC (03)(19)(S3) SRB MISALND.  
 (B90301) MSFC 5731(A3)FC (03)(19)(S3) SRB MISALND.



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
 $(A)MACH = .90$

DATA SET SYMBOL	CONFIGURATION DESCRIPTION
{BB0000}	MFC 5731 (A3IFC) [03] [19] [S3]
{BB0030}	MFC 5731 (A3IFC) [03] [19] [S3]
{BB0300}	MFC 5731 (A3IFC) [03] [19] [S3]

REFERENCE INFORMATION	6-1980	SO.	IN
SREF	5.3130	IN.	
LREF	5.3130	IN.	
BREF	2.5490	IN.	
XMRP	.0000	IN.	
YMRP	.0000	IN.	
ZMRP	.0040		
SCALE			



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

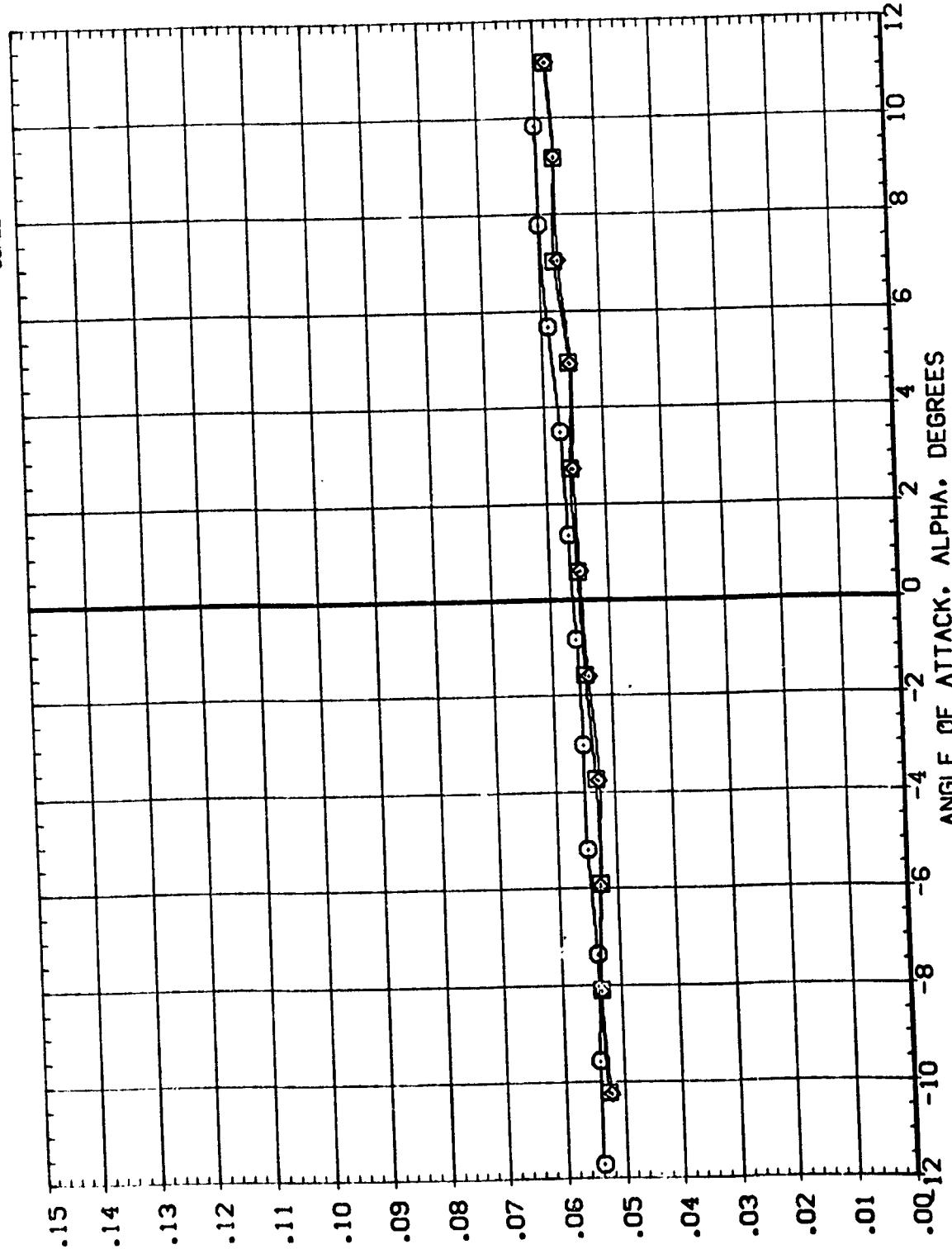
$$(\text{B})_{\text{MACH}} = 1.05$$

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 MSEC 5731(131FC) (03)(T9)(S3) MISALND.  
 MSEC 5731(131FC) (03)(T9)(S3) SRB MISALND.  
 MSEC 5731(131FC) (03)(T9)(S3) SRB MISALND.

(B90000) (B90300) (B90300) (B90301)

REFERENCE INFORMATION  
 SREF 6.1980 SD. IN  
 LREF 5.3130 IN  
 BREF 5.3130 IN  
 XRP 2.5490 IN  
 YRP .0000 IN  
 ZRP .0040 IN

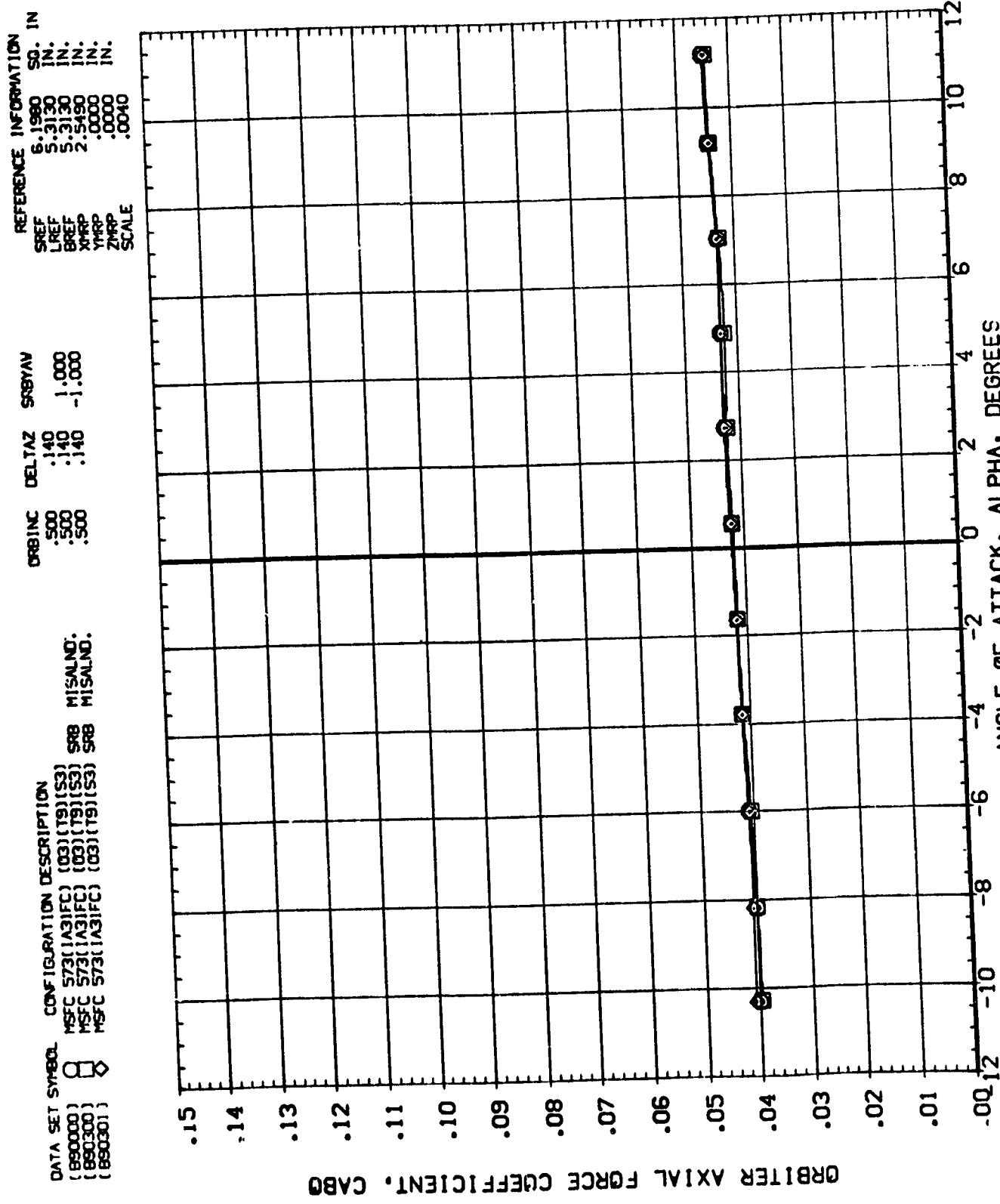
SCALE



ORBITER AXIAL FORCE COEFFICIENT, CABO

### EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(C)<sub>MACH</sub> = 1.25



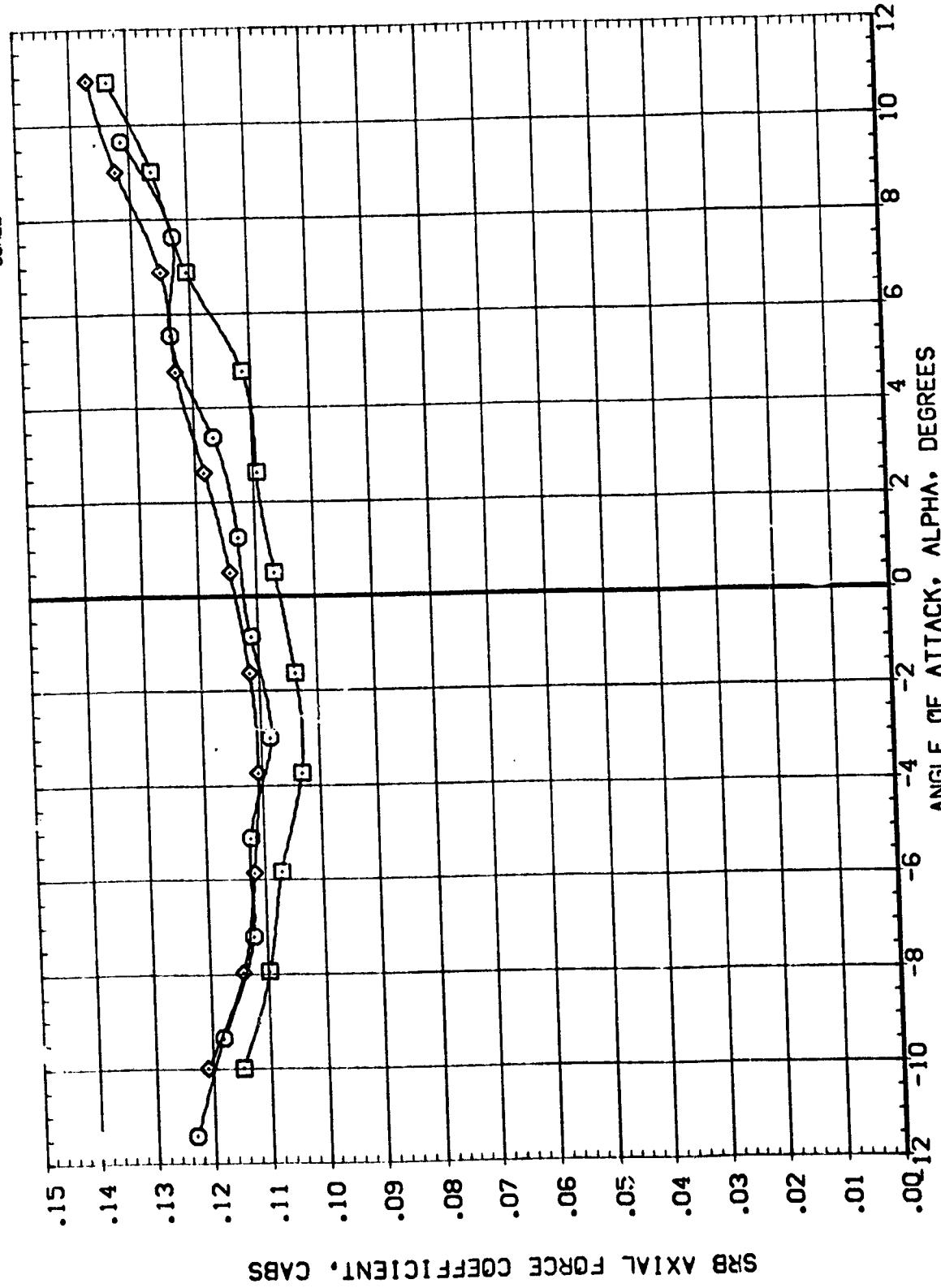
EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(D)MACH = 1.46

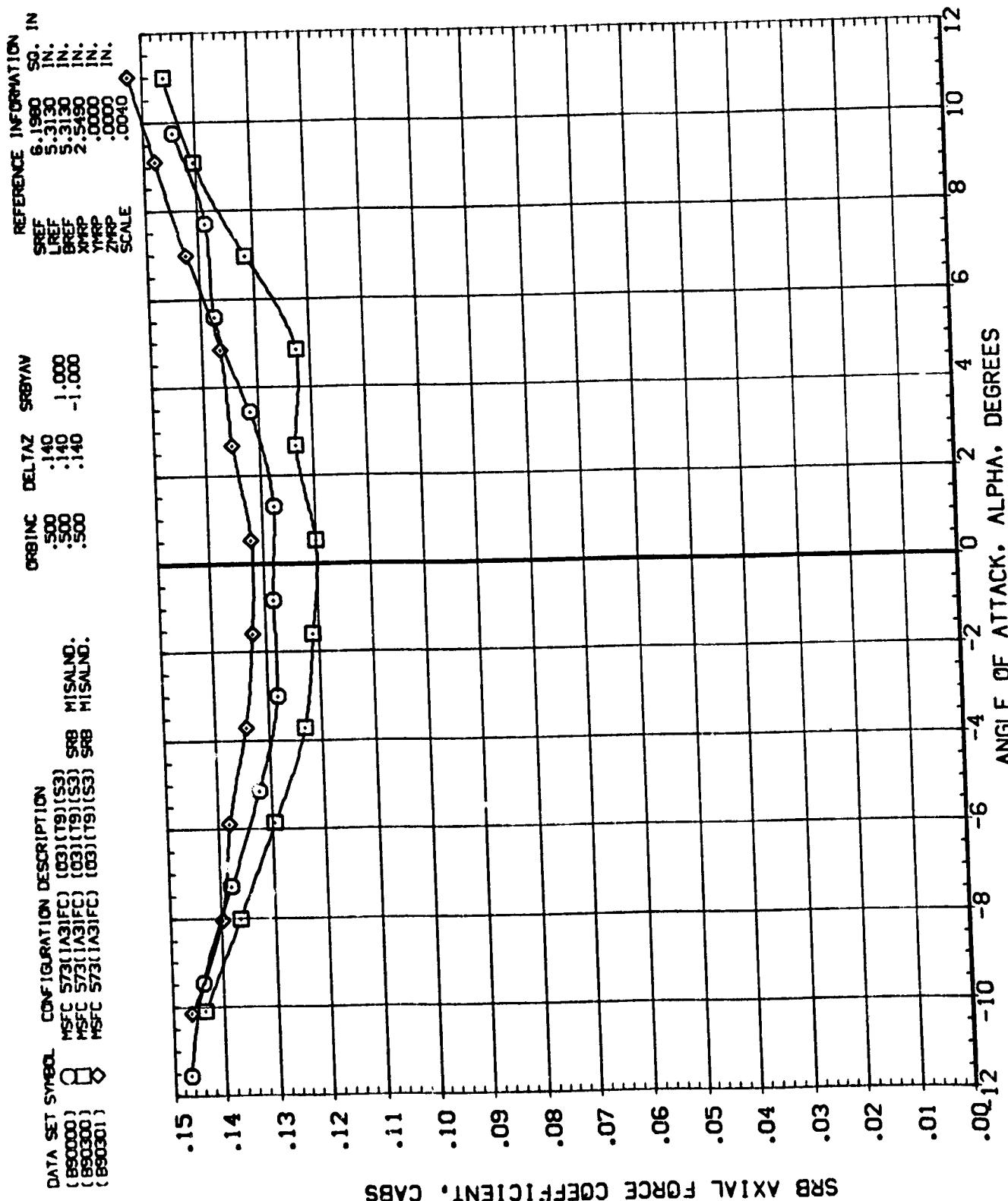
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 NSFC 573 (A3IFC) (S3) MISALND.  
 NSFC 573 (A3IFC) (S3) MISALND.  
 NSFC 573 (A3IFC) (S3) MISALND.

SRB INC. DELTAZ SRBYAW  
 .500 .140 .140  
 .500 .140 -1.000  
 .500 .140 .0000

REFERENCE INFORMATION  
 SREF 6.1980 SQ. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRF 2.5490 IN.  
 YMRF .0000 IN.  
 ZMRF .0040 IN.  
 SCALE



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
 (A) MACH = .90



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

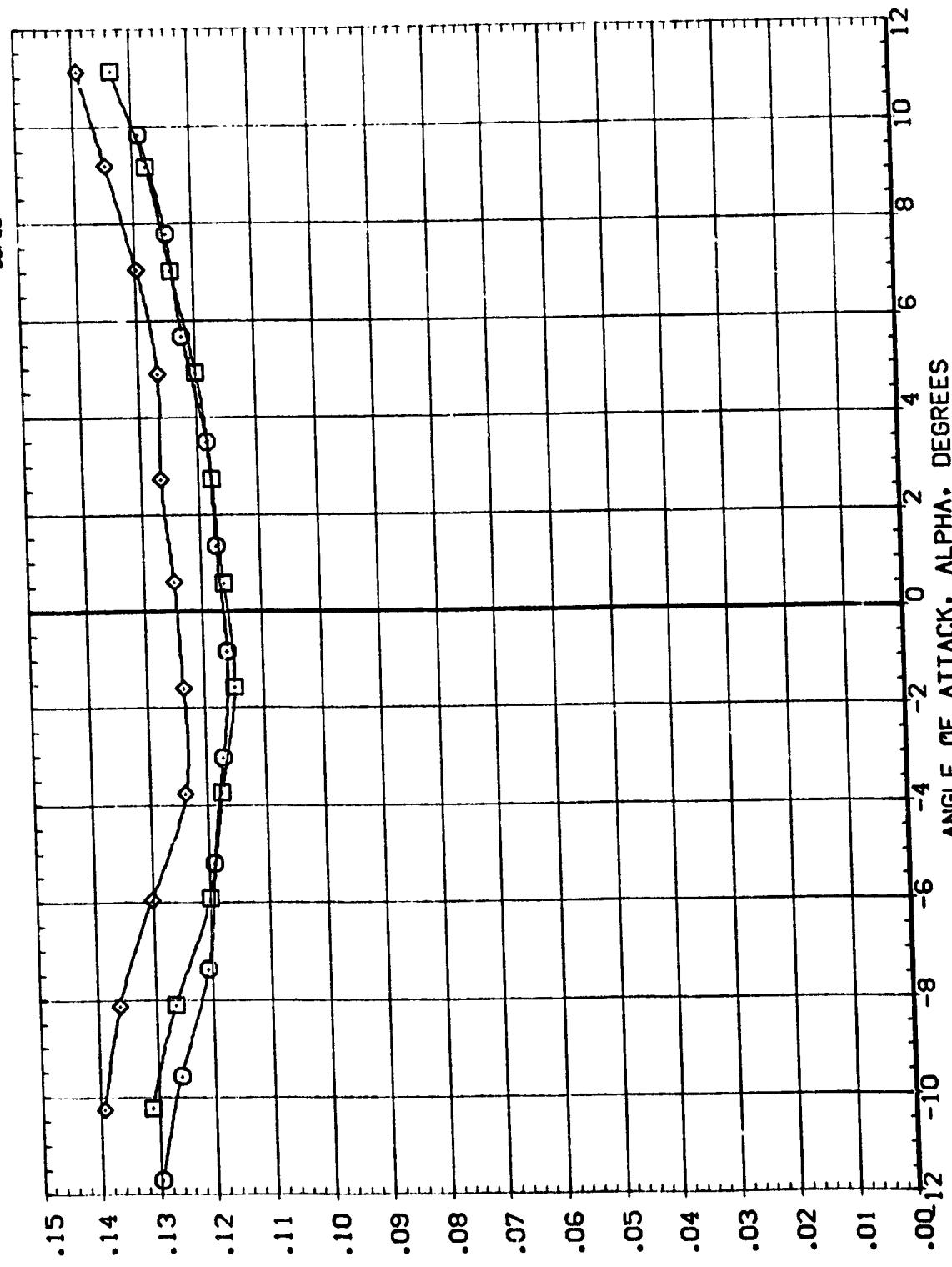
(B)MACH = 1.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(B60000)	NSFC	5731(A3)FC	(03)(T9)(S2)	SRB MISALNO.
(B903C0)	NSFC	5731(A3)FC	(03)(T9)(S3)	SRB MISALNO.
(B903C1)	NSFC	5731(A3)FC	(03)(T9)(S3)	SRB MISALNO.

REFERENCE INFORMATION  
SC. IN

SREF	6.1980
LREF	.5.3130
BREF	.5.3130
XMRP	2.5450
YMRP	.0000
ZMRP	.0000
SCALE	.0040



SRB AXIAL FORCE COEFFICIENT, CABs

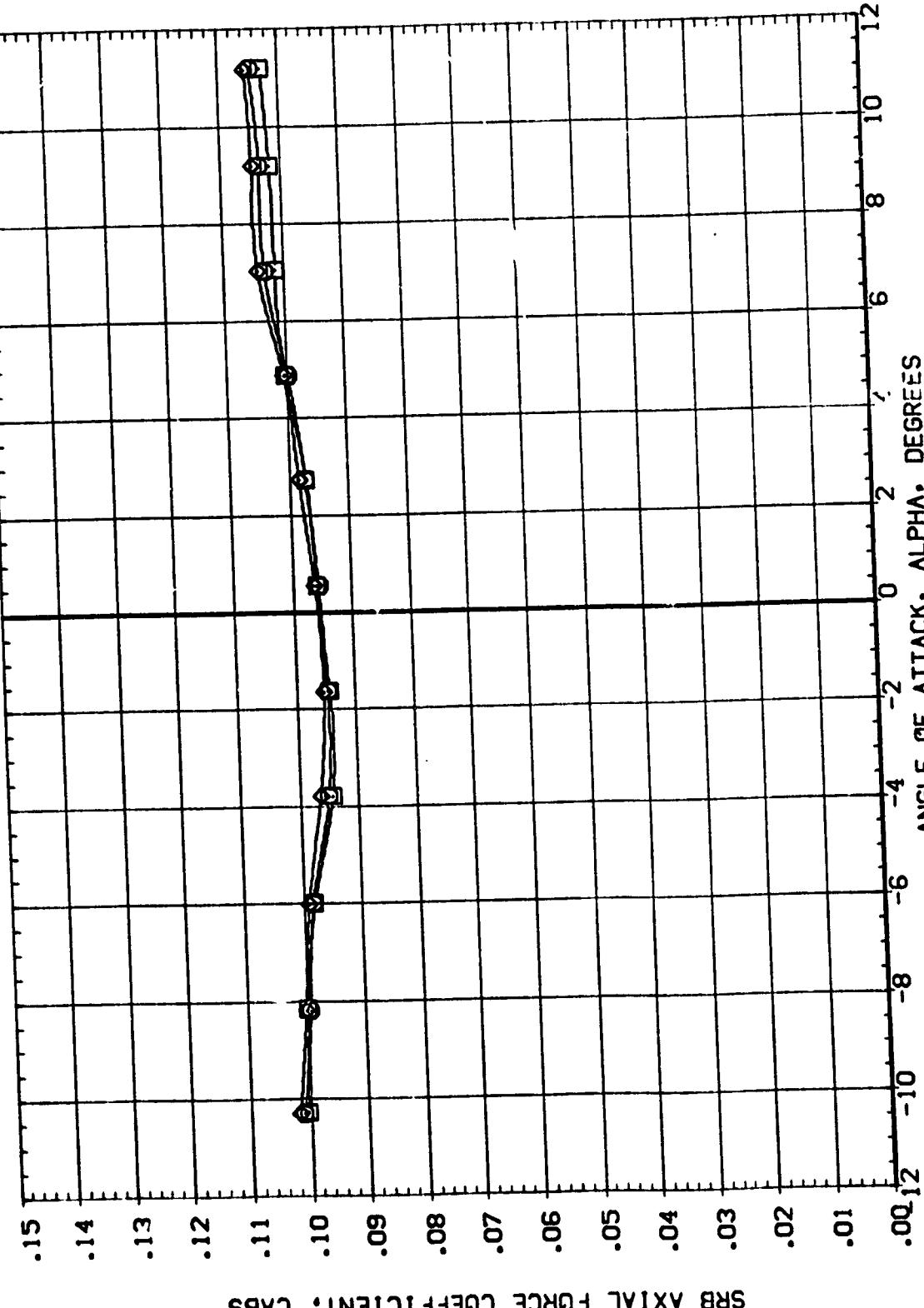
### EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(C)<sub>MACH</sub> = 1.25

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90300) S73(1A3)FC (03)(19)(S3) SRB MISALNO.  
 (B90300) MSFC S73(1A3)FC (03)(19)(S3) SRB MISALNO.  
 (B90301) MSFC S73(1A3)FC (03)(19)(S3) SRB MISALNO.

REFERENCE INFORMATION

SREF	6.1980	SO. IN
LREF	5.3130	IN.
BREF	5.3130	IN.
XMRP	2.5490	IN.
YMRP	.0000	IN.
ZMRP	.0040	IN.
SCALE		



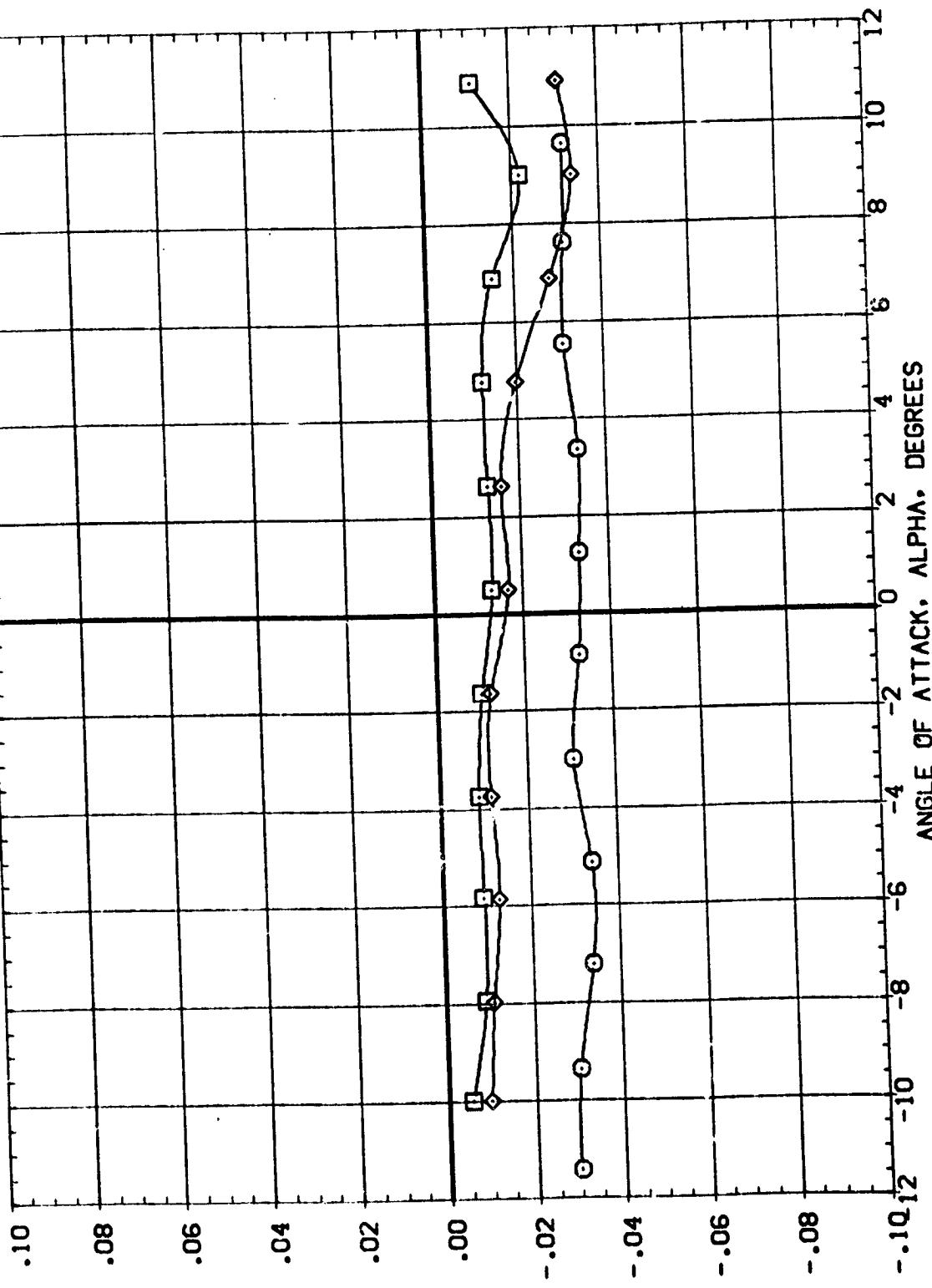
### EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

$$(\text{D})\text{MACH} = 1.46$$

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (890000) MSFC S73([A3]FC) [03][T9][S3] SRB MISALND.  
 (890300) MSFC S73([A3]FC) [03][13][S3] SRB MISALND.  
 (89C301) MSFC S73([A3]FC) [03][T8][S3] SRB MISALND.

REFERENCE INFORMATION  
 SRBF 6.1980 SQ. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE

ORBINC DELTAZ SRBYAW  
 .500 .140 .1000  
 .500 .140 -1.000  
 .500

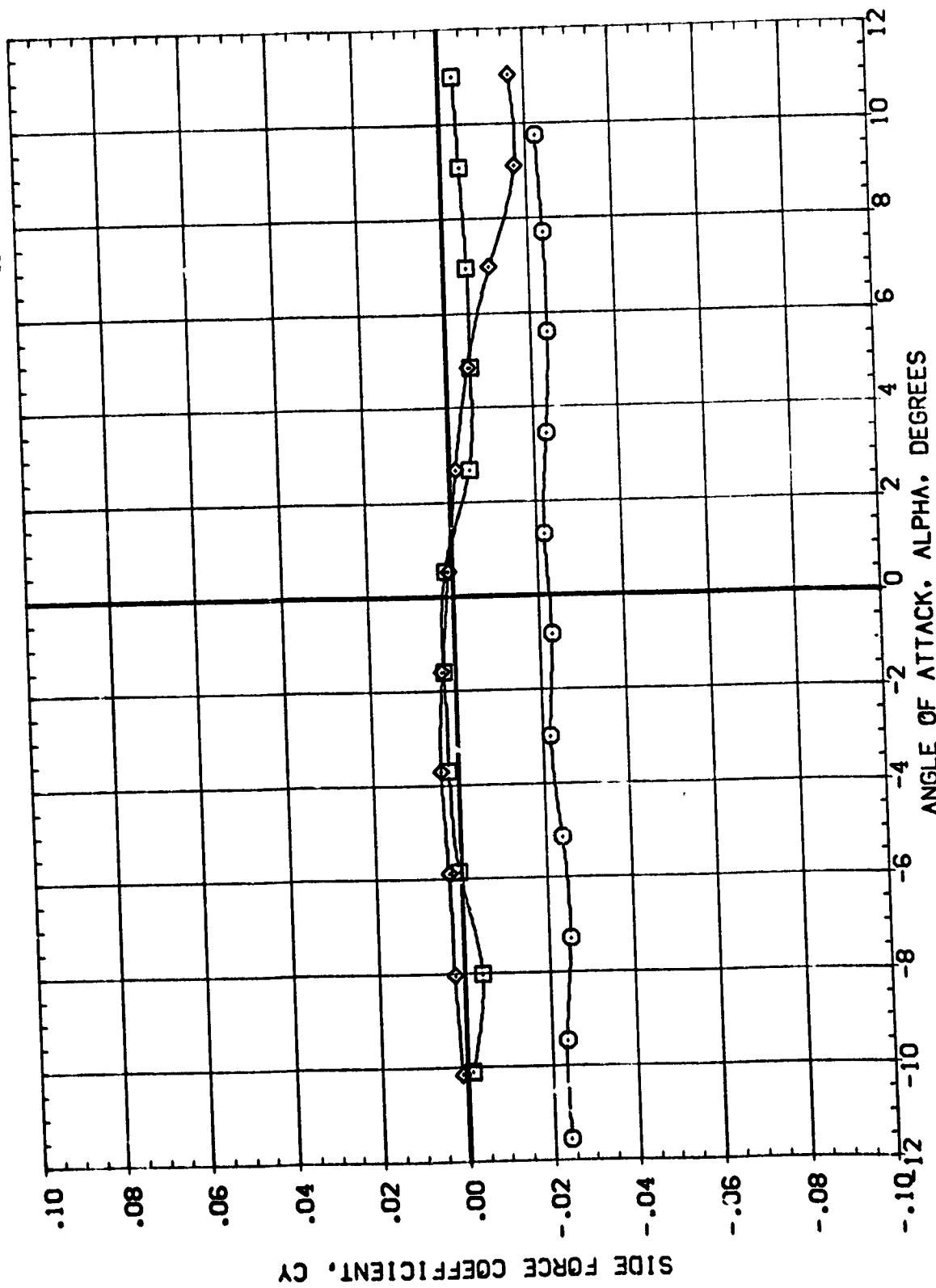


SIDE FORCE COEFFICIENT, CX

EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
 (MACH = .90)

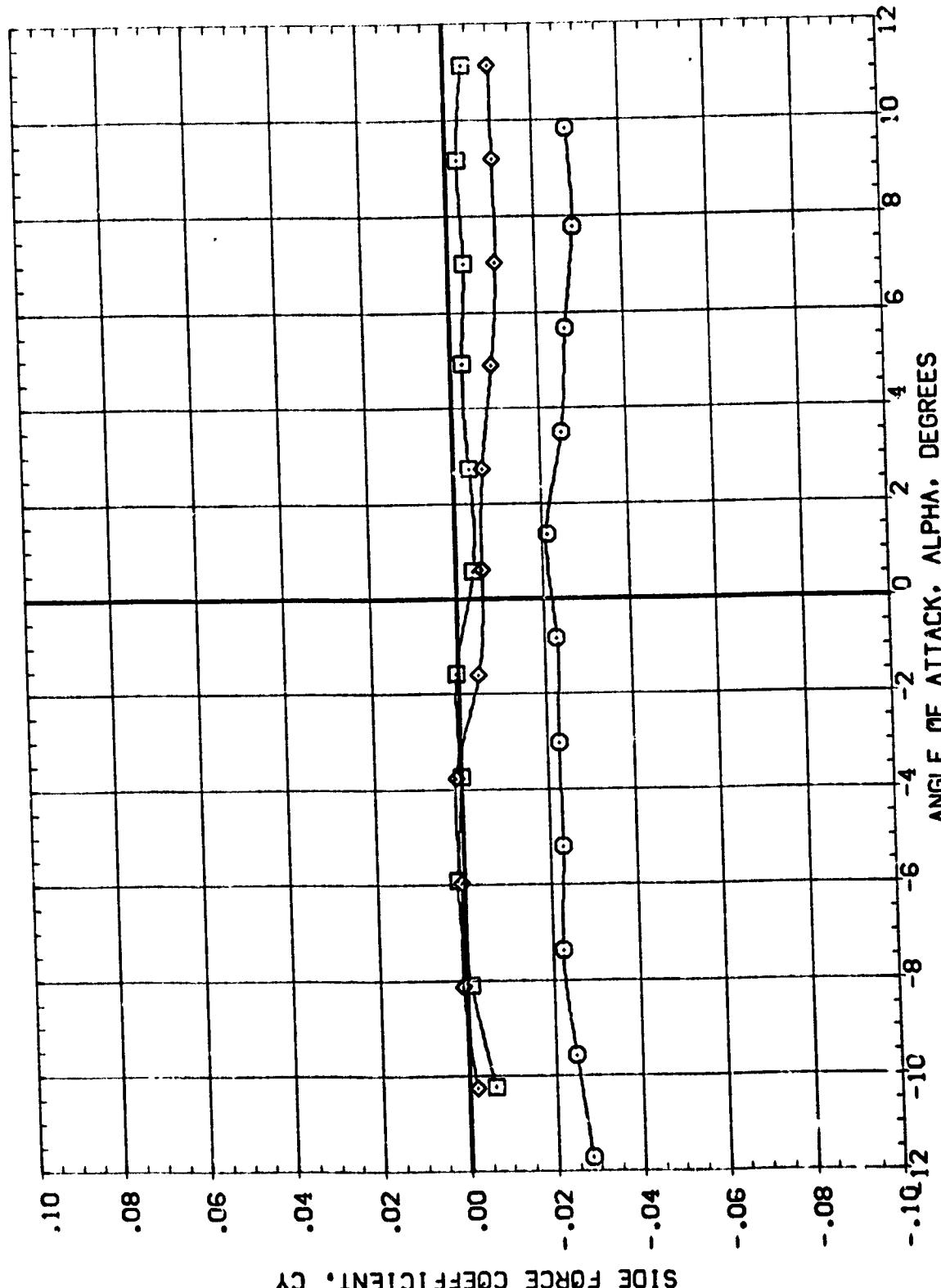
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 6900000 MSFC 5731(A3)FC (03)(T9)(S3)  
 6903001 MSFC 5731(A3)FC (03)(T9)(S3)  
 6903011 MSFC 5731(A3)FC (03)(T9)(S3)

REFERENCE INFORMATION  
 SREF 6.1980 S3. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
(B)MACH = 1.05

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	ORB INC	DELTAZ	SIDE YAW	REFERENCE INFORMATION
{890000}	MSFC S731A31FC [03][T9][S3]	.500	.140	.1000	6.1980 SD. IN.
{890300}	MSFC S731A31FC [03][T9][S3]	.500	.140	-1.000	5.3130 IN.
{890301}	MSFC S731A31FC [03][T9][S3]	.500	.140	.0000	5.3130 IN.
					2.5480 IN.
					.0000 IN.
					.0040 IN.

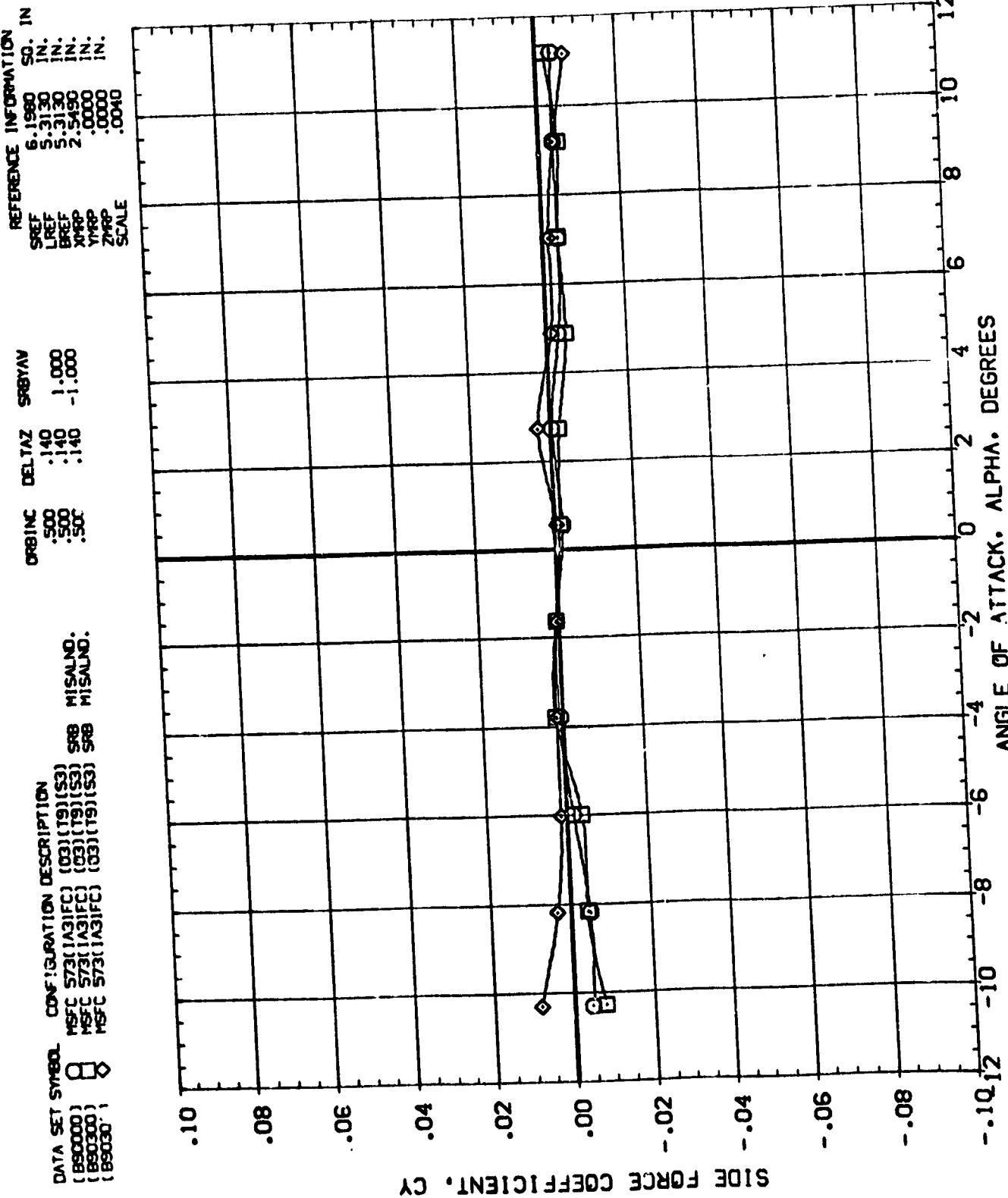


EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(C)<sub>MACH</sub> = 1.25

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REFERENCE INFORMATION  
 DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B890000) MSFC S731(A3)FC (03)(19)(S3) SRB MISALND.  
 (B890300) MSFC S731(A3)FC (03)(19)(S3) SRB MISALND.  
 (B89030) MSFC S731(A3)FC (03)(19)(S3) SRB MISALND.



### EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

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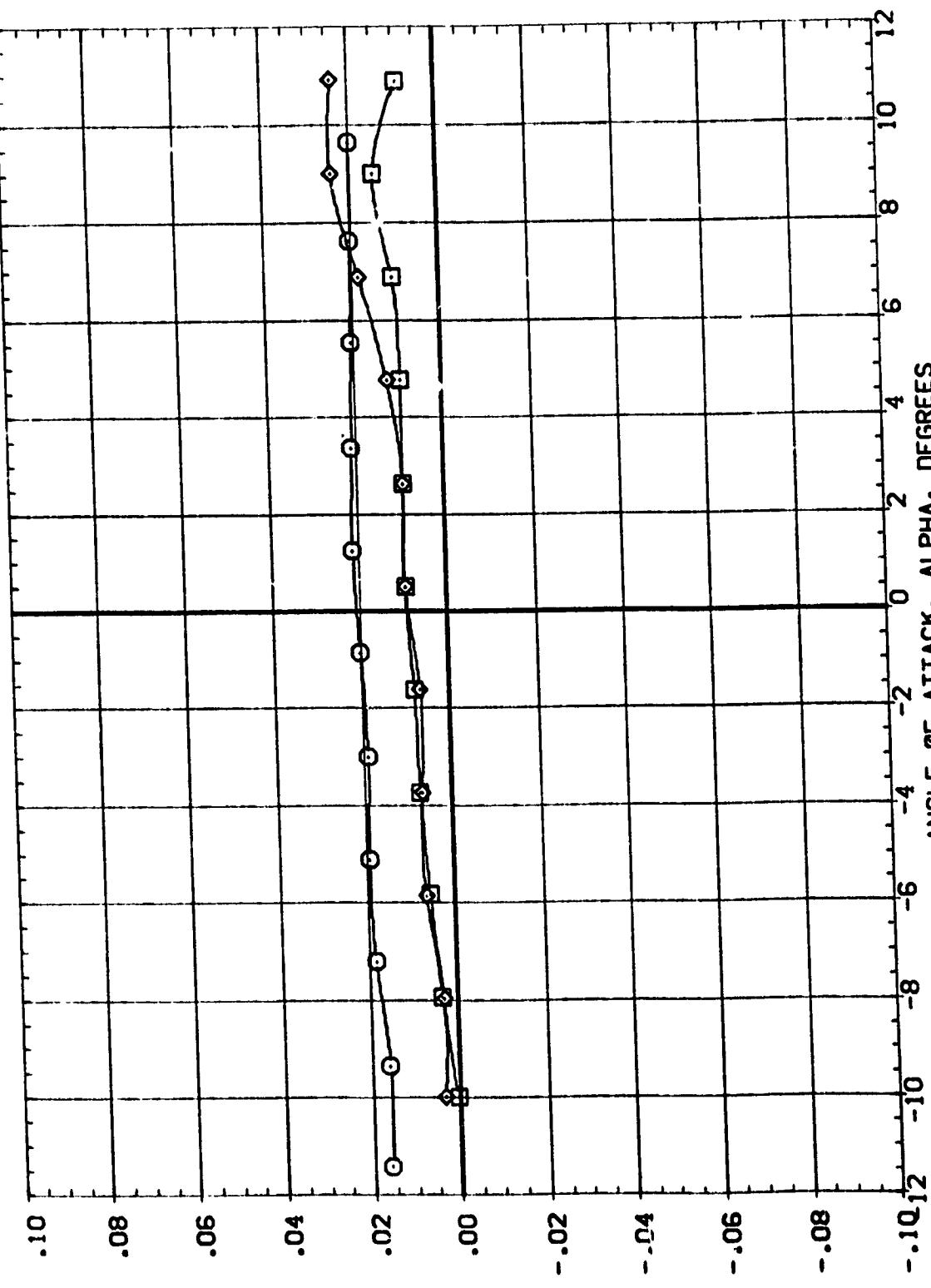
(CD)MACH = 1.46

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) MSFC 573(A31FC) (03)(19)(53) SRE MISALND.  
 (B90300) MSFC 573(A31FC) (03)(19)(53) SRE MISALND.  
 (B90301) MSFC 573(A31FC) (03)(19)(53) SRE MISALND.

DATA INC DELTAZ SRBYAV  
 .500 .140 1.000  
 .500 .140 -1.000  
 .500 .140 .000  
 .500 .140 -.0040

REFERENCE INFORMATION IN  
 SREF 6.1980 SD.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.

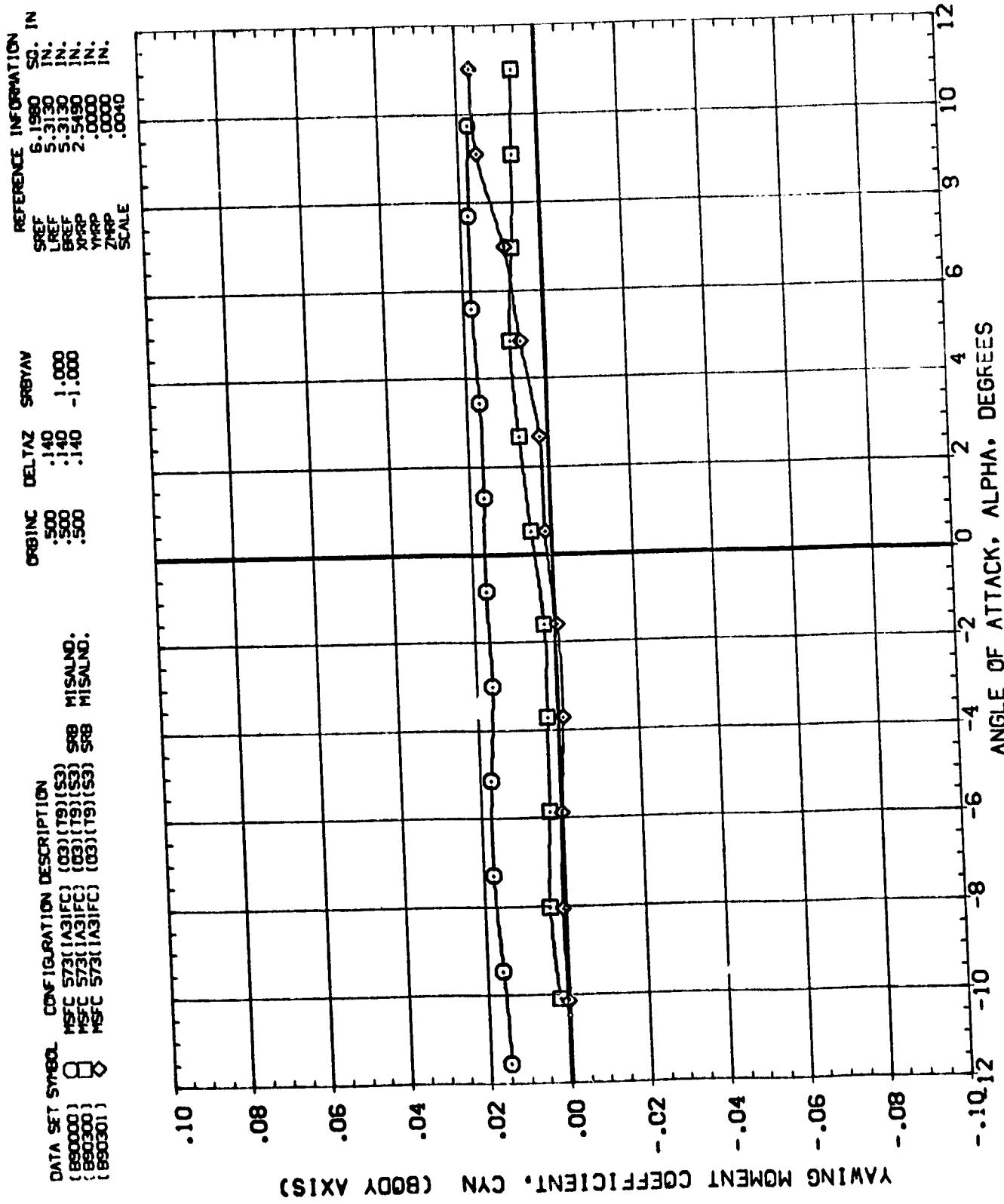
SCALE



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

( $\lambda$ )<sub>MACH</sub> = .90

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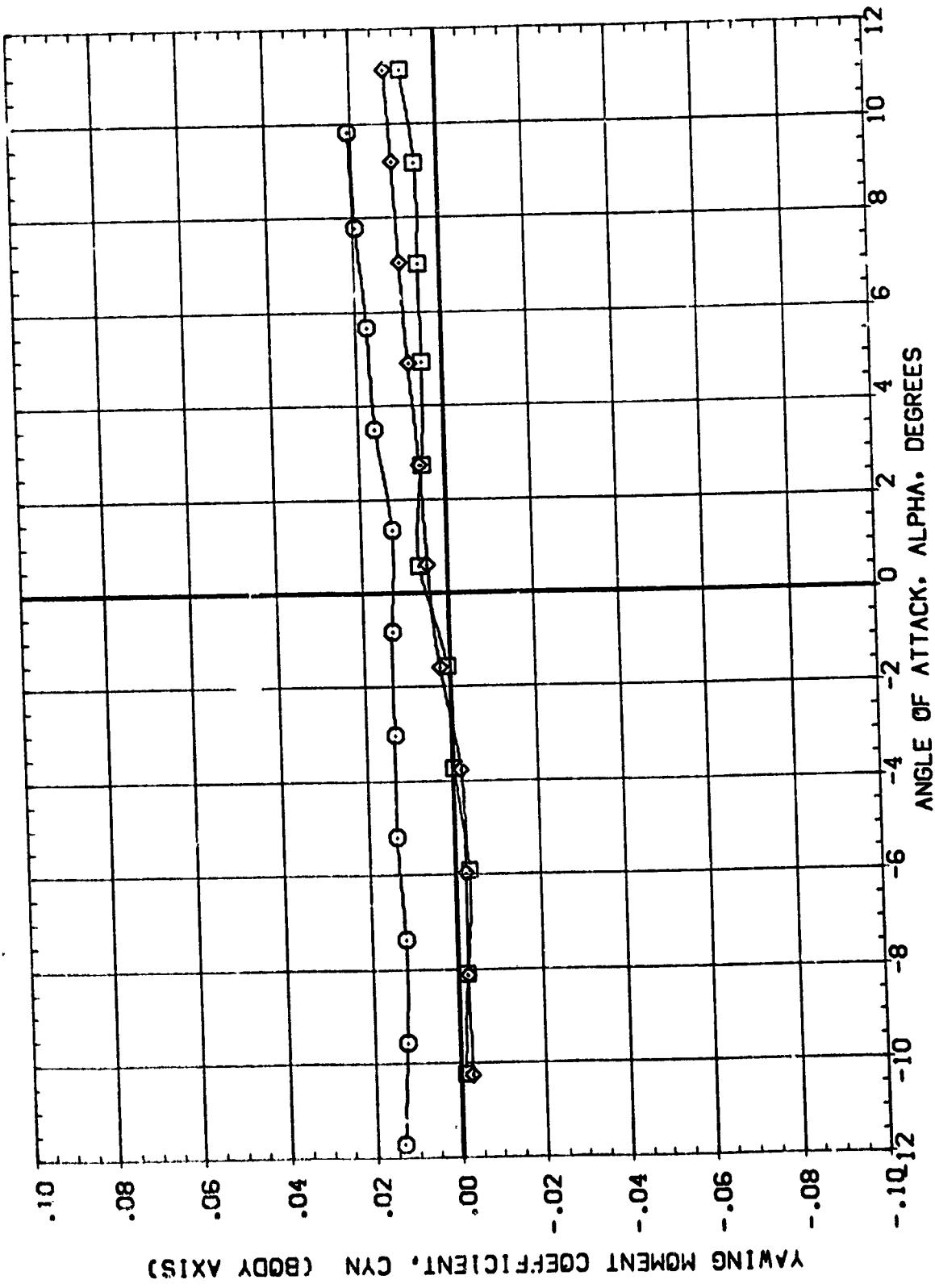


EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B60000) MSFC S73(1A3)FC (03)(19)(S3) SRB MISALD.  
 (B60300) MSFC S73(1A3)FC (03)(19)(S3) SRB MISALD.  
 (B60301) MSFC S73(1A3)FC (03)(19)(S3) SRB MISALD.

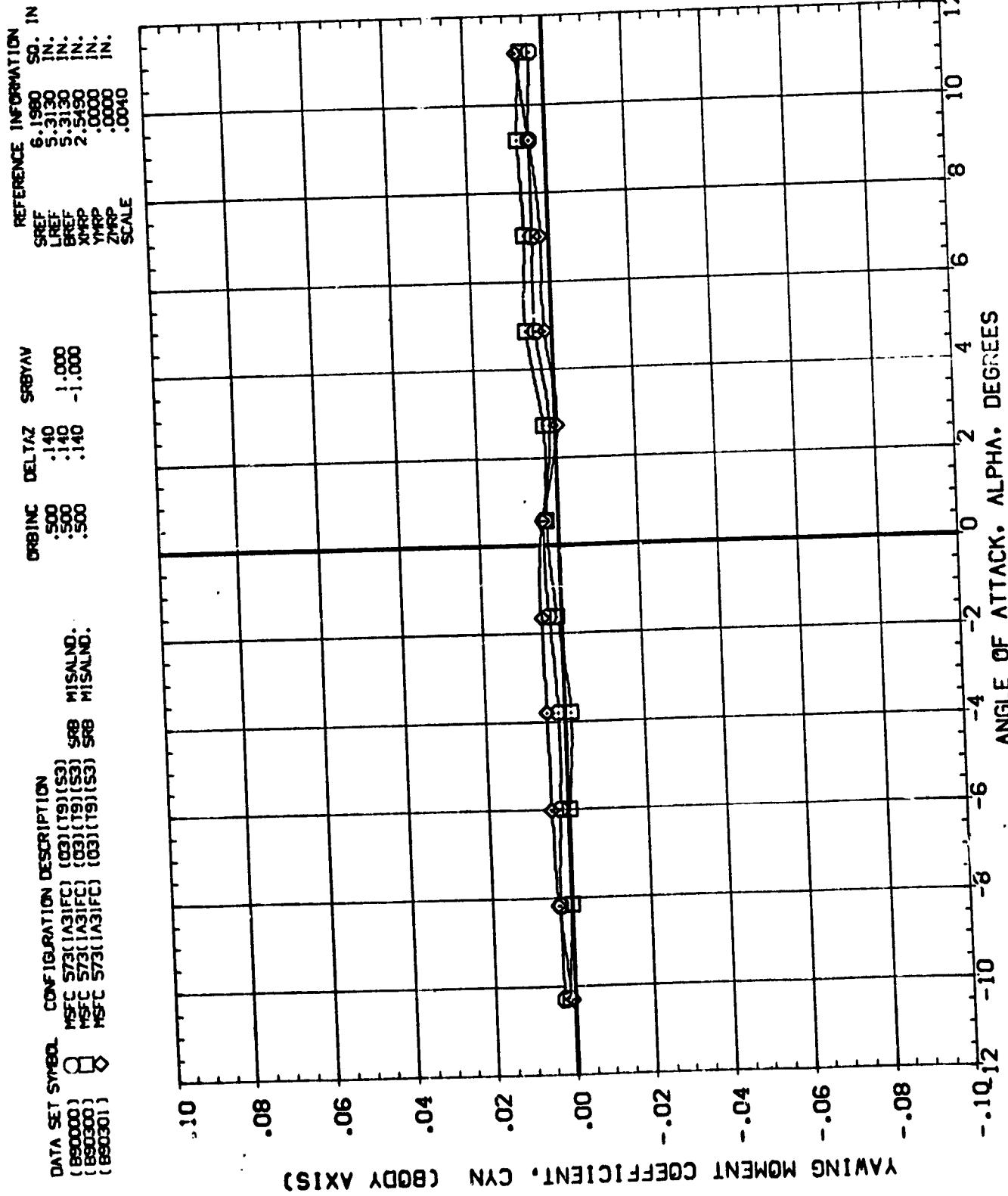
REFERENCE INFORMATION  
 SRREF 6.1980 SQ. IN.  
 LRREF 5.3130 IN.  
 BRREF 5.3130 IN.  
 XMRP 2.5190 IN.  
 YMRP .0000 IN.  
 ZMRP .0000 IN.  
 SCALE .00 0



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

$$(\text{C})\text{MACH} = 1.25$$

DATA SET SWB. CONFIGURATION DESCRIPTION  
 MSFC S73[1A3]FC [03][19][S3] MISALND.  
 MSFC S73[1A3]FC [03][19][S3] SRB MISALND.  
 MSFC S73[1A3]FC [03][19][S3] SRB MISALND.  
 (890000) (890300) (890301)



EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
 (D)MACH = 1.46

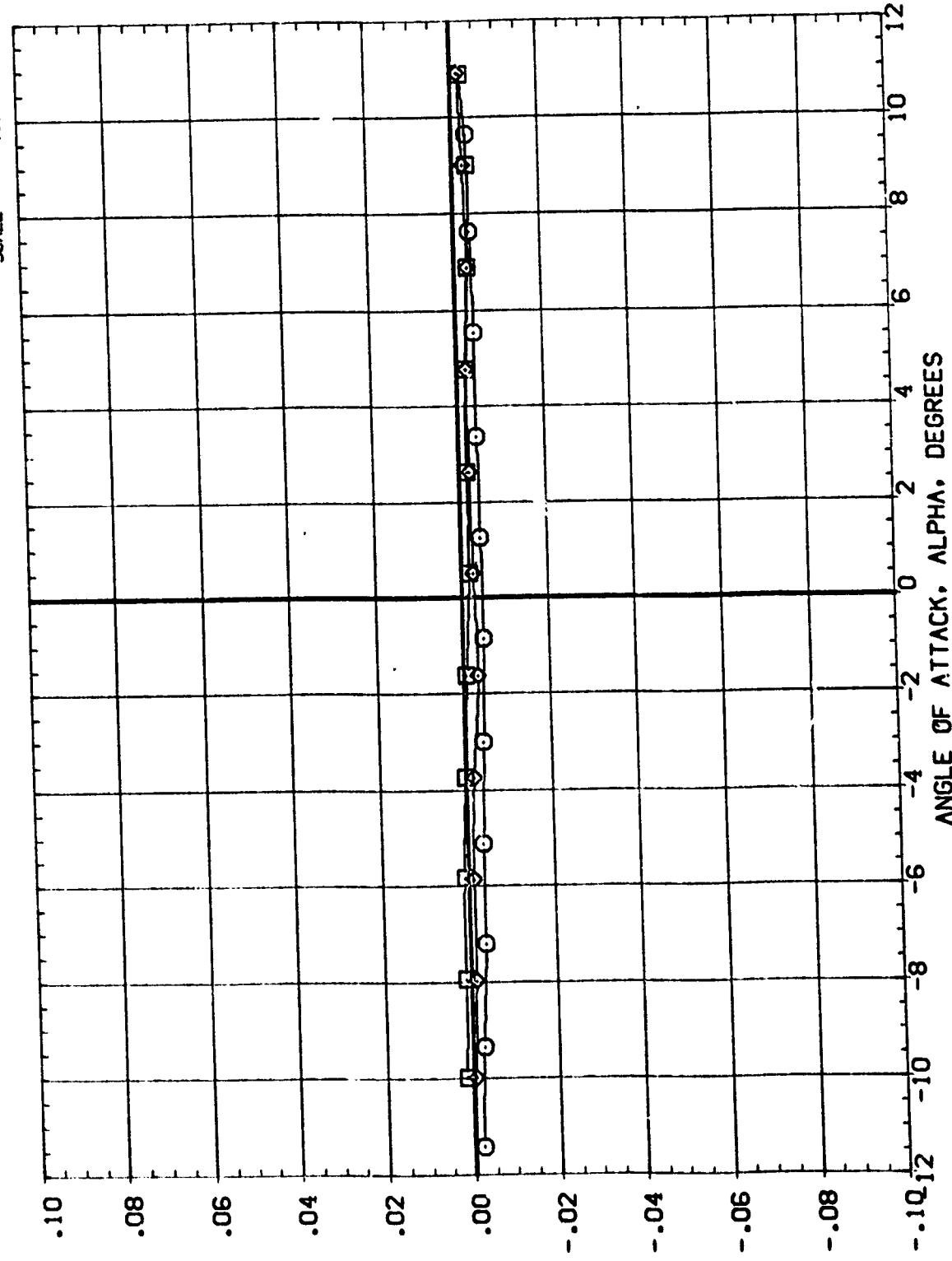
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DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (890000C) MSFC S73([A3]FC) [(03)(T9)(S3)] MISALND.  
 (89030CC) MSFC S73([A3]FC) [(03)(T9)(S3)] SRB MISALND.  
 (890301) MSFC S73([A3]FC) [(03)(T9)(S3)] SRB

REFERENCE INFORMATION  
 ORBINC DELTAZ SRBYAW  
 :500 :140 :1.000  
 :500 :140 :1.000  
 :500 :140 -1.000  
 :500 :140 :0.000  
 :500 :140 :0.000  
 :500 :140 :0.040

SREF LREF BREF XMP YMP ZMP  
 6.1980 5.3130 5.3130 2.5490 .0000 .0000  
 IN. IN. IN. IN. IN. IN.

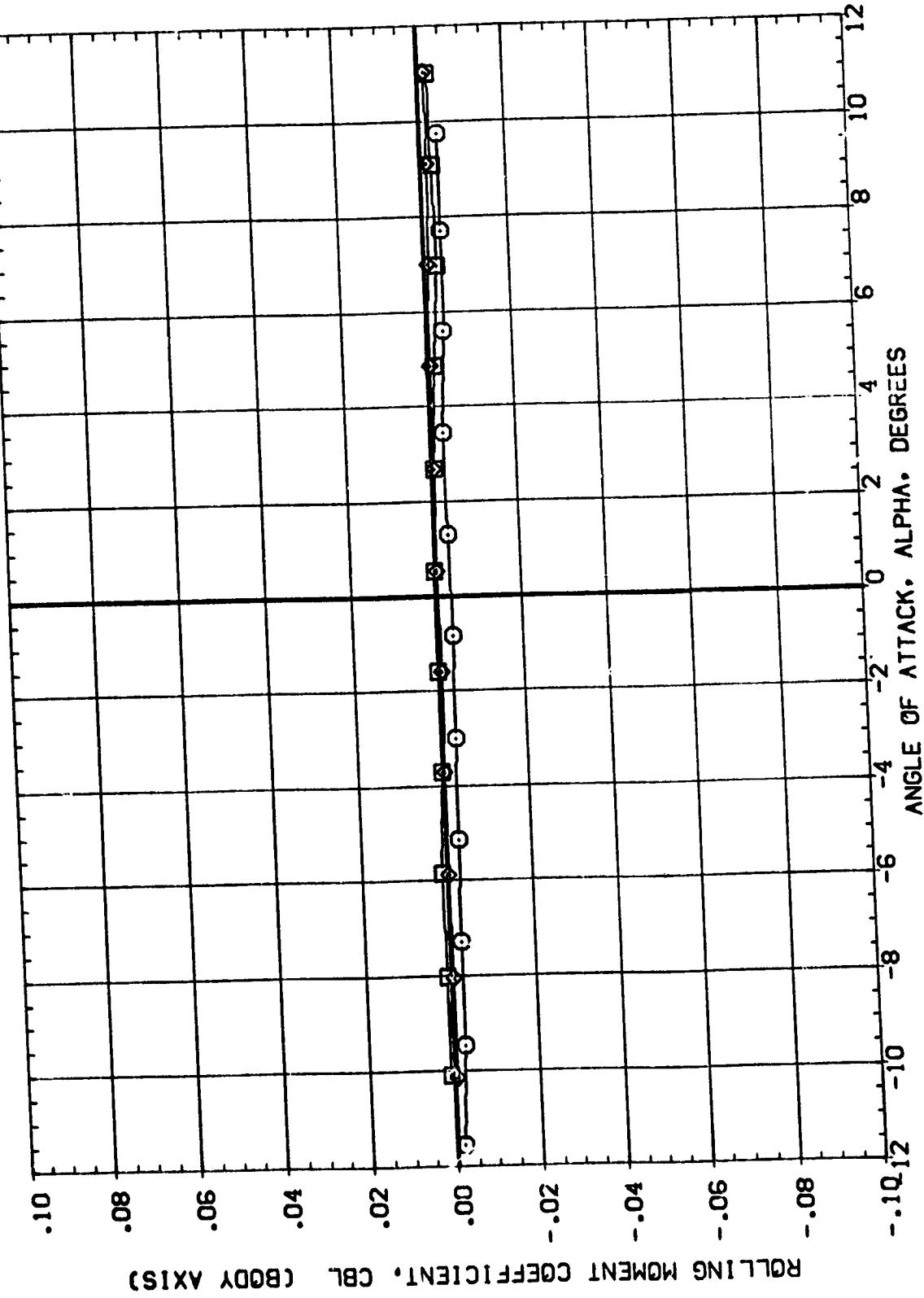
SCALE



ROLLING MOMENT COEFFICIENT, CBL (BODY AXIS)

EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS  
 (A)MACH = .90

DATA SET SYMBOL		CONFIGURATION DESCRIPTION		REFERENCE INFORMATION	
{B90000}		{B90300}		SREF	6.1980 SO. IN.
{B90301}		{B90302}		LREF	5.3130 IN.
				BREF	5.3130 IN.
				XMRP	2.5490 IN.
				YMRP	.0000 IN.
				ZMRP	.0040 IN.
				SCALE	

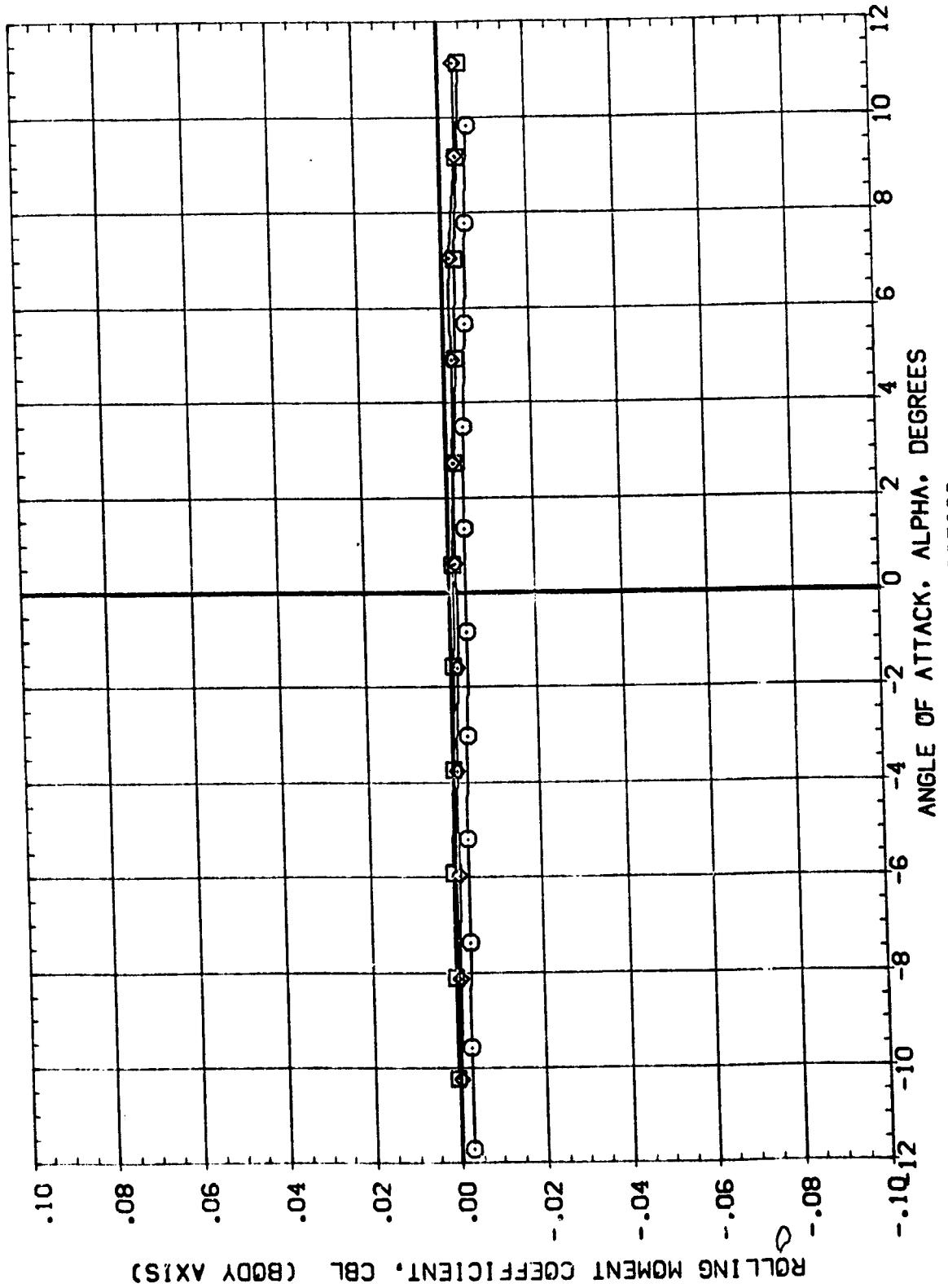


EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

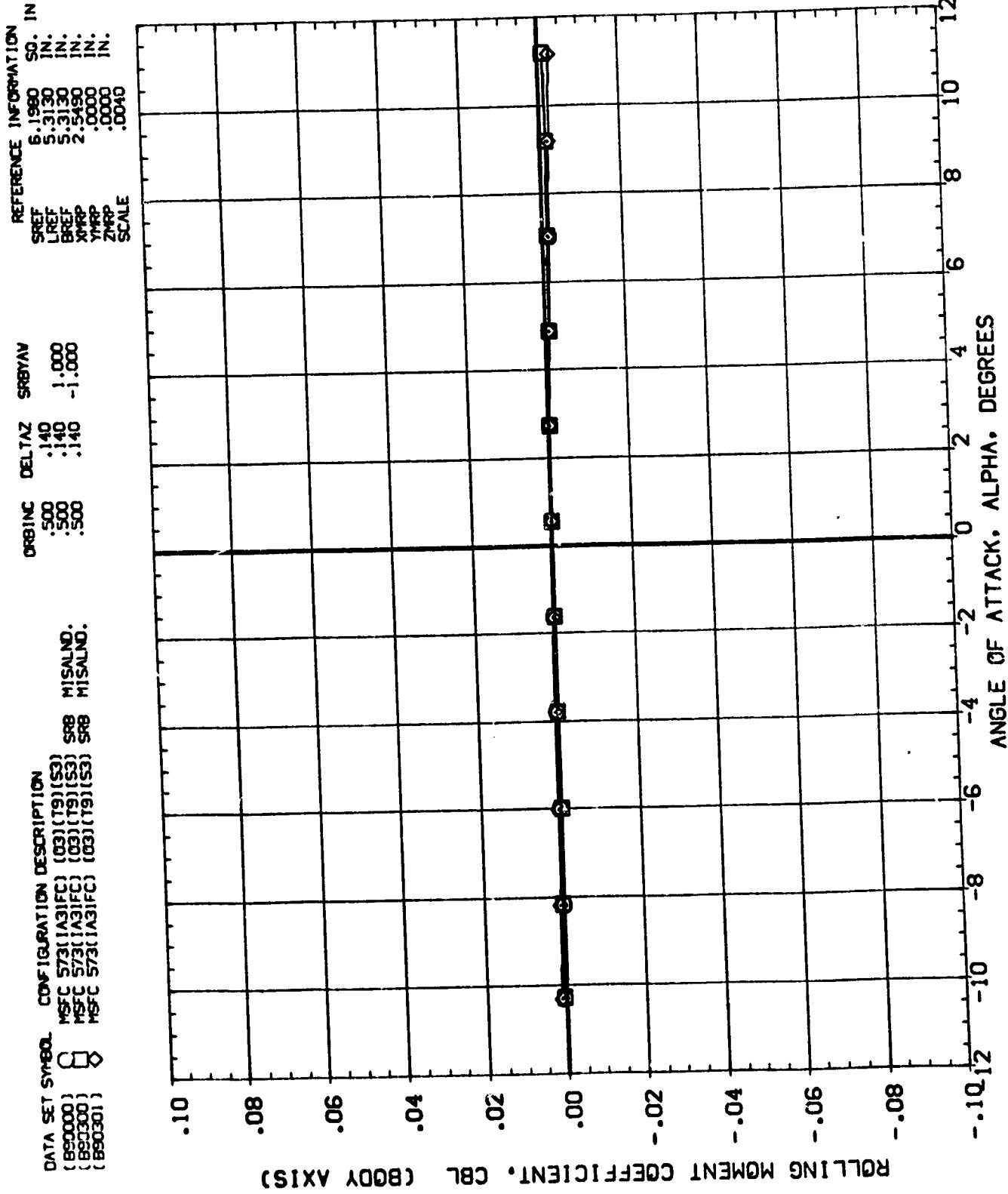
DATA SET SYMBOL COEF. FRACTION DESCRIPTION  
 { 890000 } MSEC 573(1A3)FC (03)(19)(S3) SRB MISALNO.  
 { 890300 } MSEC 573(1A3)FC (03)(19)(S3) SRB MISALNO.  
 { 890311 } MSEC 573(1A3)FC (03)(19)(S3) SRB MISALNO.

REFERENCE INFORMATION  
 SREF 6.1980 SO. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE



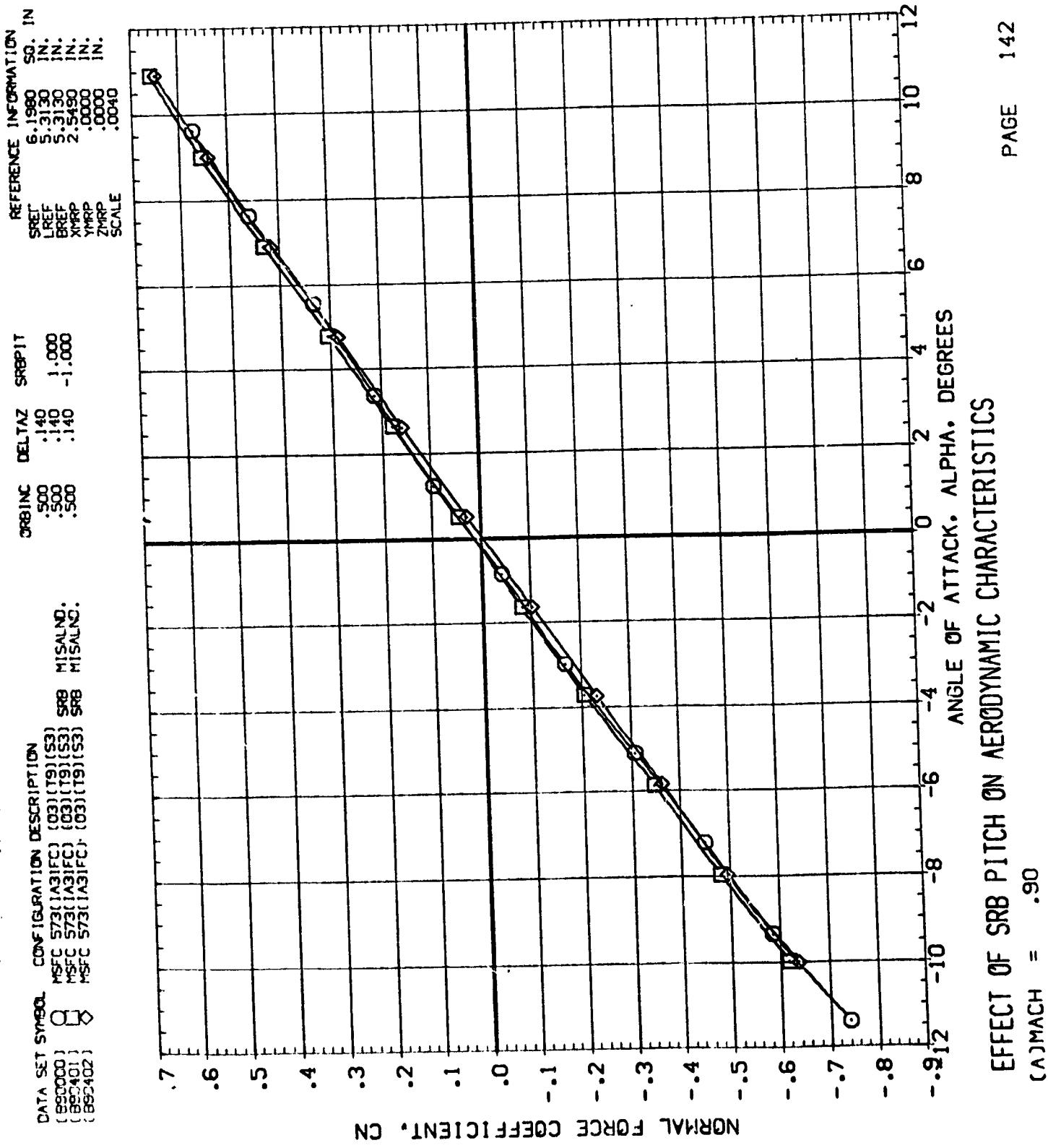
### EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

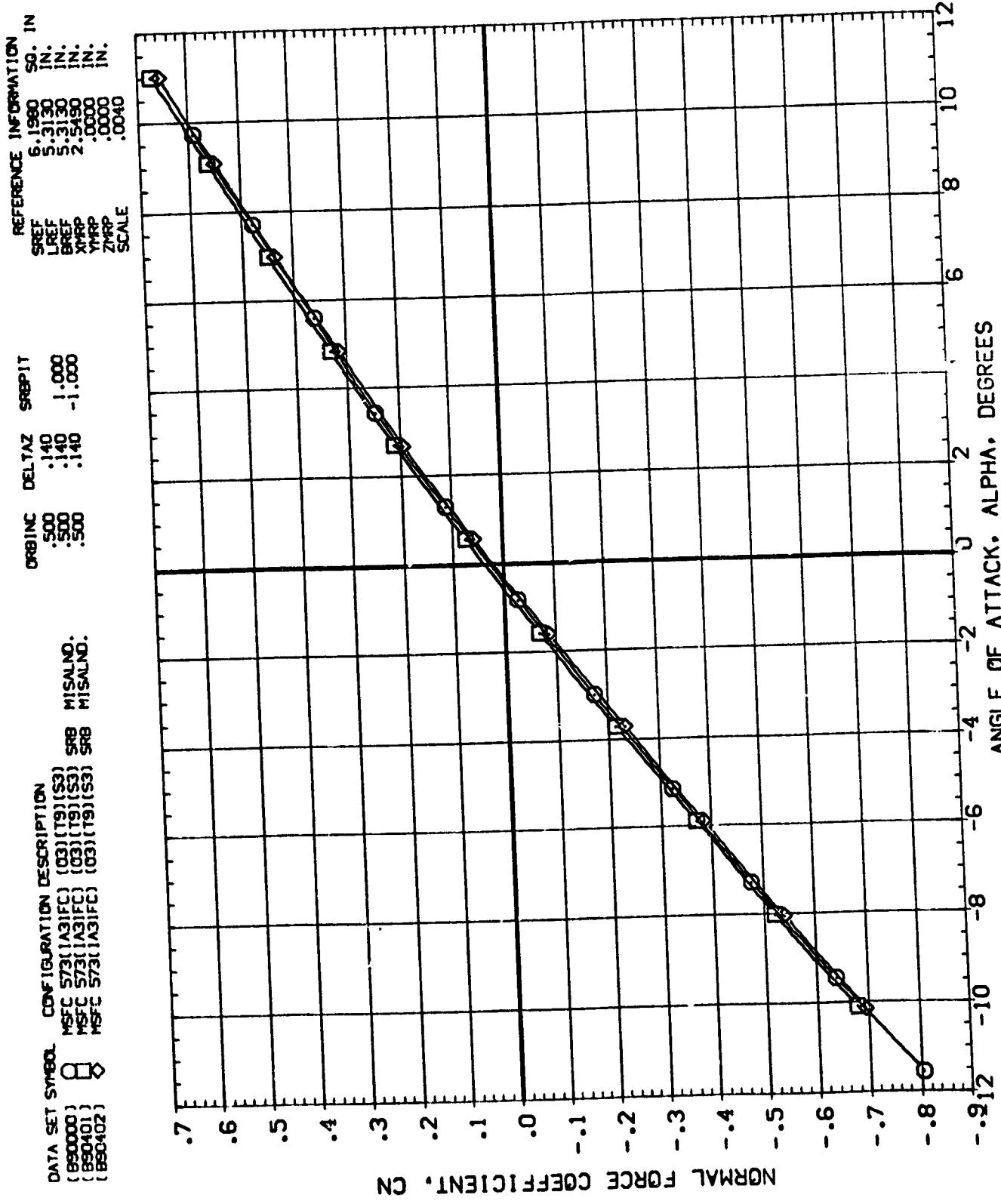


EFFECT OF SRB YAW ON AERODYNAMIC CHARACTERISTICS

(D)MACH = 1.46

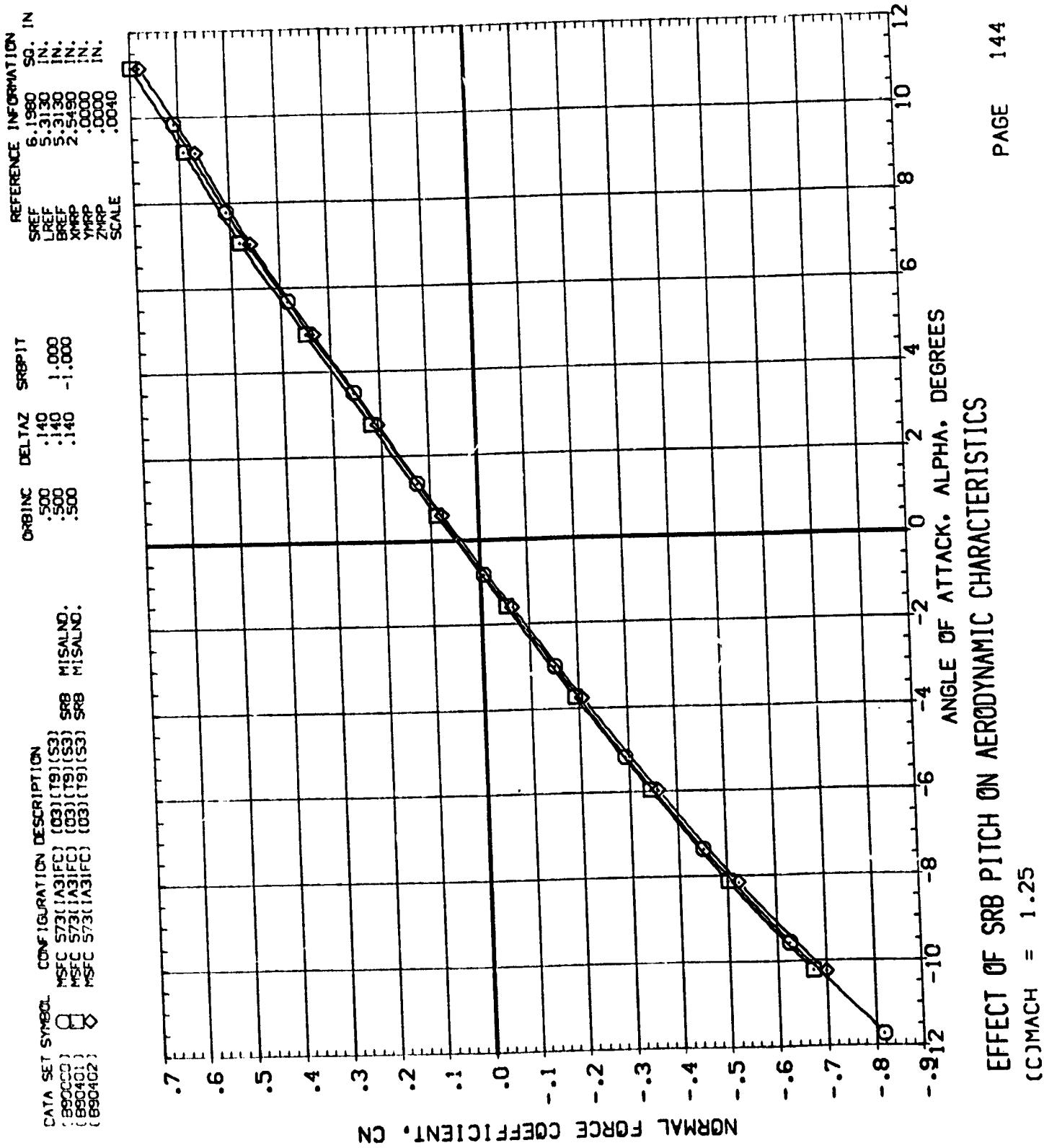


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EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

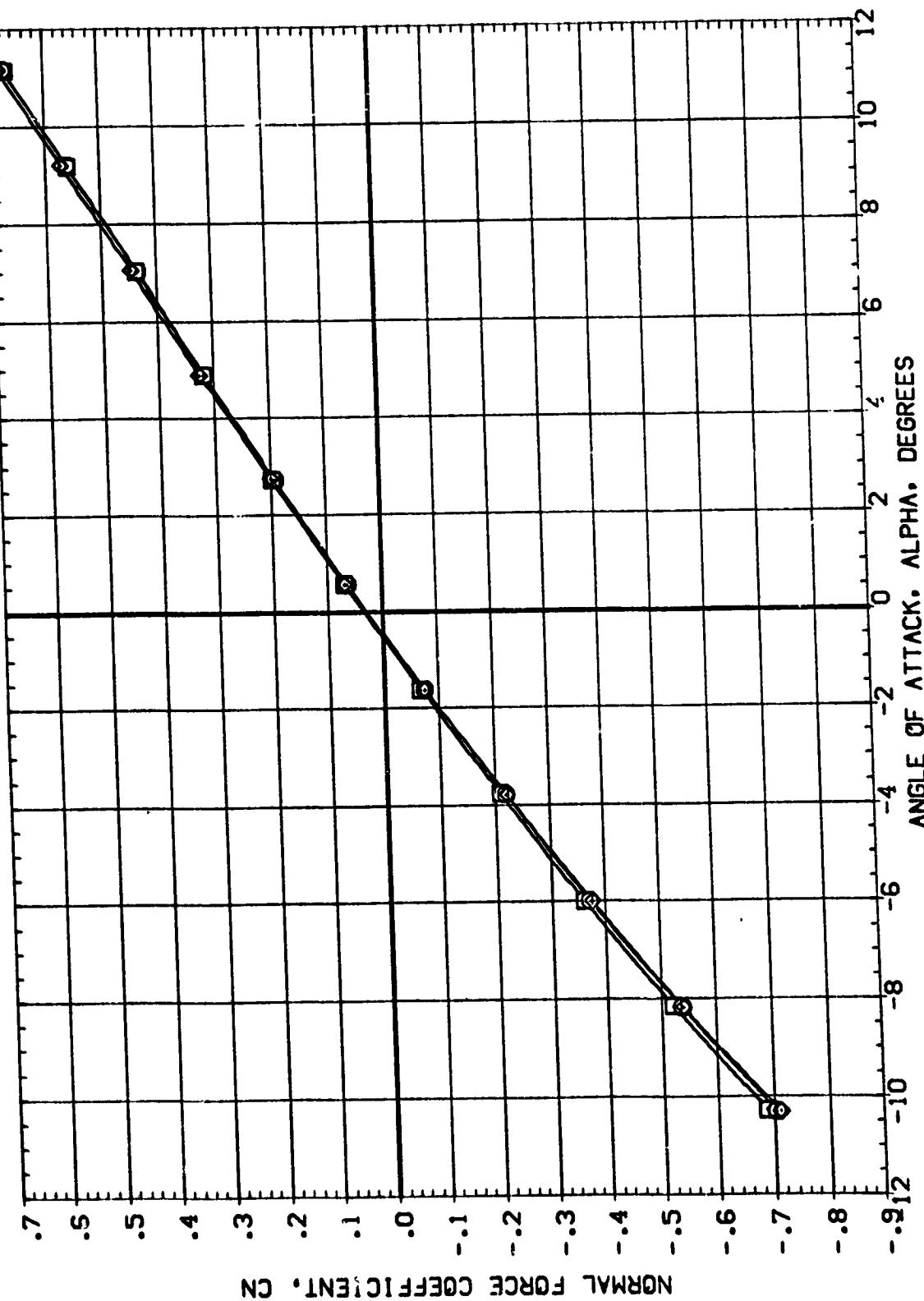
$(B)MACH = 1.05$



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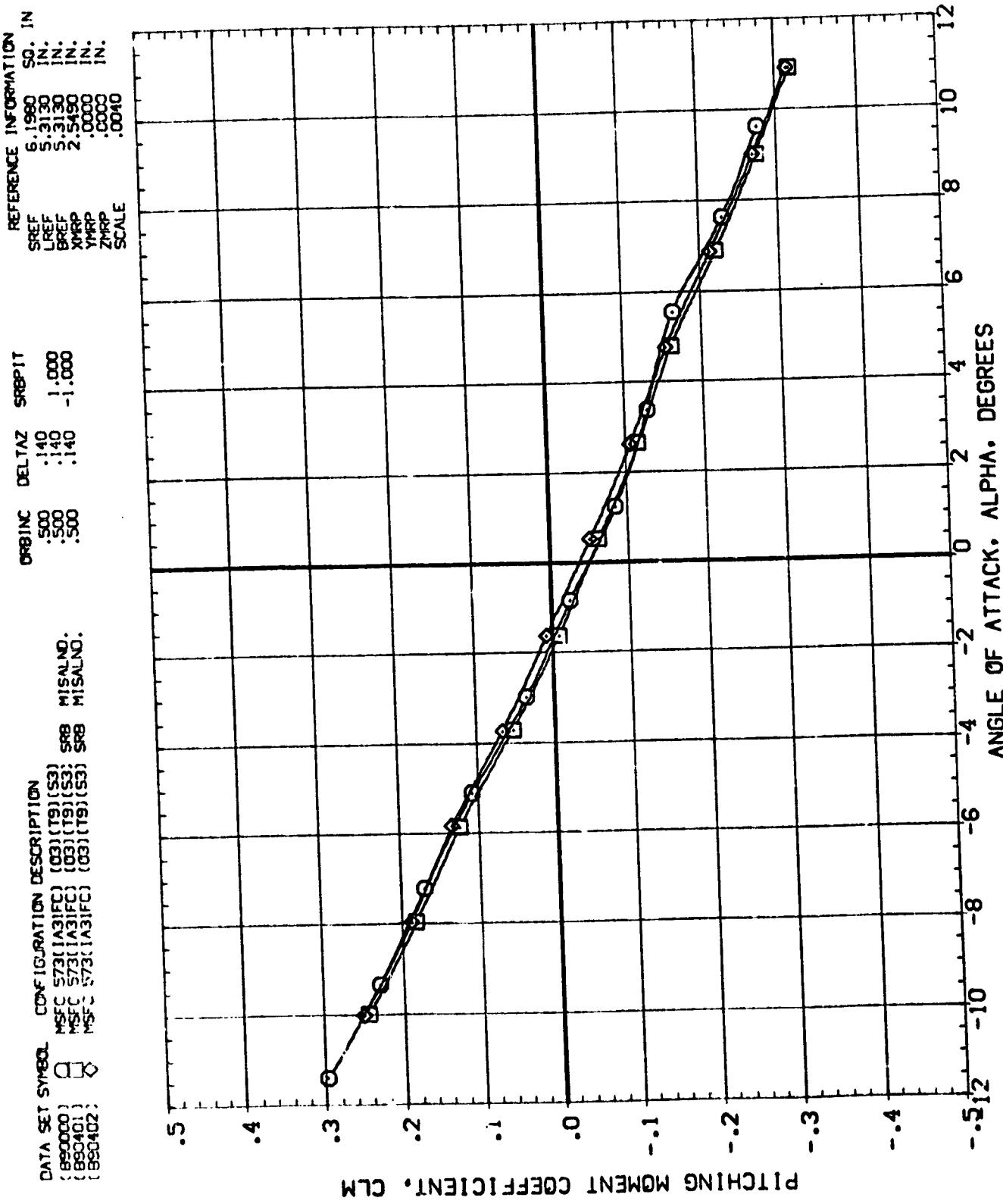
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B890000) MSC 5731(A3IFC) (03)19)(S3) SRB MISLAND.  
 (B890401) MSC 5731(A3IFC) (03)19)(S3) SRB MISLAND.  
 (B890402) MSC 5731(A3IFC) (03)19)(S3) SRB MISLAND.

REFERENCE INFORMATION  
 SRREF 6.1980 SO. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.



EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(D)MACH = 1.46



EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 $(\text{MACH}) = .90$

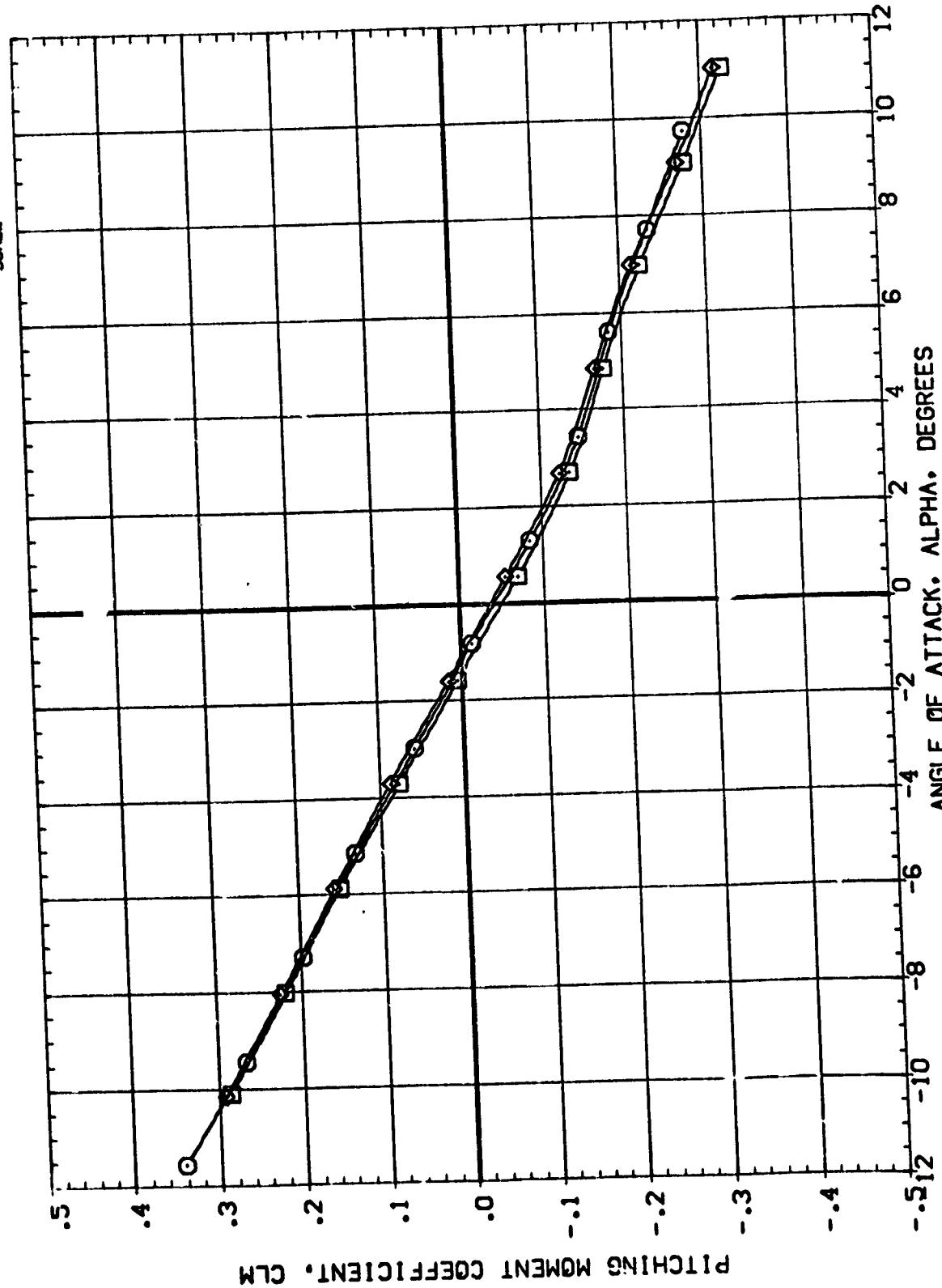
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DATA SET SYMBOL CONFIGURATION DESCRIPTION

(B)500000	NSFC	5731(1A3)FC	(03)(T9)(S3)	SRB MISLAND.
(B)50401	NSFC	5731(1A3)FC	(03)(T9)(S3)	SRB MISLAND.
(B)50402	NSFC	5731(1A3)FC	(03)(T9)(S3)	SRB MISLAND.

REFERENCE INFORMATION

SRB INC	DELTAZ	SRBPIT	SO. IN
.500	.140	1.000	6.1980
.500	.140	-1.000	5.3130
			5.3130
			2.5450
			.0000
			.0040
			SCALE

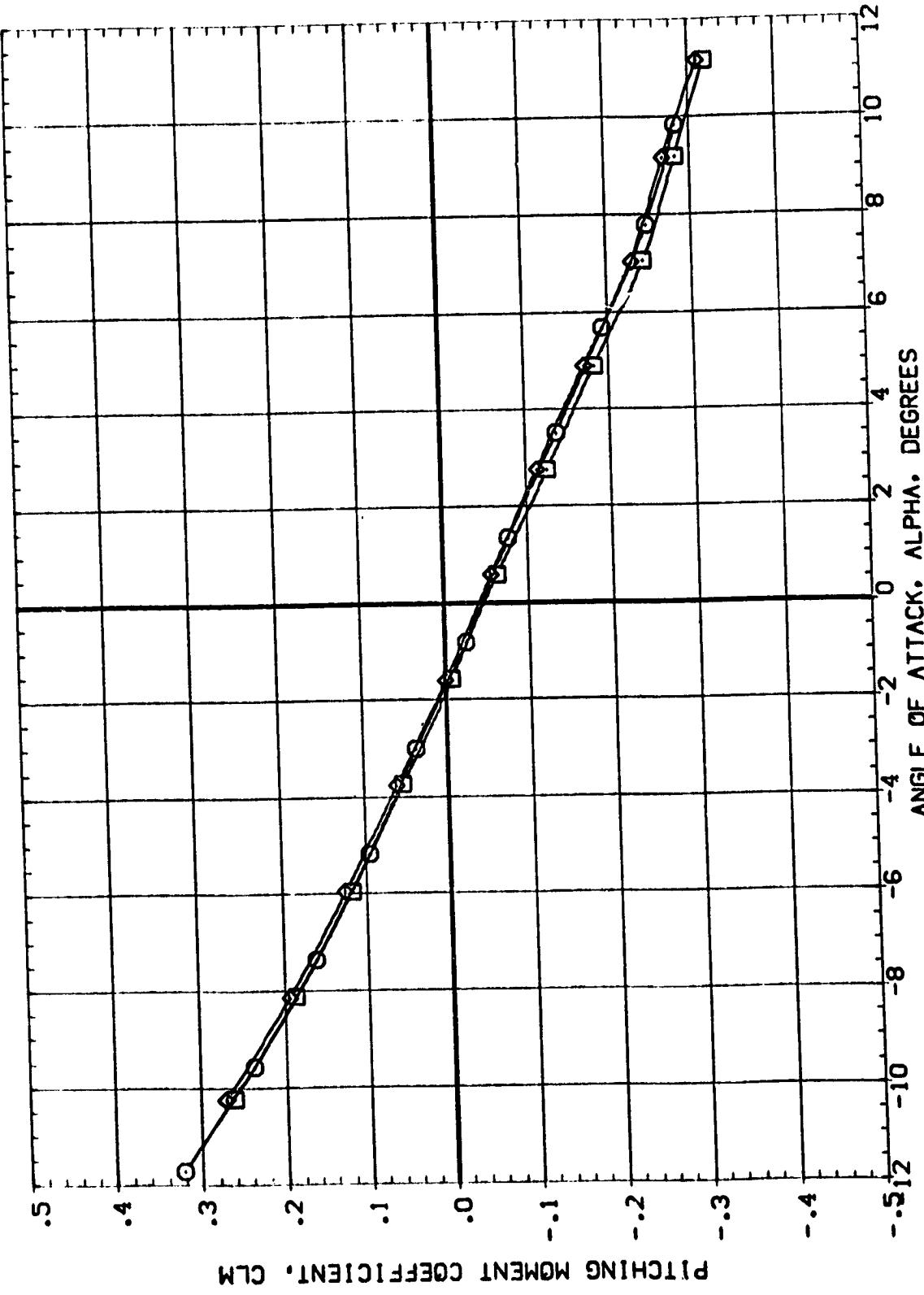


EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

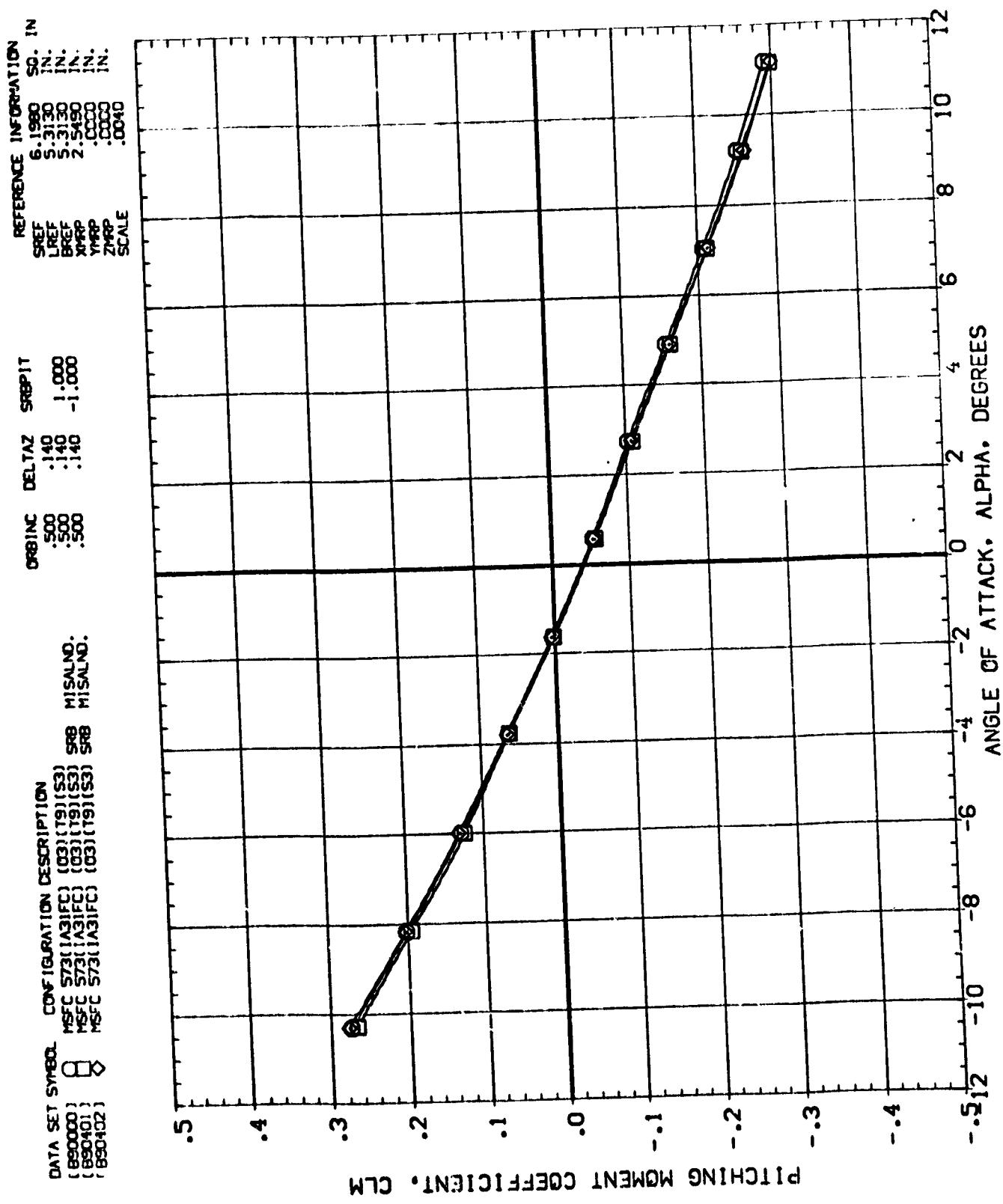
DATA SET SYMBOL      CONFIGURATION DESCRIPTION  
 (690000)      (03)(T9)(S3)  
 (690401)      MSFC ST73(1A3)FC  
 (690402)      MSFC ST73(1A3)FC

REFERENCE INFORMATION  
 ORBINC DELTAZ SRBPIT  
 .500 .140 1.000  
 .500 .140 -1.000  
 .500 .140 .000  
 .000 .000 .0040



EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 (C)MACH = 1.25

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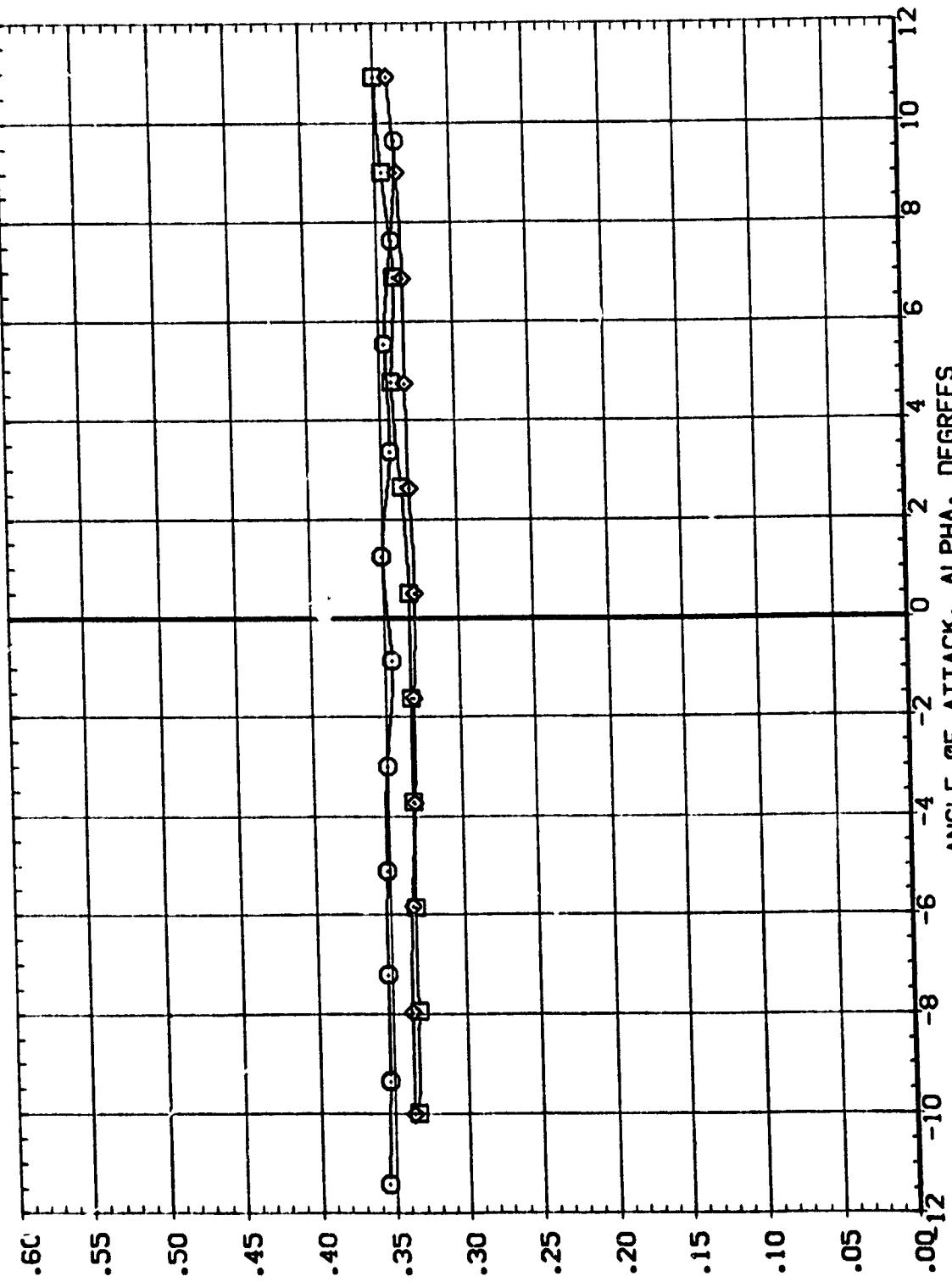


EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

$(D)_{MACH} = 1.46$

DATA SET SYMBOL	CONFIGURATION DESCRIPTION
	MSFC 573([A3]FC) (03)[T9]S3
	MSFC 573([A3]FC) (03)[T9]S3
	MSFC 573([A3]FC) (03)[T9]S3

	REFERENCE	INFORMATION	SD.	IN.
DRB INC	SREF	6.1980	.90	IN.
DETAZ	LREF	5.3130	.30	IN.
SRBPI7	BREF	5.3130	.30	IN.
	XMRP	2.5490	.49	IN.
	YMRP	.0000	.00	IN.
	ZMRP	.0000	.00	IN.



AXIAL FORCE COEFFICIENT, CA

## EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

$$V_{\text{MACH}} = .90$$

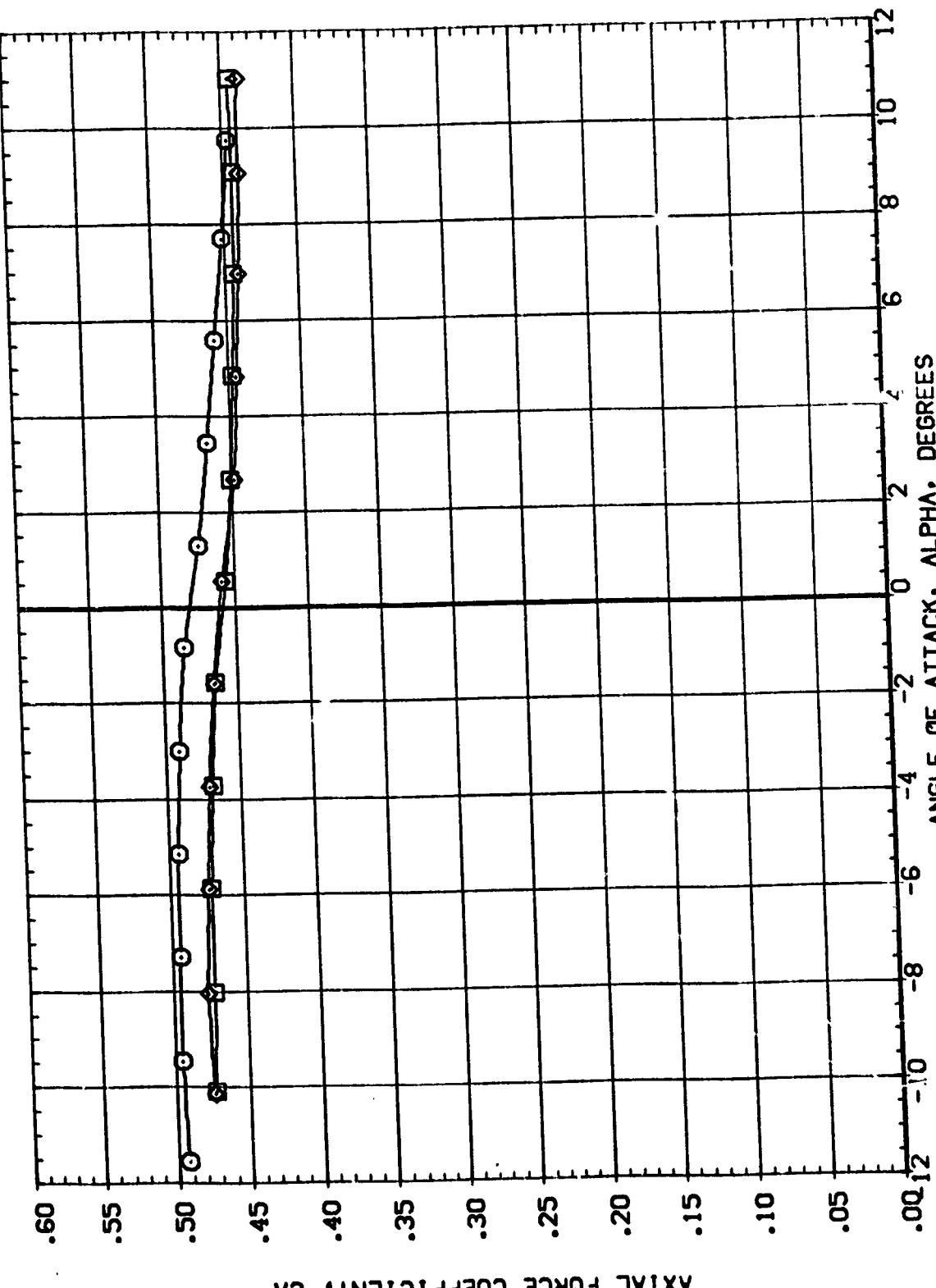
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DATA SET SYMBOL CONFIGURATION DESCRIPTION

(B50000)	NSFC 573(A3IFC)	(03)(T9)(S3)	SRB	MISALND.
(B50010)	NSFC 573(A3IFC)	(03)(T9)(S3)	SRB	MISALND.
(B50022)	NSFC 573(A3IFC)	(03)(T9)(S3)	SRB	MISALND.

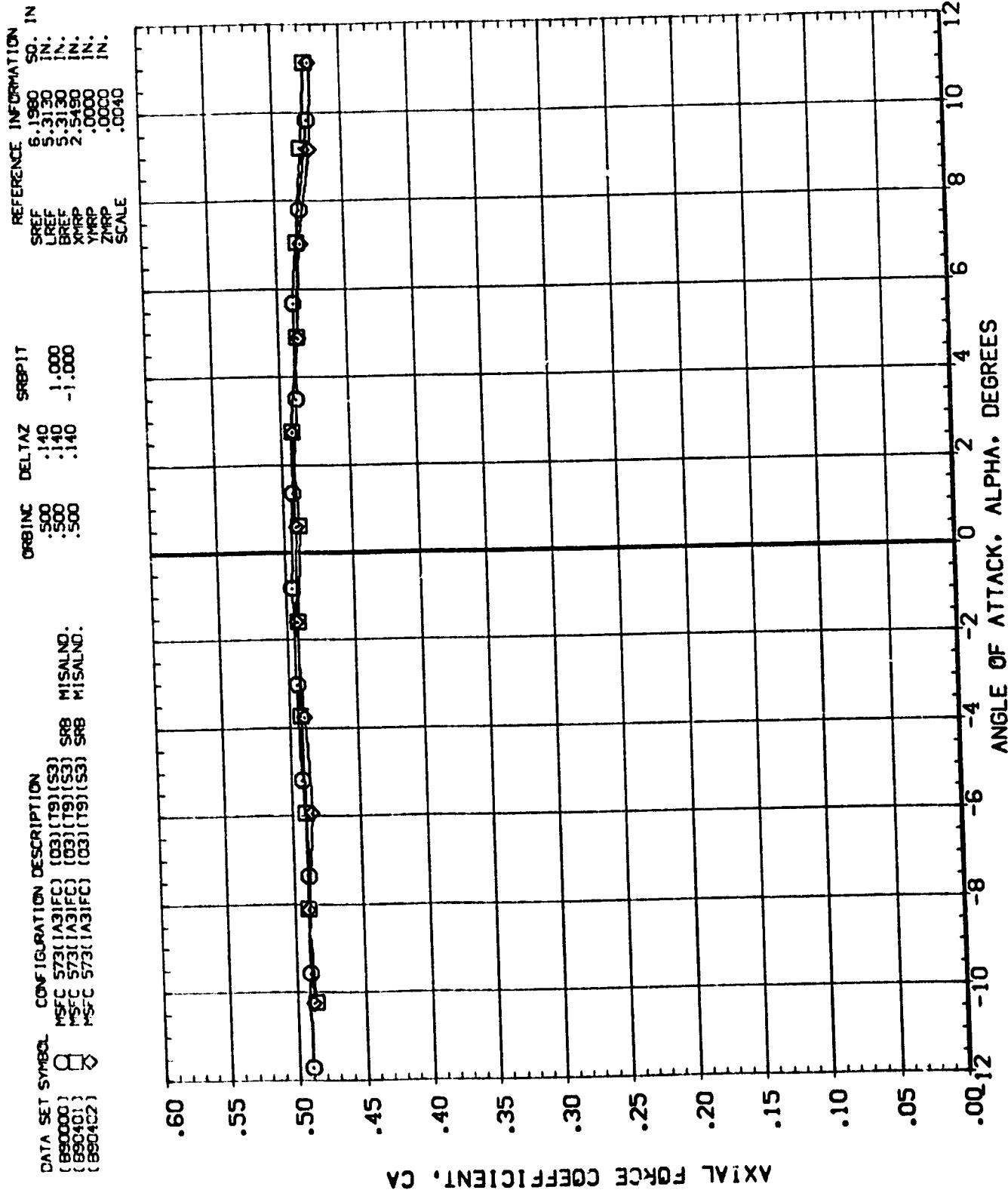
REFERENCE INFORMATION

SREF	6.1980	SD. IN
LREF	5.3130	IN.
BREF	5.3130	IN.
XMRP	2.5490	IN.
YMRP	.0000	IN.
ZMRP	.0000	IN.
SCALE	.0040	



### EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

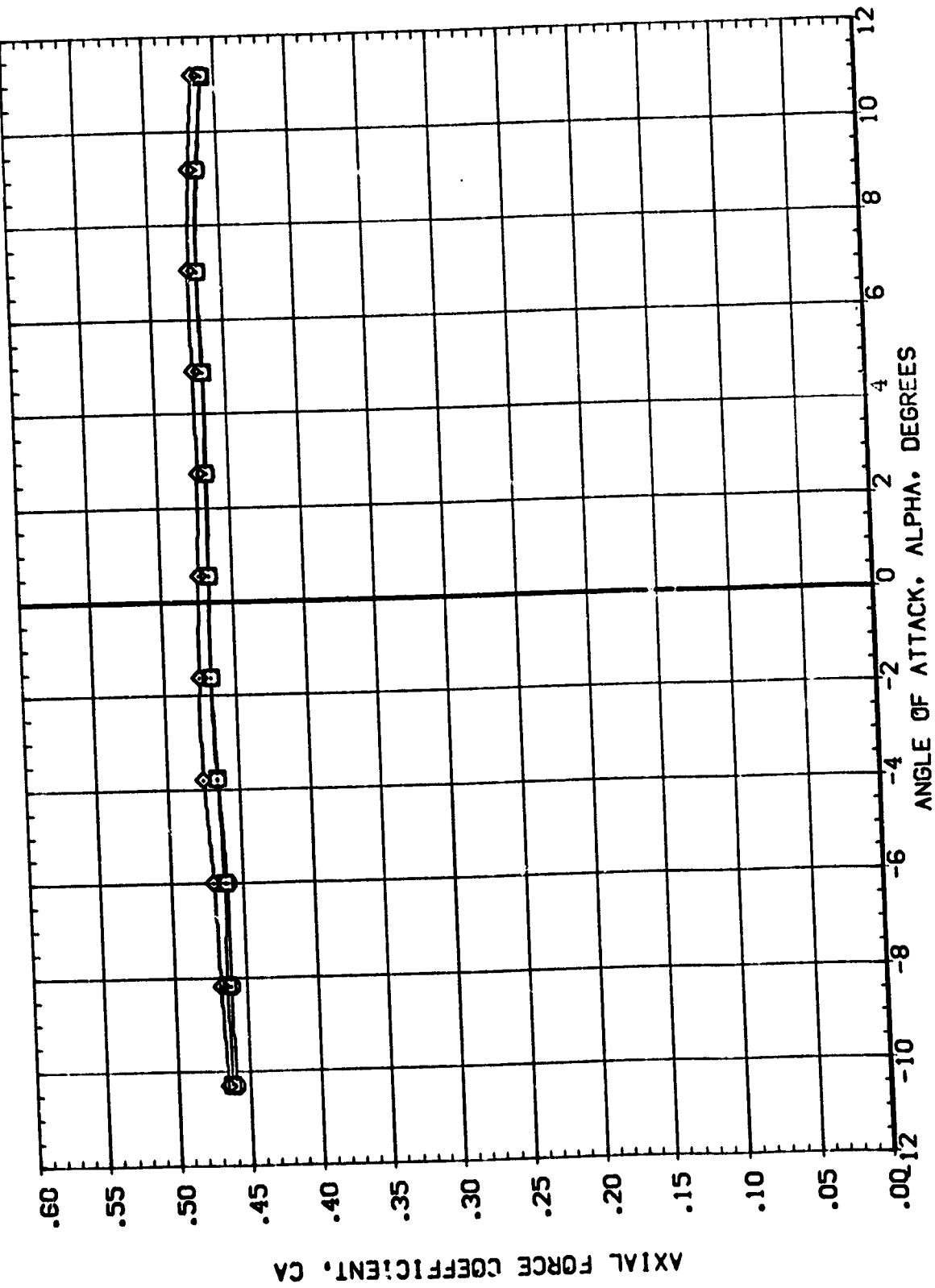
(B)MACH = 1.05



EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 $(C)_MACH = 1.25$

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (BS90000) NSFC 573(A31FC) (03)(T9)(S3) MISALND.  
 (BS90401) NSFC 573(A31FC) (03)(T9)(S3) SRB MISALND.  
 (BS90402) NSFC 573(A31FC) (03)(T9)(S3)

REFERENCE INFORMATION  
 SREF 6.1980 SD. IN  
 LREF 5.3130 IN  
 BREF 5.3130 IN  
 XRP 2.5150 IN  
 YRP .0000 IN  
 ZRP .0000 IN  
 SCALE .0040



EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

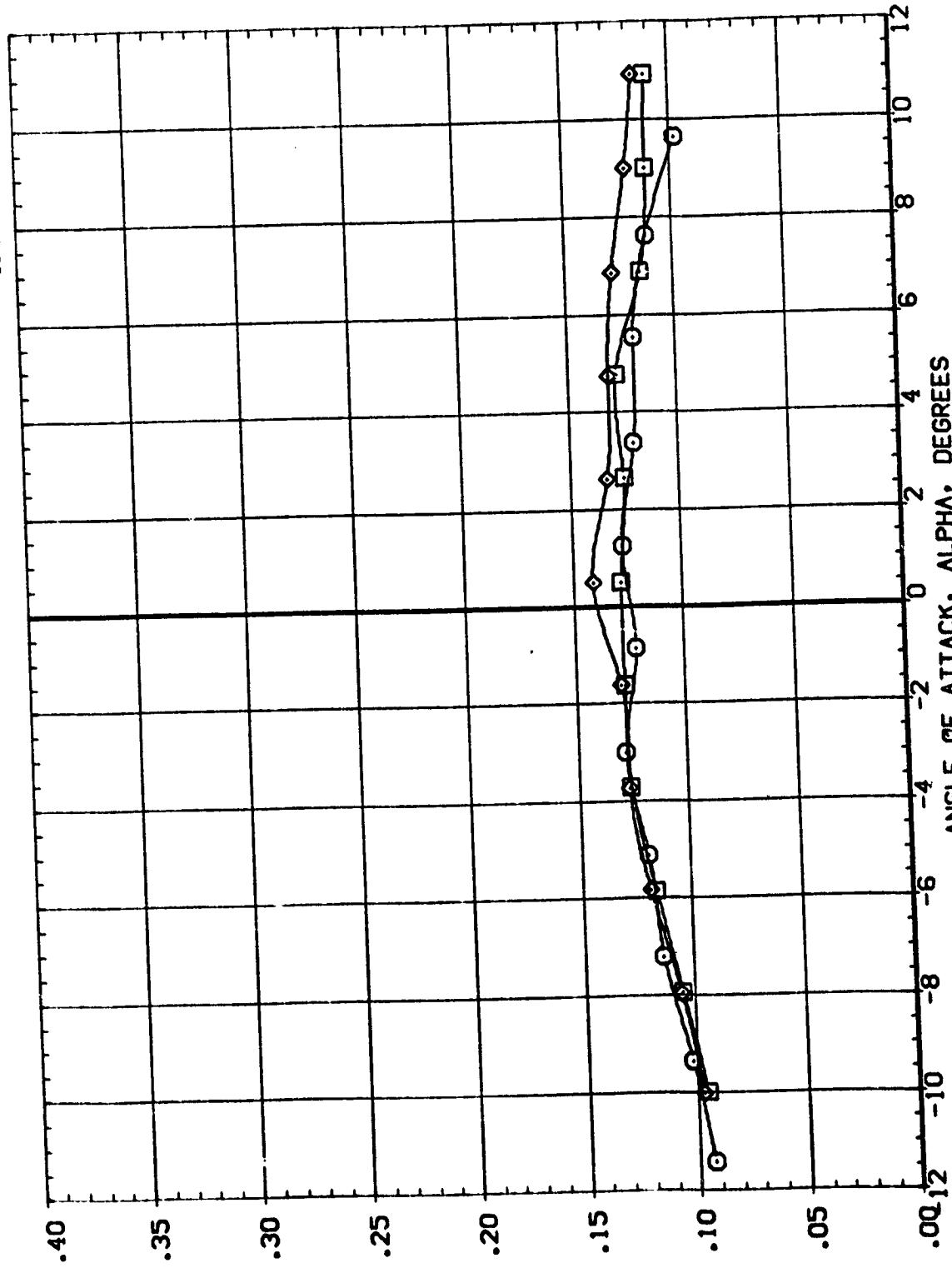
(CD)MACH = 1.46

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(880000)	NSFC 573(1A3)FC	(83)(T9)(S3)	SRB MISALND.
(8804C1)	NSFC 573(1A3)FC	(83)(T9)(S3)	SRB MISALND.
(8804C2)	NSFC 573(1A3)FC	(83)(T9)(S3)	SRB MISALND.

REFERENCE INFORMATION

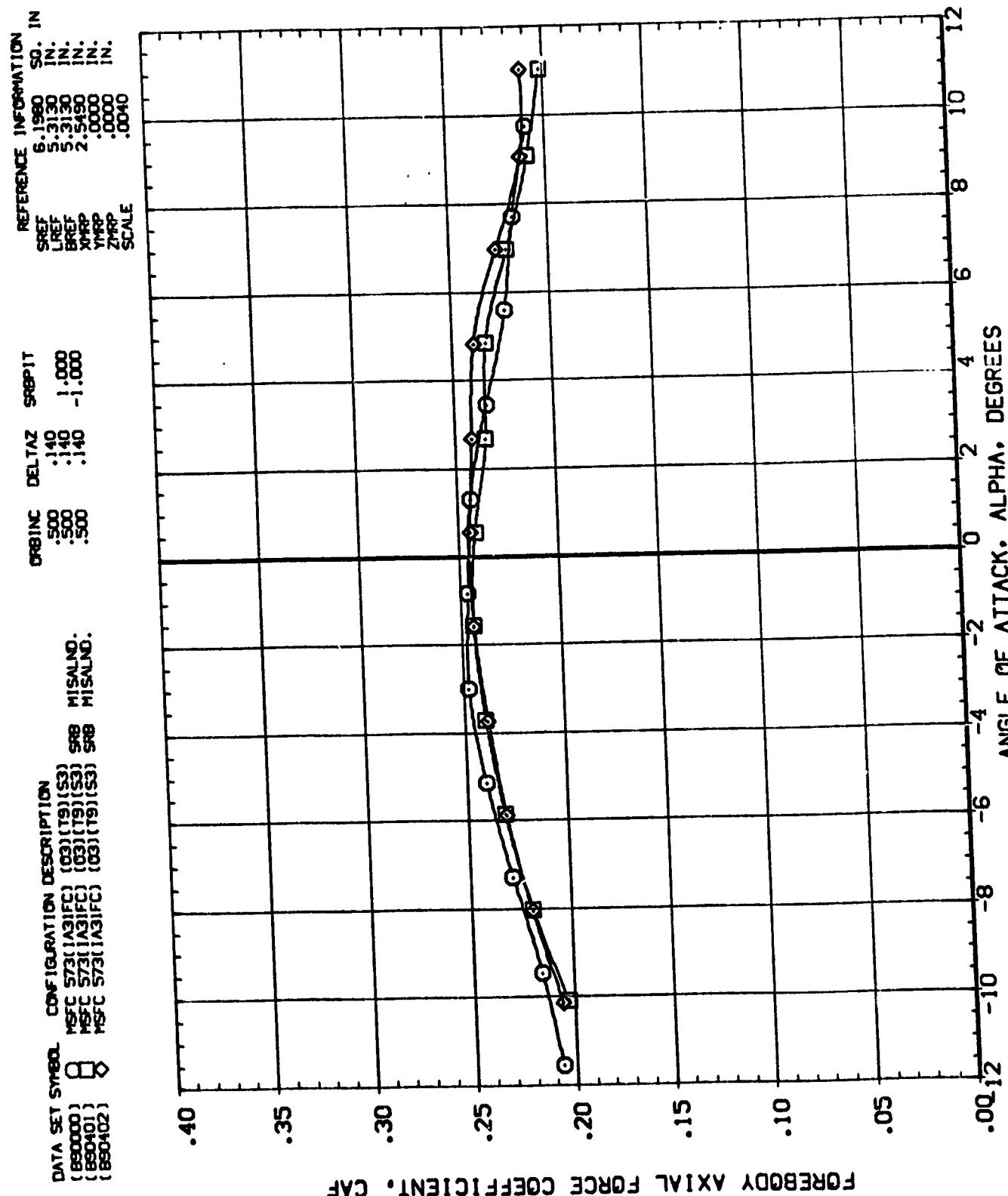
REF	6.1980 IN.
LREF	5.3130 IN.
BREF	5.3130 IN.
XRP	2.5490 IN.
YRP	.0000 IN.
ZRP	.0000 IN.
SCALE	.0040



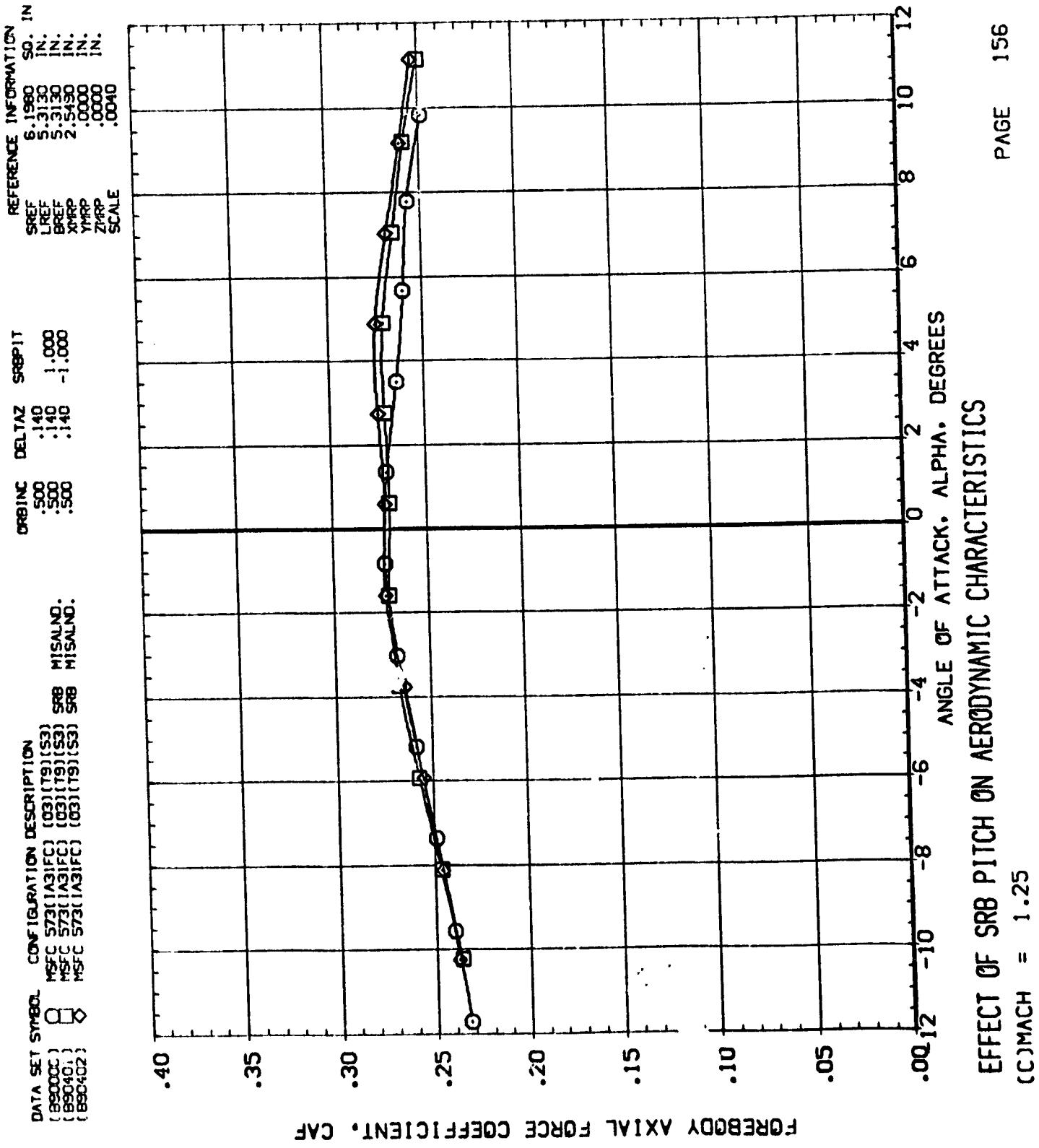
FOREBODY AXIAL FORCE COEFFICIENT, CAF

### EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(A)MACH = .90

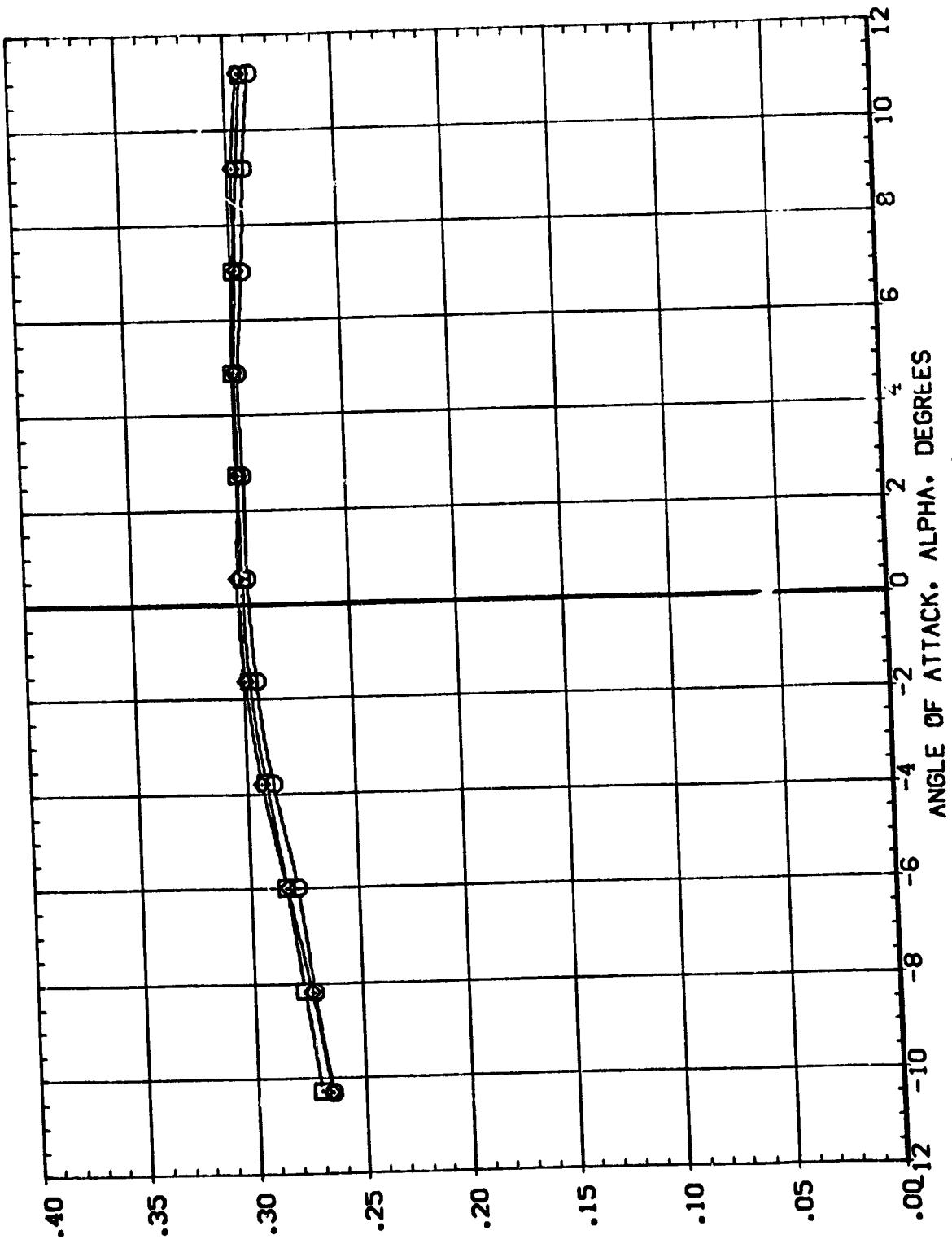


EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 $(B)_{MACH} = 1.05$



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B80000) MSC S73(1A3)FC (03)(19)(S3) MISALNO.  
 (B80401) MSC S73(1A3)FC (03)(19)(S3) SRB MISALNO.  
 (B80402) MSC S73(1A3)FC (03)(19)(S3) SRB MISALNO.

REFERENCE INFORMATION  
 DBINC DELTAZ SRBPIT  
 .500 .140 -1.000  
 .500 .140 -1.000  
 .500 .140 -1.000  
 .500 .140 -1.000  
 SCALE



FOREBODY AXIAL FORCE COEFFICIENT, CAF

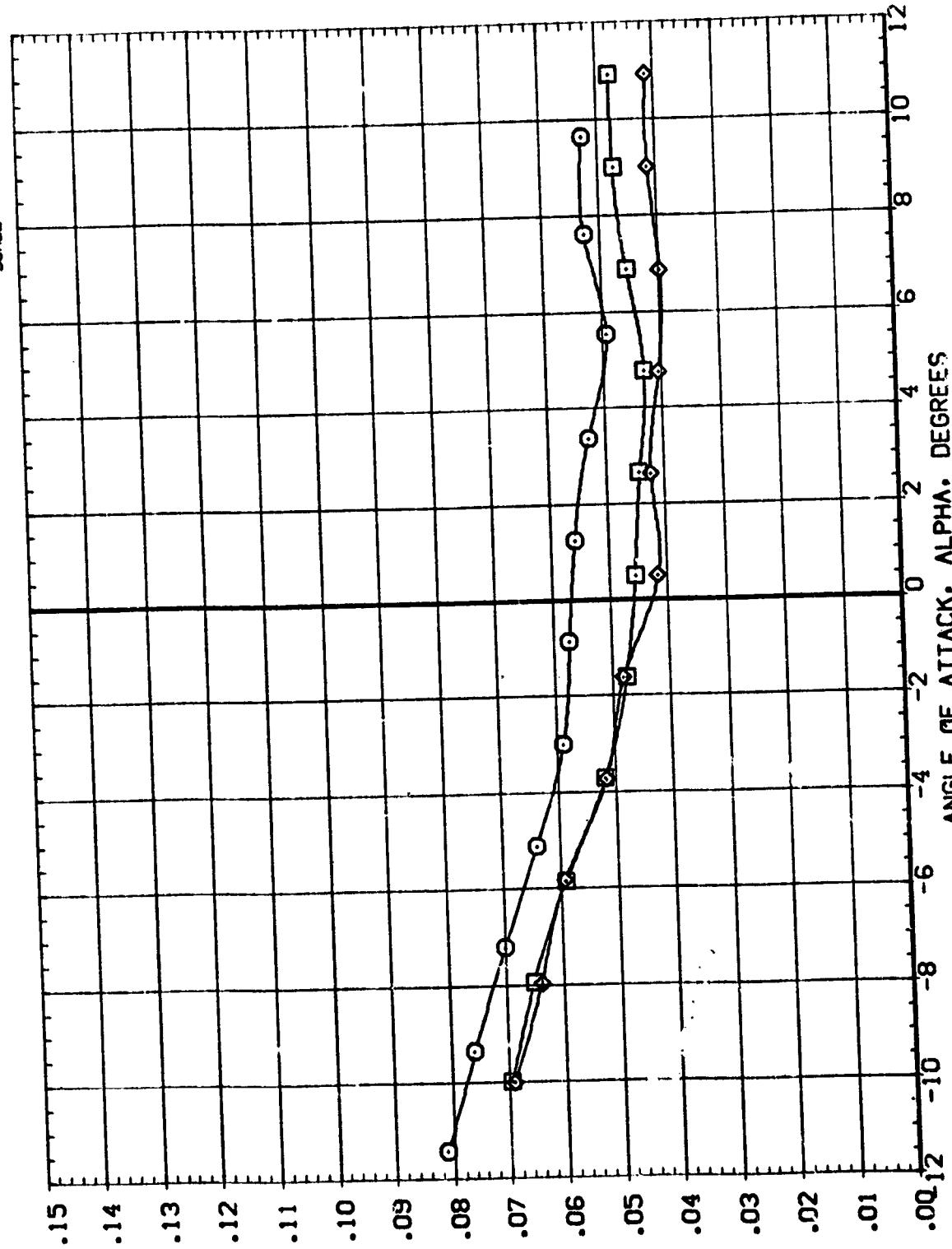
EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(CD)MACH = 1.46

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REFERENCE INFORMATION  
 SRREF 6.1980 SD. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0000 IN.  
 SCALE .0040 IN.

DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(880000)	MSFC 5731(A3)FC (63)(T9)(S3)
(880401)	MSFC 5731(A3)FC (63)(T9)(S3)
(880402)	MSFC 5731(A3)FC (63)(T9)(S3)

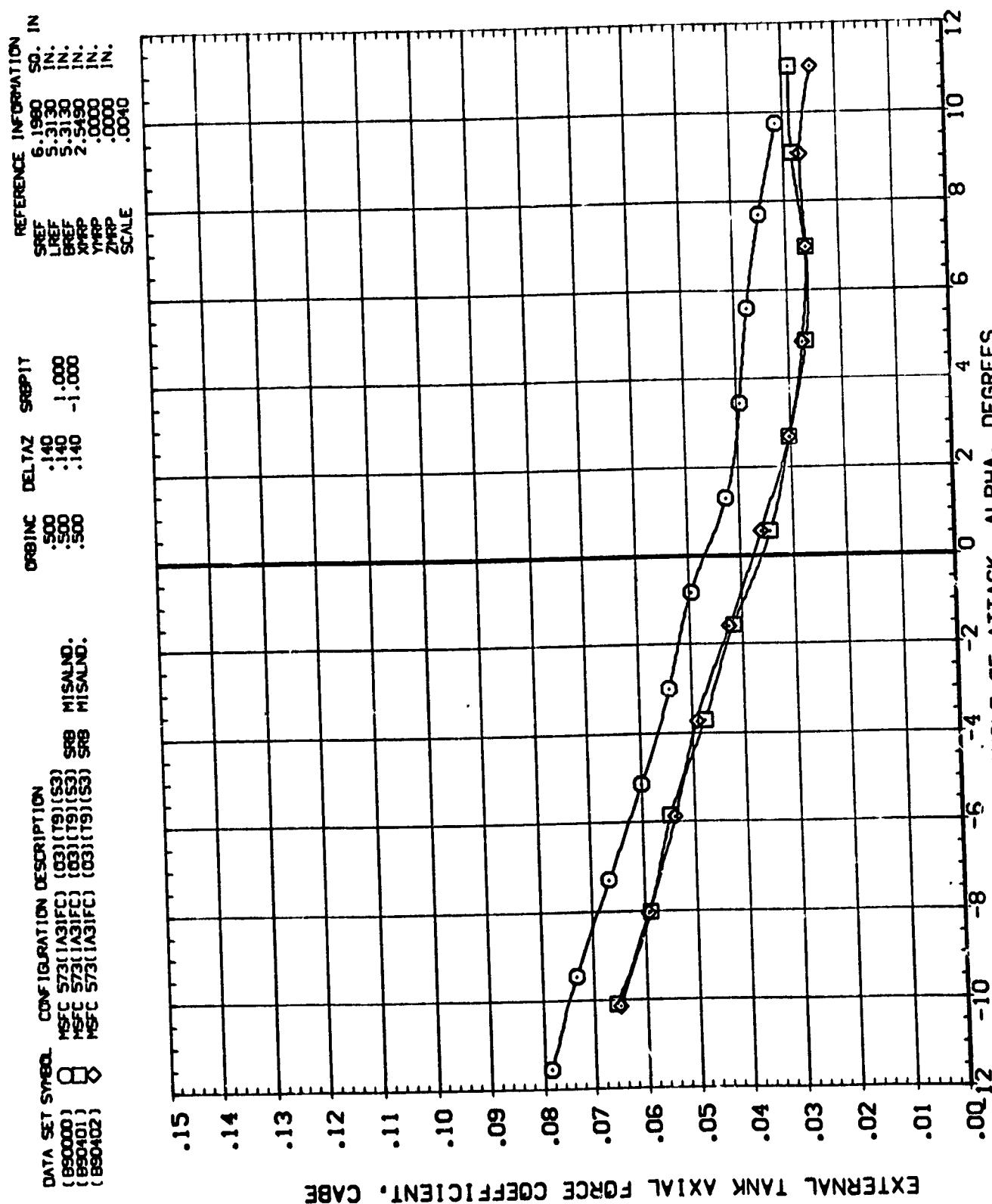


EXTERNAL TANK AXIAL FORCE COEFFICIENT. CABE

EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 $(\Delta)MACH = .90$

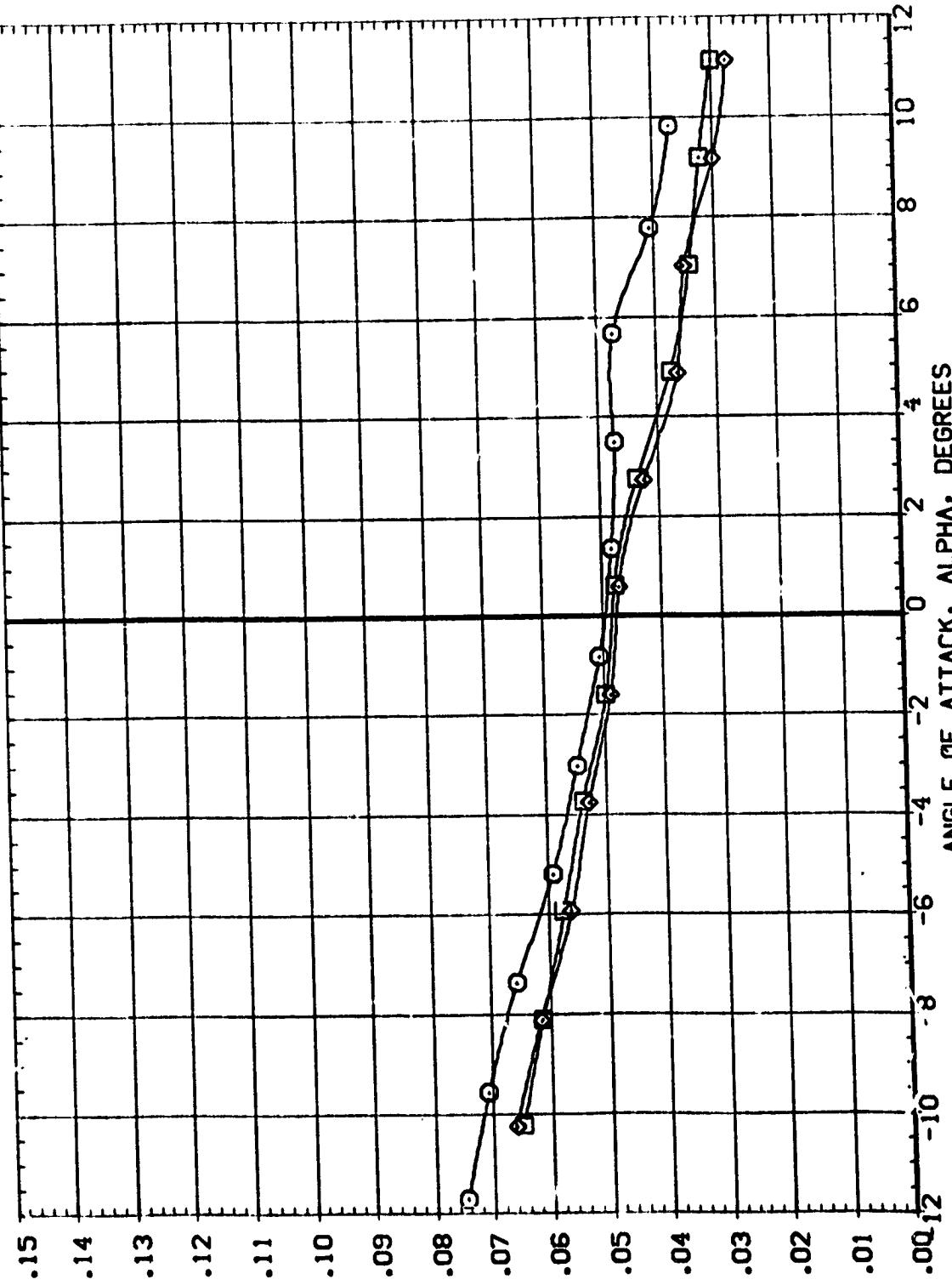
(B)MACH = 1.05

## EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS



DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) NSFC 573(1A3)FC (03)(T9)(S3)  
 (B90001) NSFC 573(1A3)FC (03)(T9)(S3) SRB MISALNO.  
 (B90402) NSFC 573(1A3)FC (03)(T9)(S3)

REFERENCE INFORMATION  
 ORB INC DELTAZ SRBPIT  
 .500 .140 .140  
 .500 .140 -1.000  
 .500 .140 -1.000  
 SRREF LREF BREF  
 6.1980 5.3130 5.3130  
 IN. IN. IN.  
 XMRP YMRP ZMRP  
 2.5490 .0000 .0000  
 IN. IN. IN.  
 .0040



EXTERNAL TANK AXIAL FORCE COEFFICIENT, CABE

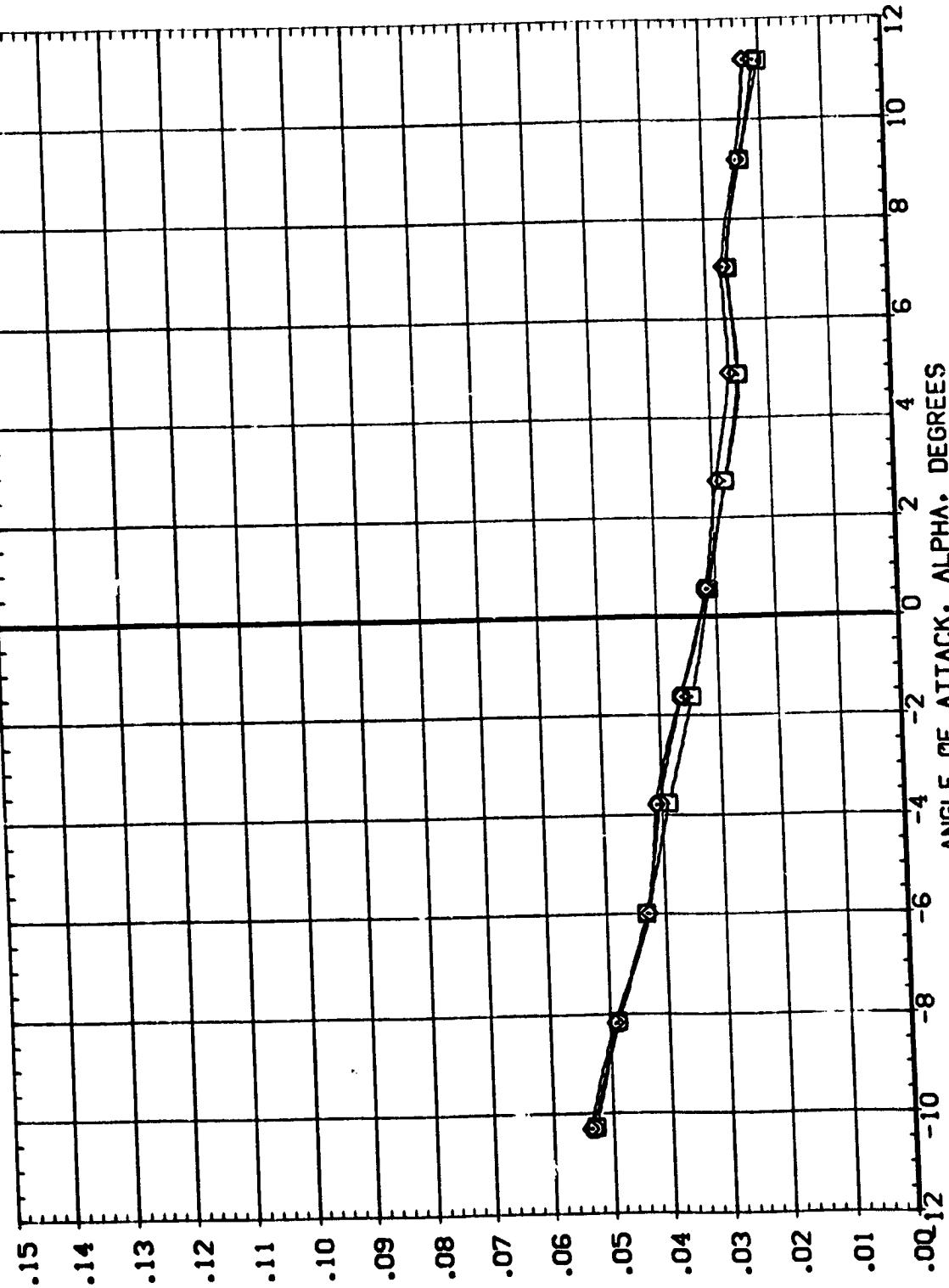
### EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 NSFC 5731(A3)FC (03)(T9)(S3) SRB MISALND.  
 (850000) (850401) (850402) NSFC 5731(A3)FC (03)(T9)(S3) SRB MISALND.

ORBITIC	DELTAZ	SRBPIT	REFERENCE INFORMATION
.500	.140		SREF 6.1980 SC. IN
.500	.140		LREF 5.3130 IN.
			BREF 5.3130 IN.
			XRP 2.5490 IN.
			YRP .0000 IN.
			ZRP .0040 IN.
			SCALE

EXTERNAL TANK AXIAL FORCE COEFFICIENT, CABE

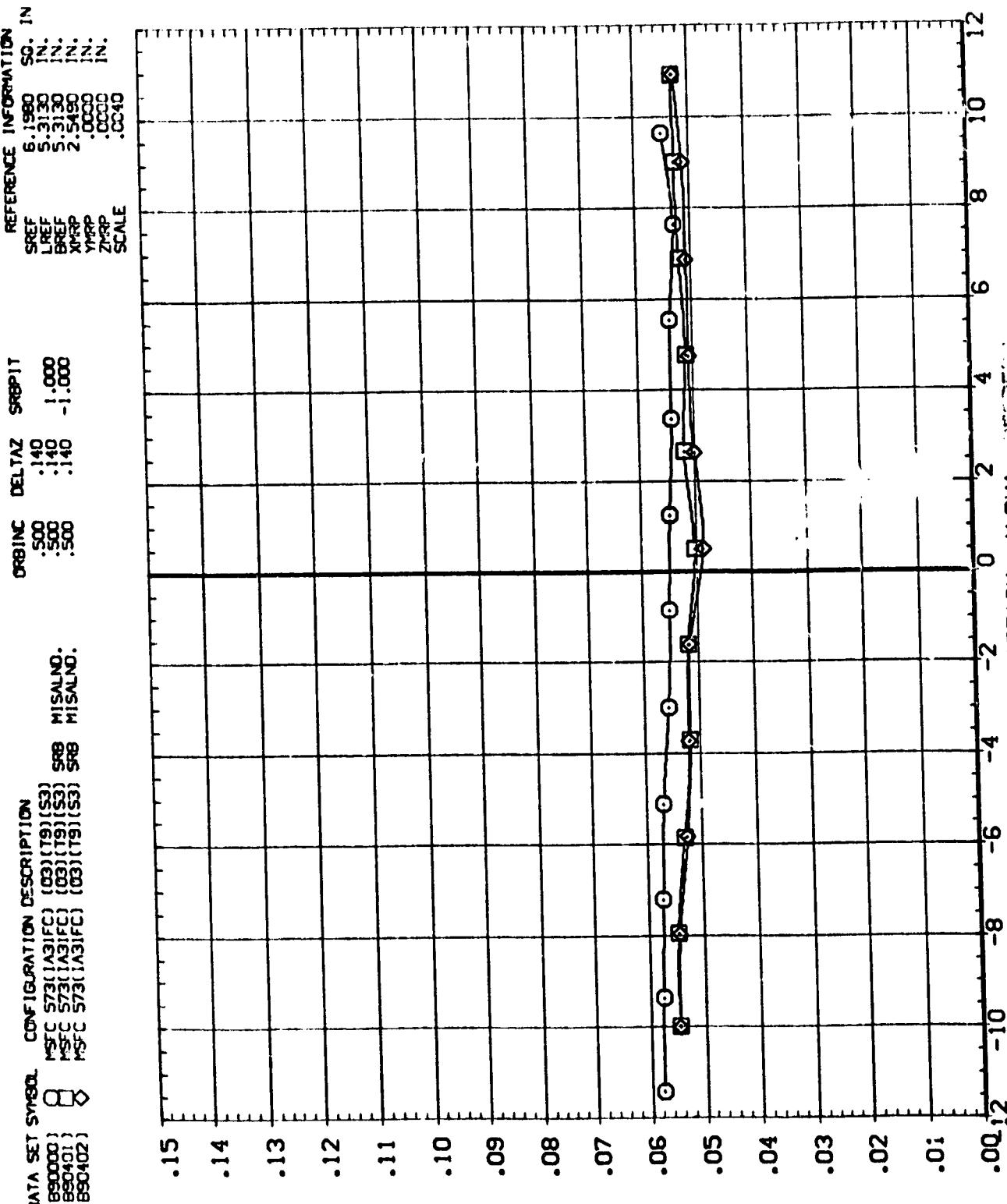


EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

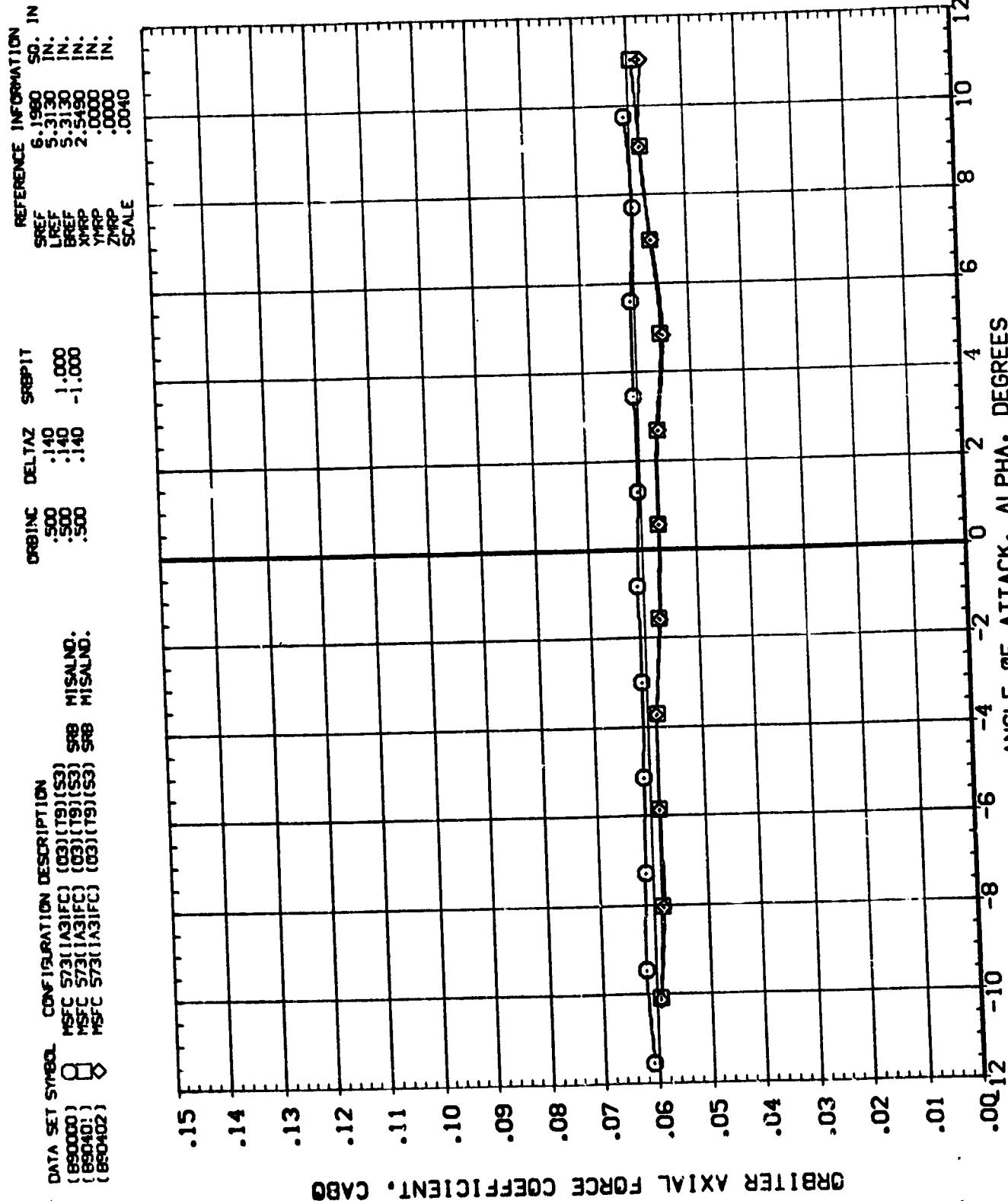
(MACH = 1.46

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DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) SFC S731(A3)FC (03)(T9)(S3) SRB MISALND.  
 (B90401) SFC S731(A3)FC (03)(T9)(S3) SRB MISALND.  
 (B90402) SFC S731(A3)FC (03)(T9)(S3) SRB MISALND.



EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTIC  
 $(V/MACH = .90)$

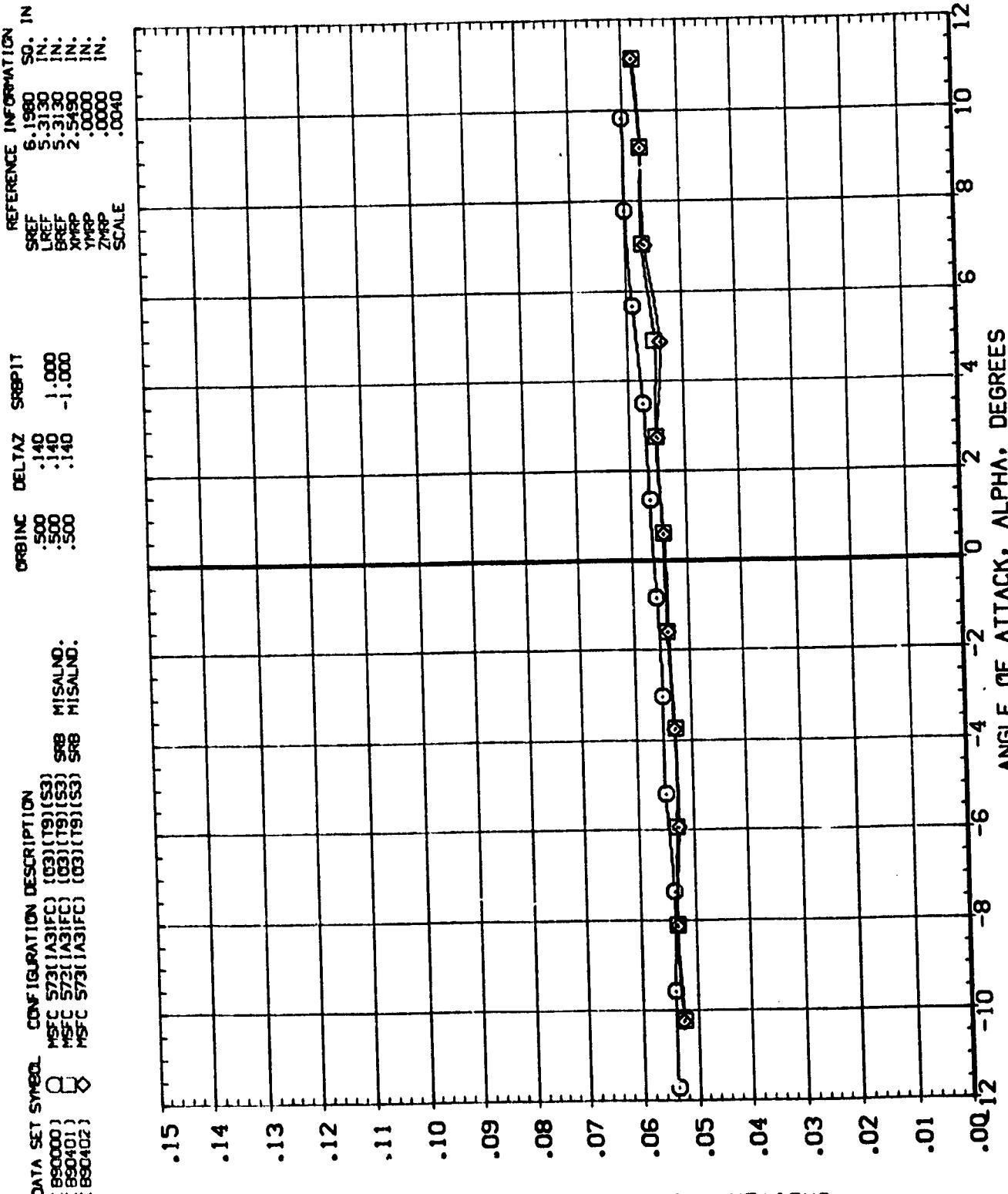


EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

DATA SET SUMMARY      CONFIGURATION DESCRIPTION

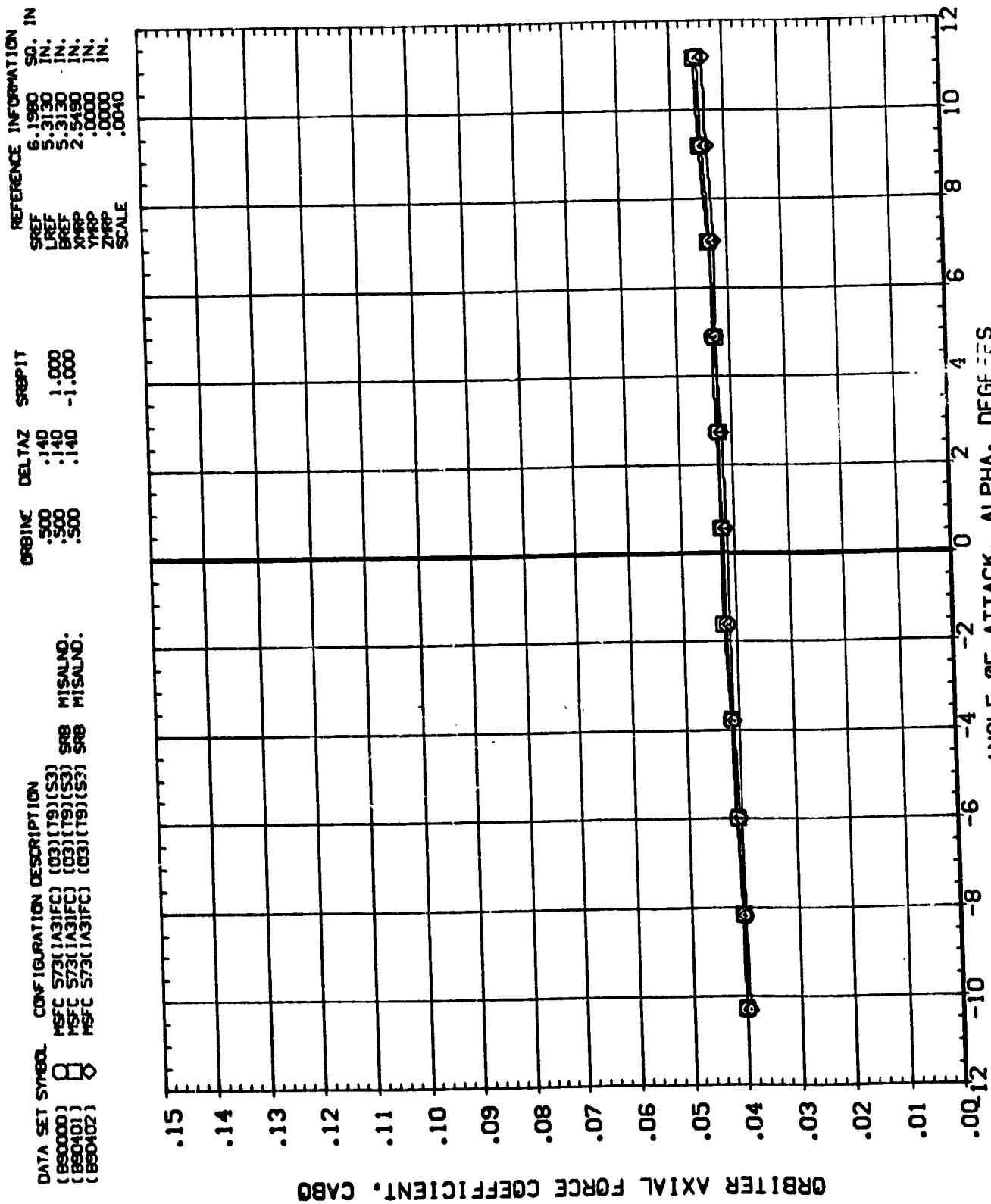
890000	MSFC 5731(A31FC)	(63) 179 (53)	SRB	MISALN.
890401	MSFC 5732(A31FC)	(63) 179 (53)	SRB	MISALN.
890402	MSFC 5733(A31FC)	(63) 179 (53)	SRB	MISALN.



### EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

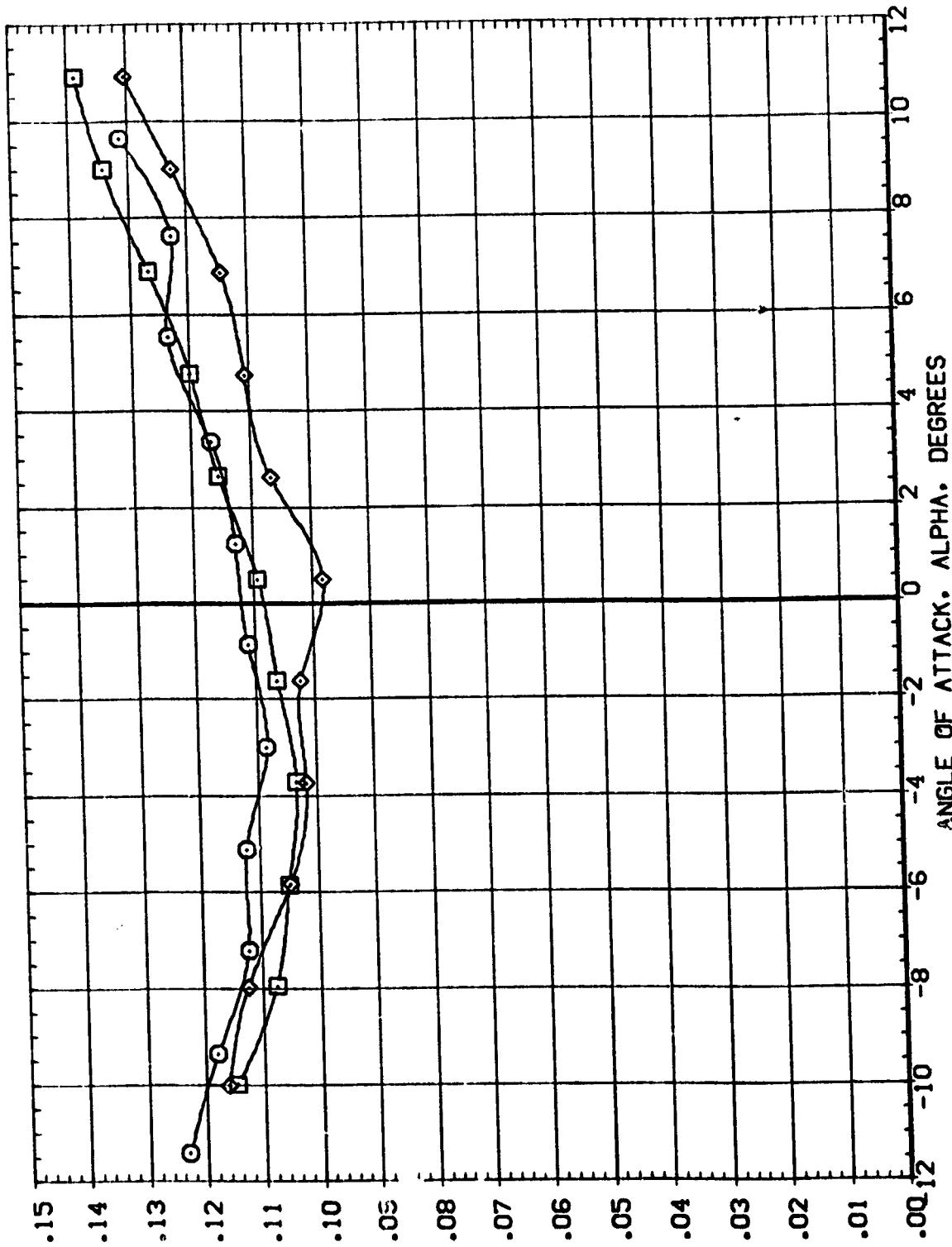
$(C)_{MACH} = 1.25$

EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 $(D)MACH = 1.46$



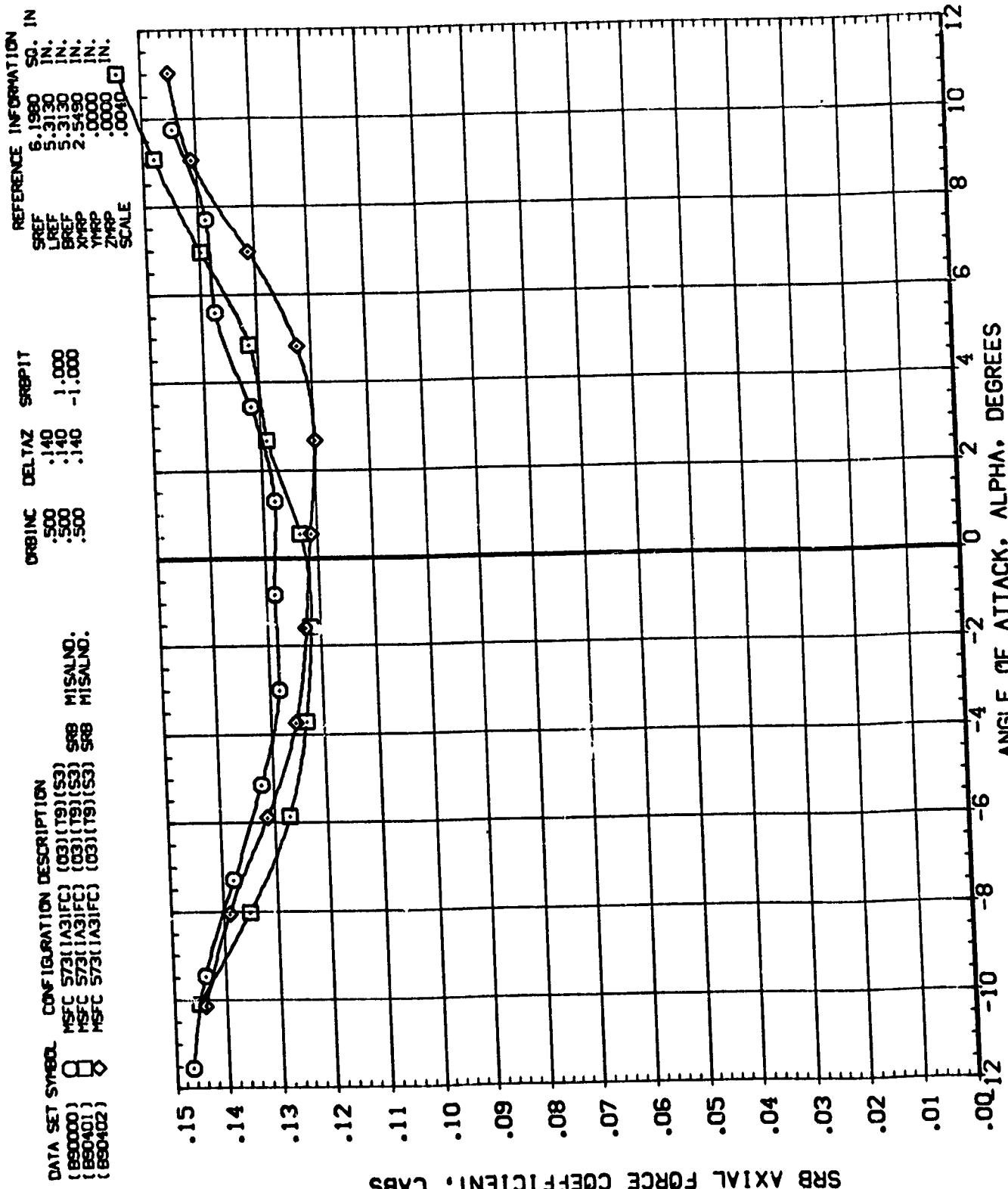
DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) O NSFC 573(A3IFC) (03)(T9)(S3) SRB MSLND.  
 (B90401) X NSFC 573(A3IFC) (03)(T9)(S3) SRB MSLND.  
 (B90402) D NSFC 573(A3IFC) (03)(T9)(S3) SRB MSLND.

REFERENCE INFORMATION  
 SREF 6.1980 SO. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XHPP 2.5490 IN.  
 YHPP .0000 IN.  
 ZHPP .0040 IN.  
 SCALE



SRB AXIAL FORCE COEFFICIENT, CABs

EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 $(\Delta)MACH = .90$

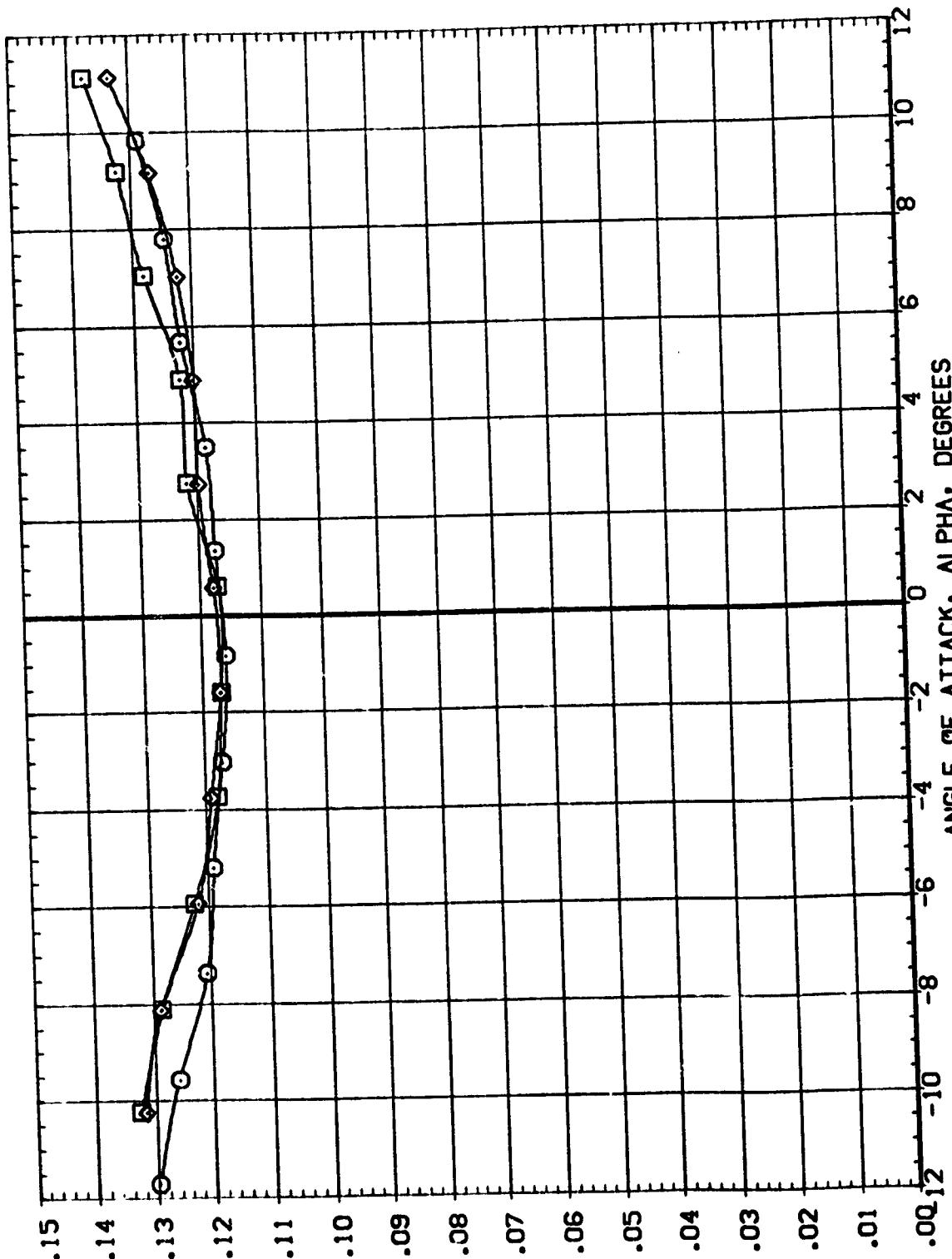


EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (890000) MSFC 5731(A3)FC (03)(T9)(S3) SRB MISALNO.  
 (890401) MSFC 5731(A3)FC (03)(T9)(S3) SRB MISALNO.  
 (890402) MSFC 5731(A3)FC (03)(T9)(S3) SRB MISALNO.

REFERENCE INFORMATION  
 SRBF 6.1980 SQ. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMP 2.5490 IN.  
 YMP .0000 IN.  
 ZMP .0040 IN.  
 SCALE

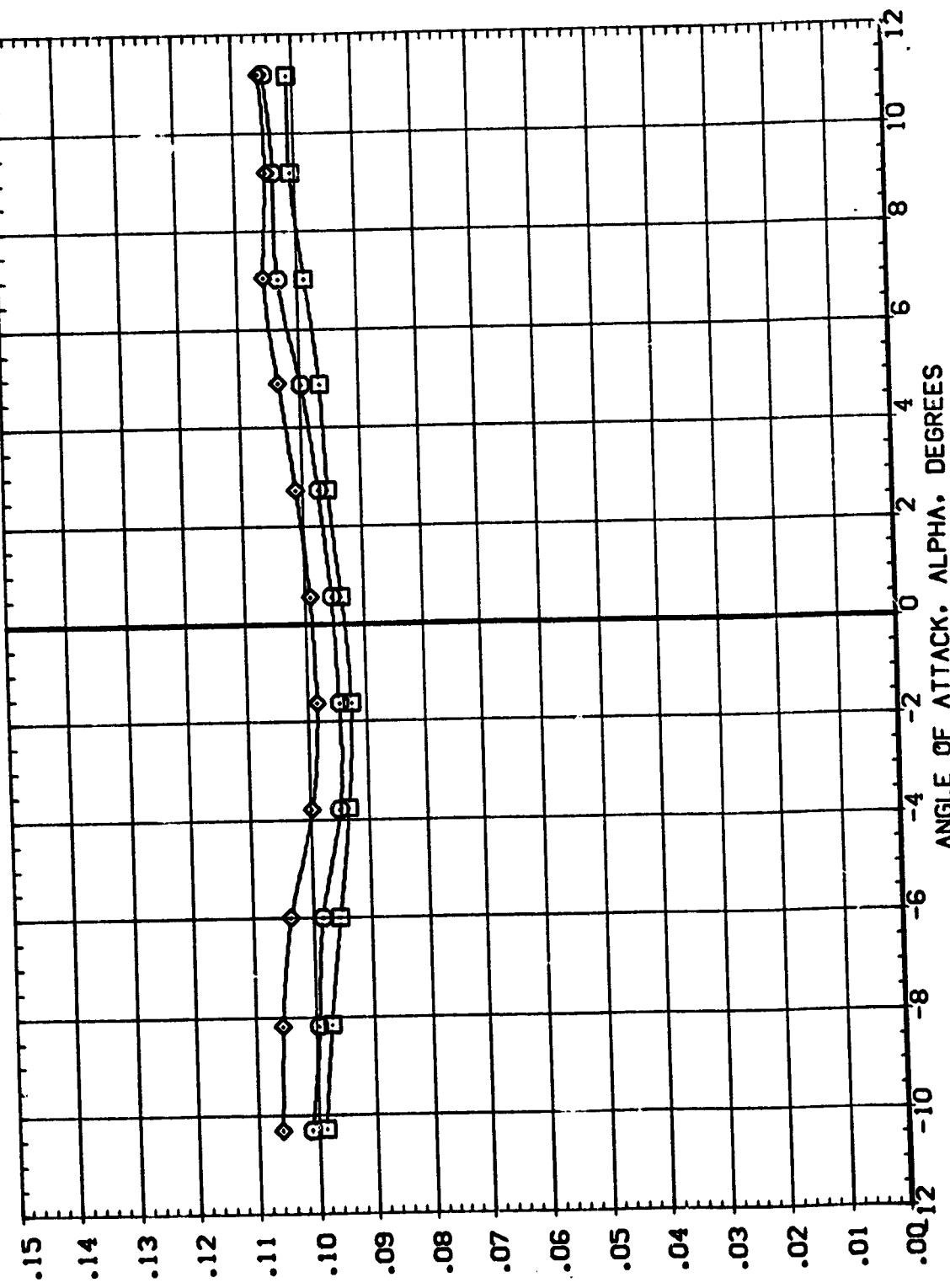


SRB AXIAL FORCE COEFFICIENT, CABS

EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 (C)<sub>MACH</sub> = 1.25

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (650000) MSFC 5731(A3)FC (03)(19)(S3) SRB MISALD.  
 (650401) MSFC 5731(A3)FC (03)(19)(S3) SRB MISALD.  
 (650402) MSFC 5731(A3)FC (03)(19)(S3) SRB MISALD.

REFERENCE INFORMATION  
 SREF 6.1980 SO. IN  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.

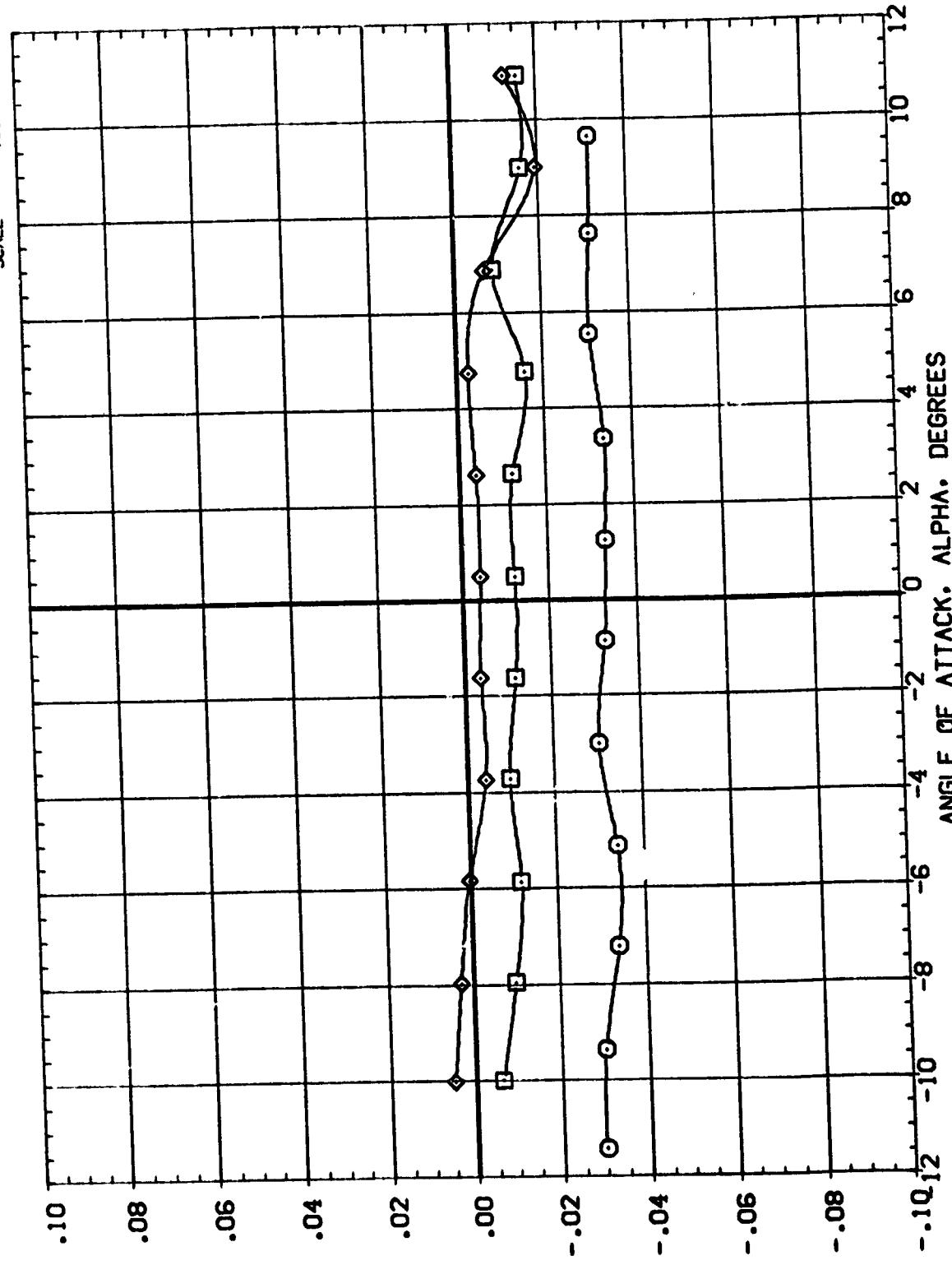


SRB AXIAL FORCE COEFFICIENT, CABs

EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 (D)<sub>MACH</sub> = 1.46

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	SRB11	SRB17
(B90000)	MFC S73[(A3)FC] (03)(T9)(S3)	.500 .500 .500	.140 -1.000 .140
(B90101)	MFC S73[(A3)FC] (03)(T9)(S3)	.500	-1.000
(B90102)	MFC S73[(A3)FC] (03)(T9)(S3)	.500	.0000

REFERENCE INFORMATION  
 SO. IN  
 SREF 6.1980  
 LREF 5.3130  
 BREF 5.3130  
 XRP 2.5590  
 YRP .0000  
 ZRP .0040  
 SCALE



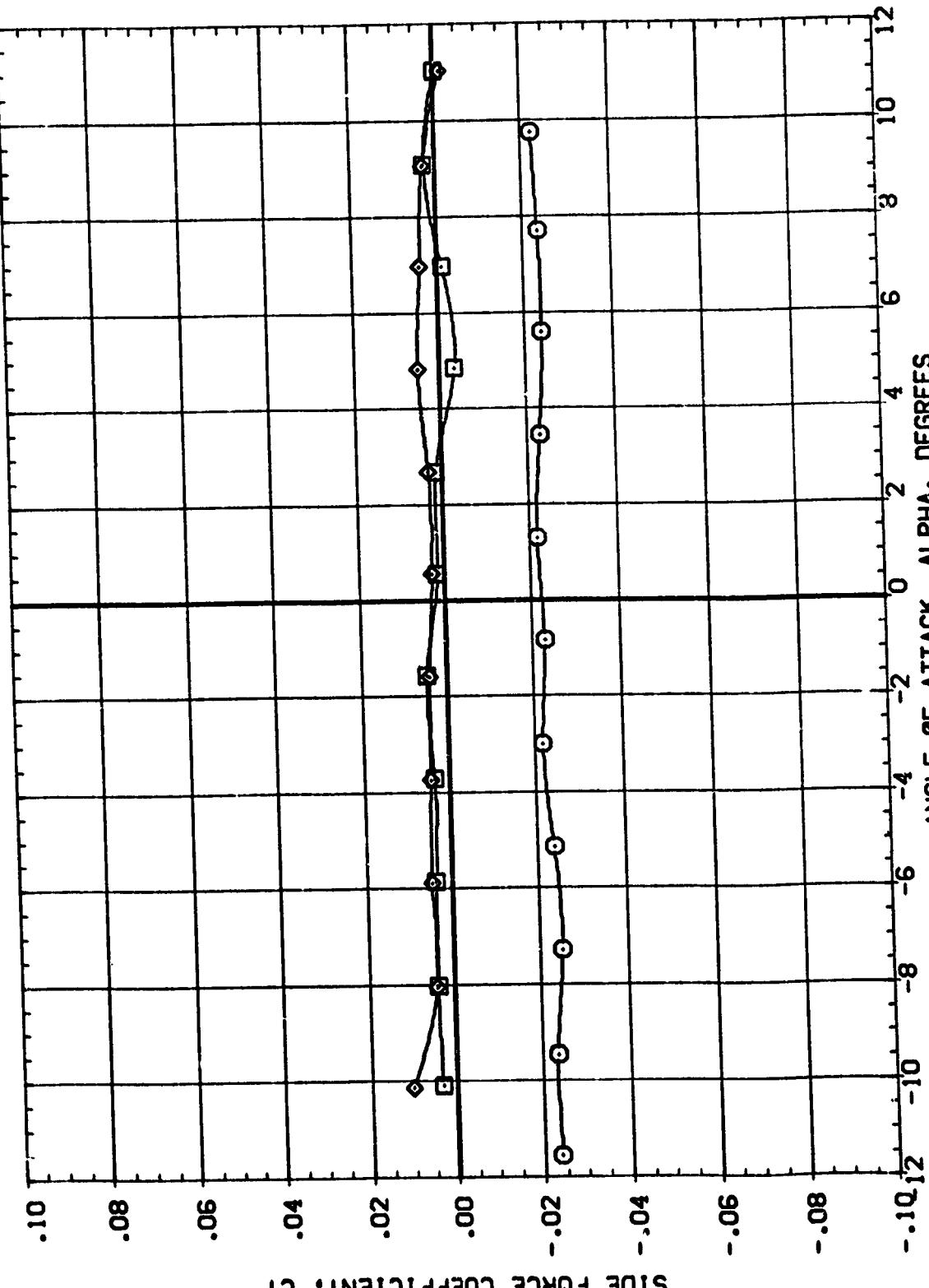
SIDE FORCE COEFFICIENT, C<sub>y</sub>

### EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(A)MACH = .90

DATA SET SYMBOL	CONFIGURATION DESCRIPTION
	[03][19][53] SRB MIS AND.

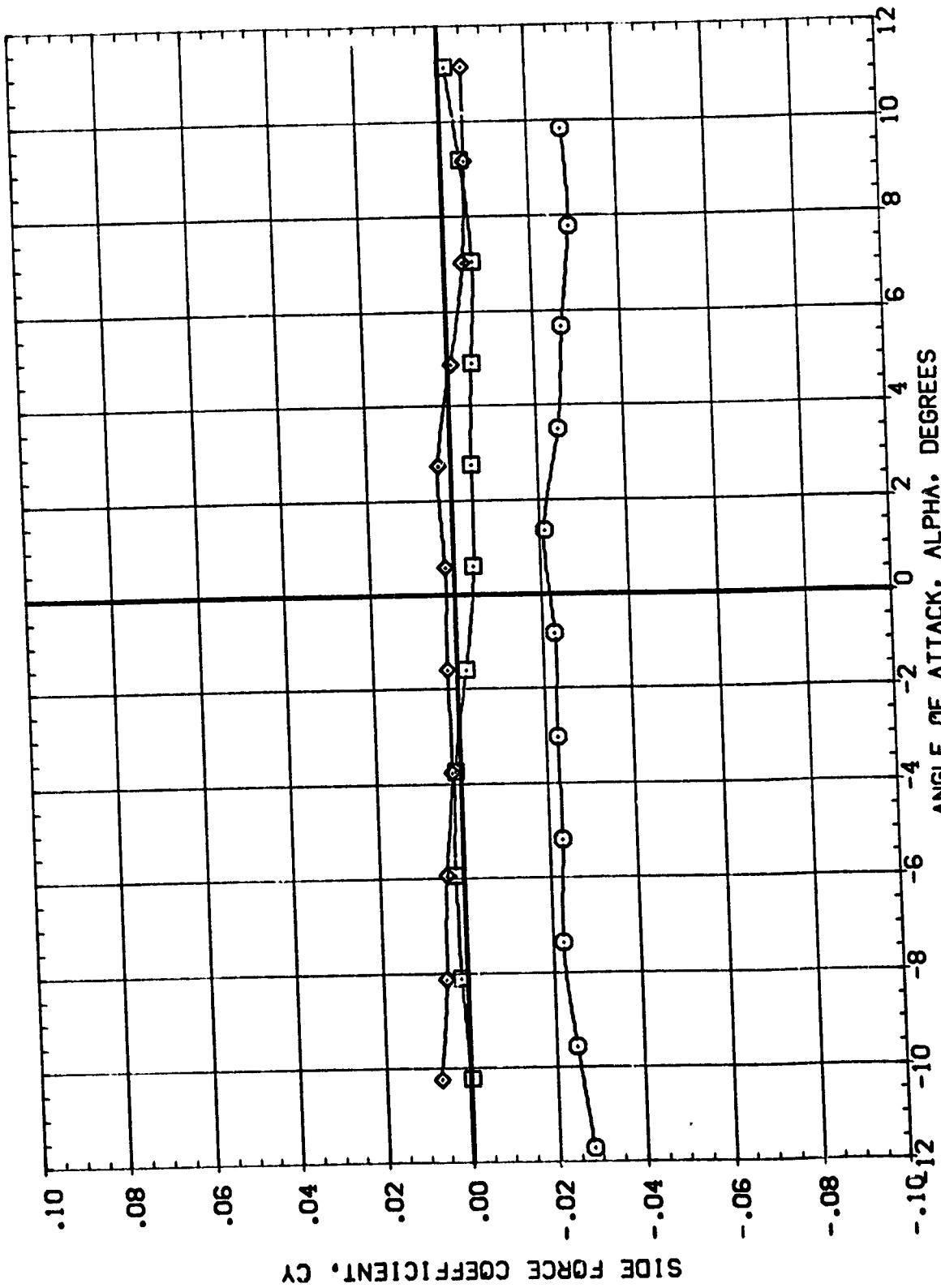
REFERENCE INFORMATION  
 SREF 6.1980 SQ. IN.  
 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5490 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE



EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

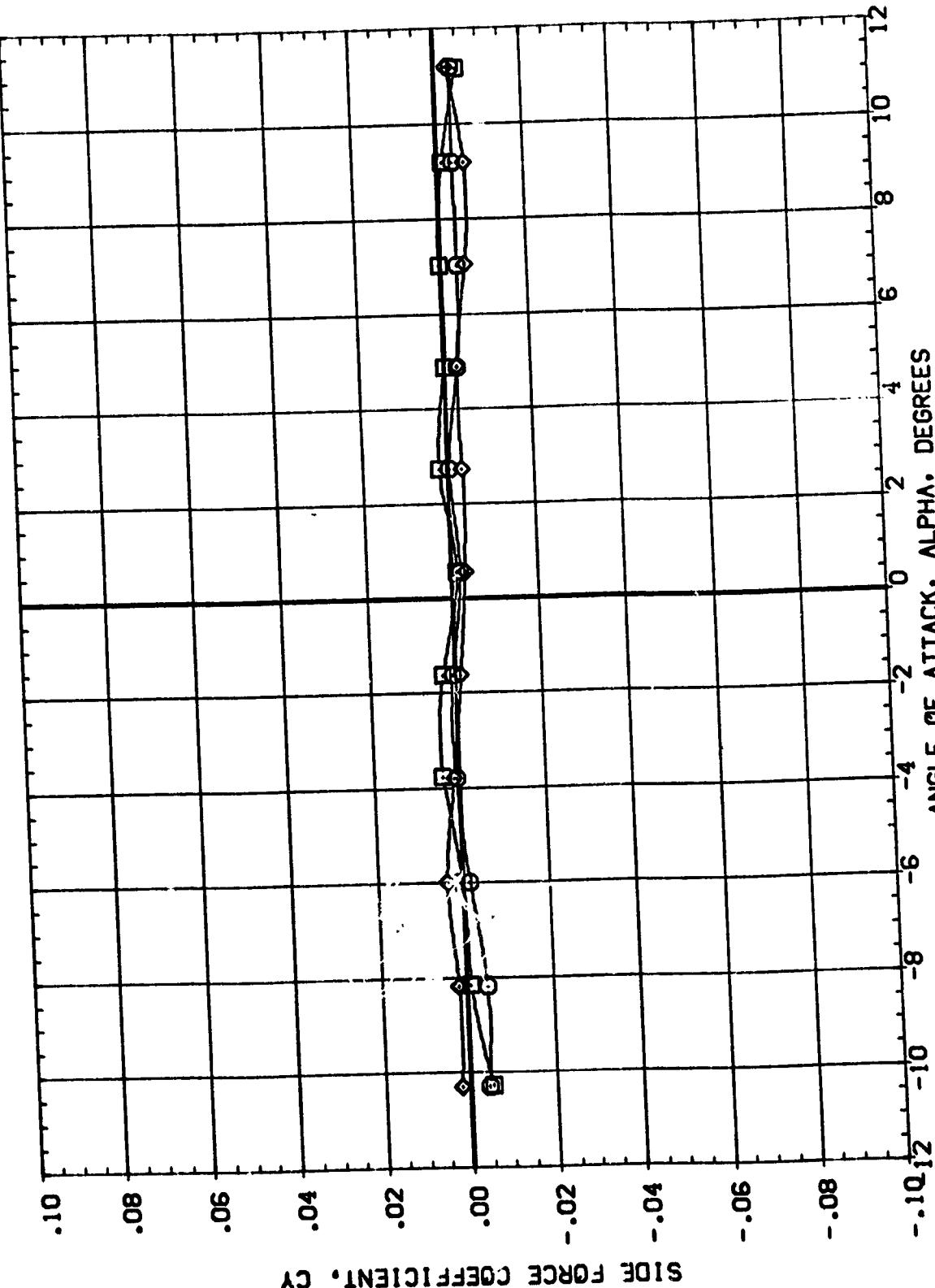
DATA SET SYMBOL	CONFIGURATION DESCRIPTION	SRB INC.	DELTA Z	SRB P/T	REFERENCE INFORMATION
B50000	MSFC 5731(A3) [03] (T9) (S3)	.500	.140		6.1980 SD. IN
B50401	MSFC 5731(A3) [03] (T9) (S3)	.500	.140	1.000	5.3130 IN.
B50402	MSFC 5731(A3) [03] (T9) (S3)	.500	.140	-1.000	5.3130 IN.



### EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

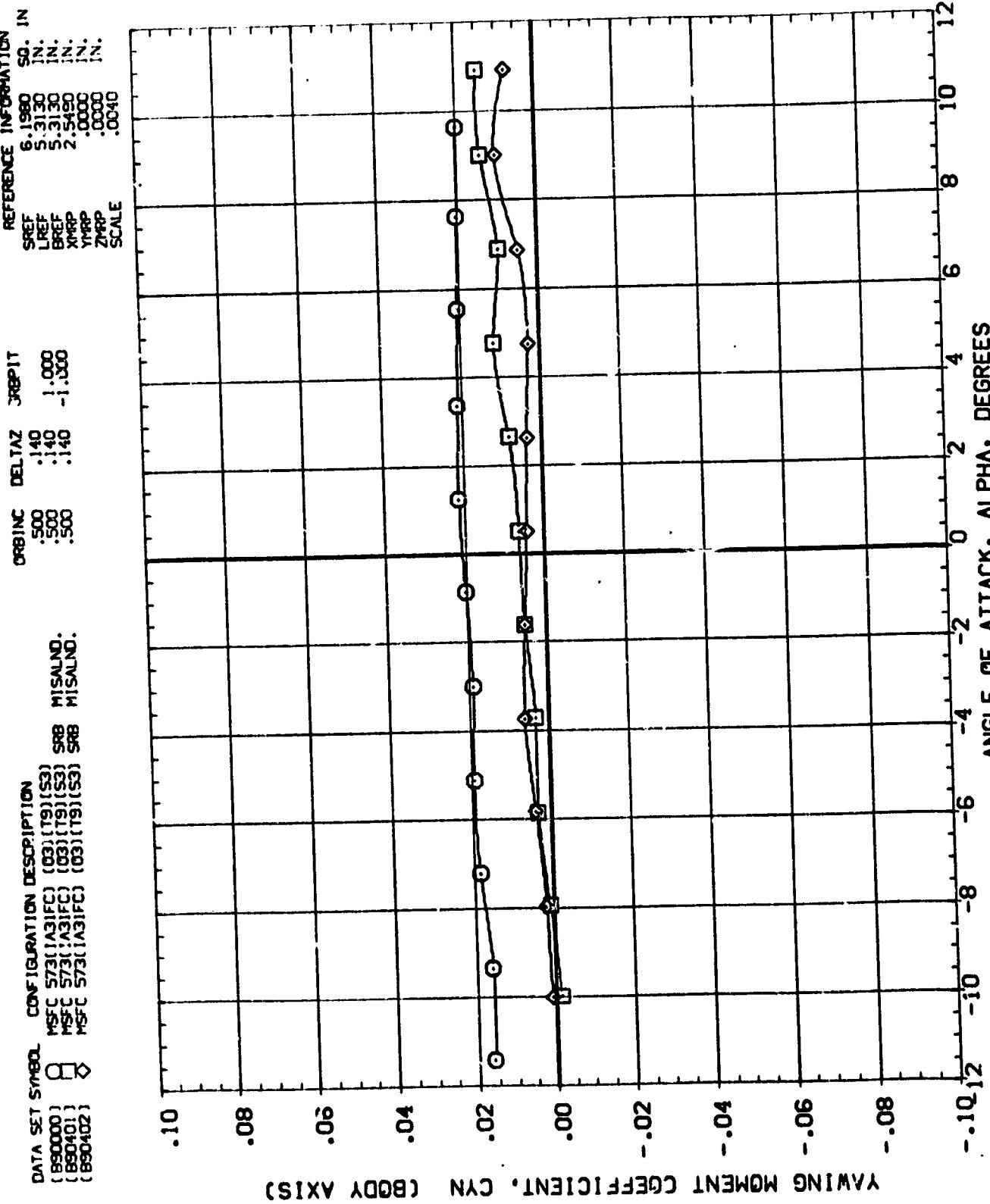
$$(C)MACH = 1.25$$

DATA SET SYMBOL		CONFIGURATION DESCRIPTION	REFERENCE INFORMATION
	NSFC	5731(A3)FC (03)(T9)(S3)	SREF 6.1980 50. IN
	NSFC	5731(A3)FC (03)(T9)(S3)	URF 5.3130 1.N.
	NSFC	5731(A3)FC (03)(T9)(S3)	SREF 5.3130 1.N.
	NSFC	5731(A3)FC (03)(T9)(S3)	XRP 2.5490 1.N.
	NSFC	5731(A3)FC (03)(T9)(S3)	YRP .0000 1.N.
	NSFC	5731(A3)FC (03)(T9)(S3)	ZRP .0040 1.N.
			SCALE

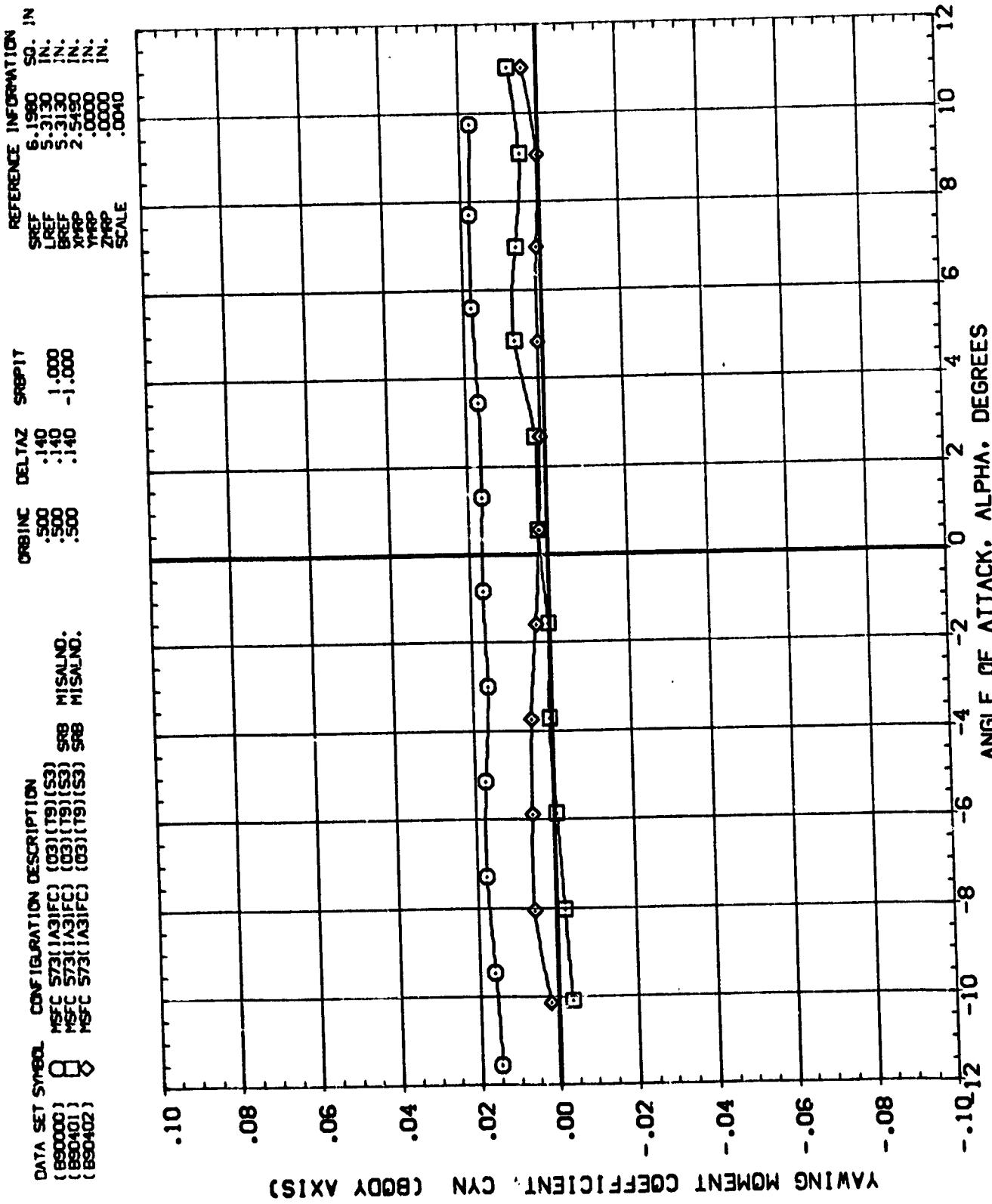


EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 $(\text{MACH} = 1.46)$

REFERENCE INFORMATION  
 DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) A MSC 5731(A3IFC) (03)(T9) (S3)  
 (B90401) D MSC 5731(A3IFC) (03)(T9) (S3) SRB MISALND.  
 (B90402) O MSC 5731(A3IFC) (03)(T9) (S3) SRB MISALND.  
 (B90403) X MSC 5731(A3IFC) (03)(T9) (S3) SRB MISALND.



EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 $(\text{A})\text{MACH} = .90$



EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

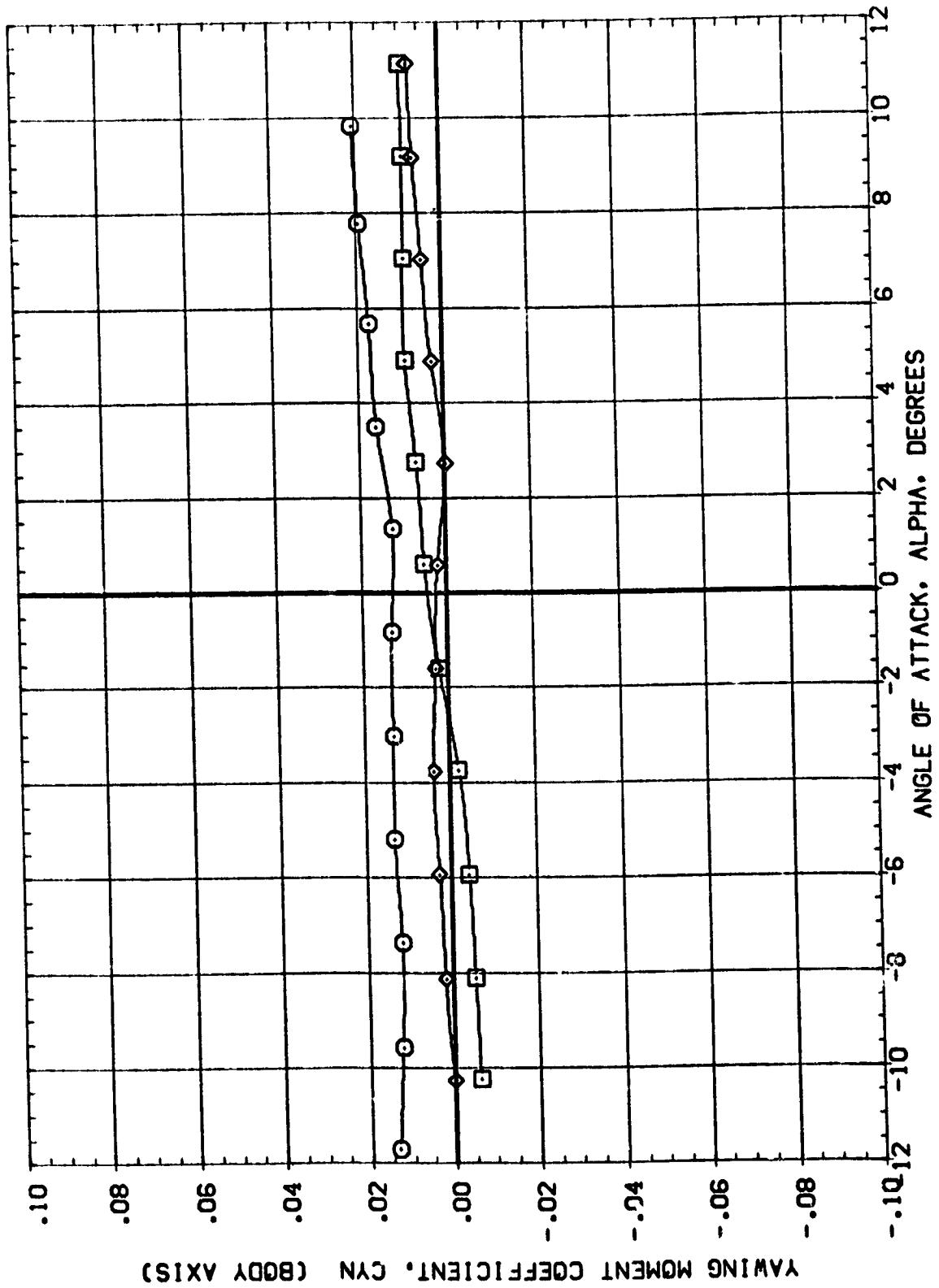
(B)MACH = 1.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(B90000)	NSFC	S73([A3]FC)	(03)(19)(S3)	SRB	MISALNO.
(B90401)	NSFC	S73([A3]FC)	(03)(19)(S3)	SRB	MISALNO.
(B90402)	NSFC	S73([A3]FC)	(03)(19)(S3)	SRB	MISALNO.

REFERENCE INFORMATION

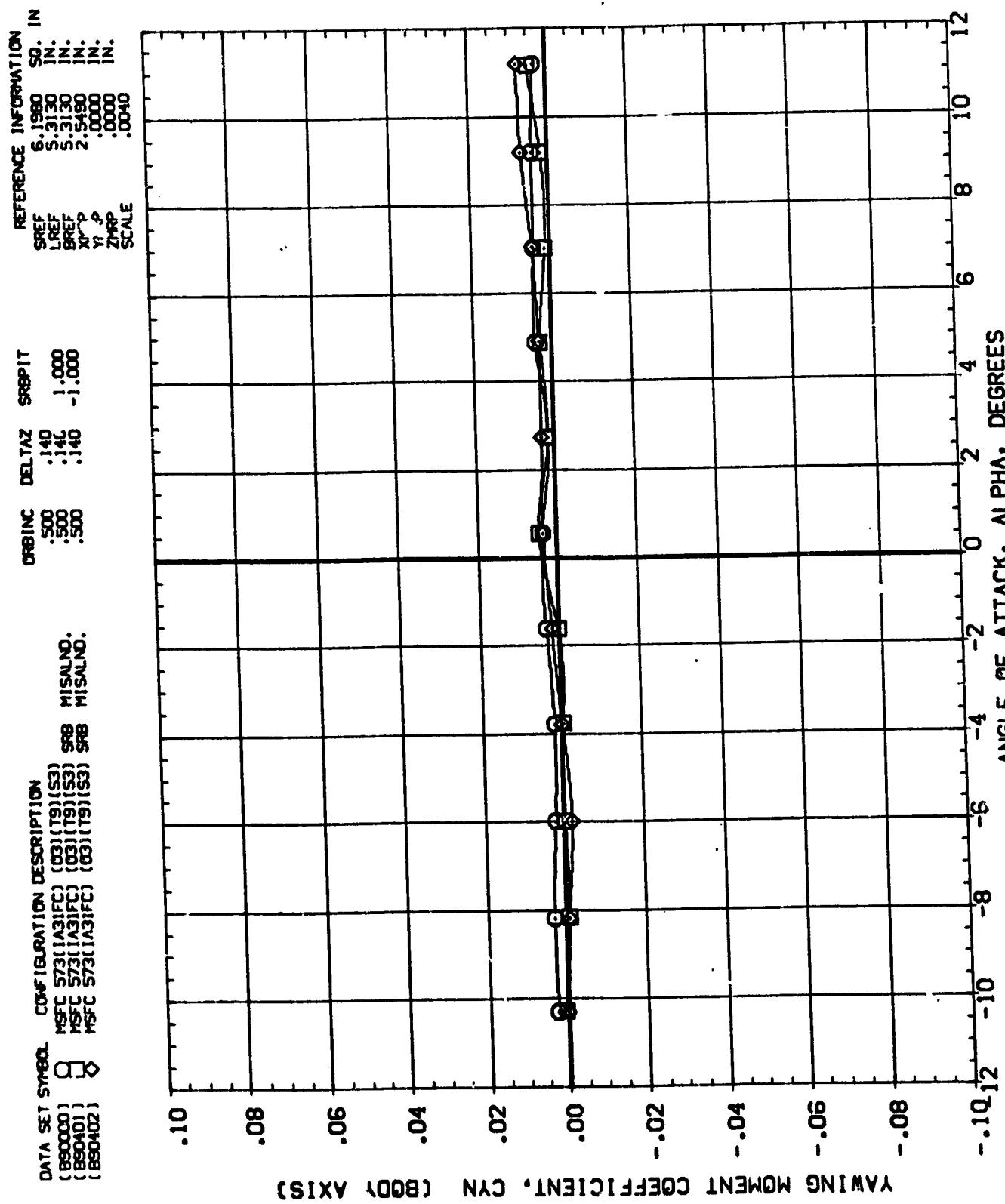
SRBF	6.1980	SG. IN.
LREF	5.3130	IN.
BREF	5.3130	IN.
XMRP	2.5490	IN.
YMRP	.0000	IN.
ZMRP	.0000	IN.
SCALE	.0040	



### EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

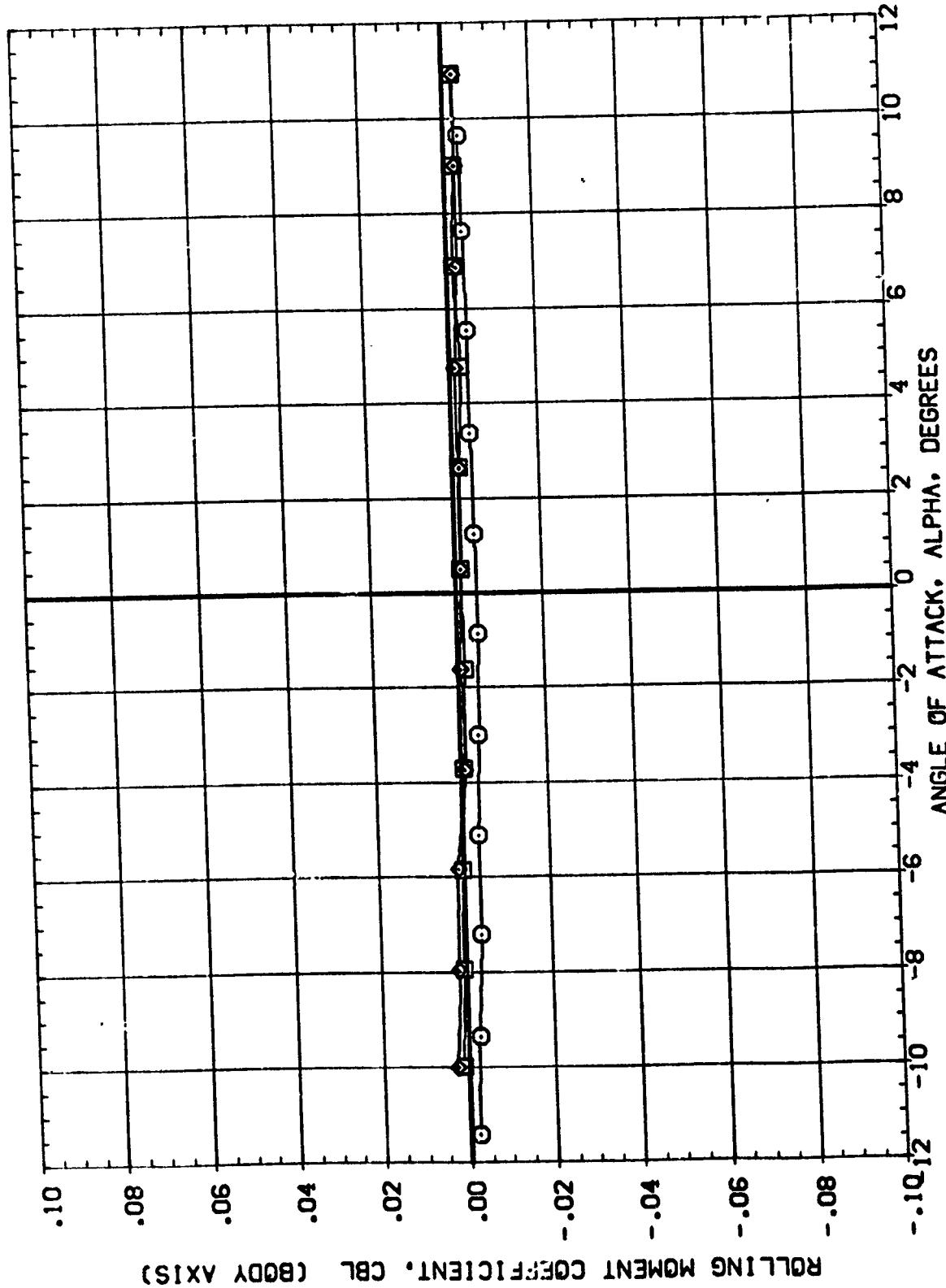
PAGE 176



EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 $(\text{D})_{\text{MACH}} = 1.46$

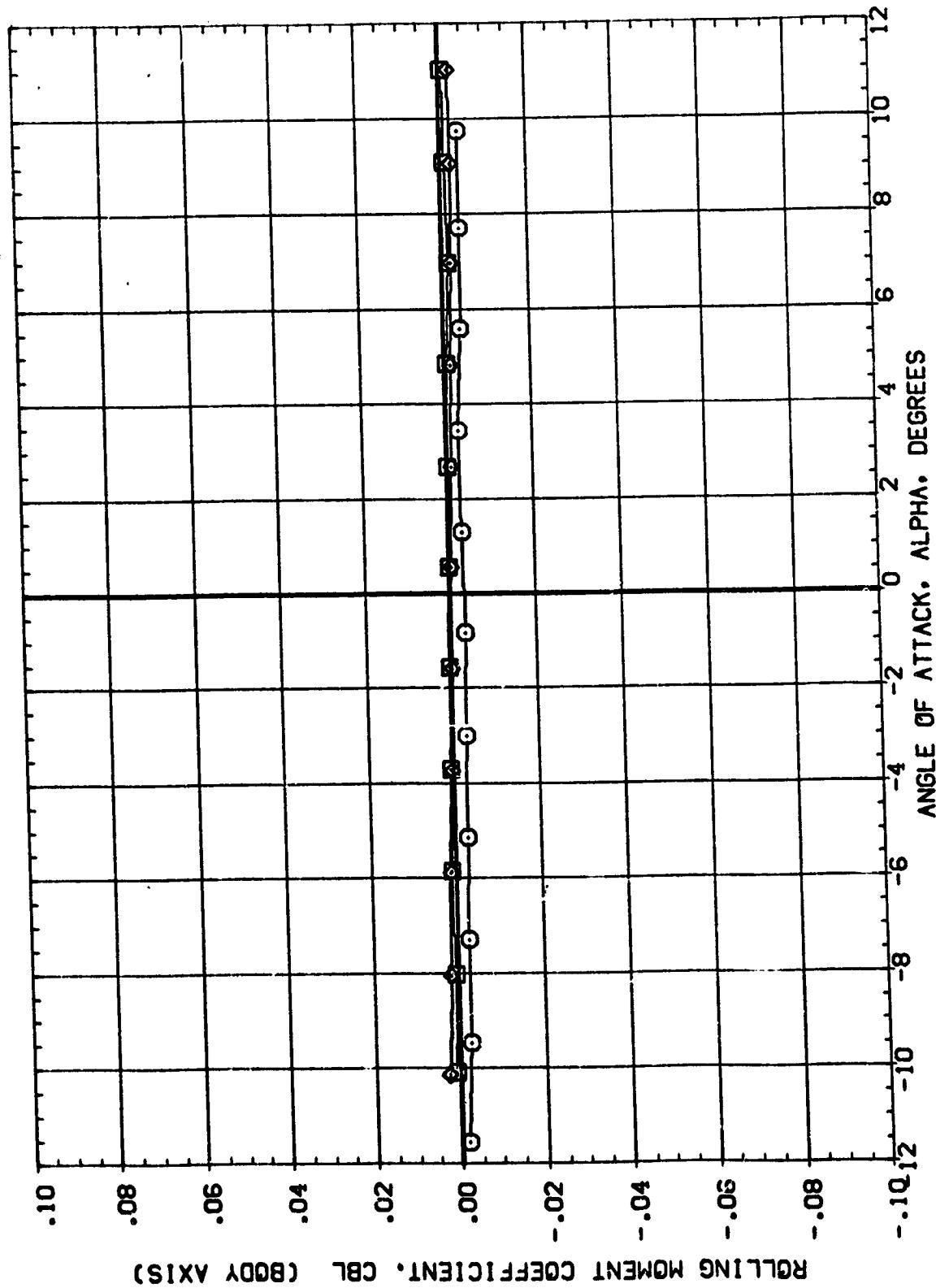
REFERENCE INFORMATION  
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 LREF 5.3130 IN.  
 BREF 5.3130 IN.  
 XMRP 2.5450 IN.  
 YMRP .0000 IN.  
 ZMRP .0040 IN.  
 SCALE

DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(B80000)	MSFC ST3(A31FC) (03)(19)(S3)
(B80401)	MSFC ST3(A31FC) (03)(19)(S3) SRB MISALND.
(B80402)	MSFC ST3(A31FC) (03)(19)(S3) SRB MISALND.



EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 $(\text{AJMACH}) = .90$

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	REFERENCE INFORMATION
(850000)	MSFC S73(IA3)FC (03)(T9)(S3)	SREF 6.1930 SG. IN.
(850401)	MSFC S73(IA3)FC (03)(T9)(S3)	LREF 5.330 IN.
(850402)	MSFC S73(IA3)FC (03)(T9)(S3)	BREF 5.330 IN.
		XMRP 2.530 IN.
		YMRP .0000 IN.
		ZMRP .0000 IN.
		SCALE .0010

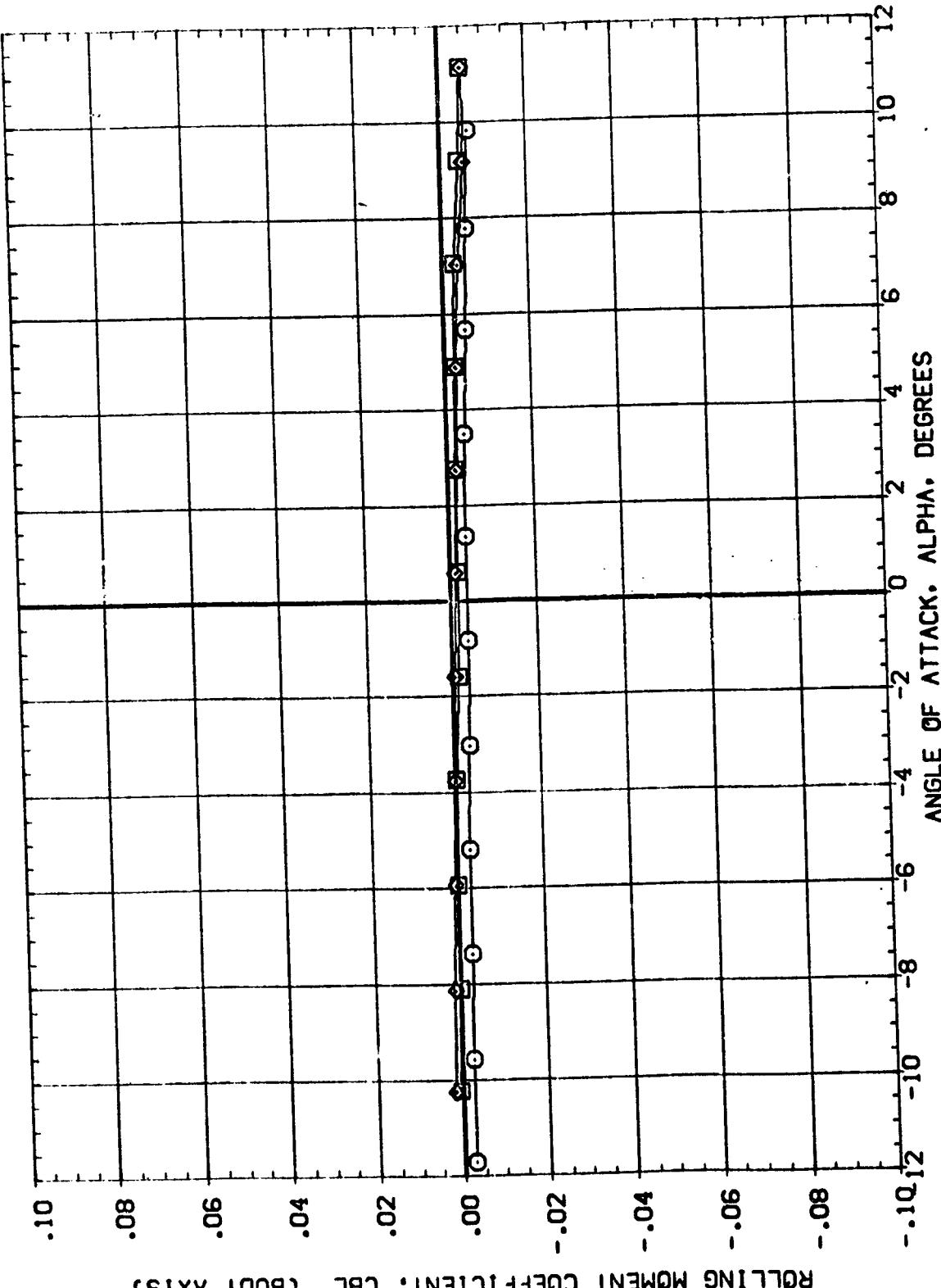


### EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(B)MACH = 1.05

DATA SET SYMBOL CONFIGURATION DESCRIPTION  
 (B90000) MSFC 5731(A3)FC (03)(T9)(S3) SRB MISALND.  
 (B90401) MSFC 5731(A3)FC (03)(T9)(S3) SRB MISALND.  
 (B90402) MSFC 5731(A3)FC (03)(T9)(S3) SRB

REFERENCE INFORMATION  
 SREF 6.1980 SG. IN  
 LREF 5.3130 IN  
 BREF 5.3130 IN  
 XMRP 2.5590 IN  
 YMRP .0000 IN  
 ZMRP .0000 IN  
 SCALE .0040



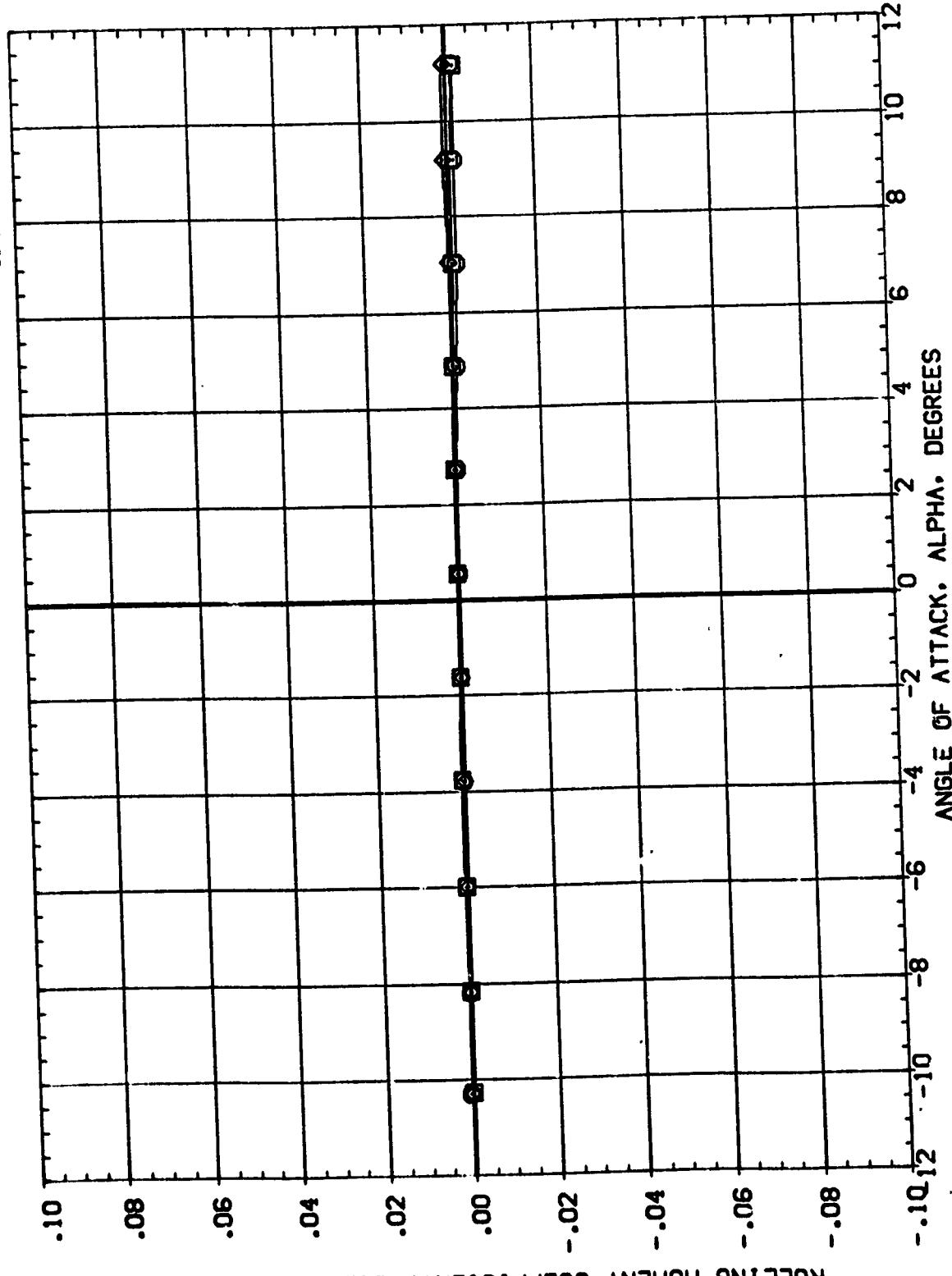
EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS

(C)MACH = 1.25

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DATA SET SYMBOL      CONFIGURATION DESCRIPTION  
 NSFC 5731 (A3) IFC (03) (T9) (S3)      MISALND.  
 (B90422)      NSFC 5731 (A3) IFC (03) (T9) (S3)      SRB MISALND.  
 (B90422)      NSFC 5731 (A3) IFC (03) (T9) (S3)      SRB MISALND.  
 (B90422)

REFERENCE INFORMATION  
 SREF      6.1950 SD. IN  
 LREF      5.3130 IN  
 BREF      5.3130 IN  
 XRP      2.5190 IN  
 YRP      .0000 IN  
 ZRP      .0000 IN  
 SCALE      .0040



ROLLING MOMENT COEFFICIENT, CBL (BODY AXIS)

EFFECT OF SRB PITCH ON AERODYNAMIC CHARACTERISTICS  
 $(D)MACH = 1.46$

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**APPENDIX**

**TABULATED SOURCE DATA**

**Plotted data tabulations are available  
from the DMS on request.**

DATE 29 OCT 73

TABULATED SOURCE DATA, MSFC TWT 573  
MSFC 573 (IASIFC) (65) (79) (65)

PAGE 1  
(RECD 0000) (29 SEP 73 )

REFERENCE DATA

SHEEF = 0.1900 IN. IN. XREF = 2.5600 IN.  
LEEF = 5.3130 IN. YREF = .0000 IN.  
SREF = 5.3130 IN. ZREF = .0000 IN.  
SCALE = .0040

RUN NO. 1/0 RNL = 6.27 GRADIENT INTERVAL = -5.00/ 5.00

	CN	CLW	CT	CYN	CL	CAF	OSO	CAB	CABE	CABG
MACH	.74100	.29600	-.02800	.01600	-.00200	.00310	.01220	.05770	.06100	.12310
.901	-11.420	.23010	-.03020	.01600	-.00260	.01160	.01220	.05760	.05760	.11600
.901	-9.350	.59840	-.03370	.01660	-.00360	.01370	.01210	.05720	.05710	.11220
.901	-7.220	-.44840	.17020	-.03400	.01950	-.00370	.01190	.01200	.05680	.11250
.901	-5.110	-.30830	.10710	-.03400	.01920	-.00420	.01280	.01170	.05550	.10650
.901	-2.990	-.16660	.03530	-.03040	.01920	-.00420	.01170	.02500	.05720	.11120
.901	-8.860	-.04070	-.02450	-.03290	.02030	-.00470	.01220	.01150	.05460	.05580
.901	1.250	.09510	-.08550	-.03920	.02140	-.00460	.01270	.01140	.05590	.05280
.901	3.380	.21340	-.12950	-.03360	.02110	-.00460	.01200	.01140	.05390	.04920
.901	5.540	.33220	-.16360	-.03090	.02060	-.00420	.01190	.01120	.05260	.12280
.901	7.630	.45870	-.23040	-.03140	.02020	-.00370	.01210	.01160	.05460	.13130
.901	9.650	.57250	-.27600	-.03170	.02000	-.00330	.00970	-.00025	.05260	.05091
	GRADIENT	.06005	-.02619	-.00049	.00032	.00000	-.00086	-.00005	-.00025	.00129

RUN NO. 4/0 RNL = 6.54 GRADIENT INTERVAL = -5.00/ 5.00

	CN	CLW	CT	CYN	CL	CAF	OSO	CAB	CABE	CABG
MACH	-.80570	.34150	-.02400	.01600	-.00170	.02680	.01220	.06080	.07660	.14660
1.047	-11.570	.63790	-.02350	.01600	-.00260	.01620	.01310	.06180	.07530	.14390
1.047	-9.460	-.47600	.19720	-.02500	.01750	-.00270	.02200	.01300	.06140	.05790
1.047	-7.290	-.32590	.13080	-.02580	.01720	-.00300	.02400	.01310	.06110	.05590
1.047	-5.140	-.17520	.05800	-.02190	.01620	-.00320	.02470	.01290	.06090	.05410
1.047	-2.990	-.02670	-.01250	-.02250	.01680	-.00350	.02460	.01290	.06100	.04950
1.047	-8.820	1.300	-.08570	-.02160	.01650	-.00320	.02430	.01280	.06140	.04220
1.047	3.440	.24580	-.14660	-.02290	.01670	-.00370	.02340	.01280	.06050	.03750
1.047	5.600	.36310	-.18600	-.02410	.01780	-.00400	.02230	.01280	.06060	.03480
1.047	7.720	.47620	-.23420	-.02550	.01780	-.00430	.02180	.01270	.05580	.03120
1.047	9.750	.59980	-.27970	-.02240	.01750	-.00410	.02090	.01290	.06080	.03120
	GRADIENT	.06825	-.03207	-.00009	.00004	.00006	-.00006	-.00002	-.00006	.00045

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## TABULATED SOURCE DATA, NSFC TWT 573

NSFC 573 (IASIFC) (03) (19) (SS)

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(29 SEP 73 )

## REFERENCE DATA

SREF = 6.1900 SR. IN XREF = 2.5490 IN.  
 LREF = 5.3130 IN. YREF = .0000 IN.  
 ZREF = 5.3130 IN. ZREF = .0000 IN.  
 SCALE = .0040

RUN NO. 2/0 RVAL = 6.66 GRADIENT INTERVAL = -5.00/ 5.00  
 MACH ALPHA CLW CY CYN CEL CAF QBO CABE CABS  
 1.195 -11.700 -.81120 .31650 -.03420 .01650 -.00410 .23120 .01200 .05660 .07710 .13770  
 1.195 -9.560 -.62800 .24070 -.03500 .01990 -.00500 .23670 .11220 .05760 .07470 .13370  
 1.195 -7.350 -.45670 .16770 -.03440 .02070 -.00520 .24120 .01230 .05610 .07050 .12860  
 1.195 -5.160 -.29630 .09900 -.03310 .02150 -.00510 .25700 .01250 .05680 .06720 .12510  
 1.195 -3.010 -.14970 .03470 -.03330 .02180 -.00500 .26320 .01250 .05690 .06310 .12210  
 1.195 -.910 -.00360 -.02900 -.03150 .02080 -.00520 .27080 .01250 .05690 .06910 .12090  
 1.195 1.340 .13500 -.08950 -.02920 .01940 -.00510 .26860 .01240 .05690 .05550 .12060  
 1.195 3.490 .27010 -.15310 -.02750 .01760 -.00450 .26560 .01250 .05690 .02190 .12230  
 1.195 5.670 .39940 .20680 -.02630 .01630 -.00350 .26100 .01260 .05690 .04780 .12560  
 1.195 7.750 .50010 -.24380 -.02730 .01760 -.00410 .25120 .01270 .05690 .04420 .12610  
 1.195 9.840 .61190 -.28440 -.02690 .01810 -.00440 .24570 .01250 .05670 .04050 .13260  
 GRADIENT .06456 -.02887 .00091 -.00082 .00014 -.00014 -.00009 -.00001 -.00001 -.00002 -.00002

RUN NO. 3/0 RVAL = 6.65 GRADIENT INTERVAL = -5.00/ 5.00  
 MACH ALPHA CLW CY CYN CEL CAF QBO CABE CABS  
 1.248 -11.730 -.81810 .32130 -.02810 .01340 -.00270 .23210 .01140 .05390 .07460 .12960  
 1.248 -9.560 -.62710 .23760 -.02480 .01240 -.00260 .23690 .01140 .05380 .06560 .12070  
 1.248 -7.370 -.45160 .16840 -.02230 .01200 -.00300 .24900 .01140 .05490 .06910 .11910  
 1.248 -5.190 -.29470 .09570 -.02310 .01350 -.00300 .25870 .01160 .05490 .05470 .11710  
 1.248 -3.010 -.15190 .03630 -.02310 .01350 -.00350 .26730 .01170 .05500 .05470 .11600  
 1.248 -.910 -.00610 -.02520 .01350 -.02280 -.00400 .27290 .01180 .05570 .04650 .11740  
 1.248 1.340 .12630 -.07840 .01210 -.02110 .01270 -.00390 .27090 .01200 .05650 .04730 .11650  
 1.248 3.500 .25410 -.13720 .01600 -.02510 .01600 -.00430 .26460 .01210 .05690 .04710 .11220  
 1.248 5.680 .36610 -.19420 .01740 -.02570 .01740 -.00410 .26120 .01220 .05690 .04050 .12460  
 1.248 7.800 .50080 -.24630 .01940 -.02690 .01940 -.00450 .25710 .01230 .05690 .03700 .12890  
 1.248 9.890 .61150 -.28460 .02060 -.02790 .02060 -.00430 .24680 .01270 .05690 .03010 .12610  
 GRADIENT .06227 -.02652 -.00034 -.00027 -.00014 -.00006 .00005 -.00003 -.00113 -.00026

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TABULATED SOURCE DATA, NSFC TWT 573  
NSFC 573 (AS31FC) (03) (19) (S3)

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(REF0000) ( 29 SEP 73 )

REFERENCE DATA

SUPER = 6.1980 IN. XMAP = 2.5400 IN.  
LWTF = 5.3130 IN. YMAP = 0.0000 IN.  
WTF = 5.3130 IN. ZMAP = 0.0000 IN.  
SCALE = .0040

RUN NO. 5/0 RVAL = 6.45 GRADIENT INTERVAL = -5.00/ 5.00

	ALPHA	CN	CLM	CV	CIN	CI	CAF	CNB	CAB	CABE
MACH										
1.463	-10.310	-.70450	.27440	-.030400	.00270	.00030	.28920	.00850	.00110	.05320
1.463	-6.160	-.53700	.20120	-.00470	.00910	.00010	.27240	.00850	.00110	.04650
1.463	-6.000	-.37080	.12600	-.00160	.00220	.00020	.27650	.00870	.00110	.04310
1.463	-5.760	-.21970	.06370	.00770	.00160	.00050	.28310	.00880	.00110	.04400
1.463	-1.630	-.07470	.00340	.01060	.00350	.00150	.29410	.00890	.00110	.04150
1.463	.570	.06470	-.02280	-.00140	.00350	.00200	.29390	.00910	.00110	.03620
1.463	2.750	.19210	-.10350	-.00140	.00440	-.00050	.29720	.00910	.00110	.02760
1.463	4.880	.31610	-.15960	-.00310	.00430	-.00090	.29700	.00910	.00110	.02490
1.463	7.040	.43600	-.20750	-.00400	.00420	-.00150	.29450	.00920	.00110	.02630
1.463	9.210	.56090	-.25420	-.00360	.00420	-.00150	.29230	.00940	.00110	.02370
1.463	11.160	.67630	-.29140	-.00360	.00360	-.00130	.28970	.00950	.00110	.02190
	GRADIENT	.06162	-.02312	-.00040	.00044	-.0004	.00104	.00004	.00169	-.00182

PARAMETRIC DATA

BETA = .000  
DELTAZ = .140

CRINC = .000

CRINC = .140

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TABULATED SOURCE DATA, MSFC TWT 573  
MSFC 573 (1A3) (03) (19) (53) C

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(REQ'D) ( 29 SEP 73 )

## REFERENCE DATA

SREF = 6.1960 IN. IN ZHFP = 2.3490 IN.  
LREF = 5.3130 IN. YHFP = .0000 IN.  
BREF = 5.3130 IN. ZHFP = .0000 IN.  
SCALE = .00010

RUN NO. 6 / 0 RVAL =

	CN	CLM	CT	CYN	CLB	CAF	OBD	CAB	CABE
MACH	.699	-.69480	.26190	-.03460	.01700	-.00290	.13120	.01190	.00450
		-.92300	.20190	-.03360	.01910	-.00370	.13640	.01170	.00510
		-.36750	.14340	-.04240	.02530	-.00470	.14440	.01140	.00390
		-.42600	.08310	-.04130	.02340	-.00510	.15370	.01130	.00330
		-.21710	.02000	-.04110	.02320	-.00530	.15400	.01120	.00280
		-.04700	.00440	-.04260	.02420	-.00530	.15490	.01090	.00160
		-.13420	.00870	-.04310	.02350	-.00560	.15520	.01090	.00160
		.26070	-.14780	-.04560	.02570	-.00640	.15560	.01060	.00030
		4.210	.36600	-.19790	-.04170	-.02420	.14770	.00590	.00120
		6.330	.50760	-.25640	-.04570	-.02670	.14410	.00180	.00060
		8.450	.61910	-.30050	-.04460	-.02630	.14120	.001100	.00190
		10.380	.06069	-.02737	-.00048	.00024	-.00016	-.00006	-.00146
GRADIENT									

RUN NO. 7 / 0 RVAL =

	CN	CLM	CT	CYN	CLB	CAF	OBD	CAB	CABE
MACH	1.047	-.72030	.23990	-.02660	.01700	-.00260	.24040	.01270	.00610
		-.59990	.23170	-.02750	.01890	-.00340	.24810	.01260	.00970
		-.40820	.16790	-.02970	.02070	-.00420	.25510	.01270	.00990
		-.6430	-.26190	-.03070	.02080	-.00470	.25690	.01290	.00920
		-.2260	-.11100	-.02710	.02130	-.00520	.26120	.01260	.00950
		-2.150	.03050	-.03990	-.03210	-.00570	.26210	.01240	.00890
		.010	.010	-.03020	.02090	-.00650	.25580	.01210	.00730
		2.140	.16780	-.10320	.02150	-.00660	.24810	.01220	.00750
		4.220	.28030	-.16300	-.02150	-.00660	.24380	.01180	.00690
		6.420	.40610	-.20340	-.02320	-.00640	.23240	.01200	.00690
		8.540	.52150	-.25750	-.03190	-.00690	.22030	.01210	.00730
		10.460	.62640	-.30340	-.03140	-.00630	-.00125	-.00035	-.00224
GRADIENT									

## PARAMETRIC DATA

BETA = .000  
DELTAZ = .140  
CRINC = .500

DATE 29 OCT 73

TABULATED SOURCE DATA, NSFC TWT 575  
NSFC 575 (NASAFC) (80) (13) (SS) 6

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(890100) ( 29 SEP 73 )

REFERENCE DATA

SHEF = 6.1980 IN. IN XHSP = 2.5450 IN.  
LNUF = 5.3130 IN. YHSP = .0000 IN.  
SHEF = 5.3130 IN. ZHSP = .0000 IN.  
SCALE = .0040

RIN NO. 8/0 RNL = 6.67 GRADIENT INTERVAL = -.9.00/ 9.00

MACH	ALPHA	CN	CLH	CY	CYN	CDL	CAF	CDQ	CABO	CABR	CABS
1.248	-10.650	-.73920	.29720	-.02700	.01200	-.00270	.25000	.01120	.05260	.06490	.12070
1.248	-9.700	-.55330	.20910	-.02650	.01290	-.00330	.26810	.01140	.05370	.06930	.11760
1.248	-8.520	-.38610	.13920	-.02690	.01190	-.00430	.27200	.01150	.05440	.06960	.11470
1.248	-7.350	-.24050	.07710	-.02530	.01700	-.00460	.27670	.01180	.05590	.06350	.11360
1.248	-2.160	-.09550	.01510	-.03080	.01780	-.00540	.28290	.01180	.05660	.04960	.11210
1.248	.030	.04620	-.04630	-.03230	.01860	-.00620	.28460	.01190	.05610	.04610	.11360
1.248	2.160	.17360	-.10090	-.03570	.02240	-.00690	.28190	.01190	.05620	.04260	.11620
1.248	4.340	.30490	-.15910	-.03520	.02310	-.00710	.27630	.01180	.05570	.03770	.12270
1.248	6.460	.43370	-.22270	-.03620	.02450	-.00850	.27050	.01200	.05650	.03390	.12480
1.248	8.630	.54670	-.26300	-.03340	.02590	-.00950	.26430	.01190	.05620	.03130	.12480
1.248	10.560	.65400	-.30230	-.03320	.02450	-.01010	.25740	.01210	.05730	.03140	.13000
1.248	GRADIENT	.06273	-.02714	-.00077	.00077	-.00063	.00000	.00005	-.00179	-.00179	.00043

PARAMETRIC DATA

BETA = .000 DELTAZ = .140

CREINC = .500

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TRANSLATED SOURCE DATA, NSFC TWT 573

#### REFERENCE DATA

2.9450 IN.	=	.0000 IN.	=	2.9450 IN.
.0000 IN.	=	.0000 IN.	=	.0000 IN.
.0000 IN.	=	.0000 IN.	=	.0000 IN.
.0000 IN.	=	.0000 IN.	=	.0000 IN.
.0000 IN.	=	.0000 IN.	=	.0000 IN.

$$\text{GRADIENT INTERVAL} = -3.00 / 3.00$$

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#### **MAGNETIC DATA**

BETA =	.000	CRIMINC =	.900
DELTAZ =	.140	CRIMOL =	1.000

$$= \frac{5.3120 \text{ N}}{2000 \text{ m}} = 0.00265 \text{ N/m}.$$

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TABLED SOURCE DATA, NSFC TIT 370  
NSFC 573 (1A31FC) (00) (79) (65) ORB. MEASUR.

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(090200) ( 29 SEP 73 )

PARAMETRIC DATA

REFERENCE DATA

$\Delta\theta =$	0.1800 SR. IN.	$\Delta\theta P =$	2.5000 IN.
$\Delta\theta^2 =$	5.3130 IN.	$\Delta\theta P^2 =$	.0000 IN.
$\Delta\theta^3 =$	5.3130 IN.	$\Delta\theta P^3 =$	.0000 IN.
$\Delta\theta^4 =$	.0000	$\Delta\theta P^4 =$	

BETA = .000  
DELTAZ = .140  
GRBINC = .500  
GRBOL = 1.000

RUN NO. 19/0 RVL = .6.50 GRADIENT INTERVAL = -5.00/ 5.00  
GRADIENT

ALPHA	CN	CLW	CY	CYN	CL	CAF	CSD	CABO	CASE	CABS
-0.247	-66910	.26370	.00960	-.00460	.00110	.28250	.01060	.03160	.05950	.12270
-0.147	-50710	.16860	.00820	-.00400	.00060	.25100	.01100	.03260	.05800	.12260
-0.147	-34630	.00660	.00660	-.00260	.00060	.26120	.01100	.03210	.05570	.11360
-0.147	-5.940	.11460	.00560	-.00260	.00060	.26120	.01100	.03240	.05240	.11360
-0.147	-1.930	.00310	.00310	-.00150	.00060	.26120	.01100	.03210	.05130	.11360
-0.147	-3.750	-.19560	.00310	-.00150	.00060	.27650	.01100	.03320	.05130	.11360
-0.147	-1.590	-.08660	.00310	-.00150	.00060	.27750	.01100	.03350	.05130	.11360
-0.147	.600	.08660	.00310	-.00150	.00060	.27650	.01100	.03350	.04180	.11360
-0.147	2.750	.21750	.00310	-.00150	.00060	.27650	.01100	.03470	.03590	.12270
-0.147	4.910	.39150	.00310	-.00150	.00060	.27650	.01100	.03610	.03240	.12270
-0.147	7.050	.47670	.00310	-.00150	.00060	.27650	.01100	.03640	.02830	.12270
-0.147	9.210	.59230	.00310	-.00150	.00060	.28470	.01200	.03640	.02830	.13310
-0.147	11.170	.70240	.00310	-.00150	.00060	.28470	.01200	.03720	.02950	.13310
-0.147	.06267	-.02705	.00073	-.00106	.00060	.00369	.00060	.00056	.00052	.00181

RUN NO. 36/0 RVL = 6.47 GRADIENT INTERVAL = -5.00/ 5.00  
GRADIENT

ALPHA	CN	CLW	CY	CYN	CL	CAF	CSD	CABO	CASE	CABS
-10.310	-69910	.27160	.00410	-.00350	.00120	.26670	.00560	.00560	.05310	.10110
-6.160	-.53010	.19810	.00350	-.00350	.00120	.27430	.00560	.04400	.04770	.06950
-6.000	-.36830	.12720	.00140	-.00140	.00120	.28130	.00570	.04060	.04260	.06840
-5.944	-.24460	.06160	.00260	-.00260	.00120	.28130	.00560	.04140	.03960	.06440
-5.944	-3.790	-.24460	.00250	-.00250	.00120	.29620	.00560	.04240	.03510	.06470
-5.944	-1.620	-.06980	.00350	-.00350	.00120	.30160	.00600	.04240	.02960	.06350
-5.944	.570	.07120	-.05510	.00140	-.00160	.30110	.00560	.02650	.02650	.06360
-5.944	2.740	.20100	-.10680	.00140	-.00160	.30160	.00560	.04260	.02960	.06360
-5.944	4.900	.32240	-.15820	-.00180	-.00160	.23810	.00560	.04310	.02950	.06320
-5.944	7.050	.44350	-.21100	-.00350	-.00160	.23530	.00560	.04260	.02260	.06360
-5.944	9.220	.56700	-.25760	-.00350	-.00160	.23300	.00560	.04460	.01960	.06470
-5.944	11.190	.69460	-.29620	-.00350	-.00160	.23120	.00560	.04460	.01960	.06470
-5.944	.06186	-.02519	.00045	-.00045	.00120	.00056	.00056	.00056	.00056	.00185

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## TABULATED SOURCE DATA, NSFC TWT 572

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NSFC 573 (IASIFC) (03) (T9) (S9) CRB. HESLAND.

(R92021) (29 SEP 73)

## REFERENCE DATA

**SREF =** 6.1900 IN. **IN XHPP =** 2.5450 IN.  
**LREF =** 5.3130 IN. **YHPP =** .0000 IN.  
**BREF =** 5.3130 IN. **ZHPP =** .0000 IN.  
**SCALE =** .0000

RUN NO. 23 / 0 RVAL = 6.14 GRADIENT INTERVAL = -5.00/ 5.00  
 CLH CLM CY CYN CBL CAF CNEO CABD CABE CABS  
 MACH ALPHA ON .626393 .24760 .03290 -.01610 .00760 .10650 .01150 .05440 .06940 .11350  
 -.656 -10.000 -.626393 .16620 .03290 -.01540 .00750 .12400 .01110 .05230 .06290 .10510  
 -.656 -7.960 -.496393 .12970 .03290 -.01370 .00770 .12100 .01060 .05150 .06010 .10220  
 -.656 -5.850 -.35190 .03560 -.01500 .00750 .13500 .01080 .05060 .05950 .05950 .09930  
 -.656 -3.710 -.22130 .06550 .03300 -.01650 .00730 .13500 .01070 .05070 .05950 .10070  
 -.656 -1.520 -.08770 -.00693 .03300 -.01690 .00730 .13500 .01050 .05050 .05950 .10040  
 -.656 -.500 .03680 -.05790 .03180 -.01210 .00680 .13650 .01050 .04950 .04950 .10640  
 -.656 2.630 -.16690 -.11030 .03300 -.01380 .00680 .13650 .01040 .04950 .04950 .11250  
 -.656 4.770 .29670 -.16020 .02830 -.01260 .00550 .13540 .01060 .05010 .05010 .11740  
 -.656 6.880 .42610 -.22140 .02720 -.01190 .00550 .12560 .01030 .05030 .05030 .12330  
 -.656 9.010 .55200 -.27430 .02330 -.00950 .00550 .12530 .01050 .05140 .05140 .12560  
 -.656 10.910 .66100 -.32090 .01920 -.00670 .00550 .12330 .01050 .05140 .05140 .12560  
 -.656 GRADIENT .06104 -.02647 .00059 .00112 -.00021 .00265 -.00012 -.00012 -.00112 .00112

RUN NO. 23 / 0 RVAL = 6.42 GRADIENT INTERVAL = -5.00/ 5.00  
 CLH CLM CY CYN CBL CAF CNEO CABD CABE CABS  
 MACH ALPHA ON .04250 -.02020 .00630 .00630 .21250 .01240 .05670 .05640 .14390  
 -.10.110 -.68230 .28760 .04100 -.01600 .00610 .22620 .01240 .05650 .05650 .13720  
 1.048 -6.040 -.52460 .21860 .04100 -.01610 .00610 .24130 .01220 .05740 .05740 .12810  
 1.048 -5.860 -.37150 .15060 .03710 -.01610 .00610 .24760 .01230 .05800 .05800 .12320  
 1.048 -3.720 -.21750 .07770 .03760 -.01570 .00660 .25340 .01210 .05700 .05700 .12080  
 1.048 -1.590 -.08790 .04650 .03530 -.01570 .00660 .25120 .01210 .05710 .05710 .12130  
 1.048 .560 .07820 -.05640 .03560 -.01580 .00660 .24640 .01210 .05710 .05710 .12230  
 1.048 2.700 .23440 -.13510 .03570 -.01720 .00790 .24890 .01150 .05440 .05440 .12280  
 1.048 4.840 .33360 -.16020 .03570 -.01840 .00780 .23860 .01160 .05550 .05550 .13070  
 1.048 6.980 .49160 -.22620 .03330 -.01840 .00780 .22730 .01200 .05650 .05650 .13810  
 1.048 9.110 .57350 -.28220 .02630 -.01520 .00680 .21870 .01230 .05790 .05790 .14570  
 1.048 11.040 .67700 -.32710 .02690 -.01460 .00680 .21870 .01230 .05790 .05790 .14570  
 1.048 GRADIENT .08467 -.03063 -.0023 -.00011 1 -.00021 -.00023 -.00023 -.00023 -.00023

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TABULATED SOURCE DATA, NSFC TWT 573  
NSFC 573 (1A31PC) (33) (33) CFB, NSALD.

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(TWT201) ( 29 SEP 73 )

NSFC 573 (1A31PC) (33) (33) CFB, NSALD.

PARAMETRIC DATA

REFERENCE DATA

BETAP =	6.1960 IN.	XREFP =	2.9450 IN.
LREF =	5.3130 IN.	YREFP =	.00000 IN.
ZREF =	5.3130 IN.	ZREFP =	.00000 IN.
SCALE =	.0040		

RUN NO.	22 / 0	RNL =	6.53	GRADIENT INTERVAL =	-5.00/ 5.00				
	CN	CLH	CY	CYN	CBL	CAF	CBO	CABE	CABS
MACH	ALPHA	.006500	.007000	.006600	-.001200	.009100	.24360	.011000	.001400
1.247	-10.270	-.50610	.16600	.05770	-.01540	.00870	.23220	.01100	.00200
1.247	-6.130	-.30610	.11760	.03380	-.01790	.00840	.26290	.01110	.00200
1.247	-5.940	-.34790	.11760	.03100	-.01630	.00800	.27120	.01100	.00190
1.247	-5.750	-.19610	.05570	.03100	-.01460	.00750	.27770	.01110	.00200
1.247	-5.560	-.05340	.02630	.02840	-.01270	.00670	.27650	.01120	.00200
1.247	-5.370	-.08640	.06230	.02370	-.01170	.00560	.27950	.01140	.00200
1.247	-5.180	-.21510	.12160	.02150	-.01090	.00520	.27780	.01150	.00200
1.247	-4.990	-.34780	.17950	.01810	-.00970	.00450	.27600	.01170	.00200
1.247	-7.060	-.47710	.23760	.01300	-.00770	.00300	.26790	.01190	.00200
1.247	-9.200	-.56690	.27840	.01300	-.00770	.00280	.26110	.01210	.00200
1.247	-11.160	-.70090	.31940	.01140	-.00770	.00230	.26060	.01230	.00200
GRADIENT	.06289	-.02710	-.00151	-.00072	-.00035	-.00035	.00006	.00029	.00066

RUN NO.	37 / 0	RNL =	6.53	GRADIENT INTERVAL =	-5.00/ 5.00				
	CN	CLH	CY	CYN	CBL	CAF	CBO	CABE	CABS
MACH	ALPHA	-.70310	.27390	.02690	-.01340	.00780	.20810	.00840	.00970
1.434	-9.180	-.53370	.20050	.02490	-.01260	.00670	.27210	.00850	.00400
1.434	-6.030	-.36860	.12770	.02360	-.01120	.00620	.27650	.00880	.01410
1.434	-3.790	-.21690	.06310	.02410	-.01080	.00520	.26950	.00890	.01400
1.434	-1.620	-.07260	.00310	.02140	-.01020	.00360	.23950	.00900	.02500
1.434	.570	-.06560	-.09140	.02060	-.01080	.00560	.29530	.00910	.03200
1.434	2.740	.19540	-.10370	.01840	-.01050	.00440	.29560	.01090	.02620
1.434	4.900	.32010	-.15910	.01570	-.00830	.00370	.29560	.01090	.04970
1.434	7.050	.44020	-.20700	.01420	-.00750	.00310	.28240	.010940	.02700
1.434	9.220	.56360	-.23260	.01350	-.00640	.00250	.28940	.010960	.02430
1.434	11.190	.67940	-.29070	.01380	-.00590	.00200	.28570	.009660	.02140
GRADIENT	.06165	-.02499	-.00091	-.00021	-.00029	-.00029	.00005	.00021	.00159

TABULATED SOURCE DATA - NSFC TUT 573  
NSFC 573 (IAIFC) (CS) (TS) (S3) MISALD.

(NSFC000) (29 SEP 73)

## REFERENCE DATA

SHEF =	6.1900 IN.	XWRF =	2.5450 IN.
LWRF =	5.3130 IN.	YWRF =	.0000 IN.
BWRF =	.3130 IN.	ZWRF =	.0000 IN.
SCALE =	.00040		

RUN NO.	24/ 0	RNL =	6.23	GRADIENT INTERVAL = -5.00/ 5.00						
ALPHA	CN	CLN	CY	CYN	CBL	CAF	ONBO	CABO	CABE	CABS
-9.990	-.61850	.21340	-.00540	.00240	.00150	.10360	.01130	.03360	.06910	.11470
-9.980	-.46050	.18250	-.00630	.00360	.00110	.11010	.01140	.03590	.06800	.10970
-9.960	-.34710	.12140	-.00910	.00570	.00100	.11990	.01110	.03250	.06970	.10710
-9.940	-.21030	.05060	-.00870	.00740	.00020	.12260	.01100	.05170	.08240	.10310
-9.920	-.08250	-.00860	-.01000	.00820	-.00070	.12750	.01090	.06130	.08710	.10370
-9.900	-.1.650	-.06800	-.01300	.00970	-.00180	.12710	.01070	.06060	.08670	.10580
-8.980	.900	.03950	-.06800	-.00630	-.00190	.12590	.01060	.05080	.08260	.10940
-8.960	2.630	.16830	-.11060	-.01260	.00560	.12500	.01050	.04940	.08040	.11140
-8.940	4.770	.29270	-.15560	-.01210	.00760	.12210	.01110	.05220	.08190	.12060
-8.920	6.900	.42650	-.21200	-.01490	.01060	.12610	.01110	.05150	.08240	.12610
-8.900	9.020	.55220	-.26790	-.02290	.01460	.12340	.01090	.05190	.08360	.13360
-8.880	10.900	.65710	-.31190	-.03100	.00900	.12210	.01100	.05190	.08327	.13124
GRADIENT	.05913	-.02423	-.00044	.00028	-.00027	.00017	-.00006	-.00027	-.00124	

RUN NO.	25/ 0	RNL =	6.50	GRADIENT INTERVAL = -5.00/ 5.00						
ALPHA	CN	CLN	CY	CYN	CBL	CAF	ONBO	CABO	CABE	CABS
-10.100	-.67500	.28410	-.00420	.00220	.00080	.21180	.01250	.03920	.06960	.14360
-9.030	-.51690	.21520	-.00410	.00210	.00080	.22460	.01260	.03930	.06970	.13650
-9.049	-.36680	.14910	-.00490	.00310	.00120	.23710	.01240	.03870	.05440	.12940
-9.070	-.21610	.07770	-.00290	.00280	.00100	.2480	.01230	.03890	.04900	.12320
-9.049	-3.710	.00070	-.00270	.00270	.00100	.2540	.01230	.03890	.04450	.12110
-9.049	-1.610	.06940	-.00160	.00360	.00110	.2460	.01220	.03760	.03670	.12050
-9.049	.540	.06940	-.07170	-.00160	.00360	.23960	.01210	.03730	.03130	.12310
-9.049	2.690	.20760	-.13310	-.00470	.00720	.2450	.01150	.03650	.02670	.12260
-9.049	4.840	.32950	-.17920	-.00570	.00860	.2450	.01160	.03630	.02650	.13140
-9.049	6.980	.44560	-.22110	-.00590	.00740	.2360	.01180	.03760	.02850	.14080
-9.049	9.090	.56540	-.22860	-.00440	.00670	.21960	.01220	.03670	.02870	.14590
-9.049	11.010	.67290	-.33130	-.00340	.00610	.21250	.01240	-.00046	-.00270	
GRADIENT	.06935	-.02369	-.00109	.00022	-.00027	.00016	-.00006	-.00024	-.00124	

DATE 29 OCT 73

## TABULATED SOURCE DATA, NSFC TNT 573

NSFC 573 (IA31FC) (63) (19) (52) SRF MISALD.

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( 29 SEP 73 )

## REFERENCE DATA

SREF = 6.1980 SQ. IN. XREF = 2.5490 IN.  
 LREF = 5.3130 IN. YREF = .0000 IN.  
 BREF = 5.3130 IN. ZREF = .0000 IN.  
 SCALE = .0040

RUN NO. 25/ 0 RNL = 6.67 GRADIENT INTERVAL = -5.00/ 5.00

	CN	CLW	CY	CYN	CLB	CAF	CAF	CABO	CABE	CABS
MACH	ALPHA	.67200	.26059	-.005930	.000420	.000600	.28210	.01110	.08200	.05390
1.247	-10.230	-4.9560	.16380	-.000590	.000220	.000600	.25100	.01130	.05330	.05960
1.247	-9.110	-	-	.00140	-.000350	.000260	.26190	.01120	.05270	.05520
1.247	-5.920	-3.3760	.11450	.00010	-.000340	.000000	.27000	.01130	.05310	.05150
1.247	-3.740	-1.9110	.03260	.000610	.000350	.000600	.27490	.01150	.05430	.04930
1.247	-1.570	-0.4670	.00610	.000310	.000390	.000590	.27520	.01170	.05510	.04590
1.247	.587	.08590	-.00760	-.000390	.000590	.000540	.27630	.01180	.05570	.04210
1.247	2.740	.21590	-.12460	-.00360	.000360	-.000160	.27550	.01170	.05540	.03650
1.247	4.920	.34523	-.16050	-.00260	.000460	-.000270	.27570	.01220	.05740	.03240
1.247	7.030	.46760	-.24070	-.00360	.000500	-.000300	.27620	.01220	.05740	.02830
-47	9.180	.58000	-.28350	-.00260	.000530	-.000400	.26600	.01210	.05710	.02730
1.247	11.130	.63170	-.32470	-.002410	.000630	-.000460	.25740	.01230	.05620	.01340
GRADIENT	.06180	-.02894	-.00044	.000270	-.000350	.00057	.00005	.00026	-.00172	.00037

RUN NO. 38/ 0 RNL = 6.46 GRADIENT INTERVAL = -5.00/ 5.00

	CN	CLW	CY	CYN	CLB	CAF	CAF	CABO	CABE	CABS
MACH	ALPHA	.69200	.27400	-.00770	.00190	.00090	.26400	.00840	.03960	.05270
3.462	-10.300	-6.170	-.32400	-.00410	.000300	.00040	.27040	.00850	.04830	.05980
3.462	-6.000	-3.6370	.12450	-.00370	-.00040	-.00020	.27950	.00860	.04200	.05800
3.462	-3.780	-2.1250	.06060	.00180	-.00150	.00000	.28770	.00880	.04150	.05940
3.462	-1.620	-.07070	.00050	.00090	.00110	.00000	.29340	.00890	.04180	.05630
3.462	.570	.06950	-.16560	-.00140	.00310	.000310	.29560	.00890	.04220	.05710
3.462	2.780	.20100	-.10830	-.00220	.00310	-.00020	.29650	.00890	.04260	.02870
3.462	4.490	.32230	-.16710	-.00490	.00660	-.00260	.29710	.00900	.04250	.02570
3.462	7.050	.44500	-.21240	-.00390	.00620	-.00110	.29580	.00910	.04300	.02650
3.462	9.210	.56840	-.25960	-.00470	.00710	-.00140	.29520	.00930	.04410	.02320
3.462	11.180	.68110	-.29570	-.00160	.00580	-.00110	.28880	.00940	.04460	.02130
GRADIENT	.06178	-.02536	-.00074	.00084	-.00008	.00101	.00002	.00013	-.00181	.00068

NSFC 573 (IA31FC) (06) (T9) (S3) S6B MTSANO.

(R90901) (29 SEP 73)

## REFERENCE DATA

SREF = 6.1900 SR. IN XREF = 2.4490 IN.  
 LREF = 5.3530 IN. YREF = .0000 IN.  
 ZREF = 5.3530 IN. ZREF = .0000 IN.  
 SCALE = .0040

RUN ID. 29/ n RVAL = 6.19 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CN	CLW	CY	CTN	CD	CAF	CBD	CAB	CBE	CBS
.902	-10.000	-.63760	-.29470	-.10960	.00360	-.00050	.03960	.01150	.05620	.07110	.11200
.902	-7.970	-.49550	-.19180	-.01080	.00310	-.00110	.10360	.01420	.06260	.06380	.11420
.902	-5.850	-.35490	-.12960	-.01260	.00570	-.00120	.11950	.01100	.05190	.05750	.11160
.902	-3.730	-.22100	-.05980	-.01140	.00690	-.00180	.12700	.01100	.05180	.05240	.11060
.902	-1.630	-.08060	-.00730	-.01200	.00680	-.00350	.12960	.01070	.05060	.04970	.11160
.902	-9.00	.04020	-.06220	-.01680	.00960	-.00270	.12610	.01070	.05070	.04950	.11450
.902	2.630	.17130	-.11410	-.01570	.00650	-.00240	.13060	.01060	.05030	.04660	.11640
.902	4.760	.29880	-.15860	-.01980	.01270	-.00220	.12490	.01060	.05050	.04450	.11220
.902	6.900	.42860	-.21300	-.02680	.01690	-.00250	.12930	.01070	.05050	.04550	.11250
.902	9.020	.55040	-.26740	-.03370	.02420	-.00240	.11360	.01110	.05250	.04110	.13230
.902	10.930	.68911	-.31357	-.03560	.02960	-.00160	.11340	.01110	.05250	.04250	.13700
	GRADIENT	.0068	-.04237	-.00096	.00068	-.00031	-.00017	-.0002	-.00011	-.00091	.00148

RUN ID. 27/ n RVAL = 6.49 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CN	CLW	CY	CTN	CD	CAF	CBD	CAB	CBE	CBS
1.048	-10.150	-.60070	-.29240	.00020	.00010	-.000070	.21670	.01220	.05740	.04450	.14620
1.048	-8.050	-.44770	-.22330	.00250	.00060	-.00060	.25110	.01210	.05710	.05380	.13990
1.048	-5.900	-.37790	-.15460	.00310	.00000	-.00000	.23190	.01250	.05610	.05580	.13790
1.048	-3.720	-.22190	-.07940	.00420	-.00110	-.00030	.23860	.01250	.05650	.05010	.13430
1.048	-1.550	-.07260	.00730	.00020	-.00060	-.00070	.24310	.01240	.05630	.04430	.13250
1.048	.940	.08661	-.06700	.00080	-.00160	-.00220	.24220	.01240	.05640	.03740	.13210
1.048	2.650	.20630	-.13260	-.00440	.00216	-.00070	.23520	.01210	.05720	.04480	.13510
1.048	4.840	.33260	-.17880	-.00300	.00610	-.00050	.23550	.01160	.04470	.02880	.13670
1.048	6.950	.45340	-.22400	-.01060	.00960	-.00010	.22410	.01160	.04577	.02660	.14240
1.048	9.120	.57310	-.27500	-.01760	.01550	-.00160	.21570	.01220	.04780	.02810	.14790
1.048	11.020	.67170	-.32190	-.01640	.01640	-.00050	.20860	.01250	.04790	.02660	.15250
	GRADIENT	.05485	-.03060	-.00160	.00068	-.0002	-.00010	-.00010	-.00011	-.00023	.00034

DATE 29 OCT 73

TABULATED SOURCE DATA, NSFC TWT 573  
 NSFC 573(LASIFC) (CS) (T9) (S3) SRR MSLWD.

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 (22 SEP 73)

## REFERENCE DATA

SREF = 0.1960 SR. IN XHFP = 2.5490 IN.  
 LREF = 5.3130 IN. YHFP = .0000 IN.  
 DREF = 5.3130 IN. ZHFP = .0000 IN.  
 SCALE = .0040

RUN NO. 28/0 RNL = 6.59 GRADIENT INTERVAL = -5.00/ 5.00

	CN	CLW	CT	CYN	CLB	CAF	CBO	CABE	CABS
1.245	ALPHA	.70160	.27340	-.003170	-.000400	.29600	.01110	.03220	.06710
1.245	-10.280	-.32320	.19770	.00000	-.00230	.24650	.01130	.03330	.06360
1.245	-6.140	-.32320	.19770	.00070	-.00240	.29950	.01120	.03290	.06340
1.245	-5.980	-.36100	.12240	.00000	-.00190	.27080	.01110	.03250	.06460
1.245	-5.760	-.20780	.06250	.00120	-.00240	.27560	.01140	.03380	.06590
1.245	-5.760	-.05880	-.00220	-.001460	-.00160	.27560	.01160	.03470	.0710
1.245	-1.800	-.05880	-.00220	-.00090	.00470	-.00160	.05920	.04490	.12690
1.245	.600	.08520	-.06140	-.00670	.00650	-.00160	.27610	.01170	.03630
1.245	2.740	.21570	-.12250	-.00570	-.00570	.27590	.01170	.03630	.12690
1.245	4.900	.35060	-.18340	-.00940	.00790	-.00170	.27590	.01200	.03630
1.245	7.050	.47820	-.24330	-.01090	.00940	-.00160	.26980	.01210	.03700
1.245	9.200	.59660	-.28220	-.01070	.01060	-.00350	.26200	.01210	.03740
1.245	11.440	.69570	-.32220	-.01030	.01590	-.00360	.25510	.01230	.03800
GRADIENT	.08424	-.02830	-.00108	.00107	-.00028	.00049	.00031	-.00171	.00039

RUN NO. 39/0 RNL = 6.45 GRADIENT INTERVAL = -5.00/ 5.00

	CN	CLW	CT	CYN	CLB	CAF	CBO	CABE	CABS
1.463	ALPHA	.71350	.27910	.000650	.00020	.26530	.00860	.04070	.05350
1.463	-10.360	-.54330	.20470	.00350	.00350	.27240	.00860	.04060	.04860
1.463	-6.150	-.37900	.13220	.00150	.00440	.27900	.00870	.04100	.04310
1.463	-6.010	-.37900	.13220	.00150	.00440	.28200	.00880	.04160	.04390
1.463	-5.800	-.22250	.06660	.00440	.00480	.29480	.00850	.04200	.04520
1.463	-1.640	-.06150	.00650	.00070	.00520	.29750	.00220	.04260	.03070
1.463	.560	.06210	-.05130	-.00070	.00410	-.00210	.29790	.00310	.04310
1.463	2.730	.16720	-.10110	-.00320	-.00070	-.00020	.29740	.00910	.02440
1.463	4.880	.30650	-.15120	-.00130	.00190	-.00100	.29350	.00360	.04330
1.463	7.030	.42710	-.20310	-.00160	.00210	-.00160	.29130	.00840	.04450
1.463	9.200	.55110	-.24960	-.00370	.00420	-.00220	.28780	.00950	.04480
1.463	11.170	.66650	-.28760	-.00700	.00700	-.00270	.28780	.00101	.00032
GRADIENT	.06154	-.02930	-.00013	-.00054	-.00017	-.00017	.00004	-.00173	.00019

## TABLED SOURCE DATA, NSFC TWT 573

NSFC 573 (IASIFC) (CS) (19) (53) SR8 MISALND.

(NSFC01) (29 SEP 73)

## REFERENCE DATA

SREF = 6.1900 SD. IN. ZREF = 2.5490 IN.  
 LREF = 5.3130 IN. YREF = .0000 IN.  
 BREF = 5.3130 IN. ZREF = .0000 IN.  
 SCALE = .0040

RUN NO.: 30/ 0 RVAL = 6.16 GRADIENT INTERVAL = -2.00/ 5.00

MACH	ALPHA	CN	CLW	CY	CYN	CLB	CAF	CBG	CABE	CABS
.901	-10.010	-.00000	.24480	-.00050	-.00140	.00130	.00010	.01160	.00460	.06960
.901	-7.900	-.48000	.18223	-.00090	.00050	.00060	.01050	.01160	.00460	.06520
.901	-5.890	-.34640	.12410	-.01160	.00350	.00360	.11030	.01120	.00500	.05910
.901	-3.710	-.20530	.05420	-.00980	.00360	-.00050	.12820	.01090	.00160	.01040
.901	-1.620	-.06010	-.00260	-.01160	.00560	-.00140	.12760	.01080	.00160	.01030
.901	.510	.04610	-.06350	-.01220	.00560	-.00120	.12610	.01060	.00200	.01020
.901	2.640	.17740	-.11750	-.01240	.00400	-.00150	.12550	.01090	.00170	.01050
.901	4.760	.30320	-.16270	-.01590	.01360	-.00240	.12790	.01080	.00110	.01200
.901	6.880	.43210	-.22140	-.00910	.01000	-.00140	.11460	.01100	.00200	.01260
.901	9.010	.59600	-.27460	-.01590	.01400	-.00220	.11170	.01110	.00280	.01340
.901	10.920	.69790	-.31870	-.01540	.01470	-.00220	.11060	.01120	.00260	.013670
GRADIENT										
		.06000	-.02654	-.00060	.00069	-.00017	.00005	-.00004	-.00004	-.00007

RUN NO.: 32/ 0 RVAL = 6.44 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	CN	CLW	CY	CYN	CLB	CAF	CBG	CABE	CABS
1.046	-10.100	-.67630	.28690	.00340	-.00360	.00320	.01250	.05920	.06960	.14520
1.046	-6.030	-.52170	.21900	.00400	-.00210	.0070	.21960	.01240	.05850	.13510
1.046	-5.870	-.36840	.15160	.00380	-.00070	.0060	.23160	.01240	.05650	.12710
1.046	-3.710	-.21490	.07750	.00360	-.00060	.00590	.24000	.01220	.05620	.12330
1.046	-1.570	-.06640	.01050	.00450	-.00400	.00300	.24420	.01210	.05720	.12480
1.046	.530	.07450	-.08960	.00170	.00240	.00000	.24200	.01200	.05690	.13410
1.046	2.650	.21010	-.13510	.00140	.00270	-.00390	.23510	.01200	.05650	.12670
1.046	4.840	.35060	-.17810	-.00360	.00720	-.00060	.23360	.01170	.05520	.13140
1.046	6.980	.44970	-.22490	-.00120	.00640	-.00140	.22140	.01200	.05660	.12620
1.046	9.100	.58440	-.27910	-.00260	.00490	-.00100	.20950	.01220	.05600	.13580
1.046	11.030	.67220	-.32560	-.00030	.00740	-.00070	.20210	.01250	.05620	.14790
GRADIENT										
		.06004	-.00047	-.00063	.00073	-.00013	-.00003	-.00006	-.00031	-.00251

DATE 29 OCT 73

TABULATED SOURCE DATA, NSFC TWT 573  
NSFC 573(1A3FC) (65) (19) (53) SEE MSLND.

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(R50401) ( 29 SEP 73 )  
REF ID: A64460

## REFERENCE DATA

SHEP =	0.1000 IN.	XREF =	2.5000 IN.
LREF =	5.3100 IN.	YREF =	.0000 IN.
SREF =	5.3100 IN.	ZREF =	.0000 IN.
SCALE =	.0040		

RUN NO. 31 / 0 RAVL = 6.34 GRADIENT INTERVAL = -3.00 / 5.00

MACH	ALPHA	CN	CM	CR	CN	CM	CR	CL	CAF	CBO	CAB	CAB
1.245	-10.250	-67480	.2520	.00050	-.00820	.00060	.00060	.23690	.01110	.05240	.04690	.13260
1.245	-8.120	-.50000	.10610	.00140	-.00550	.00000	.00000	.24690	.01130	.05330	.04630	.12750
1.245	-5.950	-.36320	.11710	.00200	-.00430	.00000	.00000	.25790	.01120	.05300	.04570	.12260
1.245	-3.740	-.19270	.05340	.00220	-.00220	-.00040	-.00040	.26690	.01120	.05260	.04560	.11860
1.245	-1.990	-.03110	.00720	-.00200	-.00220	-.00140	-.00140	.27100	.01140	.05390	.04490	.11700
1.245	.600	.06910	-.06430	-.00460	-.00550	-.00560	-.00560	.26990	.01150	.05420	.04770	.11710
1.245	2.750	.22070	-.12550	-.00460	-.00680	-.00690	-.00690	.27140	.01170	.05510	.04370	.12160
1.245	4.900	.36000	-.18360	-.00560	-.00900	-.00910	-.00910	.27140	.01170	.05510	.04760	.122250
1.245	7.050	.46000	-.24370	-.00650	-.00910	-.00910	-.00910	.26590	.01200	.05690	.04420	.12820
1.245	9.200	.59080	-.26360	-.00440	-.00890	-.00890	-.00890	.25850	.01210	.05690	.04220	.13220
1.245	11.150	.69770	-.36340	-.00520	-.00940	-.00940	-.00940	.25040	.01230	.05790	.04000	.13760
1.245	GRADIENT	.06285	-.02741	-.00076	.00125	-.00020	.00043	.00006	.00025	-.00177	.00064	

RUN NO. 40 / 0 RAVL = 6.45 GRADIENT INTERVAL = -3.00 / 5.00

MACH	ALPHA	CN	CM	CR	CN	CM	CR	CL	CAF	CBO	CAB	CAB
1.463	-10.250	-.68740	.26600	-.00330	.00070	.00000	.00000	.27690	.00850	.04910	.04820	.09660
1.463	-8.160	-.52190	.19610	-.00100	-.00100	.00010	.00010	.26350	.00850	.04930	.04930	.09730
1.463	-5.950	-.35990	.12420	.03100	-.00400	.00040	.00040	.29120	.00860	.04930	.04310	.09350
1.463	-3.770	-.21990	.06350	-.00360	-.00360	.00060	.00060	.29730	.00860	.04440	.04240	.09250
1.463	-1.620	-.05770	.00360	.00300	.00010	.00030	.00030	.29900	.00900	.04240	.04240	.09100
1.463	.570	.05880	-.05630	-.00130	.00460	-.00220	-.00220	.30000	.00900	.04250	.02780	.09550
1.463	2.730	.19720	-.10690	.00210	.00150	.00150	.00150	.30070	.00900	.04270	.02510	.09660
1.463	4.800	.31850	-.16390	.00000	.00000	-.00010	-.00010	.29850	.00920	.04390	.02610	.09670
1.463	7.037	.43710	-.21240	.00020	.00110	.00110	.00110	.29550	.00950	.04480	.02570	.10060
1.463	9.190	.59970	-.26020	-.00110	.00160	-.00070	-.00070	.29260	.00960	.04540	.02020	.10070
1.463	11.160	.67350	-.26910	-.00510	.00490	-.00180	-.00180	.00002	.00100	-.00157	.00043	
	GRADIENT	.06104	-.02580	-.00039	.00040	-.00006	-.00006					

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## TABULATED SOURCE DATA, NSFC TWT 973

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NSFC 973 (1A31FC) (CB) (T9) (S3) SRB HISLND.

(R99402) (29 SEP 73)

## REFERENCE DATA

SREF = 6,1960 SH. IN. XREF = 2,5490 IN.  
 LINEF = 5,3130 IN. YREF = .0000 IN.  
 ZREF = 5,3130 IN. ZREF = .0000 IN.  
 SCALE = .0040

RUN NO. 35/0 RVAL = 6.16 GRADIENT INTERVAL = -5.00/ 5.00

	CLN	CLM	CT	CTN	CLB	CAF	CBD	CAB	CBE	CBS
MACH	ALPHA	.693910	.29160	.00510	.00060	.00600	.01190	.00450	.00920	.11630
.697	-10.030	-.493900	.18010	.00300	.00200	.00170	.00150	.00430	.00930	.11250
.697	-7.570	-.36510	.13360	.00040	.00410	.00130	.01110	.00250	.00930	.10460
.697	-5.340	-.23130	.08670	-.00410	.00630	-.00100	.01270	.01100	.00150	.10470
.697	-3.750	-.10260	.00760	-.00370	.00560	-.00070	.02930	.01090	.003150	.10240
.697	-1.650	-.02680	.025150	-.00420	.00460	-.00130	.04110	.01050	.00660	.04150
.697	-5.00	-.19910	-.10680	-.00380	.00400	-.00140	.03690	.01060	.00930	.04220
.697	2.530	-.38420	-.15600	-.00260	.00310	-.00130	.03220	.01070	.00930	.04040
.697	4.750	-.41700	-.21400	-.00700	.00510	-.00220	.02290	.01070	.00270	.03970
.697	6.380	-.54340	-.27090	-.01950	.00193	-.00250	.02350	.01060	.00110	.04130
.697	9.010	-.64740	-.31460	-.01260	.00160	-.00230	.01670	.01110	.00260	.04140
.697	10.520	-.68031	-.02646	.00014	-.00098	-.00006	.00356	-.00004	-.00022	.00049
GRADIENT										

RUN NO. 35/0 RVAL = 6.45 GRADIENT INTERVAL = -5.00/ 5.00

	CLN	CLM	CT	CTN	CLB	CAF	CBD	CAB	CBE	CBS
MACH	ALPHA	-.69370	-.25600	.01010	.00200	.00200	.02050	.00260	.00930	.14400
1.048	-10.130	-.53960	.22500	.00430	.00300	.00170	.02100	.00230	.00610	.13600
1.048	-8.060	-.36380	.15770	.00480	.00570	.00130	.02310	.00230	.00630	.13130
1.048	-5.860	-.23360	.08670	.00450	.00540	.00040	.02360	.00240	.00630	.12520
1.048	-3.720	-.10260	.00760	-.00370	.00570	-.00020	.02410	.00210	.00700	.12270
1.048	-1.600	-.02680	.025150	-.00420	.00410	-.00130	.02440	.00200	.00670	.12130
1.048	2.540	-.38420	-.15600	-.00260	.00310	-.00130	.02410	.00190	.00620	.11990
1.048	4.760	-.41700	-.21400	-.00700	.00510	-.00220	.02410	.00190	.00480	.12260
1.048	6.390	-.54340	-.27090	-.01950	.00193	-.00250	.02350	.00190	.00640	.13100
1.048	9.020	-.64740	-.31460	-.01260	.00160	-.00230	.01670	.00110	.00260	.14100
1.048	10.530	-.68031	-.02646	.00014	-.00098	-.00006	.00356	-.00004	-.00022	.00049
GRADIENT										

## NSFC 573 (IASIFC) (05) (19) (53) SRE MSSAUD.

## PARAMETRIC DATA

## REFERENCE DATA

SREF = 6.1960 IN. 20RP = 2.3490 IN.  
 LREF = 5.3130 IN. 11RP = .0000 IN.  
 RREF = 5.3130 IN. 20RP = .0000 IN.  
 SCALE = .0040

	RUN NO.	34/ 0	RNL =	6.56 GRADIENT INTERVAL = -3.00/ 5.00		CABO	CABE	CABS
	CLW	CR	CYN	CBL	CNF	ONBO	.03230	.13170
MACH	ALPHA	ON	.0070	.00190	.23600	.01130	.03230	.13170
	-10.200	-.69520	.27140	.00300	.24590	.01130	.03230	.13170
1.243	-9.150	-.52230	.19430	.00290	.25530	.01120	.03230	.13170
1.243	-5.950	-.39960	.12490	.00250	.26410	.01120	.03230	.13170
1.243	-3.750	-.20810	.06100	.00210	.27300	.01130	.03230	.13170
1.243	-1.500	-.06450	-.00110	.00250	.00310	-.00160	.04700	.11790
1.243	.590	.07680	-.35740	.00220	.00320	.00130	.05410	.11990
1.243	2.740	.20340	-.11290	.00310	-.00220	.02790	.01160	.04250
1.243	4.850	.33390	-.17170	-.02060	-.00270	.02750	.01140	.05400
1.243	7.040	.49520	-.23200	-.00430	.00470	-.00370	.02680	.05640
1.243	9.160	.55330	-.28960	-.00530	.02650	-.00540	.02630	.02990
1.243	11.150	.67920	-.31090	-.00590	.00740	-.00540	.02570	.02740
GRADIENT	.08237	-.02672	-.00224	-.00026	-.00033	.00113	.00003	.0014

	RUN NO.	41 / 0	RNL =	6.46 GRADIENT INTERVAL = -3.00/ 5.00		CABO	CABE	CABS
	CLW	CR	CYN	CBL	CNF	ONBO	.03640	.10390
MACH	ALPHA	ON	.00210	.00000	.00020	.25580	.00850	.04600
	-10.350	-.71170	.27530	.00200	-.00060	.02730	.00840	.04600
1.462	-9.150	-.53970	.20520	.00360	.00010	.28340	.00850	.04600
1.462	-6.000	-.37450	.13130	.00360	-.00210	.29310	.00870	.04630
1.462	-3.750	-.22110	.06800	.00120	-.00030	.28920	.00870	.04630
1.462	-1.500	-.07370	.01090	.00130	.00180	.00000	.30140	.04130
1.462	.570	.07000	-.00530	-.00320	.00350	.00000	.00880	.03150
1.462	2.740	.19140	-.10650	-.00340	.00320	.00000	.00890	.04200
1.462	4.900	.32480	-.15970	-.00390	.00340	.00000	.00930	.04250
1.462	7.050	.44690	-.21310	-.00570	.00460	.00020	.00920	.04340
1.462	9.210	.57110	-.26290	-.00640	.00560	.00070	.00930	.04360
1.462	11.160	.68340	-.30120	-.00280	.00740	.00000	.00004	.00164
GRADIENT	.08274	-.02586	-.00046	.00041	.00000	.00003	.00018	.00047

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TABULATED SOURCE DATA, NSFC TNT 573

NSFC 573 (1A21FC) (03) (19) (53) 6

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(030500) ( 29 SEP 73 )

REFERENCE DATA

XREF = 6.1980 58. IN XREF = 2.5450 IN.  
YREF = 5.3130 IN. YREF = .0000 IN.  
ZREF = 5.3130 IN. ZREF = .0000 IN.  
SCALE = .0040

REF. DATA  
RUN NO. 13/ D RVAL = 0.25 GRADIENT INTERVAL = -5.00/ 5.00  
  
MACH ALPHA CN CLM CY CYN CLR CYP CZN CYP CNE CNE  
.308 -.030 -.02950 -.04450 -.05130 -.03750 -.04450 -.05250 .05110 .05250 .0510 .0510  
GRADIENT .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000

PARAMETRIC DATA

BETA = .000 CRITINC = .500  
DELTAZ = .140